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***Evaluation of high temperature  
gas cooled reactor performance:  
Benchmark analysis related to initial  
testing of the HTTR and HTR-10***



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EVALUATION OF HIGH TEMPERATURE GAS COOLED REACTOR PERFORMANCE:  
BENCHMARK ANALYSIS RELATED TO INITIAL TESTING OF THE HTTR AND HTR-10

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## FOREWORD

The Co-ordinated Research Project (CRP) on Evaluation of High Temperature Gas Cooled Reactor (HTGR) Performance was initiated by the IAEA in 1998 on the recommendation of the Technical Working Group on Gas Cooled Reactors. This CRP was established to foster the sharing of research and associated technical information between participating Member States in the ongoing development of the HTGR as a future source of nuclear energy for high temperature process heat applications and the production of electricity.

The IAEA has facilitated an extensive programme that addresses the technical development of advanced gas cooled reactor technology. This CRP complements other recently completed CRPs in validating safety and performance capabilities of the HTGR. For this CRP, computer codes and models are verified through actual test results from operating reactor facilities. Specifically, the IAEA is grateful to the Japan Atomic Energy Research Institute for providing the HTTR, China's Institute of Nuclear Energy Technology for the HTR-10 and the Russian Federation's OKBM and the Kurchatov Institute for the ASTRA critical facility in support of this CRP.

The focus of this TECDOC is on core physics benchmarks in conjunction with initial testing of the HTTR and HTR-10 and also selected thermal hydraulic benchmarks on the HTTR. The following Member State national institutions participated in the performance of this CRP:

- Institute of Energy Technology (INET), Beijing, China
- SACLAY (CEA), Gif-sur-Yvette, France
- Research Centre Juelich (ISR), Juelich, Germany
- National Nuclear Energy Agency (BATAN), Serpong, Indonesia
- Japan Atomic Energy Research Institute (JAERI), Oarai, Japan
- Nuclear Research and Consultancy (NRG), Petten, Netherlands
- OKBM/Kurchatov Institute, Nizhny Novgorod, Russian Federation
- Pebble Bed Modular Reactor (PBMR), Centurion, South Africa
- Department of Nuclear Engineering, Hacettepe University, Ankara, Turkey
- Oak Ridge National Laboratory (ORNL), Oak Ridge, TN, United States of America

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### *EDITORIAL NOTE*

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## CONTENTS

CHAPTER 1. INTRODUCTION .....	1
CHAPTER 2. HIGH TEMPERATURE TEST REACTOR (HTTR) REACTOR PHYSICS BENCHMARKS .....	3
2.1. HTTR general information .....	3
2.1.1. Facility description .....	3
2.1.2. Reactor physics benchmark problem descriptions .....	25
2.2 Reactor physics benchmark analysis and results .....	27
2.2.1 Japan .....	27
2.2.2. France .....	50
2.2.3. Germany .....	69
2.2.4. Indonesia .....	93
2.2.5. Netherlands .....	98
2.2.6. Russian Federation .....	117
2.2.7. Turkey .....	131
2.2.8. United States of America .....	136
2.3. References to Chapter 2 .....	141
CHAPTER 3. HIGH TEMPERATURE TEST REACTOR THERMAL HYDRAULIC BENCHMARKS .....	145
3.1. Thermal hydraulic benchmark information .....	145
3.1.1. Description of systems for VC and LP benchmarks .....	145
3.1.2. HTTR thermal hydraulic benchmark problem descriptions .....	156
3.2. Thermal hydraulic benchmark analysis/results .....	157
3.2.1. Japan .....	157
3.2.2. Russian Federation .....	171
3.2.3. South Africa .....	183
3.2.4. United States of America .....	208
3.2.5. France .....	218
3.3. References to Chapter 3 .....	224
CHAPTER 4. THE HIGH TEMPERATURE GAS COOLED REACTOR TEST MODULE CORE PHYSICS BENCHMARKS .....	227
4.1. HTR-10 general information .....	227
4.1.1. Facility description .....	227
4.1.2. Benchmark problem descriptions .....	239
4.2. HTR-10 benchmark problem analysis/results .....	244
4.2.1. China .....	244
4.2.2. Indonesia .....	254
4.2.3. Japan .....	261
4.2.4. Russian Federation .....	265
4.2.5. Netherlands .....	271
4.2.6. United States of America .....	280
4.2.7. Turkey .....	290
4.2.8. France .....	295
4.2.9. Germany .....	304

4.2.10 South Africa .....	314
4.3. References to Chapter 4 .....	322
CHAPTER 5. RESULTS, CONCLUSIONS AND RECOMMENDATIONS .....	327
5.1. HTTR reactor physics benchmarks .....	327
5.1.1. Collation of codes and models .....	327
5.1.2. Collation of benchmark analysis results.....	328
5.1.3. Discussion of results and general conclusions .....	332
5.1.4. Recommendations .....	333
5.2. HTTR thermal hydraulic benchmarks.....	333
5.2.1. HTTR vessel cooling.....	333
5.2.2. HTTR loss of off-site electric power.....	335
5.3. HTR-10 reactor physics benchmarks .....	336
5.3.1. Analysis methods and models .....	336
5.3.2. Collation of results .....	338
5.3.3. Analysis and conclusions .....	341
5.3.4. Recommendations .....	342
APPENDIX: RELATED IAEA PUBLICATIONS .....	343
CONTRIBUTORS TO DRAFTING AND REVIEW .....	345

## Chapter 1

### INTRODUCTION

Of paramount significance in the development of new high temperature gas cooled reactor (HTGR) concepts is the predicted capability for this advanced nuclear plant to achieve a high degree of safety through reliance on passive safety features. Because of this, the investigation and validation of the safety and operational aspects of the HTGR were the primary focus for many of the coordinated research programmes (CRPs) initiated by the IAEA in the 1990s. These included:

- the neutronic physics behaviour of the HTGR core,
- fuel performance and fission product behaviour, and
- the ability of the HTGR to dissipate decay heat by natural transport mechanisms under accident conditions

The principal tools utilized in these CRPs included scientific research and engineering development through analytical evaluation of benchmark problems, application of new and/or existing computer codes and models and utilization of test apparatus and loops for specific component validation.

The next important step in bringing this advanced nuclear power programme from concept to actuality is to verify system performance and safety under actual HTGR operating conditions. It is the need for validation via testing in nuclear reactors that was the stimulus for the IAEA to initiate this CRP on “Evaluation of High Temperature Gas Cooled Reactor Performance”.

The principal facilities utilized in the performance of this CRP included Japan’s High Temperature Engineering Test Reactor (HTTR), China’s High Temperature Test Module (HTR-10), and Russia’s ASTRA critical assembly. Selected codes and models associated with the Gas Turbine Modular Helium Reactor (GT-MHR) and Pebble Bed Modular Reactor (PBMR) plant designs are also investigated within the scope of this CRP.

The objectives of this CRP include:

1. Validation of analytical codes and performance models to actual operating conditions of HTGRs,
2. Formulation of research and development code-to-experiment benchmark activities for inclusion into the test programmes for the HTTR, the HTR-10 and the ASTRA facilities,
3. Investigation of analytical codes and models associated with future HTGR gas turbine plants utilizing code-to-code benchmark problems, and
4. Demonstration of HTGR safety characteristics.

Overall, this TECDOC addresses performance based HTGR code and model verification during startup, steady state and transient operational conditions of the HTR-10 and HTTR test facilities are being evaluated. Also included within the scope of this CRP is the utilization of national research facilities and computer models to investigate the areas of core physics, safety characteristics of the HTGR, fission product release and transportation behaviour, thermal hydraulics, control response and high temperature component performance.

This TECDOC documents the benchmark problem results obtained by Chief Scientific Investigators (CSIs) from China, France, Germany, Indonesia, Japan, the Netherlands, Russia, South Africa, Turkey and the United States for the following areas:

- Reactor physics benchmark analysis of the HTTR including initial criticality, control rod worth, excess reactivity, scram reactivity and temperature coefficient of reactivity,
- Reactor physics benchmark analysis of the HTR-10 including initial criticality, temperature coefficient of reactivity and control rod (including differential) worth,
- Selected thermal hydraulic benchmark analysis for the HTTR including vessel cooling and loss of off-site electric power.

The presentation of this TECDOC is by facility and benchmark type with Chapter 2 devoted to HTTR core physics, Chapter 3 to the HTTR thermal hydraulic benchmark problems associated with vessel cooling and loss of electric power. Chapter 4 is devoted to HTR-10 core physics benchmarks. Each of these chapters includes a description of the test reactor, an overview of the benchmark problems being addressed, individual analysis by each Member State and a review of the actual test results of each problem as performed on the test reactor. Chapter 5 provides a collation of the results, general conclusions and recommendations for code and model improvements determined as the result of participation in the CRP.



## Chapter 2

### HIGH TEMPERATURE ENGINEERING TEST REACTOR (HTTR) REACTOR PHYSICS BENCHMARKS

#### 2.1. HTTR GENERAL INFORMATION

The High Temperature Engineering Test Reactor (HTTR) of the Japan Atomic Energy Research Institute (JAERI) is a graphite-moderated and helium gas cooled reactor with an outlet temperature of 950°C and a thermal output of 30 MW [2-1].

The major objectives of the HTTR are to establish and upgrade the technological basis for advanced high temperature gas-cooled reactors (HTGRs) and to conduct various irradiation tests for innovative high temperature basic researches.

The construction was completed on May 1996. The fuel loading of was started on July 1, 1998 from core periphery. The first criticality was attained in annular type core of 19 columns on Nov.10, 1998. The first full power operation with an average core outlet temperature of 850°C was completed on 7 December 2001, and operational licensing of the HTTR was approved on 6 March 2002.

##### 2.1.1. Facility Description

###### 2.1.1.1. Background

In June 1987, the Japanese Atomic Energy Commission (JAEC) issued a revised “Long Term Program for Development and Utilization of Nuclear Energy” stressing that Japan should proceed to develop more advanced reactor technologies in parallel with the upgrading of existing nuclear reactors. It was recognized in this programme that the HTGR was not to be incorporated into the present existing power plant system, but the benefits that could be derived, such as its inherent safety and production of high temperature heat, are remarkable and should be pursued. Therefore, the promotion of R&D on the HTGR in Japan is quite significant from the viewpoint of a new nuclear technology frontier.

Within this programme, the early construction of a test reactor in place of the experimental Very High Temperature Reactor (VHTR) was recommended based on the estimated length of time (~ 10 years) to construct the plant and perform associated testing. Based on the conclusions by a special committee investigating the HTGR R&D plan for the JAEC, demand of nuclear heat applications to ~ 1,000°C is expected to become strong in the early part of the 21<sup>st</sup> century. The committee requested that a test reactor making the most use of Japanese technology be designed and built to test and study advanced HTGRs for the future. Accordingly, the test reactor should have an in-core irradiation region equipped with the capability to test the threshold of fuel failure and for the irradiation of various materials.

The committee’s report also stated that the reactor outlet coolant temperature should be 950°C, which is the highest temperature attainable considering the current technology level. For irradiation tests, a prismatic block type core structure was proposed, with a thermal power rating of 30MW for securing adequate regions for irradiation tests at high temperature [2-6, 2-7].

Based on these suggestions, JAERI suspended the previous programme where the experimental VHTR was defined as an initial step toward nuclear heat application development and proceed with the design and R&D specifically necessary to prepare the safety analysis report of the HTTR, with the objectives to:

- Establish and upgrade the technology base of the HTGR
- Perform innovative basic research in the field of high temperature engineering
- Demonstrate high temperature heat applications and utilization achieved from nuclear heat.

The Japanese government approved proceeding with the HTTR in its 1989 fiscal year budget. Construction began in March 1991 following submittal by JAERI of the HTTR safety analysis report and subsequent review by the Science and Technology Agency, and then, by the Nuclear Safety Commission.

#### *2.1.1.2. General design features of the HTTR*

The reactor core is designed to keep all specific safety features within the graphite blocks. The intermediate heat exchanger (IHX) is equipped to supply high temperature clean helium for process heat application systems, and the instrumentation and control system is designed to allow operations that simulate accidents and anticipated operational occurrences (AOOs). As the HTTR is the first HTGR in Japan and a test reactor with various purposes, it incorporates specific aspects regarding the safety design. JAERI established the following safety design principles for the HTTR in reference to the “Guidelines for Safety Design of LWR Power Plants”, but taking into account the significant safety characteristics of the HTGR and the corresponding design requirements as a test reactor:

- Coated fuel particles shall not fail during normal operation and AOOs. To satisfy this principle, the maximum fuel temperature, including systematic and random uncertainties, shall not exceed 1600°C for any AOO.
- The reactor shall be shut down safely and reliably during operation using the control rod system. Furthermore, a reserved shutdown system (RSS) which is independent of the control rod system shall be provided.
- A severe accident resulting from control rod ejection must be avoided.
- The residual heat after reactor shutdown shall be removed safely and reliably for any AOO or accident.
- A containment vessel (C/V) shall be provided to prevent fission product release and excessive air ingress into the core in case of a depressurization accident.
- The pressure in the pressurized water cooling system (PWCS) shall be controlled so as to be lower than that of the primary helium gas to prevent a large water ingress into the core in case of rupture of a heat transfer tube in the primary pressurized water cooler (PPWC).
- The helium gas pressure in the secondary helium cooling system (SHCS) shall be controlled to be slightly higher than that of the primary helium gas to prevent fission product leakage from the primary cooling system (PCS) to the secondary due to a crack in a heat transfer tube in the IHX.
- The pressure and heat resisting functions of the structures, where the high pressure and high temperature coolant is contained, are separated to reduce mechanical loads on the high temperature metal structures [2-7].

Safe and reliable shutdown of the reactor from any operational condition is achieved with the control rod system. Furthermore, a reserved shutdown system composed of B<sub>4</sub>C/C

pellets is provided. The power control and normal reactor shutdown of the HTTR are achieved with 16 pairs of control rods or 15 pairs when the center column of the core is used for an irradiation test. The control rod system can achieve subcriticality from any operational condition and maintain subcriticality under cold core conditions including the postulated event of a pair of control rods stuck in the operational position. The major design specifications of the HTTR are shown in Table 2.1. The reactor outlet coolant temperature at the full power is set at both 850° and 950°C. The reactor operational mode at 850°C is defined as "rated operation" and at 950°C is "high temperature test operation" because operation of the HTTR is not allowed at 950°C for full life of the initial core. Tests such as the safety demonstration tests and irradiation tests are allowed only in the rated operation mode. The high temperature nuclear process heat utilization system will be operated at the high temperature test operational mode. The design life of permanent structural components in the HTTR plant is based on 20 years with a load factor of 60 % of full power operation.

The HTGR has excellent safety capabilities with respect to the accidental release of fission products. Nevertheless, the HTTR is required to have a containment vessel to meet Japanese safety design guidelines for the light water nuclear power plants.

### 2.1.1.3. HTTR plant layout and cooling system

The HTTR is located on JAERI's Oarai Research Establishment site which is approximately 100 kilometers north of the Tokyo metropolitan area and is near the Pacific Ocean. The plant area is 200 m x 300 m in size. The shortest distance between the HTTR reactor core and site boundary is about 280 m in the southwest direction. As illustrated in Figure 2.1, the HTTR plant arrangement is comprised of the reactor building, spent fuel storage building, a machinery building, cooling towers, exhaust stack, a high temperature process heat utilization system and other auxiliary facilities. The reactor building of 48 m x 50 m in size is situated in the central area of the plant. The exhaust stack of 80 m in height is north of the reactor building for the air ventilated from the reactor building to be released to the atmosphere. The heat utilization system will be constructed south of the reactor building [2-8].

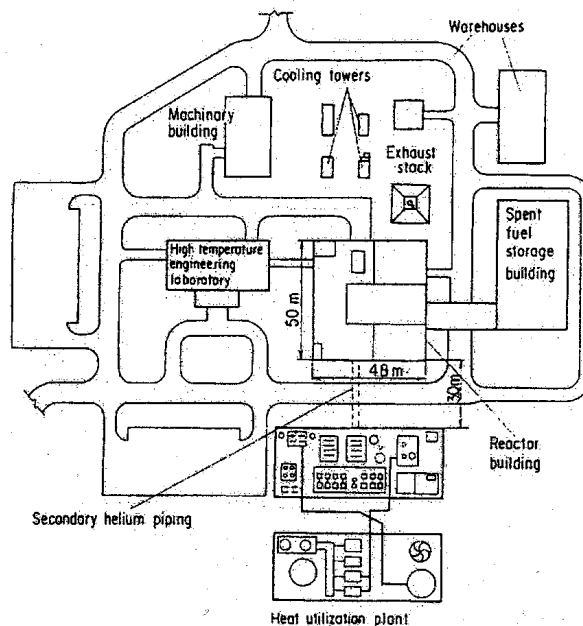


FIG. 2.1. HTTR plant arrangement.

The reactor building includes five levels with three floors underground (Figure 2.2). A steel reactor containment vessel of 18.5 m in diameter and 30 m in height is installed in the center of the reactor building. A refueling hatch is attached to the C/V above the reactor pressure vessel (RPV). Functions of the C/V are to:

- 1) Contain fission products (FPs) as one of the multiple barriers against FP release into the atmosphere, and
- 2) Limit the amount of air ingress into the core in a primary pipe rupture accident.

The reactor pressure vessel (RPV) is formed as a vertical cylinder, with a hemispherical top and bottom head closures and 31 standpipes. The top head closure is bolted to a flange of the vessel cylinder. The standpipes include "control rod (CR) stand-pipes", "irradiation stand-pipes", and standpipes for instrumentation. The irradiation standpipes are utilized to introduce specimens and experimental equipment into the core. A thermal shield is attached to the inner surface of the top head closure to prevent the closure from overheating in a depressurization accident such as a primary helium pipe rupture. The RPV is of 2-1/4Cr1Mo steel normalized and tempered.

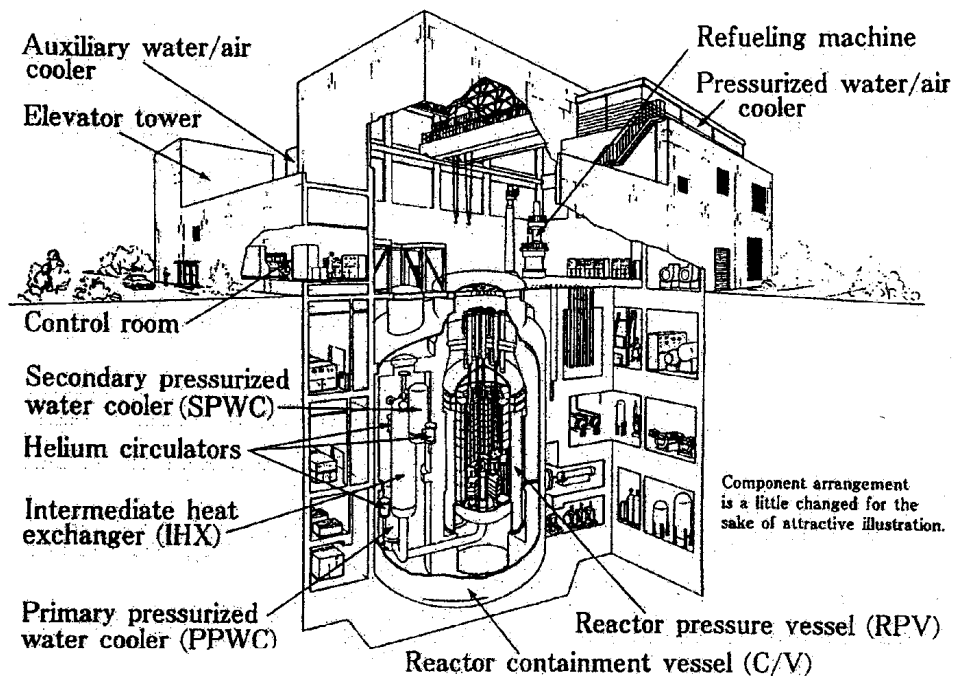


FIG. 2.2. HTTR reactor building.

The double containment concept (RPV + C/V) was applied to the HTTR, because the safety features of the HTGR have not been developed in Japan. The HTTR has multiple barriers to fission product release, namely, the fuel coatings, the RPV boundary, the C/V and the reactor building. Most HTGRs being designed in other countries also include these barriers except the C/V. Some compartments surrounding the C/V in the reactor building serve as confinement, or service area. The service area is maintained at a slightly negative pressure to atmosphere by a ventilation and an air conditioning system during normal operation and accident conditions. The barriers of the C/V and the service area significantly reduce the off-site radiation dose in an accident condition such as a primary helium pipe rupture. Major components such as primary cooling system components as well as the RPV (Figure 2.3) are contained within the C/V.

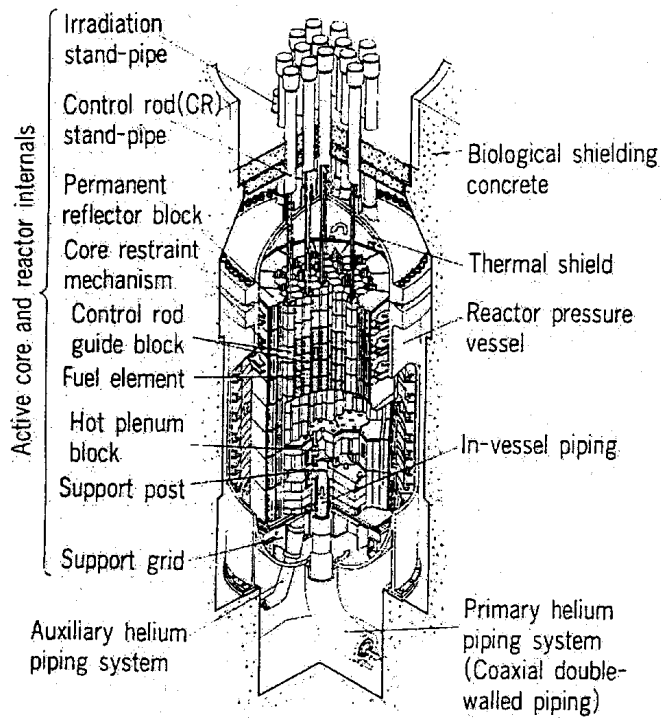


FIG. 2.3. HTTR pressure vessel and internals.

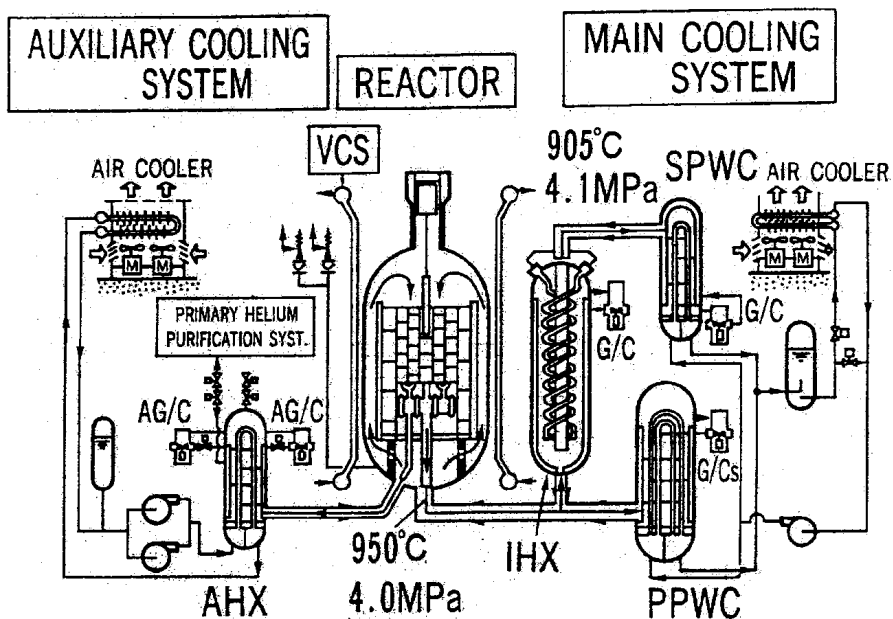


FIG. 2.4. HTTR cooling system.

The flow diagram of the reactor cooling system is shown in Figure 2.4. This system is composed of a main cooling system (MCS), an auxiliary cooling system (ACS) and two reactor vessel cooling systems (VCSs). The MCS removes the heat energy from the reactor core during the normal operation, while the ACS and VCSs are functioned as engineered safety features and remove the residual heat energy after a reactor scram. As the core restraint mechanism requires protection against thermal damage from reactor heat during normal operation and anticipated operational occurrences, the ACS functions as protection by forced cooling the restraint mechanism.

During a reactor scram, gas circulators of the MCS are shut to protect the heat transfer tubes of the two pressurized water coolers against overheating. In an AOO and accident condition when forced cooling of the core is available, the ACS automatically starts in response to the reactor scram signal. The VCS functions as a residual heat removal system when forced circulation in the primary cooling system is no longer available due to a rupture of its piping system. It also operates during normal operation to cool the reactor shielding concrete wall.

The MCS consists of an IHX, a primary pressurized water cooler (PPWC), a secondary pressurized water cooler (SPWC) and pressurized water/air cooler. The MCS has two operational modes; “single loaded operation”, and “parallel loaded operation”. The PPWC functions to remove the reactor heat of 30 MW during the single loaded operation, while during parallel load operation the IHX removes 10MW and the PPWC removes 20MW. The SPWC serves the function of removing the heat from the IHX. The heat removed by the PPWC and the SPWC is transported through the pressurized water at 3.5MPa. The pressurized water is then cooled down by the air cooler. In the HTTR reactor plant, the reactor heat of 30 MW is eventually transferred to the atmosphere by the pressurized water and the air cooler. During normal operation, the pressure of the secondary helium is controlled to always be 0.1 MPa higher than that of the primary helium at the IHX heat transfer tubes in order to reduce the pressure load on the tubes and to protect for accidental leakage of radioactive materials into the secondary helium. The water pressure is controlled so that a large amount of water can not ingress into the core with a PPWC tube rupture accident.

The auxiliary helium transfers a small fraction of the reactor heat to pressurized water. Eventually, the reactor heat is dissipated to the atmosphere at the auxiliary water/air cooler. The ACS consists of an auxiliary heat exchanger (AHX), two auxiliary helium circulators and an air cooler. At the AHX, the auxiliary helium is cooled by water. During normal operation, a small flow of auxiliary helium (~ 200 kg/h) passes through the AHX to the primary helium purification system so as to remove impurities contained in the reactor coolant. With a reactor scram, while the reactor coolant pressure boundary remains intact, the auxiliary helium cooling system automatically starts and transfers the residual heat from the core to the auxiliary air cooler. The AHX has heat transfer capacity of approximately 3.5MW.

Two vessel cooling systems are provided as protection of the reactor core and the RPV against thermal damage by residual heat after a reactor scram when the ACS cannot, or fails to, cool the core. Each of these systems is capable of controlling temperatures of the core and RPV within safe limits and consists of water-cooled panels surrounding the RPV with two cooling water systems. Cooling tubes with fins form the panels and are arranged so that adjacent tubes do not belong to the same system and a tube failure will not danger the RPV and core. The heat removal rate from the RPV to the panels is designed as 0.6 MW so as to effectively remove heat to meet the requirement for the maximum allowable normal fuel temperature of 1495°C and also 0.3 MW or more with an accident condition where the reactor core is not cooled by the ACS. The VCS is also an engineered safety feature equipped with two independent complete sets which are backed up with an emergency power supply. It is operated even during normal operation in order to cool the biological shielding concrete wall.

The IHX is a helically coiled counter flow type heat exchanger. To minimize constraints of axial and radial thermal expansion of the helically coiled heat transfer tubes, a floating hot header with a combination of a central hot gas duct passes through the central space inside the helix bundle. An assembled tube support allows free thermal expansion of a

helix in the radial direction. The primary helium enters the IHX through the inner pipe of the primary concentric hot gas duct attached to the bottom of the IHX. It flows up outside the tubes thereby transferring nuclear heat of 10 MW to the secondary helium and flows back to the annular space between the inner and outer shells, The secondary helium flows down inside the heat transfer tubes and flows up through the center as hot gas. A double-walled shell with thermal insulation attached to the inside surface of the inner shell provides reliable separation of the heat resisting and pressure retaining functions. Cold helium flowing through the annulus brings uniform temperature distribution throughout the outer shell which serves the function of being the pressure retaining member.

#### 2.1.1.4. HTTR core configuration

The annular core is one of the promising core types for future HTGRs because of high inherent safety characteristics for loss of coolant accidents [2-2, 2-3, 2-4]. The decay heat removal is enhanced by introduction of the annular core because the heat transfer pass will be shortened due to thinning of the fuel region. As a result, the fuel temperature on a loss of coolant accident can be maintained less than the fuel temperature limit of 1600°C by the vessel cooling system (VCS) surrounding the reactor pressure vessel. The decay heat will be transferred radially through the fuel regions, side reflector blocks and reactor pressure vessel to the cooling panel of the VCS by heat conduction, radiation and convection without any active cooling system.

Table 2-1. Specification of the HTTR

Thermal power	30 MW
Outlet coolant temperature	950°C
Inlet coolant temperature	395°C
Primary coolant pressure	4 MPa
Core structure	Graphite
Equivalent core diameter	2.3 m
Effective core height	2.9 m
Average power density	2.5 W/cm <sup>3</sup>
Fuel	UO <sub>2</sub>
Uranium enrichment	3 to 10 wt%
Type of fuel	Pin-in-block
Burn-up period (efpd)	660 days
Coolant material	Helium gas
Flow direction in core	Downward
Reflector thickness	
Top	1.16 m
Side	0.99 m
Bottom	1.16 m
Number of fuel assemblies	150
Number of fuel columns	30
Number of pairs of control rods	
In core	7
In reflector	9

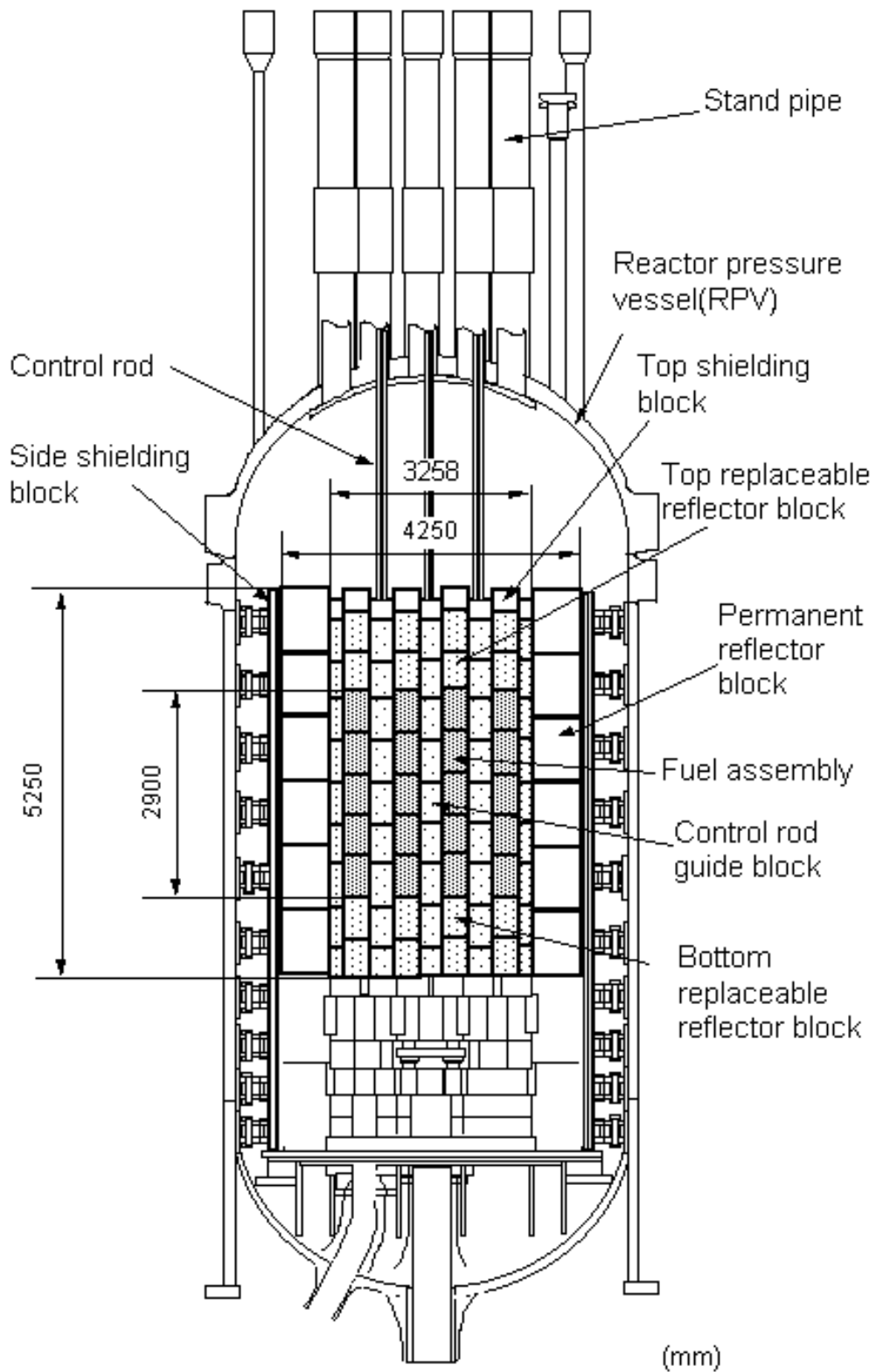


FIG.2.5. Vertical view of the HTTR.



Three different types of cores were formed during fuel loading for start-up core physics experiments; thin and thick annular cores were achieved at 18 and 24 fuel column loaded cores, respectively, and the fully-loaded core consists of 30 fuel columns. These three core types were proposed for benchmark problems within this CRP. The effective diameter and height of the HTTR are 230 and 290 cm, respectively. The core size of the HTTR corresponds to about one half of that of the future HTGRs [2-2, 2-3, 2-4]. Moreover, the high excess reactivity of the HTTR is similar to that of the future HTGRs because the large excess reactivity is necessary to compensate for power operation effects of temperature, xenon, burnup, etc.. Therefore, the benchmark problems of the HTTR's start-up core physics experiments are useful for verification of design codes of the future HTGRs.

The major specifications of the HTTR are given in Table 2-1 [2-1]. The reactor consists of core components and reactor internals. They are arranged in the reactor pressure vessel (RPV), which is 13.2m in height and 5.5m in diameter. Figure 2.5 shows the vertical cross section of the core and reactor internals structure. The core consists of core components which are prismatic hexagonal blocks 580mm in height and 360mm in width across the flats. These include fuel assembly blocks, control rod guide blocks, replaceable reflector blocks, and irradiation blocks. The core components are piled up cylindrically to form the core. The reactor internals consist of graphite and metallic core support structures and shielding blocks. They support and arrange the core components within the RPV.

The active core, 290cm in height and 230cm in effective diameter, consists of 30 columns and 7 control rod guide columns. Horizontal cross sections of the HTTR are shown in Figures 2.6 and 2.7. An additional 9 control rod columns are arranged among the adjacent reflector graphite columns. The replaceable reflector region adjacent to the active core consists of 9 control rod columns, 12 replaceable reflector columns, and 3 irradiation columns, which are surrounded by permanent reflector blocks. Each fuel column consists of 2 top reflector blocks, 5 fuel assemblies, and 2 bottom reflector blocks.

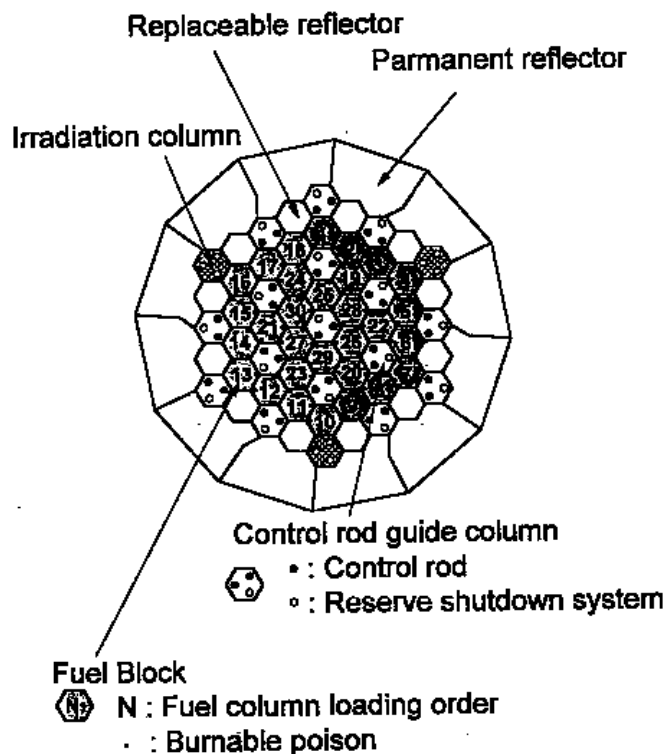


FIG. 2.6. Fuel column loading order and horizontal view of HTTR core.

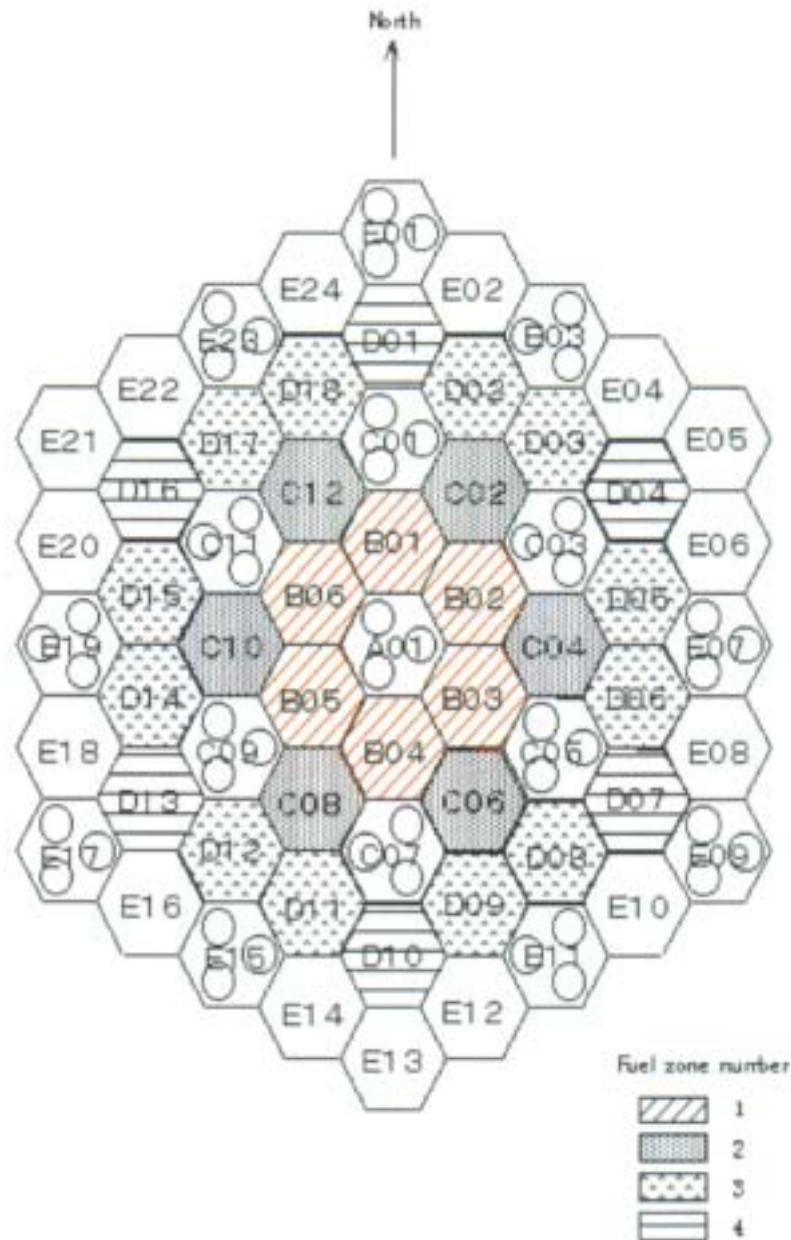


FIG. 2.7. HTTR fuel column name and zone number.

Figure 2.8 shows the structure of the pin-in-block type fuel. A fuel rod consists of a graphite sleeve containing 14 fuel compacts. The fuel rods are inserted into the coolant channels of the fuel graphite blocks. Each fuel compact contains about 13,000 coated fuel particles (CFPs) embedded in the graphite matrix. The number of uranium enrichments is 12. The highest and lowest enrichments are 9.9 and 3.4 wt%, respectively. Fuels of higher-enriched uranium are placed in the upper and outer core regions to reduce the maximum fuel temperature. Burnable poisons (BPs) made of boron carbide and carbon are inserted into two of three holes below the dowel pins in the fuel graphite block. The coolant gas flow is downward through annular channels formed by the graphite block and the fuel rod.

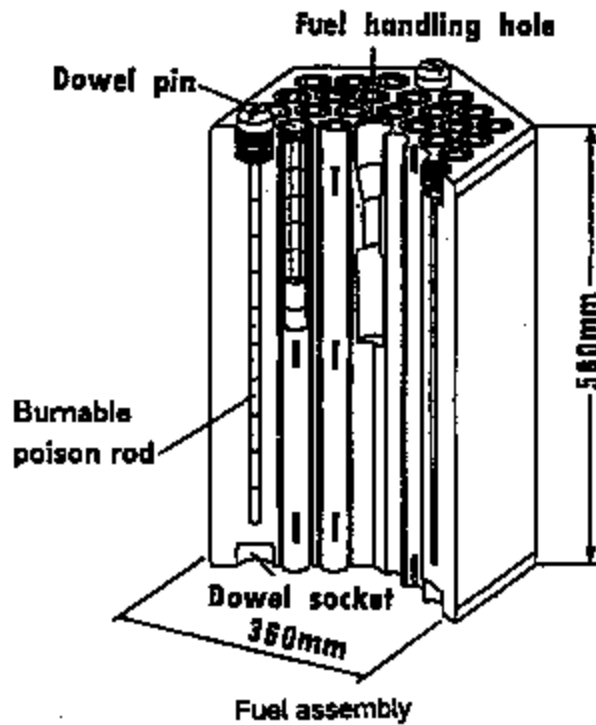
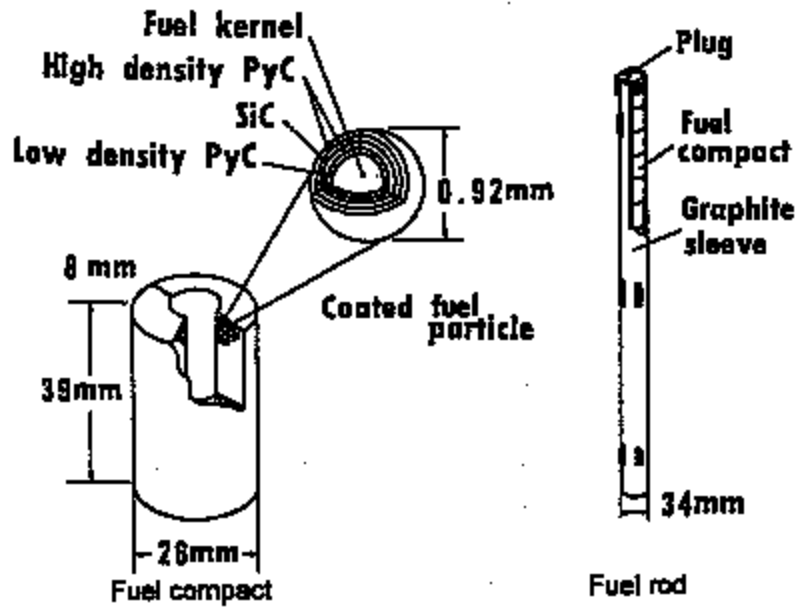


FIG. 2.8: Structure of fuel assembly.

The top of the control rod guide columns and irradiation columns are 100mm lower than that of the fuel columns because the 9<sup>th</sup> layer block of these columns are 480mm in height. Each horizontal gap width between two columns is 2mm in average.

Reactivity is controlled by 16 pairs of control rods. A pair of control rods is individually moved by a control rod drive mechanism located in stand-pipes connected to the hemispherical top head closure of the RPV. The control rods are inserted into two channels of the control rod guide columns in the active core and in the replaceable reflector regions.

## ***Fuel loading scheme***

Before fuel loading, the whole fuel region in the core is filled with graphite dummy blocks. There are two types of dummy blocks: graphite blocks with three large holes and those with three small holes. The graphite blocks with three large holes were actually used as control rod guide columns for out-pile seismic tests of the core structure. The others with three small holes were newly produced as dummy fuel blocks. The primary coolant system is filled with helium at normal atmospheric pressure and is not in operation during fuel loading. The core is at room temperature. Fuel loading is carried out by replacing the dummy blocks with the fuel assemblies, column by column. The fuel loading scheme is shown in Figure 2.6. The fuel blocks are loaded from the periphery to the center, and thin and thick annular cores are made at 18 and 24 fuel-column-loaded core, respectively.

The approach to criticality was observed by monitoring the inverse of the neutron multiplication fraction. The core is regarded as critical when the neutron density is maintained constant after removing the temporary neutron source. After the first criticality, the increment in reactivity was measured by the inverse kinetic method. The excess reactivity of the core was obtained by adding all increments of the reactivity from the first criticality to the fully-loaded core.

## ***Configuration and composition of components***

For the calculations, the detailed data of the active core, the adjacent replaceable reflector region and the permanent reflector region were necessary. Configuration of the components in the above mentioned regions is described as follows: The position of blocks in the core is noted by vertical position number and column number. The vertical number is 1, 2, . . . , 9 as from the top blocks in the 1<sup>st</sup> layer to the bottom blocks in the 9<sup>th</sup> layer, with the column number named according to Figure 2.9. For example, “4C05“ means that the block is placed at the 4<sup>th</sup> block from the top, the 2<sup>nd</sup> ring from the core center, and the 5<sup>th</sup> block from the north in clockwise direction. In all, 30 fuel columns are grouped concentrically into 4 fuel zones as shown in Figure 2.9.

## ***Fuel assembly***

A fuel assembly consists of fuel rods, two burnable poison (BP) rods and a fuel graphite block. Each fuel rod consists of a graphite sleeve and 14 fuel compacts containing coated fuel particles (CFPs). The fuel rods are inserted into vertical holes of 41mm diameter in the fuel graphite block and form annular coolant channels between the vertical holes and the fuel rods. Figure 2.8 shows the structure of fuel assembly. There are two types of fuel graphite blocks having 31 or 33 fuel rods. The number of different uranium enrichments are 12 in the core. The uranium enrichment of all compacts in a fuel assembly is not changed. The fuel assembly is classified by the uranium enrichment, the number of fuel rods and the type of BPs. The fuel assembly arrangement in the core is shown in Figure 2.9.

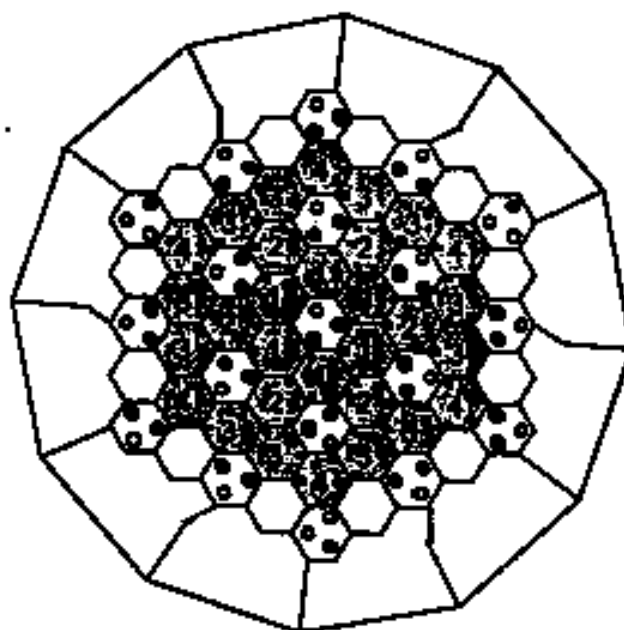
A CFP consists of a spherical fuel kernel of low enriched  $\text{UO}_2$  with TRISO coating. The TRISO coating consists of a low-density, porous pyrolytic carbon (PyC) buffer layer adjacent to the fuel kernel, followed by high density isotropic PyC layer, a SiC layer and a final outer PyC layer. The CFPs are embedded in graphite matrix of the fuel compact. The CFPs are classified into 12 sorts by their uranium enrichments. The form of the CFPs is given in Figure 2.10.

Layer number from top block	Layer number from top fuel block	Items	Fuel zone number			
			1	2	3	4
3	1	Uranium enrichment (wt%)	6.68	7.82	8.98	9.61
		Number of fuel rod of graphite block	33	33	31	31
		Type of burnable poisons	H-I	H-I	H-I	H-I
4	2	Uranium enrichment (wt%)	5.18	6.25	7.19	7.82
		Number of fuel rod of graphite block	33	33	31	31
		Type of burnable poisons	H-II	H-II	H-II	H-II
5	3	Uranium enrichment (wt%)	4.29	5.18	5.91	6.25
		Number of fuel rod of graphite block	33	33	31	31
		Type of burnable poisons	H-III	H-III	H-III	H-III
6	4	Uranium enrichment (wt%)	3.30	3.86	4.29	4.79
		Number of fuel rod of graphite block	33	33	31	31
		Type of burnable poisons	H-I	H-I	H-I	H-I
7	5	Uranium enrichment (wt%)	3.30	3.86	4.29	4.79
		Number of fuel rod of graphite block	33	33	31	31
		Type of burnable poisons	H-I	H-I	H-I	H-I

Note:

Fuel zone number is shown in the below figure.

Type of burnable poisons is shown in Table 2-3.



 N : Fuel zone number  
 : Burnable poison

FIG. 2.9. Fuel assembly arrangement in the HTTR core.

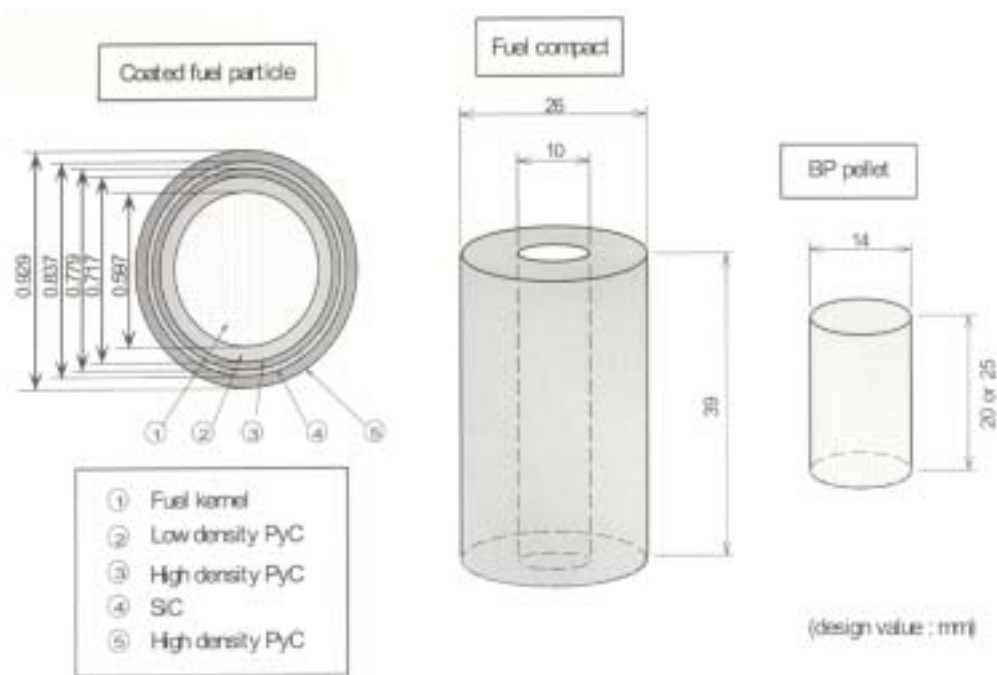


FIG. 2.10. Coated fuel particle, fuel compact and burnable poison pellet.

The fuel compact consists of CFPs and graphite matrix. The form is a hollow cylinder of 10mm in inner diameter, 26mm in outer diameter and 39mm in height.

The fuel rod consists of a graphite sleeve containing 14 fuel compacts. It is inserted into the coolant channel of the fuel graphite block. The form of the fuel rod is given in Figure 2.11. The fuel graphite block is a prismatic hexagonal block 580mm in height and 360mm in width across the flats. The block has 33 fuel holes in the fuel zones 1 and 2, and 31 fuel holes in the fuel zone 3 and 4. The fuel zone numbers are defined in Fig. 2.7. The form of the block is given in Figure 2.12 and Figure 2.13.

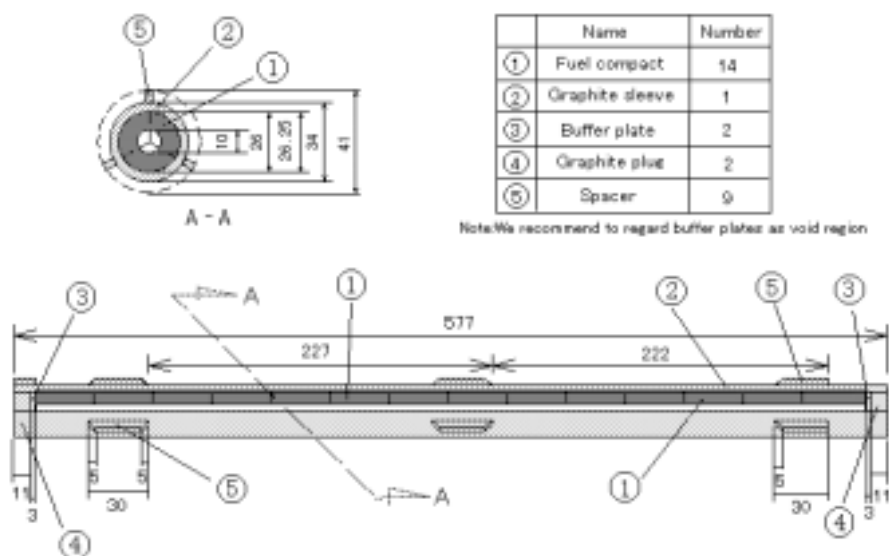


FIG. 2.11. Fuel rod.

TABLE 2 Burnable poisons and Graphite disks

Burnable poison pellet

Material	B <sub>4</sub> C-C composite	
Boron arrangements	shown in Fig.5	
Type	H-I	H-II
Density (g/cm <sup>3</sup> )	1.79	1.82
Natural boron concentration (wt%)	2.22	2.74
Diameter (mm)	13.9	13.9
<sup>10</sup> B abundance ratio (wt%)	18.7	18.7

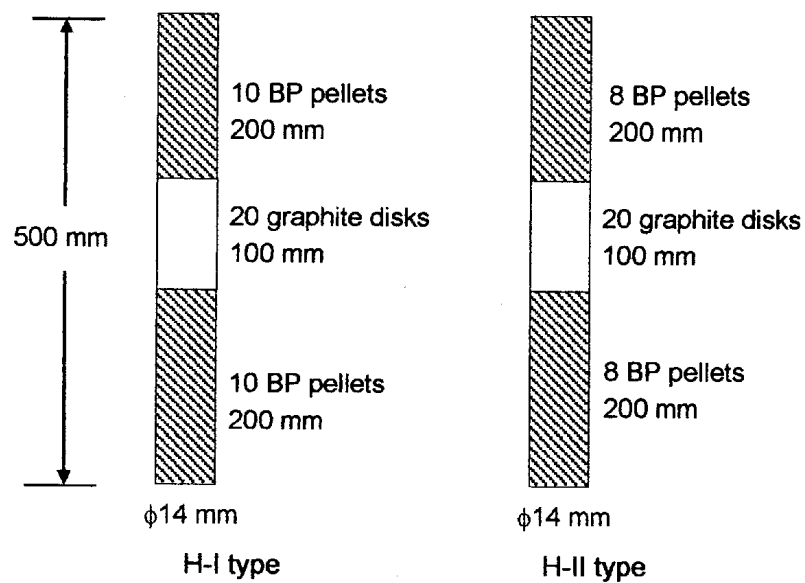
Graphite disk

Material	Graphite	
Diameter (mm)	14.0	
Density (g/cm <sup>3</sup> )	1.77	
Impurity (ppm)*	0.37	* Natural boron equivalent

Burnable poison rod

Number of BP rods in a fuel assembly	2 (shown in Fig.2)
Height	500 mm

Configuration of BP pellets and graphite disks in a BP rod are shown in the following figures



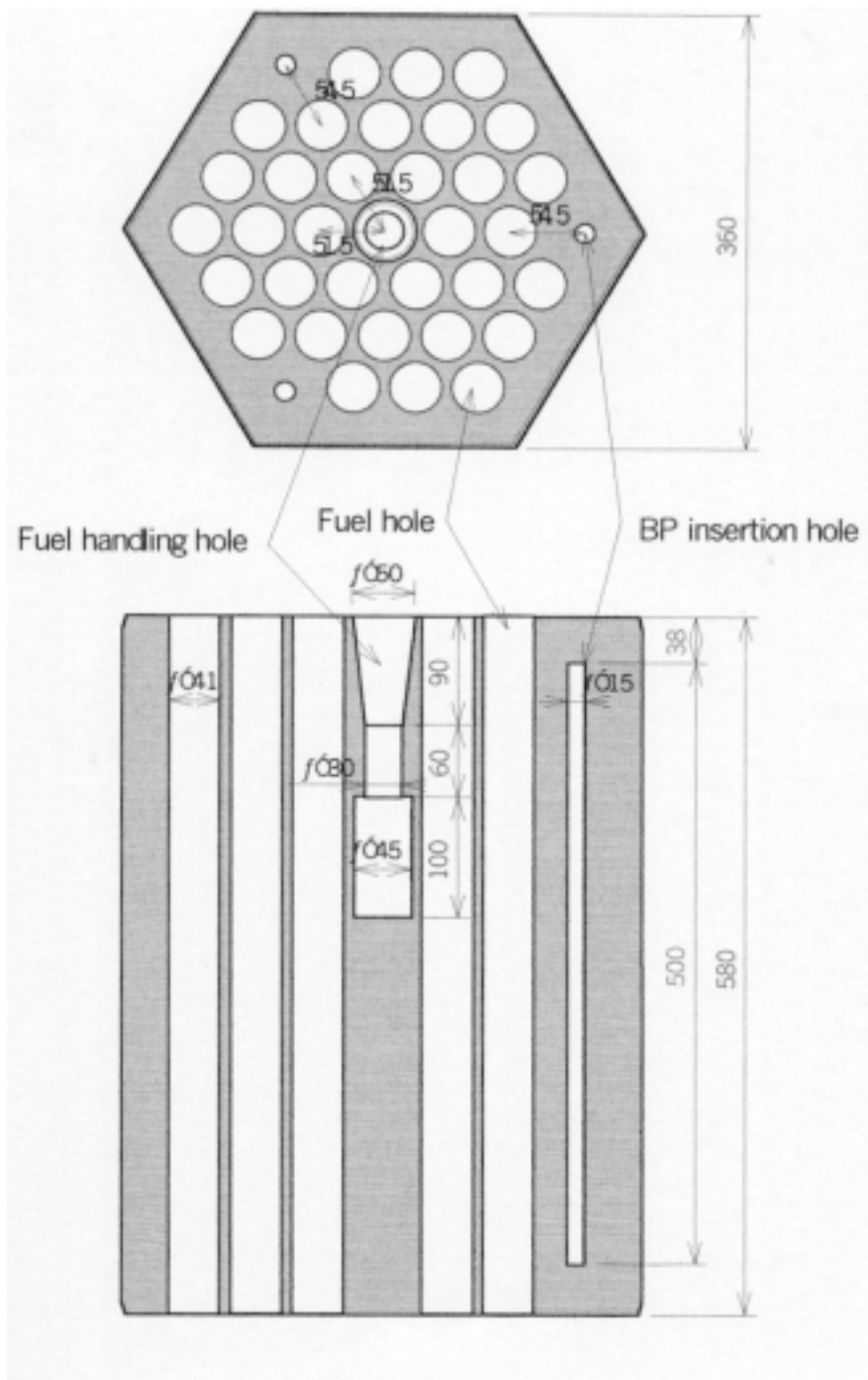


FIG. 2.12. Fuel block for 33 pin fuel assembly.



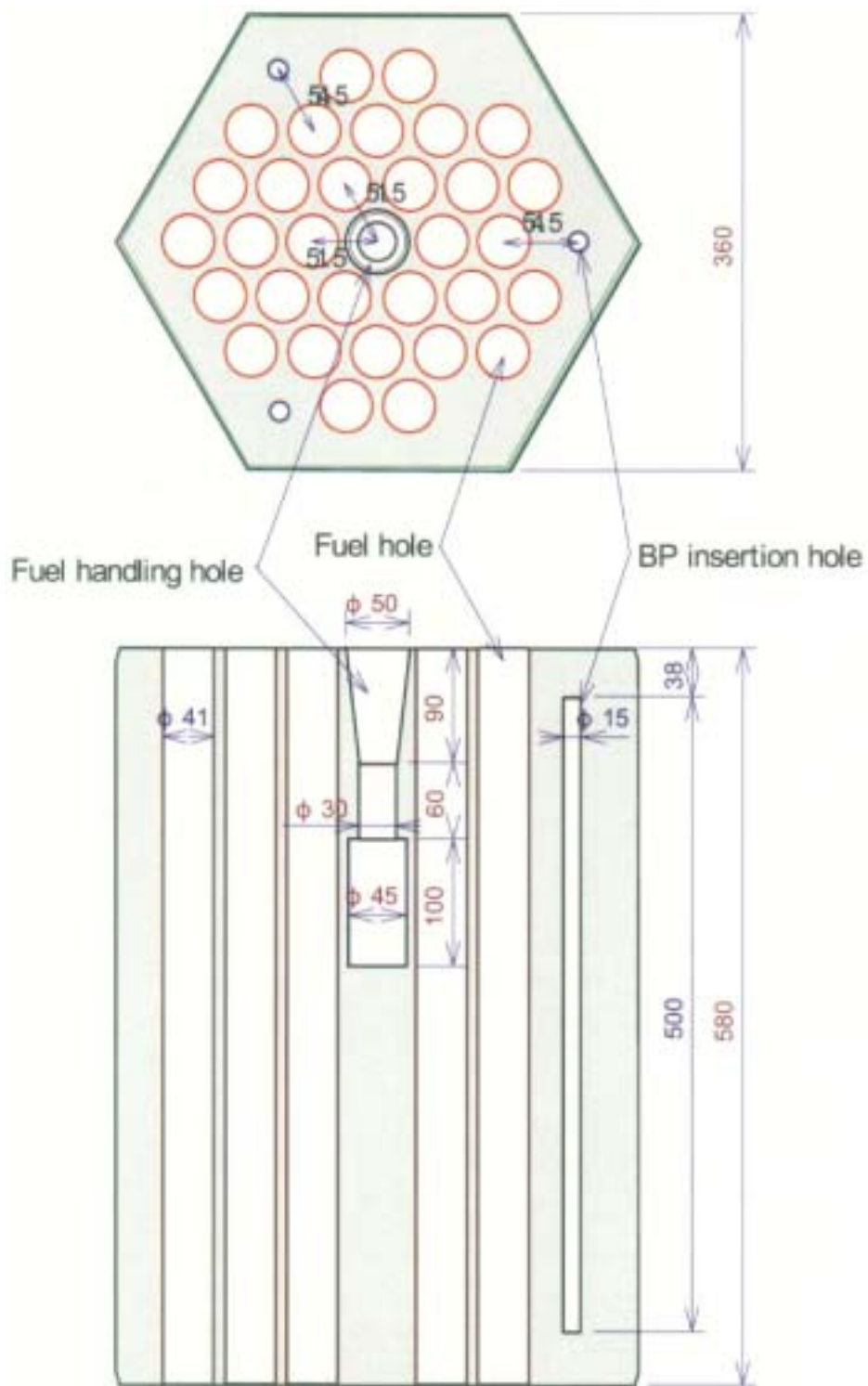


FIG. 2.13. Fuel graphite block for 31.pin fuel assembly.

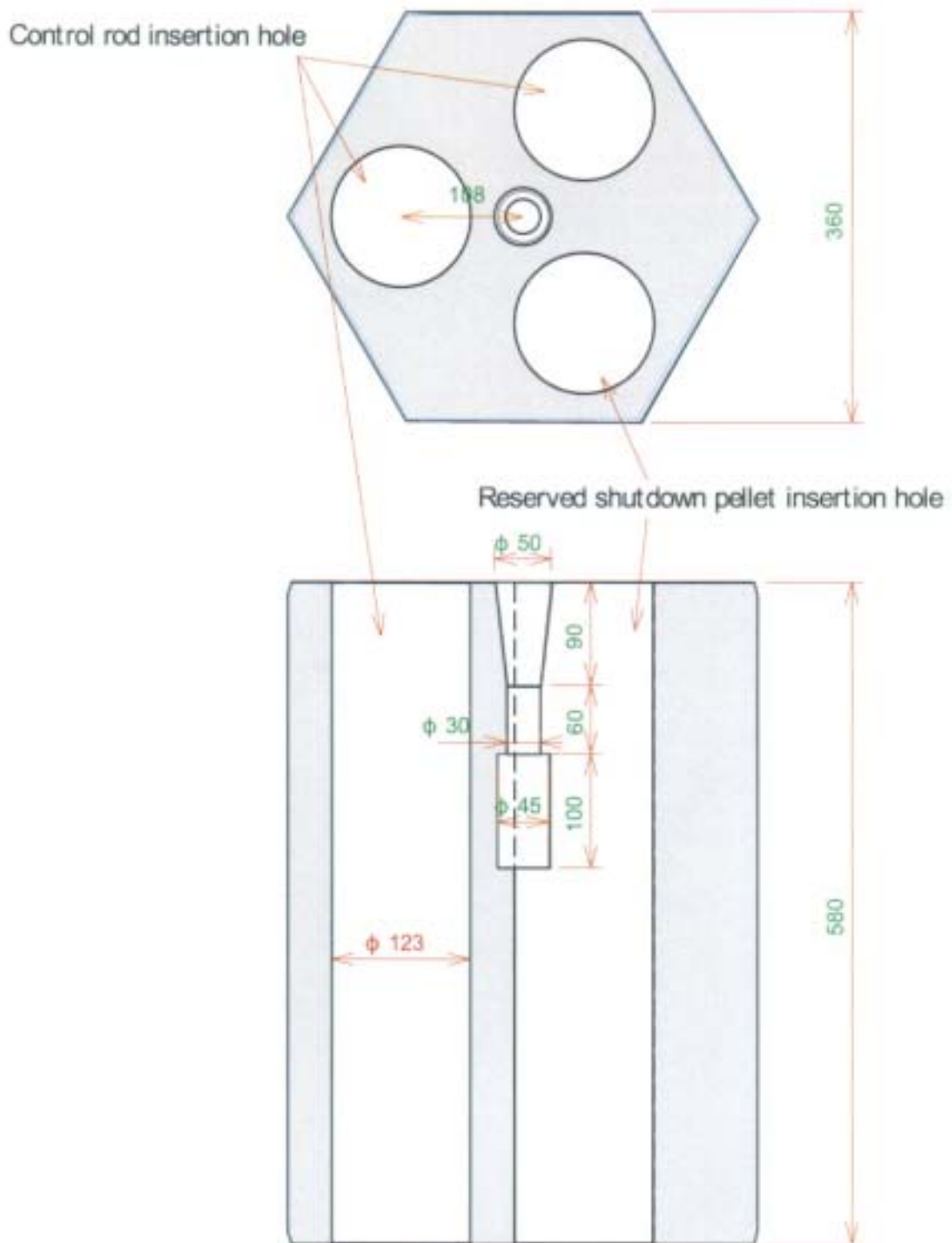
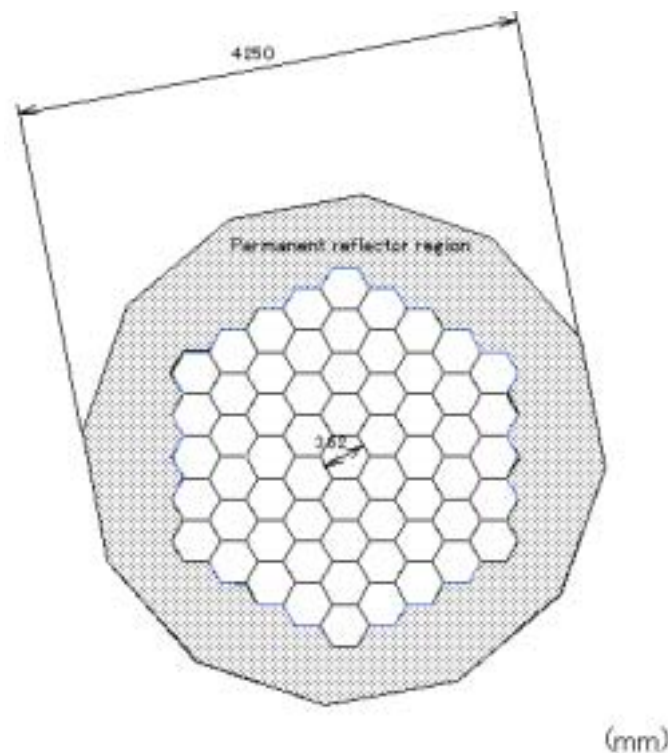


FIG. 2.14. Control rod guide block in layers 1 through 5.

Burnable poison (BP) rods are inserted into two of three BP insertion holes of a fuel graphite block as shown in Fig. 2.8. The BP insertion holes are under the 3 dowel pins of the fuel graphite block and are 15mm in inner diameter and 500mm in length. One hole without the BP rod is empty. Figure 2.6 shows positions of holes which are filled with BPs. The diameter of BP rod is 14mm. The BP rod consists of BP pellets and graphite disks as shown in Table 2.2. The graphite disks are put between the BP pellets. The compositions of the BP pellets are  $B_4C$  and C. There are two types of BP pellets: H-I with 2.22wt% of natural boron concentration and H-II with 2.74wt%. The form of the BP pellets and graphite disks is given in Figure 2.10.

### ***Control rod, reflector and dummy fuel blocks***

There are 16 control rod guide columns each consisting of 9 control rod guide blocks. The control rod guide blocks have three holes. Two of them are control rod insertion holes. The third hole is a reserve shutdown system (RSS) hole which is used for emergency. Figure 2-6 shows the position of the control rod insertion holes and RSS holes in the core and the replaceable reflector region. Top and bottom replaceable reflector blocks are placed above and below the fuel assemblies. The arrangement of coolant channels (with 23mm in inner diameter) in the top replaceable reflector blocks corresponds to that of coolant channels (with 41mm in inner diameter) in the fuel assemblies within the same column. The upper bottom replaceable reflector block (the 8<sup>th</sup> layer) has the same arrangement of coolant channels as the top replaceable reflector block. The lower bottom replaceable reflector block (the 9<sup>th</sup> layer) has 6 large coolant channels.



*FIG. 2.15. Permanent reflector region.*

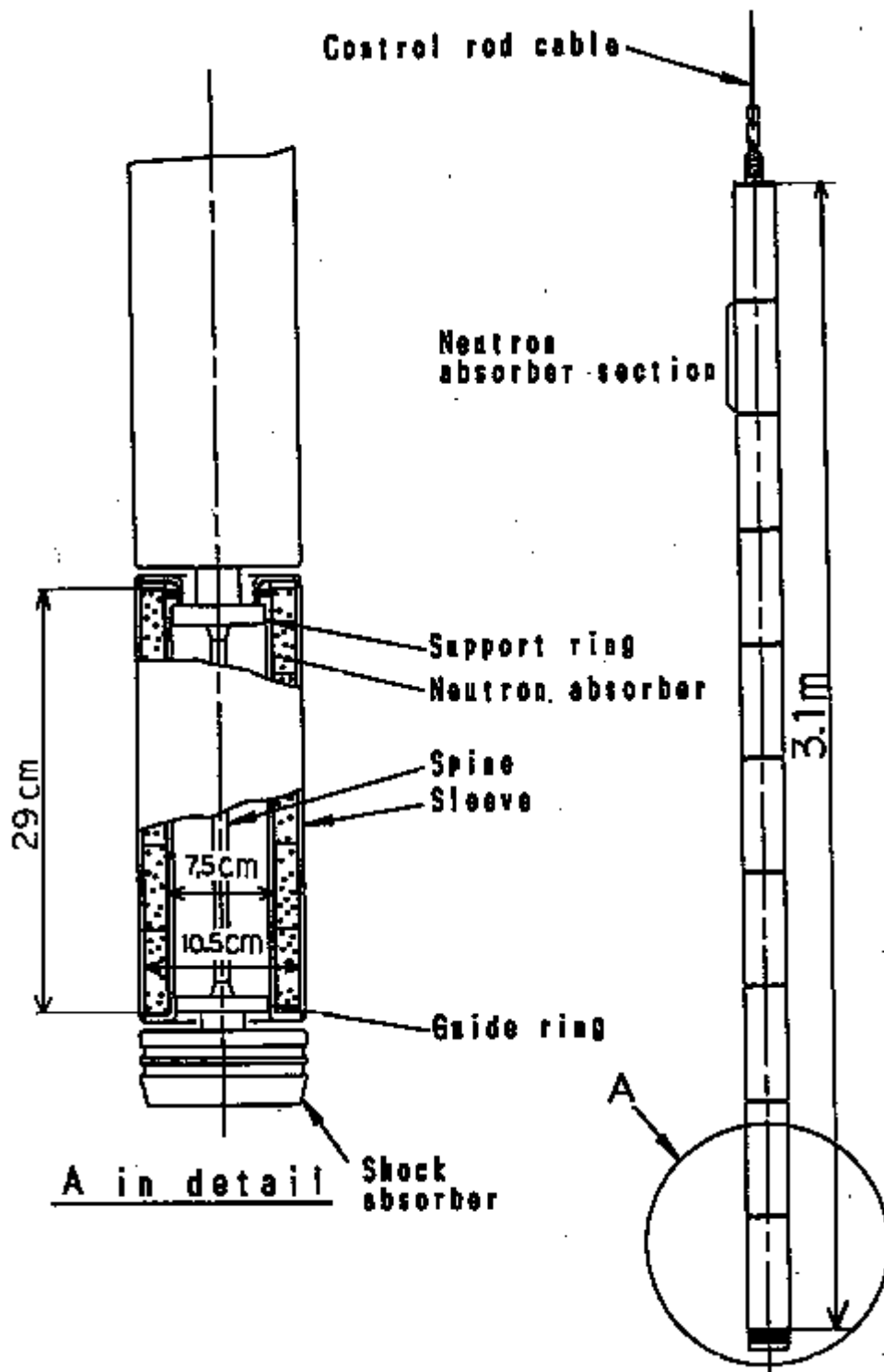


FIG. 2.16. Schematic of control rod.

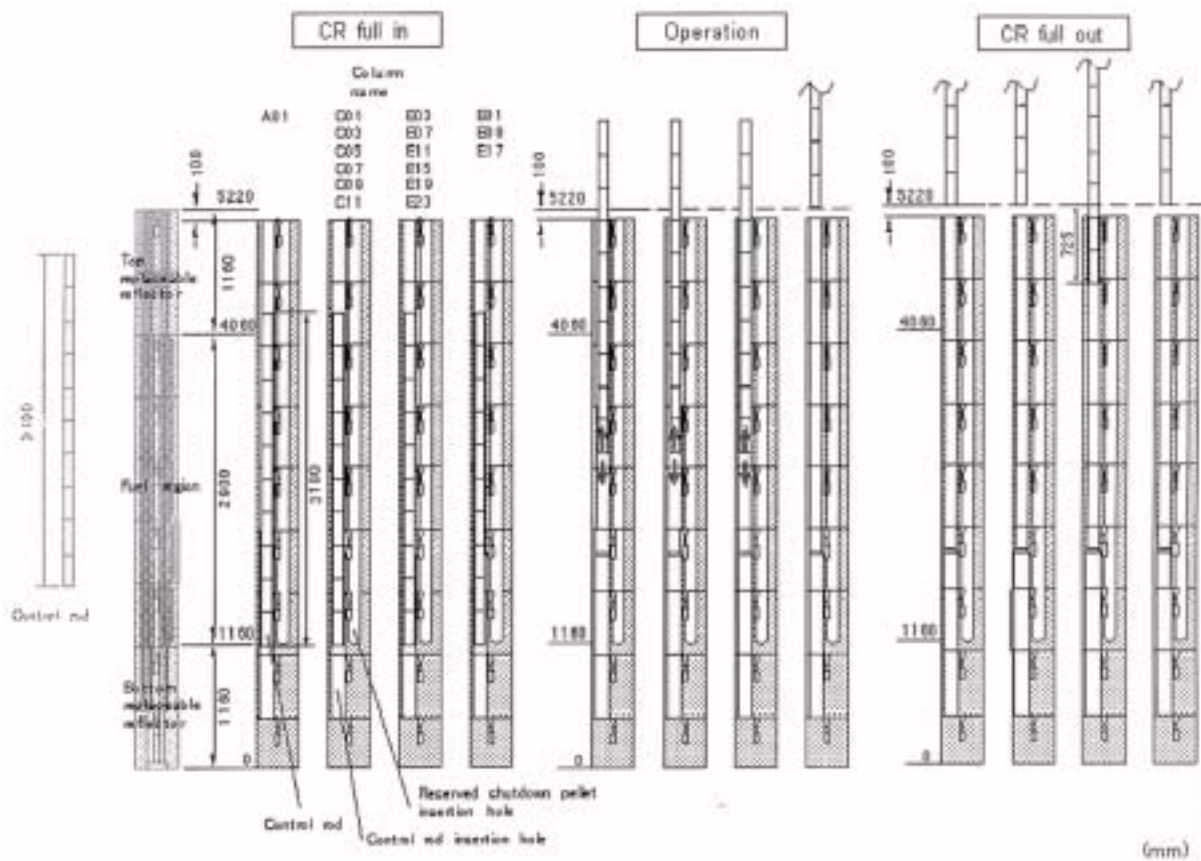


FIG. 2.17. Axial control rod position.

Dummy fuel blocks are placed in the active core before fuel loading as substitution for fuel assemblies. The external form of the dummy fuel block is the same as that of a fuel graphite block. The dummy fuel blocks contain higher impurity than the fuel graphite blocks. All dummy fuel blocks are replaced by fuel assemblies. All blocks except the bottom block of the control rod guide column and the irradiation column are 58 cm in height and 36 cm in width of across flats. The bottom blocks of the control rod guide and irradiation columns are 48 cm in height and 36 cm in width across the flats. A typical form of the control rod guide blocks for layers 1 through 5 is shown in Figure 2-14. Boron pins are installed into the lower parts of blocks in the 9<sup>th</sup> layer for neutron shielding. For benchmark problems, the neutron shielding pin region is considered as a black absorber or vacuum in calculation models.

Core components are horizontally surrounded by 12 permanent reflector blocks whose form is a large polygonal graphite block. The width across the flats of the core, including permanent reflector blocks, is 4250 mm as shown in Figure 2.15. The permanent reflector blocks have holes for irradiation tests and neutron detectors. The void fraction is 0.7 % due to the holes. Around the permanent reflector blocks, there are side shielding blocks consisting of B<sub>4</sub>C/C. The structures outside of the permanent reflector region were neglected in the calculation models.

### Control Rods

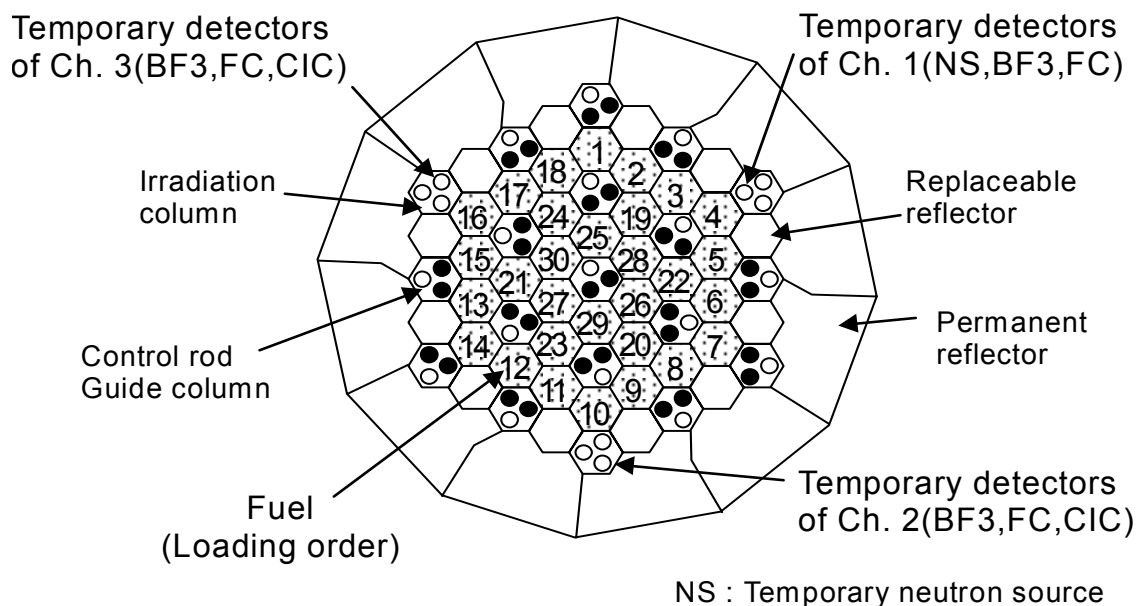
There are 16 pairs of control rods, 7 in the active core and 9 in the replaceable reflector region. A pair of control rods is inserted into holes of a control rod guide column.

Each control rod consists of 10 neutron absorber sections connected with metallic spines and support rings. Each section contains 5 sintered compacts of  $B_4C$  and C as neutron absorber in the annular space. The schematics of the control rod are shown in Figure 2. 16. The axial positions of the control rods are divided into three types as shown in Figure 2.17. When the control rods are fully inserted, the lower ends of all control rods are on the same plane with the bottom face of the 7<sup>th</sup> layer of a fuel column. In the first approach to criticality three pairs of control rods at columns E01, E09 and E17 in Figure 2.7 are fully withdrawn. These are not used to operate the reactor. The rest of 13 pairs of control rods are the withdrawn during the approach to criticality. At the critical condition, all control rod pairs except those of columns E01, E09, and E17 are adjusted so that their insertion depth will be the same. This insertion level is evaluated in the benchmark test HTTR-CR. When the control rods are fully withdrawn, their upper limit is the upper face of the 1<sup>st</sup> replaceable reflector block (over the fuel region) with the exception of those control rods (R2) at columns E03, E07, E11, E15, E19, and E23 in Figure 2.7. The control rods of these six columns have upper limits at 725mm below the top of the 1<sup>st</sup> block.

Further detailed data of the core, fuel assemblies and core internal components have been provided in [2-5].

### ***Nuclear Instrumentation***

Temporary neutron instrumentation consisting of three BF<sub>3</sub> counters, three fission counters (FC), two gamma-ray-compensated ionization chambers (CIC) was used for the startup core physics tests. Positions of detectors were not arbitrarily chosen like liquid (water)-moderated reactors. Only the existing holes were available for the solid (graphite)-moderated reactor like the HTTR. However, the holes for control rods (CRs) and reserve shutdown systems could not be used due to safety-related restrictions. Thus the detectors are located in three irradiation columns as shown in Figure 2.18.



*FIG. 2.18. Cross section of the HTTR and fuel loading order.*

The BF<sub>3</sub> is used for monitoring increase in the neutron density. The inverse of the multiplication factor was evaluated from the change in neutron density. The FC was used for measurement of the axial neutron flux distribution. The CIC was used for reactivity measurement by the Inverse Kinetic (IK) method. A temporary neutron source (Am-Be, 1.48×10<sup>4</sup>Bq) was inserted in irradiation test column. The vertical position of each detector is shown in Figure 2.19. The BF<sub>3</sub> in Ch.1 is installed about 2 m apart from the neutron source to avoid the effect of direct-flying neutrons from the source. BF<sub>3</sub> and FC in the same channel are located at the same vertical position in order to have redundancy for 1/M monitoring. FCs were movable in the vertical direction to measure the axial neutron flux distribution. The CICs were fixed in the core. The temporary neutron source was withdrawn from the core to exclude the neutron source effect on the measurement.

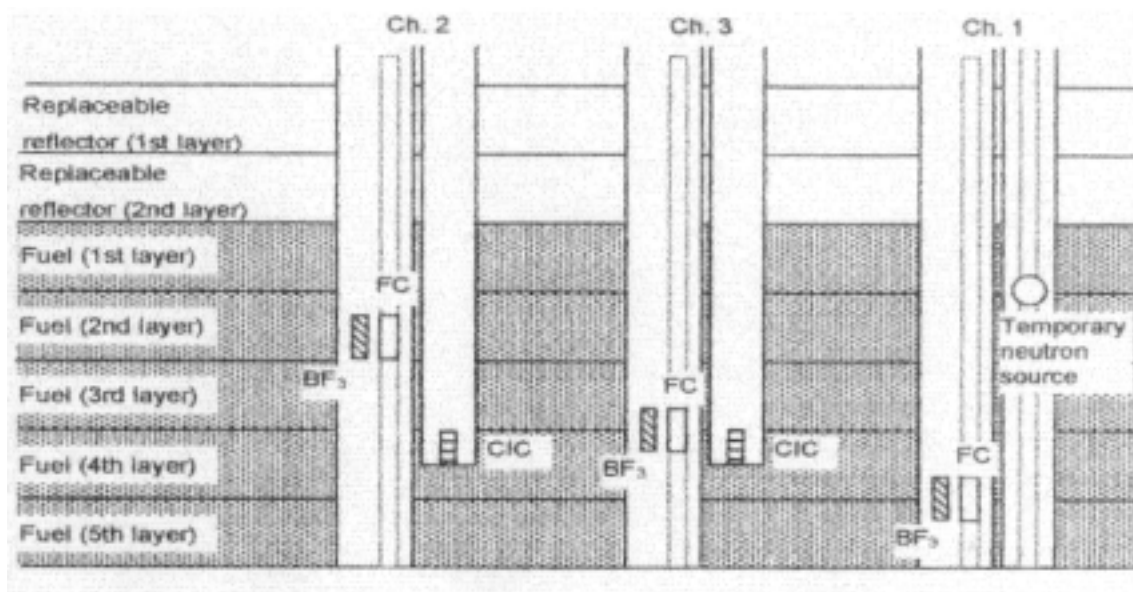


FIG. 2.19. Vertical location of temporary neutron detector system.

## 2.1.2. Benchmark Problem Descriptions

The benchmark problems addressed within this CRP include HTTR related start-up core physics tests and thermal hydraulics. General descriptions of these problems are included below. Details of the methodologies utilized and the results obtained for each problem by the individual CSIs are presented herein by each Member State.

### 2.1.2.1. Initial Criticality – [HTTR-FC] (Phase 1)

The number of fuel columns are evaluated for the first criticality, with the fuel columns charged from the outer region of the core. They are loaded clockwise, one by one. A small excess reactivity at the first criticality is also evaluated.

#### *2.1.2.2. Initial Criticality – [HTTR-FC] (Phase 2)*

The following effects are considered in the Phase 2 to improve the benchmark problem calculation accuracy:

- 1) Air in void of graphite
- 2) Revised impurity contents in dummy block
- 3) Aluminum in the temporary neutron detector holders.

#### *2.1.2.3. Control Rod Position at Criticality – (HTTR-CR)*

The control rod insertion depths are evaluated at the critical condition for the following three cases. All control rod insertion levels are adjusted on the same level except three pairs of control rods in the most outer region in the side reflectors. These three pairs of control rods should be fully withdrawn for the calculation.

- 1) 18 columns (thin annular core)
- 2) 24 columns (thick annular core)
- 3) 30 columns (fully-loaded core)

#### *2.1.2.4. Excess Reactivity – (HTTR-EX)*

The excess reactivity is evaluated for the three cases mentioned above. The room temperature of 300K is to be assumed as the moderator and fuel temperatures for the benchmark problem. One atmospheric pressure of helium is to be used as the primary coolant condition.

#### *2.1.2.5. Scram Reactivity – (HTTR-SC)*

The Scram reactivity is to be evaluated for the following two cases:

- 1) All reflector CRs are inserted at the critical condition
- 2) All CRs in reflector and core are inserted at the critical condition

The core condition for this benchmark problem is as follows:

- Fully-loaded core (30 column fuel core)
- Fresh fuel core

#### *2.1.2.6. Isothermal Temperature Coefficient (HTTR-TC)*

Isothermal temperature coefficients for the fully-loaded core are to be evaluated from the effective multiplication. The critical control rod positions are changed with temperature elevation in the real reactor operation. However, the control rod position is not to be changed in the calculation to obtain the reactivity difference. Critical control rod positions are to be evaluated at temperature of 480K.



## 2.2. REACTOR PHYSICS BENCHMARK ANALYSIS AND RESULTS

### 2.2.1. Japan

#### 2.2.1.1 Analysis methodology and model description

##### *Diffusion calculation [2-9, 2-10]*

The calculations for the benchmark problems were carried out using a nuclear characteristics evaluation code system which was developed from an HTTR nuclear design code system [2-11]. The code system consists of the DELIGHT [2-12], TWOTRAN-II [2-13] and CITATION-1000VP [2-14] codes. An outline of the calculation codes and model is summarized in Table 2-3. The program structure of the system is shown in Figure 2.20.

Table 2-3. Codes, models and nuclear data library for diffusion calculation

Items	Name of country	Japan
	Name of Institute	JAERI
Nucl. Data file	ENDF/B-III, IV	
Fuel cell code	DELIGHT	
Theory	Collision probability	
Model	Pin cell	
Cut off energy	2.38eV	
No. of groups	40	
BP cell code	TWOTRAN-II	
Theory	Transport	
Model	2-D(r-Z)	
No. of groups	6	
Control rod cell code	TWOTRAN-II	
Theory	Transport	
Model	2-D (x-y)	
No. of groups	6	
Core cal. Code	CITATION	
Model	3-D (Triangle)	
No. of groups, (Fast +Thermal)	6, (3+3)	

DELIGHT is an one-dimensional lattice burnup cell calculation code that has been developed in JAERI. TWOTRAN-II is a transport code that was used to provide average group constants of burnable absorber (BP) in fuel blocks and graphite blocks where control rods (CRs) are inserted. CITATION-1000VP is a reactor core analysis code. This code was developed from CITATION [2-15] so that nuclear characteristic analyses could be carried out with a three-dimensional whole core model of the HTTR in a short calculation time.

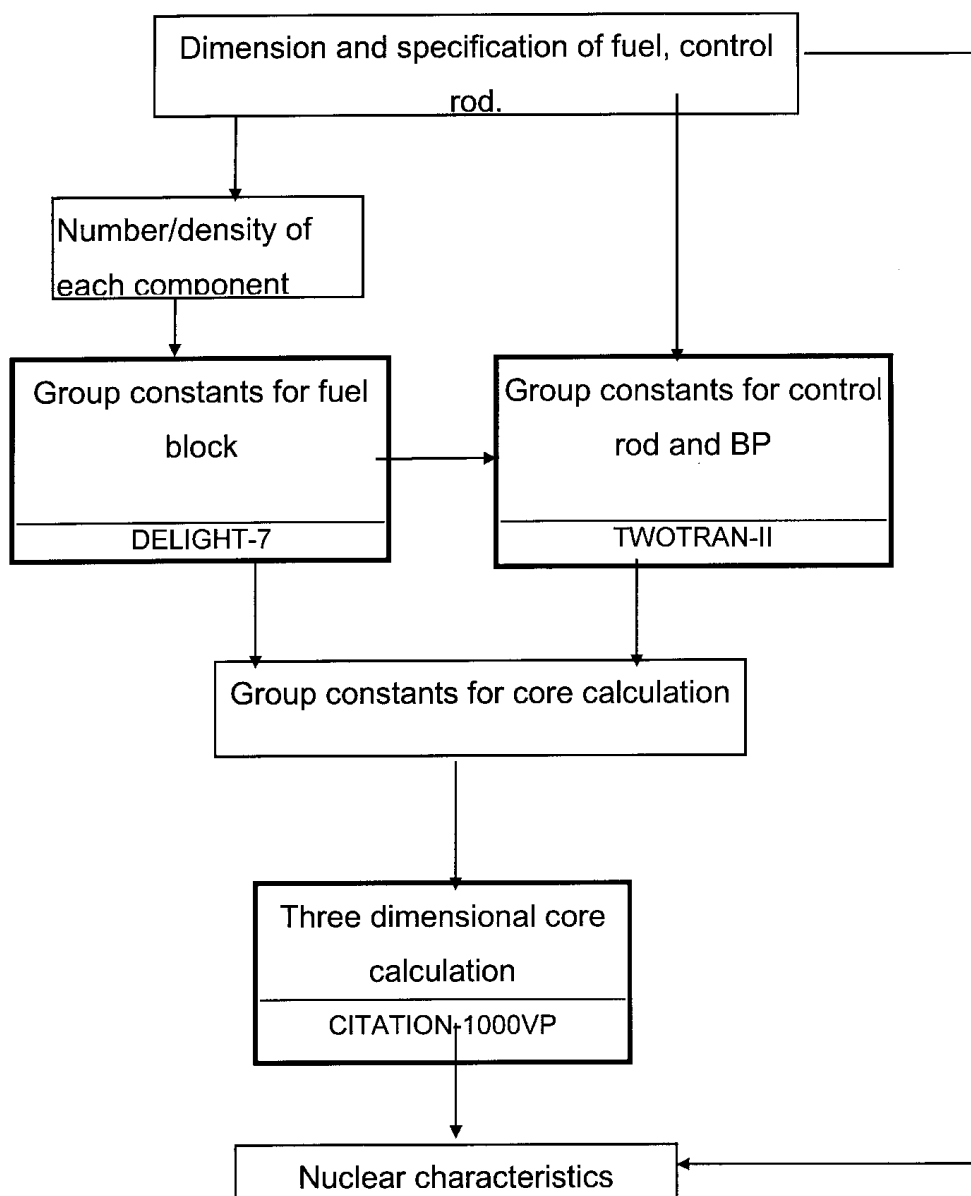


FIG. 2.20. Program structure of the HTTR nuclear characteristics evaluation code system.

DELIGHT was used to provide group constants of fuel and graphite blocks for succeeding core calculations. Resonance, neutron spectrum, neutron flux distribution, criticality, and burn-up calculations were done sequentially. Nuclear data were based on ENDF/B-IV except burn-up chain data that were extracted from ENDF/B-III. In the resonance range, the code employs intermediate resonance approximation and can consider the effect of a double heterogeneity caused by coated fuel particles (CFPs) and assembled fuel rods. The average group constants of the whole fuel block were obtained by a fuel cell calculation as follows; The group constants of the fuel rods were calculated by using a one-dimensional cylindrical fuel cell model as shown in Figure 2.21. The fuel rods in a block are located in the inner position of the fuel block, and the outer region of a fuel block is graphite rich region. To simulate the harder neutron spectrum, the area of cross section of the fuel cell was determined by the pitch of fuel rods. The pitch of fuel rods is 5.15cm. Therefore, the outer radius of fuel cell is about 2.7cm.

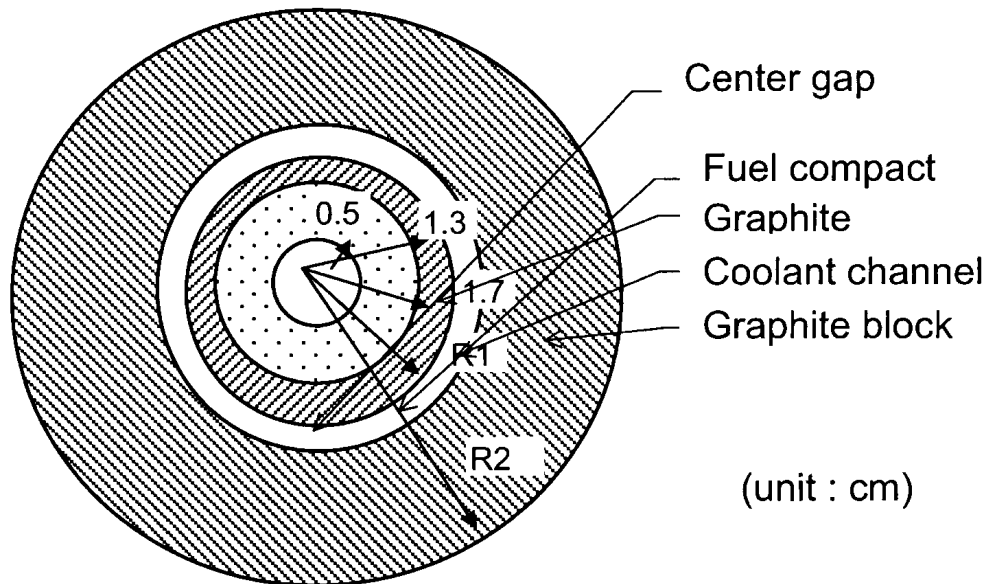


FIG. 2.21. Fuel cell model for DELIGHT code.

The group constants for BP and CR were calculated by TWOTRAN-II code.

#### BP cell model

The BP of the HTTR is a zebra type configuration. The average group constants of the zebra type BP was calculated by two-dimensional r-Z model of TWOTRAN-II code. The BP cell model is shown in Figure 2.22. The BP cell model corresponds to a quarter of a fuel block which contains half of the BP rod. BP rod consists of BP pellets and graphite pellets which are surrounded by graphite, the outer region contains homogenized fuel. Macroscopic cross section sets for the calculation were provided by the DELIGHT code.

In order to obtain the effective microscopic cross section, a homogenized region is determined to have the same cross section as the BP region in the core calculation model. Three kinds of  $\sigma_a$  for  $^{10}\text{B}$  in the BP rod are evaluated for core calculations. The first one was for the fuel block with 7.9% enrichment of  $^{235}\text{U}$ , 33 fuel pins and 2.0wt% of boron concentration in the BP rods. This  $\sigma_a$ -set was used for all BPs in the first layer of fuel blocks. The second one was for the fuel block with 6.3% enrichment of  $^{235}\text{U}$ , 33 fuel pins and 2.5wt% of boron concentration in the BP rods which was used for all BPs with 2.5wt% of boron concentration. The third one was for the fuel block with 3.9% enrichment of  $^{235}\text{U}$ , 33 fuel pins and 2.0wt% of boron concentration in the BP rods. This kind of  $\sigma_a$  was used for all BPs in the 4th and 5th layer of fuel blocks.

#### CR cell model

The average group constants of a pair of CRs and of the corresponding graphite block were obtained with the flux-weighting method. The neutron fluxes were calculated with a two-dimensional x-y model. The model is based on half of the graphite block where one CR is inserted.

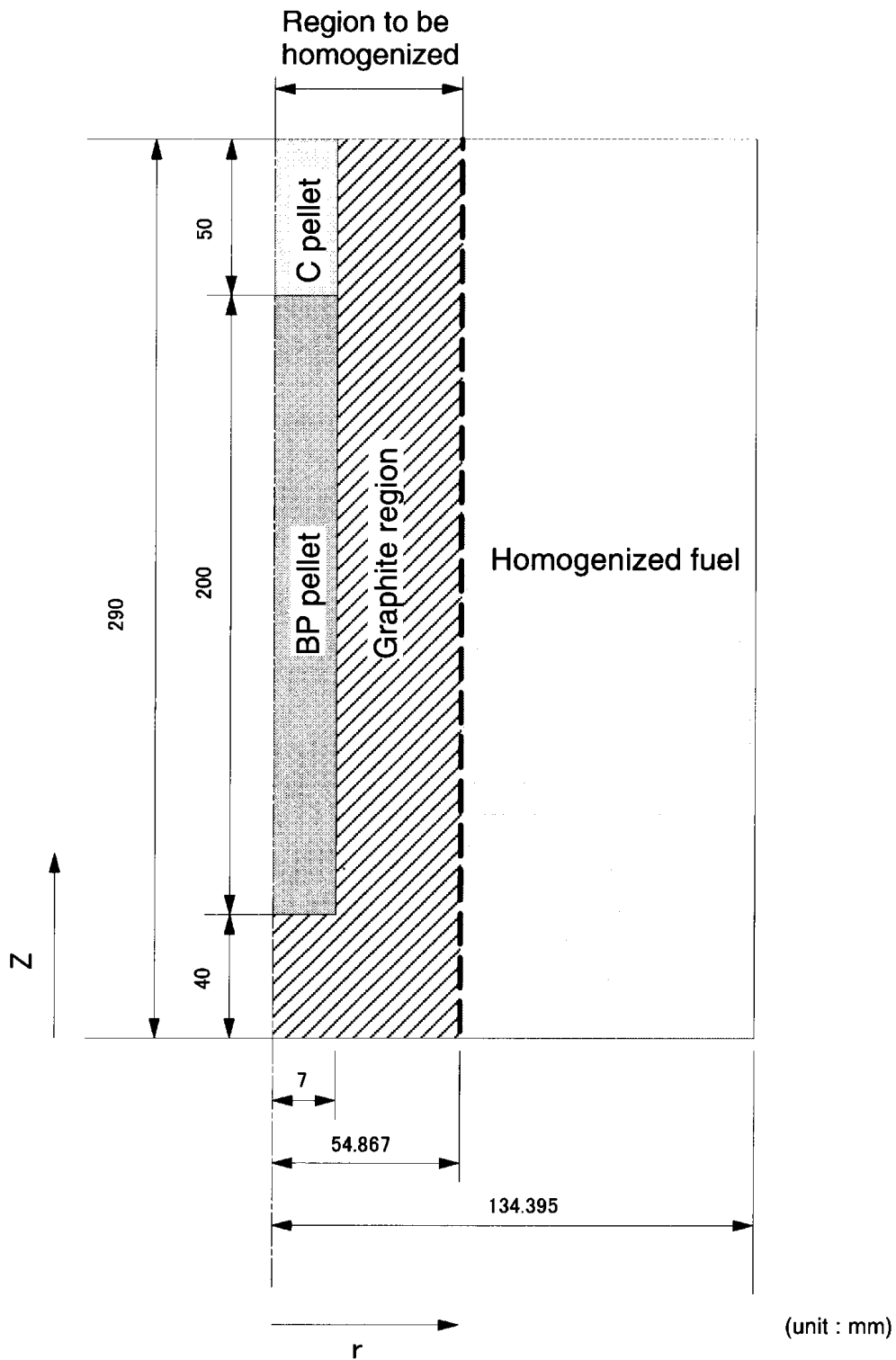


FIG. 2.22. BP cell model by TWOTRAN-II

The average group constant of CRs was obtained by smearing group constants in a region. For core calculations, two kinds of CR models were developed which are called as “CR-block model” and “CR-hex model”. The smearing regions to obtain average group constants for CRs are different in the two models. The schematics for each model is shown in Figure 2.23. The CR-block model was used for all benchmark problems. The CR-hex model was used for HTTR-SC to check the effect of model.

i) CR-block model

The average group constants of CRs were homogenized in a CR guide block region. In the core calculation for the model, CR guide block was modeled as one region. This model is the same with the model used in the previous benchmark problem.

ii) CR-hex model

In the model, one CR rod is modeled as hexagonal shape which contains CR and surrounding graphite. The average group constant of CRs was obtained by smearing a narrow area in a CR guide block. In the core calculation for the model, the CR guide block is divided into three regions. One for CRs, one for a reserved shut down system pellet insertion hole, and one for graphite region. It is possible to consider the position of CRs in a control rod guide block at core calculations using the model.

In the benchmark problems, CR-block model was used as standard model. The effects of CR-block model and CR-hex model were evaluated in the benchmark of HTTR-SC.

The CITATION-1000VP is a reactor core analysis code based on the diffusion theory. This code was improved to enable a full core model calculation of the HTTR by extending the number of zones and meshes in the original CITATION code and enhancing the calculation speed by the vectorization of the code. This code was used for the analysis of the effective multiplication factor. The neutron energy group consists of 3 fast and 3 thermal groups.

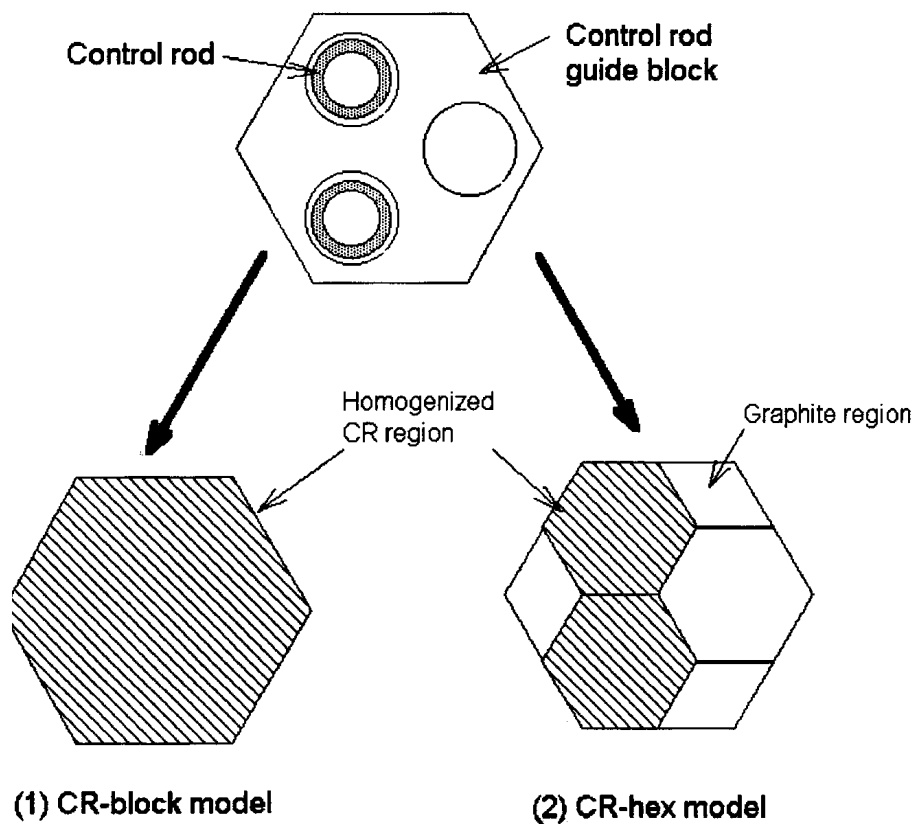


FIG. 2.23. Comparison of CR model concept.

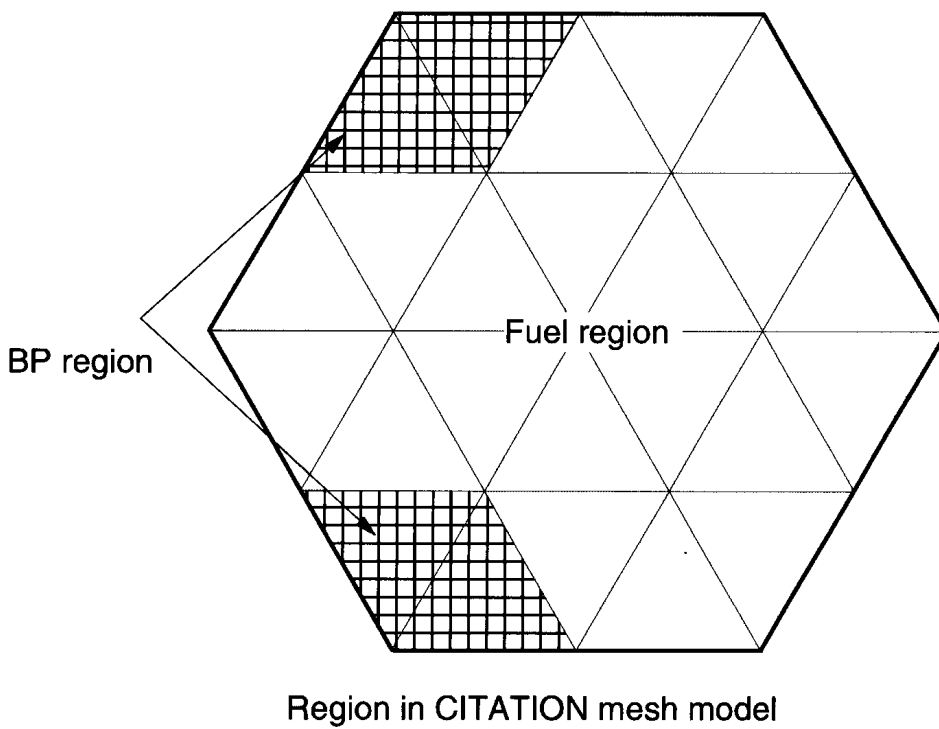
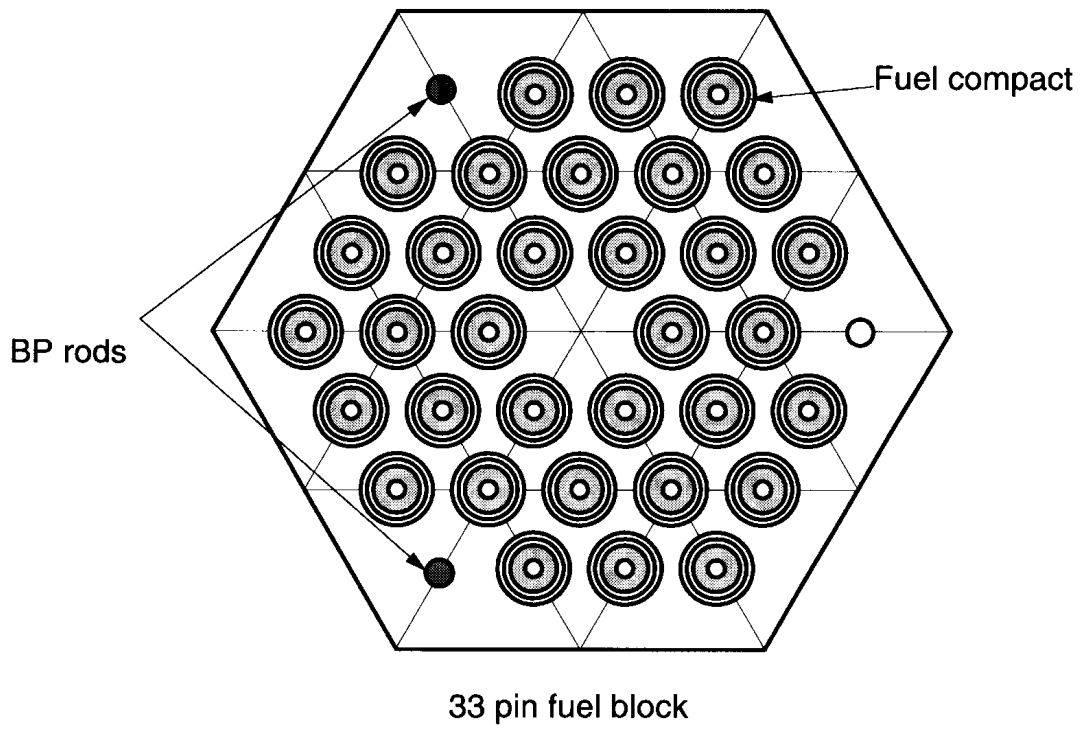


FIG. 2.24. Configuration of regions in block for core calculation mode

A fuel block is divided into 24 triangular meshes horizontally and into 4 meshes vertically for the three-dimensional whole core calculation to simulate the position of the BP rods in a fuel block. In the horizontal plane, a fuel block was divided into BP region and fuel region as shown in Figure 2.24.  $^{10}\text{B}$  and  $^{11}\text{B}$  in a BP rod were distributed only in the BP region. Nuclides of the fuel, such as  $^{235}\text{U}$ ,  $^{238}\text{U}$ , Si, were distributed only in the fuel region. Graphite is distributed homogeneously in the fuel and BP region. For the CR-block model, a CR guide block was modeled as one region.

### ***Monte Carlo calculation***

The whole core is presented with hexagonal lattices consisting of graphite reflector blocks and fuel blocks. The fuel block is also presented with hexagonal lattices consisting of fuel rods and BP rods. The fuel rod and BP rod are modeled with cylindrical body descriptions in hexagonal cells. Effective multiplication factors  $k_{\text{eff}}$  of different cores in fuel loading were calculated with MVP [2-16, 2-17] using the nuclear data of JENDL-3.2 [2-18]. The thermal neutron treatment for graphite in the JENDL-3.2 is based on  $S(\alpha,\beta)$  of ENDF/III. The core temperature of 300K is assumed. The most probable value of  $k_{\text{eff}}$  was evaluated from track length, collision and analog estimators with the method of maximum likelihood. Number of histories per batch was 20,000 for all cases. Number of batches was 150. The first 5 batches were neglected for the statistical treatments. Although large number of scattering reactions must be calculated in one history for HTGR due to the low slowing down power of graphite, the CPU time is still reasonably small for the whole core analysis. The computation was carried out on the FACOM VPP-500.

In the geometry description, the hexagonal multiple lattice capability was used for periodical arrays of fuel rods in a fuel block and of hexagonal blocks in the core. The cross-sectional model of the 18 column, 24 column and whole core is given in Figures 2.15, 2.26, 2.28 and 2.27, respectively. The outer form of the permanent reflector blocks is modeled with a cylinder. The cross-sectional model of fuel block is shown in Figure 2-28. The cylindrical geometry of fuel rods and burnable poison rods were precisely modeled with body descriptions in hexagonal cells. The voids in control rod guide blocks and dummy fuel blocks increase neutron leakage because of the neutron streaming through them. They are also modeled with body descriptions in hexagonal cells. The CFPs in the compact are treated with the following three models:

- (1) Homogeneous Model: All substances in CFPs are smeared with the graphite matrix in the compact.
- (2) Corrected-homogeneous Model: Heterogeneity effect of CFPs, evaluated with collision probability theory [2-19], is added to the result of (1)
- (3) Heterogeneous Model: Heterogeneity effect of CFPs is directly evaluated with Monte Carlo calculation using statistical geometry model based on MURATA's idea [2-20].

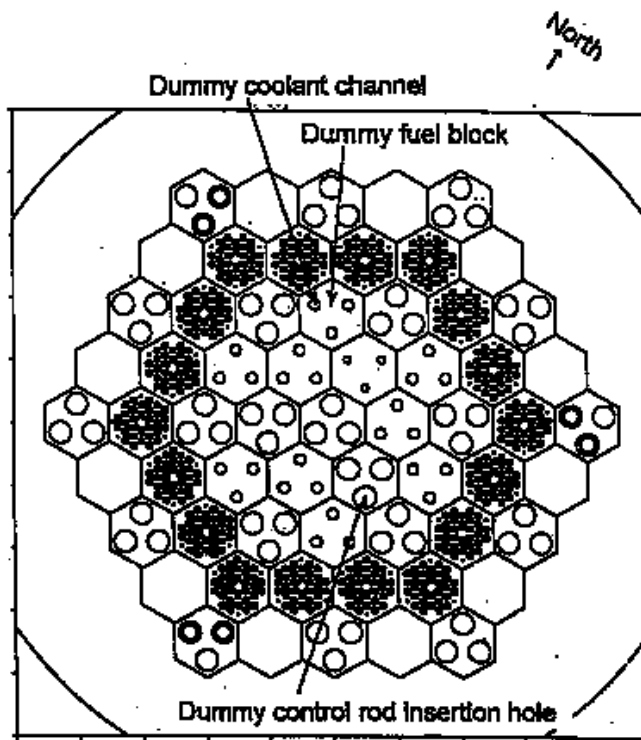


FIG. 2.25. Horizontal cross-section of 18 column core model.

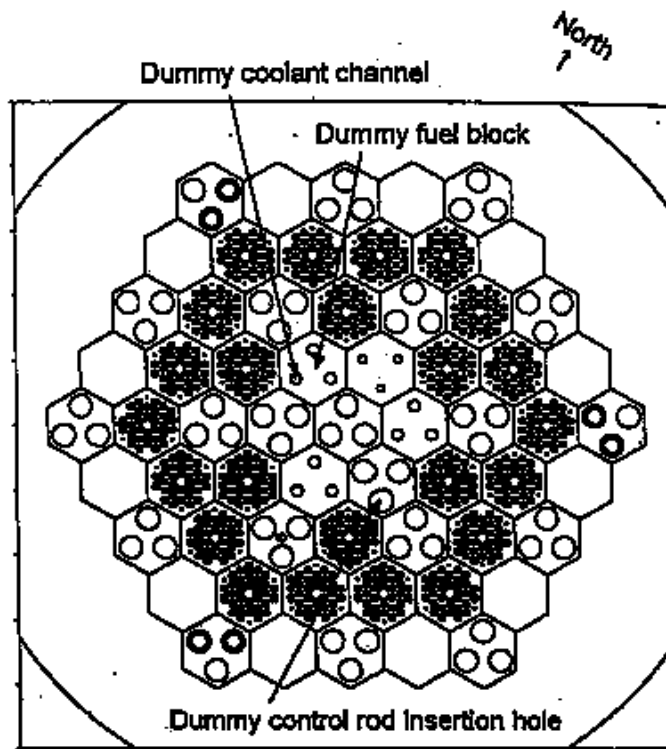


FIG. 2.26. Horizontal cross-section of 24 column core model.



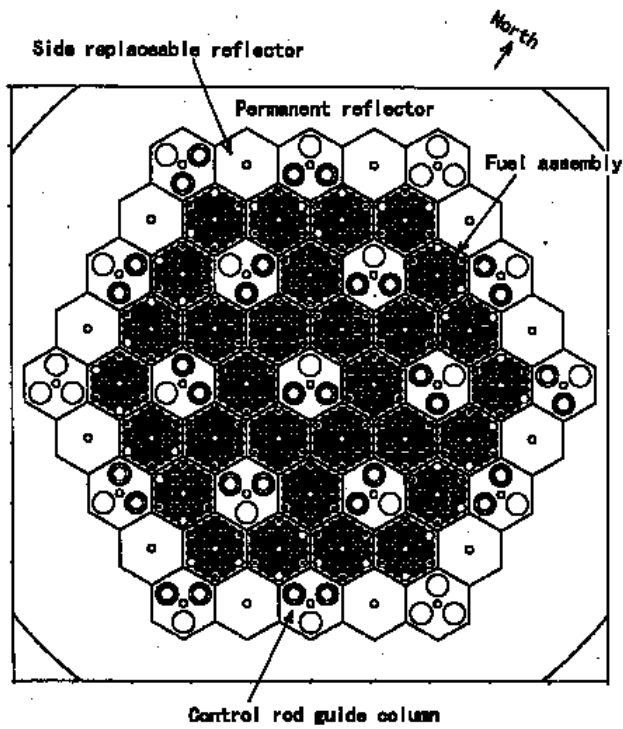


FIG.2.27. Horizontal cross-section of 30 column core model (fully-loaded core).

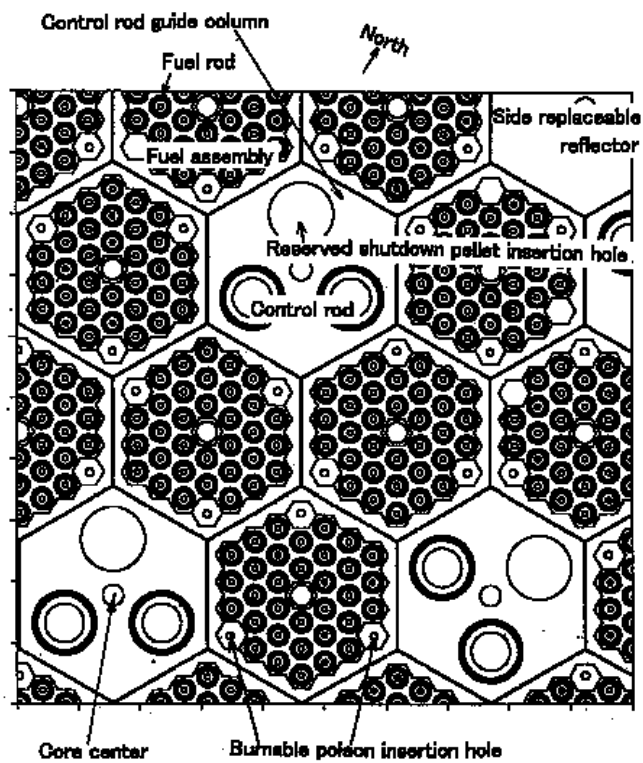


FIG: 2.28. Magnification of analytical model.

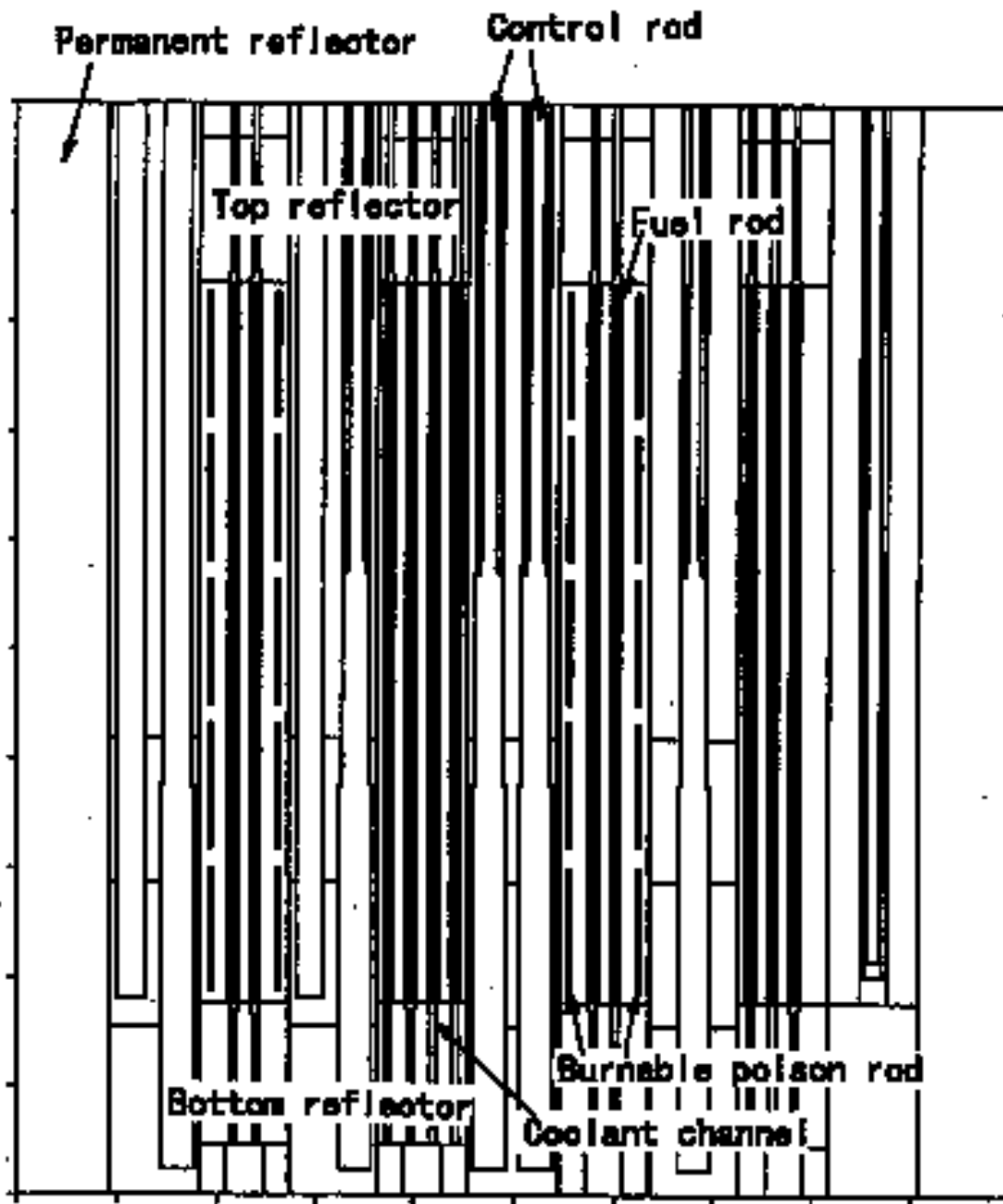


FIG. 2.29. Vertical cross-section of 30 column core model (fully loaded core).

Total amounts of material such as uranium, oxygen and carbon are conserved in the compact for each model. About 13,000 CFPs are distributed randomly in a fuel compact. The  $k_{\text{eff}}$  - calculations for fuel loading steps were carried out in the condition where CRs were fully withdrawn from the core and reflectors. The vertical cross-sectional model of core is shown in Figure 2.29.

The vertical length of each component is precisely modeled. The BP is separated in upper and lower parts in a fuel block. Graphite disks are inserted between upper and lower parts of BPs. The lengths of BP part and Graphite disk part are 40 and 10 cm, respectively.

### 2.2.1.2. Results of HTTR-FC calculation

The number of fuel columns are evaluated for the first criticality, when the fuels columns are charged from the outer region of the core. They are loaded clockwise, one by one. A small excess reactivity at the first criticality is also evaluated. For phase 2, the HTTR-FC is re-calculated considering the following effects:

- 1) Air in void of graphite
- 2) Revised impurity contents in dummy block
- 3) Aluminum in the temporary neutron detector holders.

The change in the effective multiplication factor at fuel loading was calculated. The R2 CRs remain in the top region of the core. The CR model was CR-block model. Fourteen kinds of microscopic cross section set of fuel were used for the calculations considering the kinds of  $^{235}\text{U}$  enrichment and boron concentration in the burnable absorber rod.

#### HTTR-FC by diffusion calculation

The change in effective multiplication factor at fuel loading is shown in Table 2-4 and Figure 2.30. The number of fuel column at the first criticality is 17 columns and the excess reactivity at the first criticality is about  $0.05\% \Delta k/k$ .

Table 2-4. Calculation results of HTTR-FC

No. of fuel column	$k_{\text{eff}}$ [-]	Excess reactivity [% $\Delta k/k$ ]
16	0.9921520	-0.79
17	1.0005451	0.05

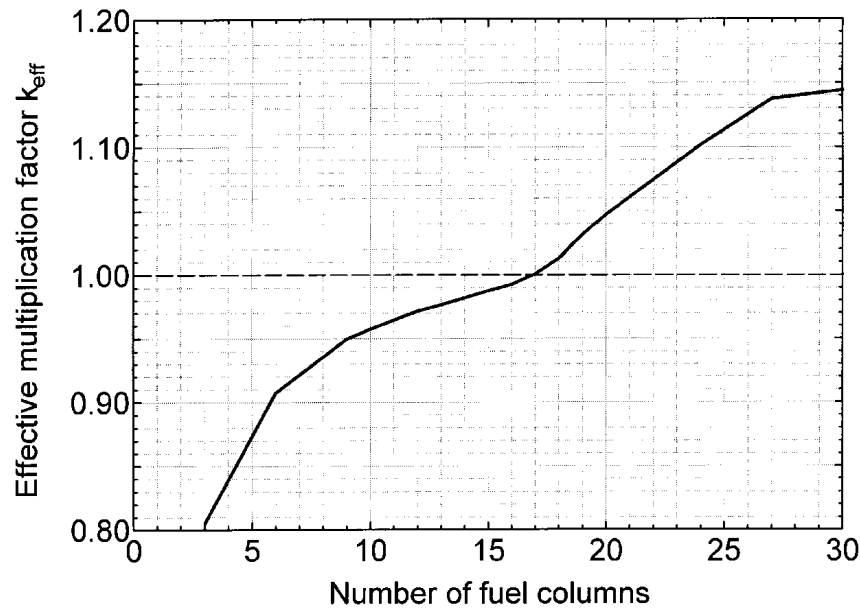


FIG. 2.30. Change in  $k_{\text{eff}}$  at fuel loading.

### HTTR-FC by Monte Carlo

The change in effective multiplication factors and excess reactivities are given in Figure 2.31 and Table 2-5. The CFP's heterogeneity effect evaluated with the collision probability theory was  $0.012 \Delta k$  for the HTTR fuel. This effect was added to the calculated  $k_{\text{eff}}$  - values for the corrected-homo. Model. The heterogeneity effect of CFP was evaluated also from difference in  $k_{\text{eff}}$  of the Homo- and Hetero- models. The results are given in Table 2-6. Six pairs of control rods in side reflector can not be withdrawn from the top reflector perfectly, because a sufficient negative reactivity addition rate must be supplied for reactor scrams. The CR insertion was simulated in the calculation. The fuel number for the first criticality is estimated for 18 fuel columns (Hetero. model).

Table 2-5. Calculated  $k_{\text{eff}}$  and excess reactivity to loaded fuel columns

Fuel columns	Heterogeneous model		Homogeneous model	
	$k_{\text{eff}}^*$	$\rho_{\text{ex}}^{**}$	$k_{\text{eff}}$	$\rho_{\text{ex}}$
12	$0.959596 \pm 0.00051$	-4.21	—	—
14	$0.97429 \pm 0.00056$	-2.64	—	—
16	$0.98461 \pm 0.00055$	-1.56	$0.98052 \pm 0.00054$	-1.99
17	$0.99506 \pm 0.00052$	-0.50	—	—
18	$1.00609 \pm 0.00047$	0.61	$1.00266 \pm 0.00050$	0.27
19	$1.01626 \pm 0.00046$	1.60	—	—
21	$1.04827 \pm 0.00047$	4.60	—	—
24	$1.09968 \pm 0.00047$	9.06	$1.09350 \pm 0.00039$	8.55
27	$1.13399 \pm 0.00044$	11.82	—	—
30	$1.14278 \pm 0.00039$	12.49	$1.13258 \pm 0.00045$	11.71

\*:  $k_{\text{eff}}$  is Effective multiplication factor. The variation shows the statistical error.

\*\*: $\rho_{\text{ex}}$  is Excess reactivity in  $\% \Delta k/k$ .

Table 2-6. Heterogeneity effects of CFP

Fuel columns	$\rho_{\text{ex}}(\text{heterogeneous}) - \rho_{\text{ex}}(\text{homogeneous})$ ( $\% \Delta k/k$ )
16	0.43
18	0.34
24	0.51
30	0.78

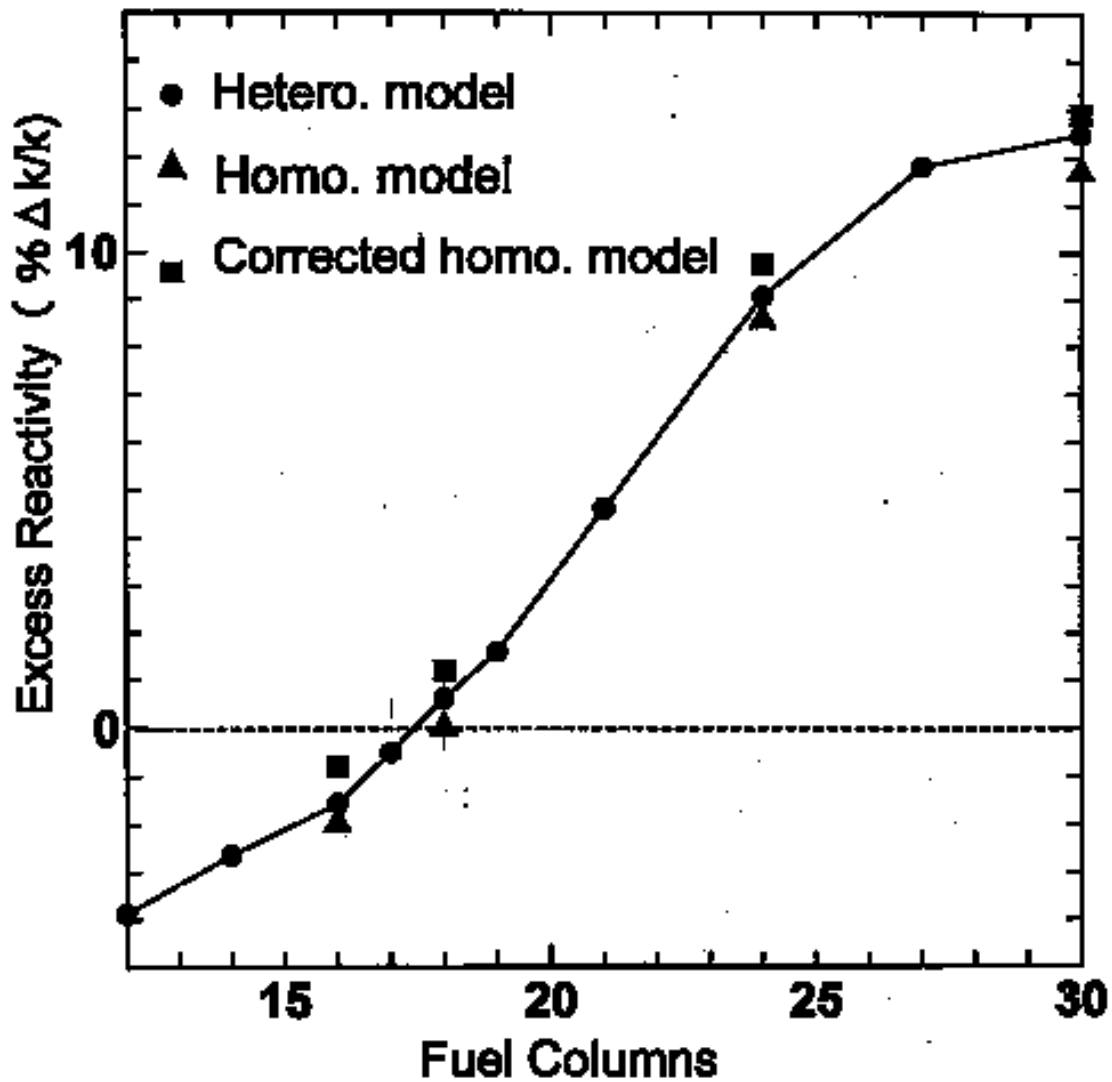


FIG.2.31. Change in excess reactivity with fuel loading.

The experimental results for HTTR-FC were as follows: The fuel loading was started from the core periphery to core center so that the annular core will be constructed on the way to the full core, to obtain the nuclear characteristics. Fuel blocks are loaded clockwise in the core periphery as shown in Figure 2.25. The fuel loading was carried out by replacing dummy blocks with fuel blocks. The annular core was made when the core was loaded with fuel columns from 18 to 24. Inverse multiplication factors ( $1/M$ ) were evaluated at 0, 6, 9, 12, 15, 16, 17 and 18 fuel column-loaded cores to predict the first criticality. The first criticality was achieved at the 19 fuel-column-loaded core.

### 2.2.1.3. HTTR-CR

The control rod insertion depths are evaluated at the critical condition for the following three cases:

- 1) 18 columns (thin annular core)
- 2) 24 columns (thick annular core)
- 3) 30 columns (fully loaded core)

All control rod insertion levels are adjusted on the same level except three pairs of control rods in the most outer region in the side reflectors. These three pairs of control rods are fully withdrawn for the calculation. (The fully withdrawn position of CRs (C, R1, R3) are over the top of the replaceable reflector. The “fully withdrawn” position of the six control rods of R2 is 725 mm below the top of the 1<sup>st</sup> replaceable reflector block.)

### ***HTTR-CR by diffusion calculation***

The control rod position at criticality for 18-columns loaded core, 24 columns-loaded core and fully loaded core are shown in Table 2-7.

Table 2-7. Calculation results of HTTR-CR

No. of fuel column	Control rod position at criticality [mm]
18	3035
24	2055
30	1665

### ***HTTR-CR by Monte Carlo calculation***

The analytical method for HTTR-CR is as same as that for HTTR-FC. The results are as follows:

Model and column number	CR position (mm)
18 (Corrected homo. model)	2810
24 (Corrected homo model)	2080
30 (Hetero model)	1800

The experimental results for HTTR-CR were as follows: The reactor was made critical in every fuel loading step after the first criticality to the full core. The steps were 21, 24, 27 and 30 columns. The CR position at every critical condition was measured to evaluate calculation accuracy. The CRs are inserted from top to bottom. The tops of C-, R1- and R2-CRs were kept the same level at critical conditions. The R3-CRs were fully withdrawn. The CR position is defined as distance from the boundary between the fuel region and the bottom reflector. The change in the critical CR position is given in the following Table 2-8:

Table 2-8. Measured critical control rod positions

Fuel column	21	24	27	30
Rod position(mm)*	2646±5	2215±5	1899±5	1775±5

\*Distance from the boundary between fuel region and the bottom reflector. Sinking of CR driving mechanism (14mm) was considered.

#### 2.2.1.4. HTTR-EX

The excess reactivity is evaluated for the three cases mentioned in HTTR-CR. The room temperature of 300K is assumed as the moderator and fuel temperatures for the benchmark problem. One atmospheric pressure of helium is used as the primary coolant condition.

#### ***HTTR-EX by diffusion calculation***

The excess reactivity at 18-columns loaded core, 24 columns-loaded core and fully loaded core are shown in Table 2-9.

Table 2-9. Calculation results of HTTR-EX

No. of fuel column	$k_{\text{eff}}$ [-]	Excess reactivity [% $\Delta k/k$ ]
18	1.0126343	1.2
24	1.1014290	9.2
30	1.1442246	12.6

#### ***HTTR-EX by Monte Carlo calculation***

The analytical method for HTTR-EX is as same as that for HTTR-FC. The results are as follows:

Model and column number	Excess reactivity (% $\Delta k/k$ )
18 (Hetero. model)	0.61
24 (Hetero. model)	9.06
30 (Hetero model)	12.5

The experimental results are as follows: Fuel addition method was applied for the excess reactivity measurement. The increment in excess reactivity was measured by IK method at 21, 24, 27, and 30 column-loaded core. The measurement was affected by the negative shadowing effect. The measured increments in excess reactivity were revised with following relation to correct the shadowing effect. The excess reactivity for each core are given in Table 2-10.

$$\rho_{ex} = \sum \Delta\rho_{ex}$$

$$\Delta\rho_{ex} = \Delta\rho_{ikm} \cdot R$$

$$R = \frac{\Delta\rho_{excal}}{\Delta\rho_{ikcal}}$$

$\rho_{ex}$  :Excess reactivity (% $\Delta k/k$ )

$\Delta\rho_{ex}$  :Increment in excess reactivity (% $\Delta k/k$ )

$\Delta\rho_{ikm}$ : Increment in excess reactivity measured with IK method (% $\Delta k/k$ )

$\Delta\rho_{excal}$ : Increment in excess reactivity calculated after definition.

All control rods are fully withdrawn from the critical rod position (% $\Delta k/k$ )

$\Delta\rho_{ikcal}$ : Increment in excess reactivity calculated after IK measurement procedure (% $\Delta k/k$ )

R: Revising factor

Table 2-10. Excess reactivity

Fuel columns	$\rho_{ikm}$ (% $\Delta k/k$ )	$\Delta\rho_{ikm}$ (% $\Delta k/k$ )	R	$\Delta\rho_{ex}$ (% $\Delta k/k$ )	$\rho_{ex}$ (% $\Delta k/k$ )
21	$2.3 \pm 0.23$	$2.3 \pm 0.23$	$1.69 \pm 0.12$	$4.0 \pm 1.1$	$4.0 \pm 1.1$
24	$5.0 \pm 0.50$	$2.7 \pm 0.27$	$1.40 \pm 0.11$	$3.7 \pm 1.0$	$7.7 \pm 2.1$
27	$7.4 \pm 0.74$	$2.4 \pm 0.24$	$1.26 \pm 0.15$	$3.0 \pm 0.9$	$10.7 \pm 3.0$
30	$8.5 \pm 0.85$	$1.1 \pm 0.11$	$1.19 \pm 0.04$	$1.3 \pm 0.3$	$12.0 \pm 3.3$

( $\beta_{eff}=0.0065$ )

### 2.2.1.5. HTTR-SC

The core condition is as follows:

- Fully-loaded core (30 column fuel core)
- Fresh fuel core

The following two cases are investigated the HTTR-SC:

#### 1) Scram reactivity of reflector CRs (R2 and R3)

The Scram reactivity of the reflector CRs is evaluated for the fully-loaded core as follows.

$$\rho_R = \frac{k_{Crit.} - k_{RCR-in}}{k_{Crit.} \cdot k_{RCR-in}}$$

Where:

$\rho_R$ : Scram reactivity of reflector CRs ( $\Delta k/k$ )

$k_{Crit.}$ : Effective multiplication factor at critical CR position

$k_{RCR-in}$ : Effective multiplication factor at CR position after scram

The effective multiplication factors is calculated at the critical CR positions and the CR positions after scram. The CR positions are given in Table 2-11. The reflector CRs are fully inserted after scram. The positions of the in-core CRs are not changed. The temperature for core and reflector is 300K.

Table 2-11. Control rod position before and after scram of reflector CRs\*

CR Group	Critical position (mm)	Position after scram (mm)	Remark
C	1775	1775	Not changed
R1	1775	1775	Not changed
R2	1775	-55**	
R3	Full out	-55	

\*:Case (T4-3005), No neutron source

\*\* :Control rods are inserted slightly into the top of the bottom reflector



2) Scram reactivity of all control rods (C, R1,R2 and R3):

The scram reactivity of all CRs is evaluated from the effective multiplication factors as follows:

$$\rho_A = \frac{k_{Crit.} - k_{ACR-in}}{k_{Crit.} \cdot k_{ACR-in}}$$

Where:

$\rho_A$ : Scram reactivity of all CRs ( $\Delta k/k$ )

$k_{Crit.}$ : Effective multiplication factor at critical CR position

$k_{ACR-in}$ : Effective multiplication factor at CR position after scram

The effective multiplication factors is calculated at the critical CR positions and the CR positions after scram. The CR positions are given in Table 2-12. The all CRs are fully inserted after scram. The temperature for core and reflector is 300K.

Table 2-12. Control rod positions before and after scram of reflector and in-core CRs\*

CR Group	Critical positions (mm)	Positions after scram (mm)	Remark
C	1775	-55**	
R1	1775	-55	
R2	1775	-55	
R3	Full out	-55	

\*:Case (T4-3005), No neutron source

\*\* :Control rods are inserted slightly into the top of the bottom reflector

The measured scram reactivity was 0.46 $\Delta k/k$

### ***HTTR-SC by diffusion calculation***

The scram reactivity should be calculated from the criticality condition with CR position at 1789mm. In the calculation model, it is difficult to set the CR position at 1789mm because the vertical mesh division is 145mm. The scram reactivity was evaluated by interpolating the results of 1740mm and 2030mm.

The calculated results of HTTR-SC by the CR-block model and the CR-hex model are shown in Table 2-13. The results of CR-block model was smaller than that of the CR-hex model. In the CR-hex model, reflector CRs are nearer to the core than in the CR-block model. Therefore, scram reactivity becomes larger.

The comparison of measured and calculated value of the scram reactivity are shown in Table 2-14. For the results of all CRs, the error of CR-block model and CR-hex model are about -7% and -4%, respectively. The difference of both models is small. Both model show good agreement with measured results.

For the results of reflector CRs, the error of CR-block model and CR-hex model are about -31% and -26%, respectively. The CR-hex model shows less error than CR-block mode, but still shows more than 20% of error. The calculation model should be further revision.

Table 2-13. Calculated results of HTTR-SC

(1) Scram reactivity of reflector CRs(R2 and R3)

CR model	CR-block model		CR-hex model		
CR position before scram [mm]	1740	2030	1740	2030	
$k_{\text{eff}}$	Before scram	1.0093164	1.0418115	1.0084132	1.0409116
	R2, R3 full in (0mm)	0.9311435	0.9599020	0.9247989	0.9534538
Scram reactivity [% $\Delta k/k$ ]	8.32	8.19	8.97	8.81	
Scram reactivity from 1789 mm [% $\Delta k/k$ ]	8.30		8.94		

(2) Scram reactivity of all CRs

CR model	CR-block model		CR-hex model		
CR position before scram [mm]	1740	2030	1740	2030	
$k_{\text{eff}}$	Before scram	1.0093164	1.0418115	1.0084132	1.0409116
	All CRs full in (0mm)	0.7069916		0.6980522	
Scram reactivity [% $\Delta k/k$ ]	42.37	45.46	44.09	47.19	
Scram reactivity from 1789 mm [% $\Delta k/k$ ]	42.9		44.6		

Table 2-14. Comparison to the measured value

(1) CR-block model

	Scram reactivity [% $\Delta k/k$ ]		Error [%]
	Measured (E)	Calculated (C)	C/E-1
Reflector CRs	12.1	8.30	-31.4
All CRs	46.3	42.9	-7.3

(2) CR-hex model

	Scram reactivity [% $\Delta k/k$ ]		Error [%]
	Measured (E)	Calculated (C)	C/E-1
Reflector CRs	12.1	8.94	-26.1.
All CRs	46.3	44.6	-3.7

**HTTR-SC by Monte Carlo Calculation**

The analytical method for HTTR-SC is the same as that for HTTR-FC. The results are given in Table 2-15:

A two-step scram will be performed in the HTTR to prohibit the high temperature exposure of the in-core CRs in full power operation. The in-core CRs are inserted when the core temperature becomes less than 750°C or 2400 seconds passes after insertion of reflector CRs. The simulation of the two-steps scram was conducted in startup core physics. The in-core CRs were inserted in the time delay of 20 seconds after insertion of the reflector CRs. The time delay was determined so that the neutron density signal from the CIC did not become too small for reactivity measurement. The reactivity was measured continuously by the IK method. Digital computer was used to evaluate the reactivity from change in neutron density signal.

Table 2-15. Scram reactivity of control rods

Items	Scram Reactivity
1) Scram reactivity of reflector CRs	$0.99908 \pm 0.00045$
$k_{crit}$	$0.91222 \pm 0.00058$
$k_{RCR-in}$	$0.0953 \pm 0.0007$
$\rho_R(\Delta k/k)$	
2) Scram reactivity of all CRs	$0.99908 \pm 0.00045$
$k_{crit}$	$0.68873 \pm 0.00064$
$k_{ACR-in}$	$0.4510 \pm 0.0008$
$\rho_A(\Delta k/k)$	

Table 2-16. CR position at two step scram\*

R	CR position [mm]			
	before scram	B	F	S
		first scram	step	second step scram
1	775±5	1	1	-
	775±5	1	1	55±5**
	775±5	1	-	55±5**
2	775±5	F	-	55±5**
3	pull out		55±5**	55±5**

\*: Sinking of CR driving mechanism was considered

\*\* : Control rods are inserted slightly below 0 mm at scram.

Table 2-17. Scram reactivity at two-step scram

Scram	Measured scram reactivity [ $\Delta k/k$ ]	Error C/E-1 (%)
Reflector CRs	$0.12 \pm 0.012$	-17
All CRs	$0.46 \pm 0.046$	10

( $\beta_{eff} = 0.007$ )

### 2.2.1.6. HTTR-TC

Isothermal temperature coefficients for fully-loaded core are evaluated from the effective multiplication factors by the following relation:

$$\rho_n = \frac{k_{n+1} - k_n}{k_{n+1} \cdot k_n} \cdot \frac{1}{(T_{n+1} - T_n)}$$

Where:  $\rho_n$ : Temperature coefficient between  $T_n$  and  $T_{n+1}$  ( $\Delta k/k/K$ )

$T_n$ : Core temperature at  $n^{\text{th}}$  measurement (K)

$T_{n+1}$ : Core temperature at  $n+1^{\text{th}}$  measurement (K)

$k_n$ : Effective multiplication factor at  $T_n$

$k_{n+1}$ : Effective multiplication factor at  $T_{n+1}$

The critical control rod positions are changed with temperature elevation in the real reactor operation. However, the control rod position should not be changed in calculation to obtain the reactivity difference. The critical positions given in Table 2-18 as CR position are to be used for calculation of effective multiplication factors. The effective multiplication factors are calculated for the following temperatures:

280, 300, 340, 380, 420, 460, 480 (K)

Temperature coefficient for following temperatures are evaluated from the effective temperature coefficients with the above relation.

290 320 360 400 440 470 (K)

Additionally, the critical control rod positions are evaluated at temperature of 480K. The insertion depths of C, R1 and R2 are at the same level. All of R3 is fully withdrawn.

Table 2-18. Control rod position for temperature coefficient evaluation

CR Group	Critical positions* (mm)
C	1777
R1	1777
R2	1777
R3	Full out

\*These values are obtained from the critical positions (T4-3005) by correcting with lifting effect of CR driving mechanism (about -14mm) and temperature effect from 25°C to 27°C (about 2mm).

### **HTTR-TC by diffusion calculation**

The isothermal temperature coefficients were evaluated using the CR-block model. It was impossible to calculate isothermal temperature coefficients directly due to mesh division of the core calculation model. Therefore, the isothermal temperature coefficients at 1776mm were evaluated by interpolating the results at 1740mm and 2030mm. In the calculation, four kinds of microscopic cross section set were used. Each layer of fuel block use same microscopic cross section set. Air in void of graphite and Al of temporary neutron detector holders were not considered.

The calculation results are shown in Table 2-19 and Figure 2.32. Due to our convenience, calculated temperature points are not same to the problem description. Therefore, isothermal temperature coefficients at the temperature point described in the problem definition are evaluated by the calculated results. The results are provided in Table 2-19. The average temperature coefficient within the temperature range is about  $-1.3 \times 10^{-4} \Delta k/k/K$ .

The criticality CR position at 480K was evaluated by the results already shown in Table 2-19. The results are shown in Table 2-20.

Table 2-19: Calculated results of HTTR-TC (by CR-block model)

(1) Effective multiplication factors

CR position [mm]	1740	2030	
Temperature [K]	300	1.0169601	1.0497657
	320	1.0145556	1.0474498
	340	1.0119939	1.0449780
	380	1.0065323	1.0396969
	420	1.0011004	1.0344445
	460	0.9957035	1.0292383
	480	0.9901534	1.0238264

(2) Isothermal temperature coefficients

CR position [mm]	Temperature coefficients [ $\Delta k/k/K$ ]			
	1740	2030	1776*	
Temperature [K]	310	$-1.16596 \times 10^{-4}$	$-1.05309 \times 10^{-4}$	$-1.15 \times 10^{-4}$
	330	$-1.24751 \times 10^{-4}$	$-1.12913 \times 10^{-4}$	$-1.23 \times 10^{-4}$
	360	$-1.34046 \times 10^{-4}$	$-1.21521 \times 10^{-4}$	$-1.32 \times 10^{-4}$
	400	$-1.34768 \times 10^{-4}$	$-1.22091 \times 10^{-4}$	$-1.33 \times 10^{-4}$
	440	$-1.35356 \times 10^{-4}$	$-1.22247 \times 10^{-4}$	$-1.34 \times 10^{-4}$
	480	$-1.40737 \times 10^{-4}$	$-1.28395 \times 10^{-4}$	$-1.39 \times 10^{-4}$

\* : Interpolated by the results of 1740mm and 2030mm.

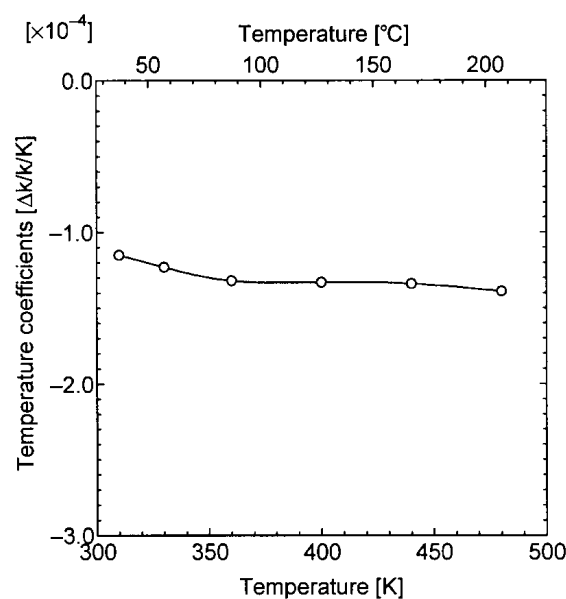


FIG. 2.32. Calculated results of temperature coefficients (CR position: 1776mm).

Table 2-20. Criticality CR position at 480K

	CR position [mm]
C, R1, R2	1825
R3	4060

***HTTR-TC by Monte Carlo Calculation***

The analytical method for HTTR-TC is as same as that for HTTR-FC. The results are given in Table 2-21:

Table 2-21. Temperature coefficients of HTTR core

Items	Temperature coefficients
Effective multiplication factor for different temperatures	
$k_{300}$	$0.99762 \pm 0.00054$
$k_{340}$	$0.99273 \pm 0.00051$
$k_{380}$	$0.98621 \pm 0.00050$
$k_{420}$	$0.97990 \pm 0.00050$
$k_{460}$	$0.97393 \pm 0.00057$
$k_{480}$	$0.97221 \pm 0.00051$
Temperature coefficients ( $\Delta k/k/K$ )	
$\rho_{320}$	$-1.23 \times 10^{-4}$
$\rho_{360}$	$-1.66 \times 10^{-4}$
$\rho_{400}$	$-1.63 \times 10^{-4}$
$\rho_{440}$	$-1.56 \times 10^{-4}$
$\rho_{470}$	$-0.91 \times 10^{-4}$
Critical control rod position at 480K	
C, R1, and R2 (mm)	1789
R3	Full out

The control rod positions at critical condition were measured for three different core temperatures (T). The core temperature was obtained by averaging core-inlet and -outlet coolant temperatures. The control rod positions and core temperatures are summarized in Table 2-22:

Table 2-22. Control rod positions and core temperature

CR group	CR position (mm)		
	T=22.7 □	T=122.0□	T=144.9□
C	1751*±5	1873±5	1903±5
R1	1751*±5	1873±5	1903±5
R2	1751*±5	1873±5	1903±5
R3	Full out	Full out	Full out

\*:Sinking of CR driving mechanism (14mm) was considered.

\*\*After taking out temporary detectors from core

The temperature effect was evaluated from CR positions, using calculated control rod worth curve. The isothermal temperature coefficients are given in the following Table:

Table 2-23. Isothermal temperature coefficient

Temperature range (□)	Iso.temp.coeff. ( $\Delta k/k/T$ )
22.7~122.0	-1.34E-4*
122.0~144.9	-1.42E-4*

## 2.2.2. France

### 2.2.2.1 General analysis method and model description

#### Codes and calculation scheme

The French reactor physics code system SAPHYR has been used in the following HTTR calculations. SAPHYR gathers several codes developed at CEA like APOLLO2 [2-23] (transport) based on a database produced with THEMIS/NJOY, CRONOS2 [2-24] (diffusion-transport), FLICA4 (3D- thermal hydraulics), ..., which are interconnected. This code system, initially dedicated to PWR calculations and research & development purposes, seems to be well adapted for the assessment of the HTGR performances and characteristics. Finally, the Monte-Carlo code TRIPOLI4 [2-25] has also been used throughout the study.

All the HTTR problems proposed in reference [2-21] have been treated considering two calculation methods: one based on a Transport – Diffusion calculation scheme and a second one based on a Transport – Monte-Carlo calculation scheme. Figure 2.33 illustrates the general procedure. The standard 172-groups or point-wise cross sections library issued mainly from JEF-2.2 are used for the calculations.

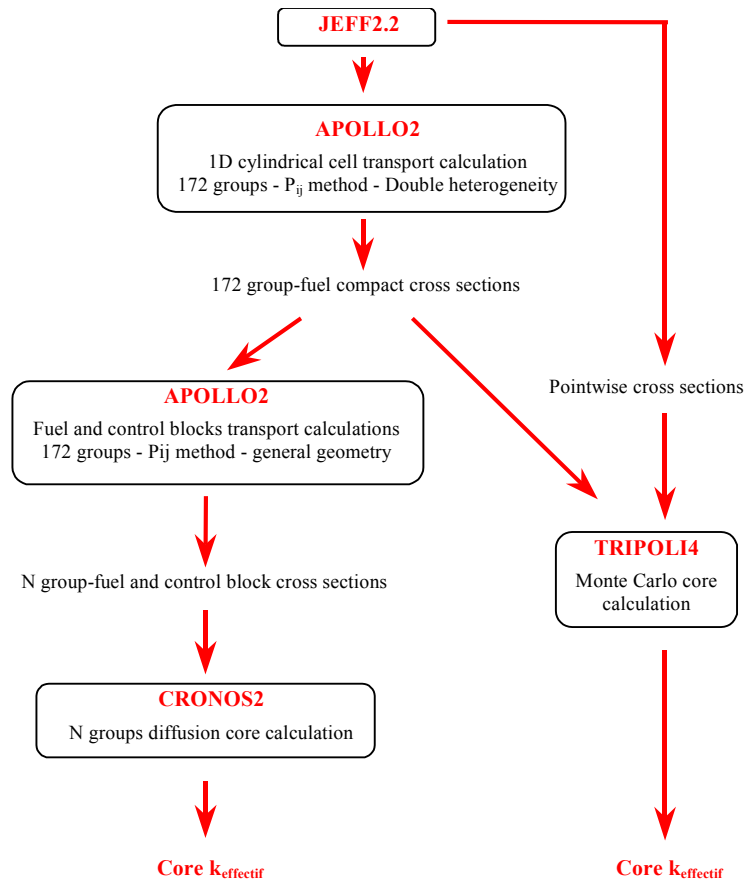


FIG. 2.33. Description of both calculation schemes.

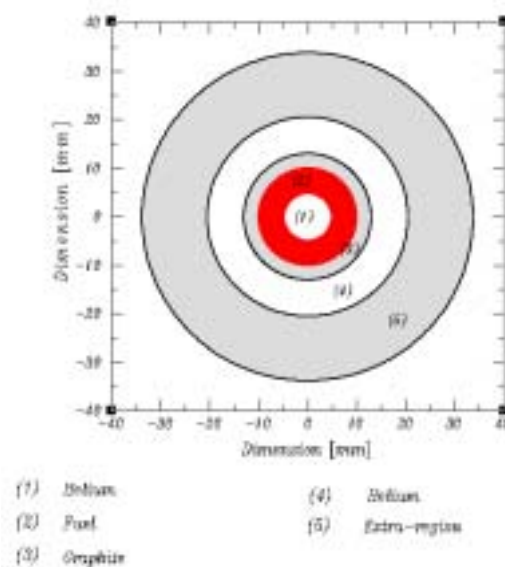
It is noteworthy that as far as the Transport-Diffusion calculations are concerned, two different approaches have been used in terms of core modelling. The first one, a rough model with strong hypotheses (homogenised fuel block, no streaming effect, ...) led to preliminary predictions of the HTTR-FC, EX and CR problems (**first results**). Then, enhancements to



improve the modelisation have been identified, assessed and finally implemented in course of revised calculations using new benchmark data proposed in [2-27]. This has been done for all the HTTR benchmarks (FC, EX, CR, SC and TC) and constitutes the second prediction (**final results**).

***Cylindrical calculation of the fuel compact***

Knowing that the stochastic geometries calculations (coated fuel particles - CFP - randomly distributed in the fuel compact) is not available in the Monte-Carlo code TRIPOLI4, a 1D-cell calculation has been performed as a first step. It takes into account a precise spherical description of the particles with all their coatings which themselves fill the annular part of the cylindrical geometry of the fuel compact (Figure 3.34) with a packing density around 30 %. The self-shielding of the uranium is calculated during this calculation stage. A collision probability method is used to solve the transport equation with 172 energy groups. The critical buckling search allows taking into account the neutron leakage by the addition of a homogeneous leakage term in the form of  $DB^2\Phi$ . The extra region of the cylindrical cell is representative of the other fuel cells disposed around with the triangular pitch of the fuel rods in the assembly.



*FIG. 2.34. Cylindrical Fuel Cell Model.*

This first stage provides **fuel compact-homogenized 172-group cross sections** for Monte-Carlo core calculations (TRIPOLI4). Therefore, in the core calculations performed by TRIPOLI4, point-wise cross sections are used everywhere in the core except in the fuel rod region where the multigroup cross sections have been generated with APOLLO2.

Although that the CFP might be directly considered in the 2D fuel element calculations, as for TRIPOLI4 the same **fuel compact averaged 172-group cross sections** have been used in the 2D transport model described hereafter. This avoids calculating the self-shielding in the 2D configuration and results in a large saving of CPU time without making severe assumption.

### Fuel block 2D-calculations

This stage must lead to an averaged flux weighted library of the different fuel blocks existing in the core. This library of n-group-constants is directly read during the core diffusion calculations. The energy structures ( $n = 2$  to 20) are given in annexe. A 172 group-collision probability method is used in the 2D-fuel block geometry (Figure 2.35). The fuel handling hole zone is modelled by a lower graphite density. As far as the burnable poisons (BP) are concerned, the  $B_4C$  and graphite pellets are **axially homogenised** with the graphite matrix in order to reduce the number of calculations from 32 to 16 and to simplify the core calculations. The impact of this axial homogenisation is important and has been evaluated hereafter.

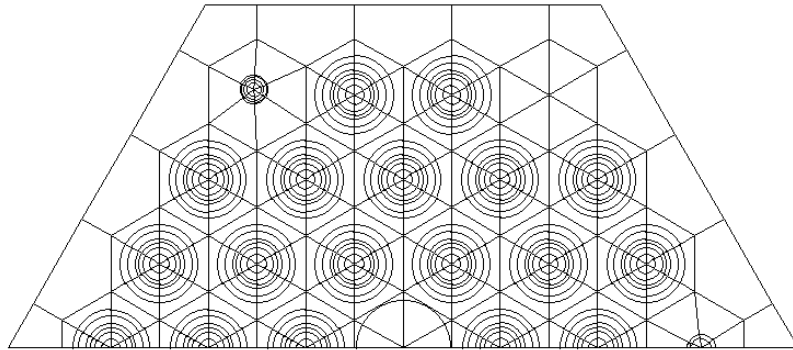


FIG. 2.35. 2D-fuel block geometry.

These 2D calculations were performed in infinite medium with a critical buckling search. A B1 homogeneous neutron leakage model has been retained as a first step. This led to **isotropic diffusion coefficients** and did not allow taking into account the streaming effect. The  $P_{ij}$  model gives a flux evaluation in each region depicted in Figure 2.35. An example of the power distribution in a fuel block is given in Figure 2.36 (the map is normalised to 100).

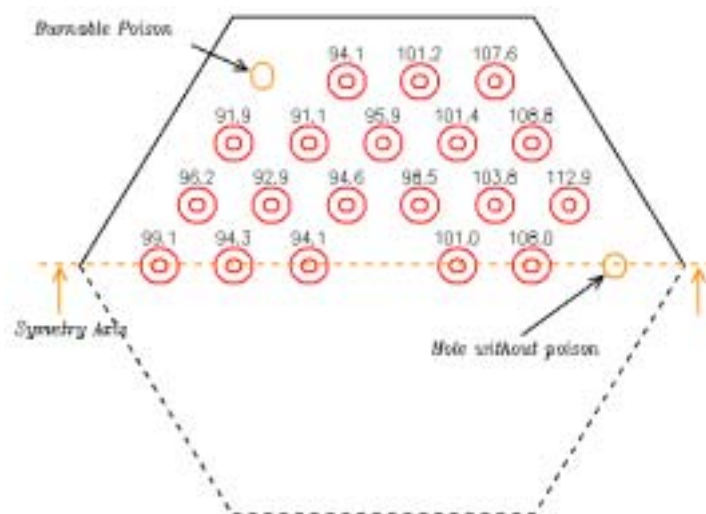


FIG. 2.36. Power Distribution in the f343320 fuel block.

The 2D-results of the fuel blocks in fundamental mode are gathered in the Table 2-24 below and an example of the average neutron flux obtained for the lower and the higher enriched fuel blocks are given in Figure 2.37.

Table 2-24. 2D-Transport Calculations on the Fuel Blocks

Fuel Block	Enrichment [% masse]	Multiplication factor	Migration Area [cm <sup>2</sup> ]
f343320	3,4	1,10650	491,05
f393320	3,9	1,14991	487,14
f433120	4,3	1,18391	486,43
f483120	4,8	1,21366	481,82
f433325	4,3	1,15656	476,90
f523325	5,2	1,20380	470,53
f593125	5,9	1,24481	469,50
f633125	6,3	1,25798	468,43
f633325	6,3	1,24962	465,50
f723125	7,2	1,28937	463,70
f793125	7,9	1,30862	457,99
f673320	4,6	1,28423	467,60
f793320	7,9	1,31733	459,52
f943120	9,4	1,36335	458,81
f993120	9,9	1,37213	455,45

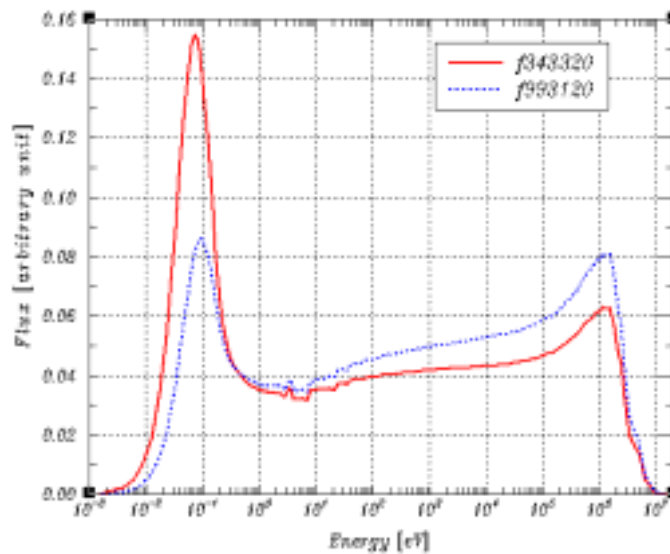


FIG. 2.37. Neutron Spectrum in the fuel block

### Control blocks 2D-calculations

From the previous calculations, the average fluxes of the fuel blocks were used as a 172 group-neutron source placed at the periphery of the control blocks to generate the averaged weighted library needed for core diffusion calculation. A similar collision probability method has been used in the 2D-geometry shown in Figure 2.38 for the control blocks with and without the control rods inserted.

As for the fuel blocks, the n-group-constants have been created by spatial homogenisation on the overall control block geometry. Indeed, among the finite element meshes available in the CRONOS2 diffusion code, it was first impossible to take into account the position and the orientation of the fuel and control blocks by an heterogeneous geometric description of these elements. Moreover, no transport-diffusion equivalence factors have been considered for the control blocks when the control rods are inserted. The averaged neutron fluxes obtained in the control blocks are given in Figure 2.39.

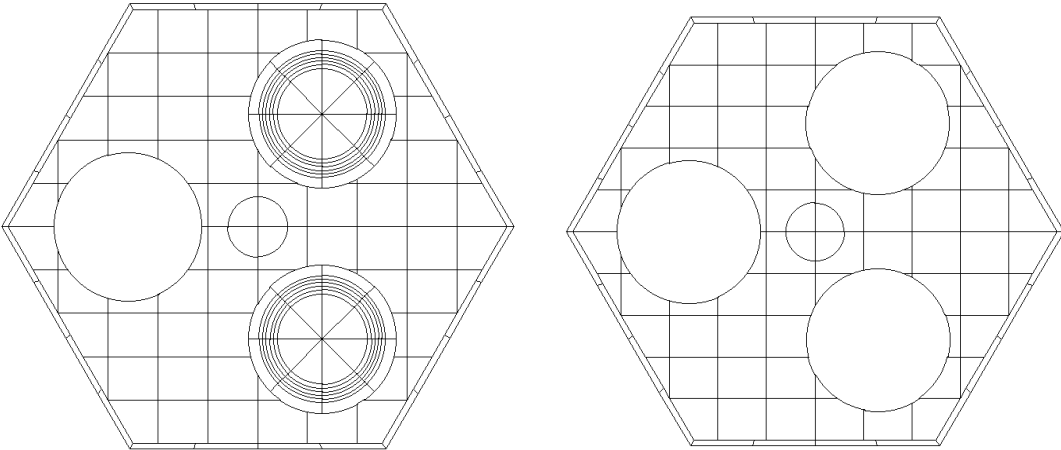


FIG. 2.38. 2D-Control Rod Blocks Geometry.

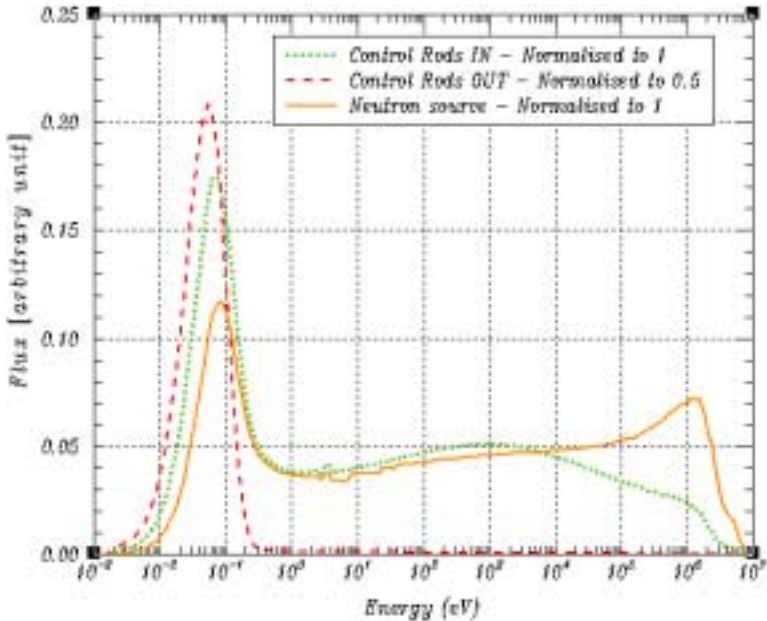


FIG. 2.39. Neutron Spectrum in the Control Rod Blocks.

### ***Reflector cross section generation***

In the diffusion calculations, the n-group-cross sections of the replaceable reflector have been evaluated with similar methods to those described in the previous section where a neutron source is placed at the periphery of the blocks. As far as the permanent reflector region is concerned, the cross sections come from a 1D-cylindrical core transport calculation (in its fully loaded configuration) performed with the  $S_n$  method.

### ***Core diffusion calculations***

An hexagonal 3D-geometry has been used for the core description. Each fuel and control block is represented by a **hexagon with a homogeneous composition**. On this hexagonal mesh, the *finite element method* provides several polynomial approximations. In this case, a second order exact integration (standard parabolic Lagrange) has been adopted horizontally and leads to 19 flux values per mesh (equivalent to a 24 homogeneous meshes in finite difference method).

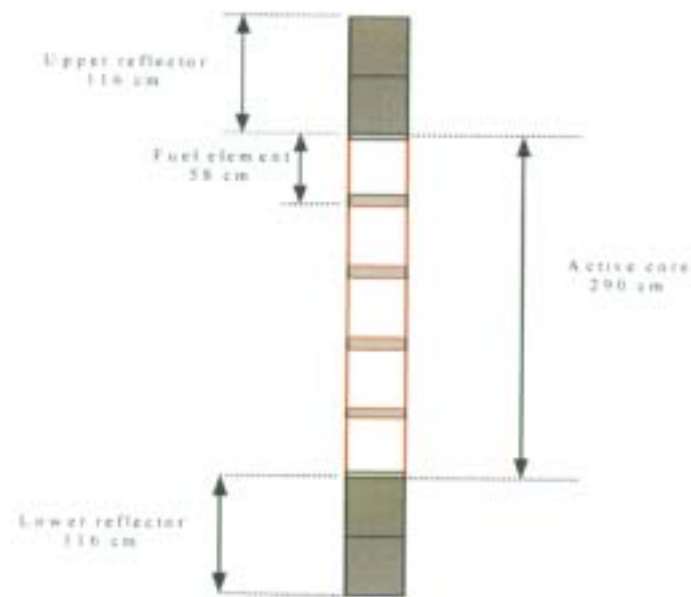


FIG. 2.40. Core Axial Description for CRONOS2 code.

Axially, a linear Lagrange method using Gauss Formula has been applied. Each fuel section (14 fuel compact stack) was divided into 6 meshes and the inter-space between fuel sections (3.4 cm of graphite) has been described explicitly per Fig. 2.40.

### ***Transport-Monte Carlo core calculations***

In spite of the homogenised fuel compacts, the core model developed here is a very detailed 3D-model of the HTTR with all components modelled explicitly. For these components, point-wise cross sections based on the JEF2.2 evaluation were used. The fuel compact was represented by a set of 172-group-constants coming from the previous transport calculation. It is noteworthy that in this case the burnable poison had the exact axial description in opposition to the homogeneous axial description used in the core diffusion calculations.

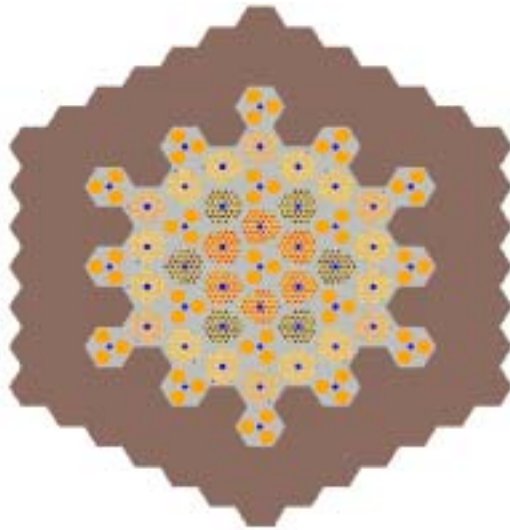


FIG. 2.41. Cross Section of Fully Loaded Core for TRIPOLI4 code.

Only the core configurations with 18 and 30 columns have been considered here. Figures 2.41 and 2.42 show a cross section view of the fully loaded core and detailed axial and radial cross section views of the control rod and fuel blocks.

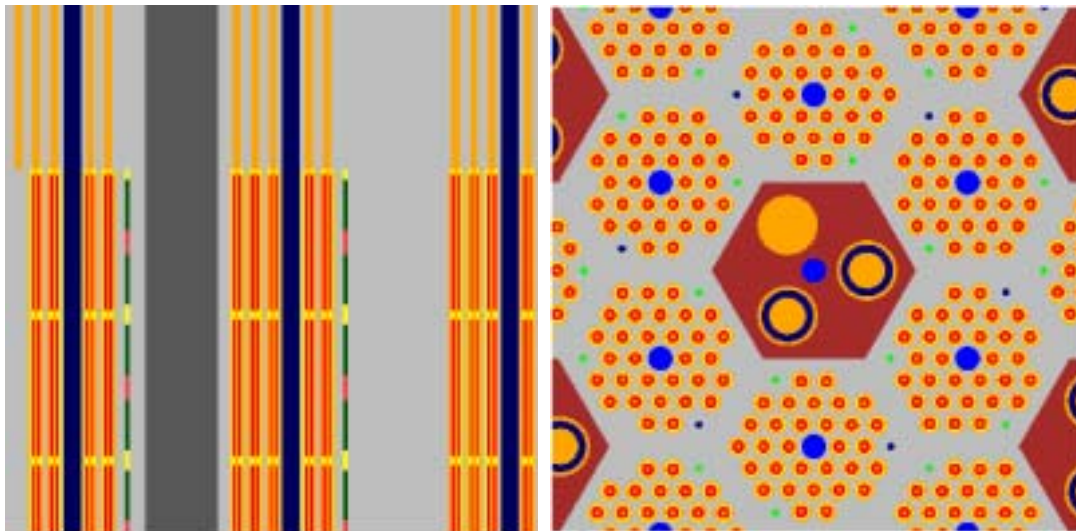


FIG. 2.42. Details of the Geometry used in the TRIPOLI4 code.

### ***Synthesis of the assumption of the initial diffusion calculation scheme***

The key factors in modelling the HTTR with the *transport-diffusion* calculation scheme are the following:

- the BP are axially homogenized
- the streaming effect is not taken into account everywhere in the core
- homogenised hexagons are considered without equivalent factor in order to respect the absorption rates of the BP and the control rod (CR) between both transport and diffusion calculation stages
- flux-weighted cross sections are used for the reflector
- CR insertion on the top reflector not taken into account

The following section provides the results taking into account these assumptions. Then, their impacts on the results have been assessed. They have been reconsidered in course of the revised calculations according to the results of these assessments.

### 2.2.2.2 First Results

#### **HTTR-FC and EX**

The calculations performed with the two different methods are gathered in Table 2-25. It is important to note that all the  $k_{\text{eff}}$ -values of the Table 2-25 are corrected with a  $\Delta k$  of -0.004 in order to take into account the presence of the control not fully withdrawn to the top of the reflector and not considered in the core models.

All the preliminary calculations underestimated the number of fuel columns needed to achieve the first criticality (Diffusion calculations: **10 columns**; Monte-Carlo calculations: **17 fuel columns**). As it can be seen in Table 2-25, the discrepancy between the calculations and the experiment at least ranges from  $\Delta k = 0.017$  to  $0.058$  at 18 fuel columns loading and from  $\Delta k = 0.01$  to  $0.033$  at full core.

It is noteworthy that the observed discrepancies decrease with increasing number of fuel columns in the core. Due to the large experimental uncertainty at 30 fuel columns loading, the differences between the calculations and the experiment are within the error bar, whereas at the thin annular core assembly the discrepancies are significant. Two reasons for the latter circumstance can be proposed. The first would be that the two steps transport-diffusion calculation based on the fundamental mode assumption would be less and less appropriate as one goes toward the annular core configuration. The second would concern the level of the actual boron impurity in the dummy fuel blocks and of the residual air (instead of helium) in the graphite pores. As far as the latter is concerned, the impurities of some dummy fuel blocks have been re-measured by JAERI and revised data [2-27] have been recommended for the recalculation of the first criticality (HTTR-FC2).

Table 2-25. Experimental, Monte-Carlo and First Diffusion Results

	<b>CRONOS2</b> 3D - (first model) • diffusion 8 gr • homog. Fuel block • no streaming • axially homog. BP	$\Delta k$ [% $\Delta k/k$ ]	<b>TRIPOLI4</b> 3D Monte-Carlo 172 gr & pointwise	$\Delta k$ [% $\Delta k/k$ ]	<b>EXPERIMENT</b>
<b>30 col</b>	<b>1.1698</b>	<b>14.5 %</b>	<b>1.14630 ± 0.0009</b>	<b>12.8 %</b>	<b>1.13630 ± (&gt; 3.6%)</b>
28	1.1691	14.5 %			
26	1.1596	13.8 %			
<b>24</b>	<b>1.1399</b>	<b>12.3 %</b>			
22	1.1158	10.4 %			
20	1.0886	8.14 %			
19	1.0745	6.93 %			
<b>18 col</b>	<b>1.0580</b>	<b>5.48 %</b>	<b>1.01710 ± 0.0009</b>	<b>1.68 %</b>	<b>subcritical</b>
16	1.0383	3.69 %			
14	1.0284	2.76 %			
12	1.0184	1.81 %			
10	1.0044	0.44 %			
9	.9970	-0.3 %			



In the course of the studies, the following reasons for the above mentioned discrepancies (especially for the simplified Transport – Diffusion calculation scheme) have been identified and quantified:

- A non-adequate treatment of the axial self-shielding in the BP rods,
- An underestimation of the neutron streaming (due to large channels of the CR blocks),
- The neglect of the detailed structure of the HTTR fuel block in the core calculations.

These main physical effects and their impacts on the core reactivity are briefly depicted in Figure 2.43 for the case of the annular core configuration. Similar tendencies can be observed for the full core configuration. Nevertheless, different absolute values (into bracket Figure 2.43) are obtained for the quantified physical effects due to the harder neutron spectrum in the fully loaded core. Indeed, the observed weight of the boron absorption in the BP is different in this case.

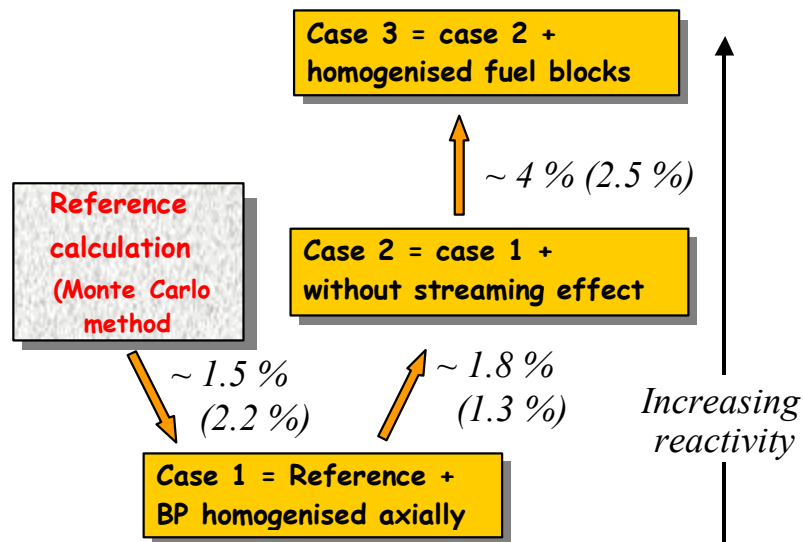


FIG. 2.43. Key factors in modelling and their impact on the reactivity for the 18 fuel columns core configuration (and full core).

For the thin annular core, starting from a best estimate calculation and neglecting the fact that the burnable poison was axially a succession of boron and graphite pellets leads to a predicted reactivity 1.5 % lower than considering the actual heterogeneous composition of these burnable poisons (3D Monte Carlo estimation). On the opposite, a core calculation that does not take into account the streaming effect will result in an increase of the reactivity of about 1.8 %. At this stage, it is interesting to note that by making two strong physical hypotheses a result not far from the best estimate calculation can be obtained. Finally, a discrepancy on the order of 4 % can be achieved if an insufficient description of the fuel block is used to model the high level of radial heterogeneity (2D Monte Carlo estimation).

Therefore, the HTTR-FC2 benchmark has been a good opportunity to implement the new enhanced methods coming from this analysis and to evaluate the progress considering the new data.



## HTTR-CR

### Transport-diffusion calculations

The control rod insertion depths have also been evaluated to achieve criticality in the three configurations recommended by the benchmark problem HTTR-CR (thin and thick annular core and fully loaded core). The results corresponding to the *transport-diffusion* method are given in the following Table 2-26 with the calculational approach illustrated in Figure 2.44.

Table 2-26. Critical Insertion Depths of the Control Rod

<i>Configuration: Number of fuel columns</i>	<i>Control rod critical positions from the bottom of the active core</i>
(thin annular core) 18	<b>251 cm</b>
(thick annular core) 24	<b>181 cm</b>
(fully loaded core) 30	<b>151 cm</b>

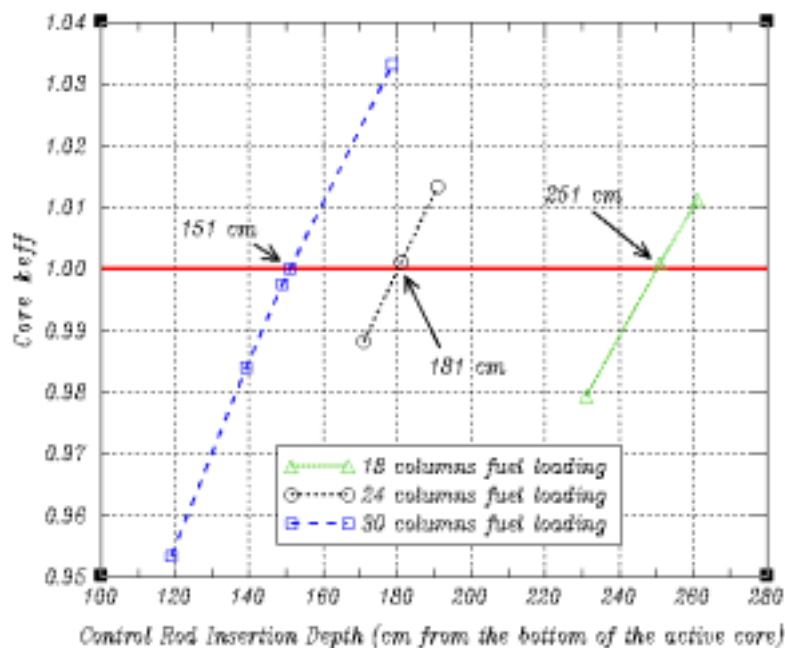


FIG. 2.44. Effective multiplication factor as a function of the control rod position.

It should be noted that the core model used here does not allow taking into account the orientation of the control rods in the homogeneous block description, as it is also the case for the position of the burnable poison in the fuel element.

### Transport-Monte Carlo Core Calculations

As far as the second method is concerned (*transport-Monte Carlo*) it is noteworthy that, as recommended by the benchmark problem, the structural materials of the control rod have not been taken into account. Conversely, a detailed axial description of the control rods has been done as depicted in Figure 2.45.

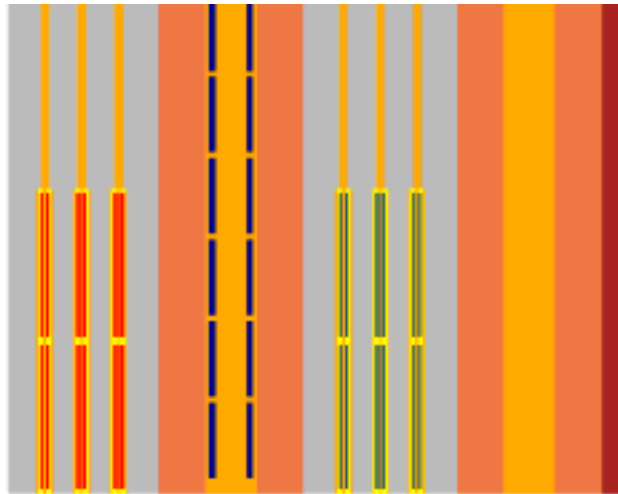


FIG. 2.45. Axial Cross Section View of the Inserted Control Rods.

The calculations aimed to tentatively evaluate the critical insertion depth of the control rod in the fully loaded core configuration have been performed. Two control rod insertion depths have been considered. The first one -178.7 cm- is the experimental value and the second one -170 cm- comes from an estimation of the control rod efficiency obtained in the diffusion calculations. The results are given in the Table 2-27 below:

Table 2-27. Reactivity for Two Control Rod Positions (Full Core)

<i>Configuration: Number of fuel columns</i>	<i>Control rod position from the bottom of the active core</i>	$k_{\text{eff}}$
(fully loaded core) 30	<b>178.7 cm</b>	<b>1.00850±0.0009</b>
(fully loaded core) 30	<b>170.0 cm</b>	<b>0.99840±0.0009</b>

From the results above, the critical rod position can be evaluated to **171 cm**. As a conclusion, one can note that, compared with the Monte Carlo, the control rod worth is overestimated with the diffusion method (without utilizing equivalence factor).

### 2.2.2.3 Modification to Model and Assessment of Improvements

#### ***New finite elements in the core diffusion model***

New finite elements recently implemented in CRONOS-2 have been used. They allow taking into account the exact position of the burnable poison in the fuel blocks and the fuel element orientation in the core.

Indeed, from the 2D transport calculations illustrated on the Figure 2.34, the fuel element was initially homogenised in one hexagonal finite element. Then, with the help of the new available finite elements, two different meshes were considered to describe the fuel elements with 24 radial meshes: 24 equilateral triangles (type I) or the cutting out depicted in Figure 2.46 (type II). Only the last one has been kept in the final model because of the fact that it is the only one that allows homogenising the poison with its associated graphite without homogenising partially the fuel compacts. Therefore, the fuel element structure is described by using three different mediums and the flux is calculated for each point described in Figure 2.46 (61 points for the hexagonal element).

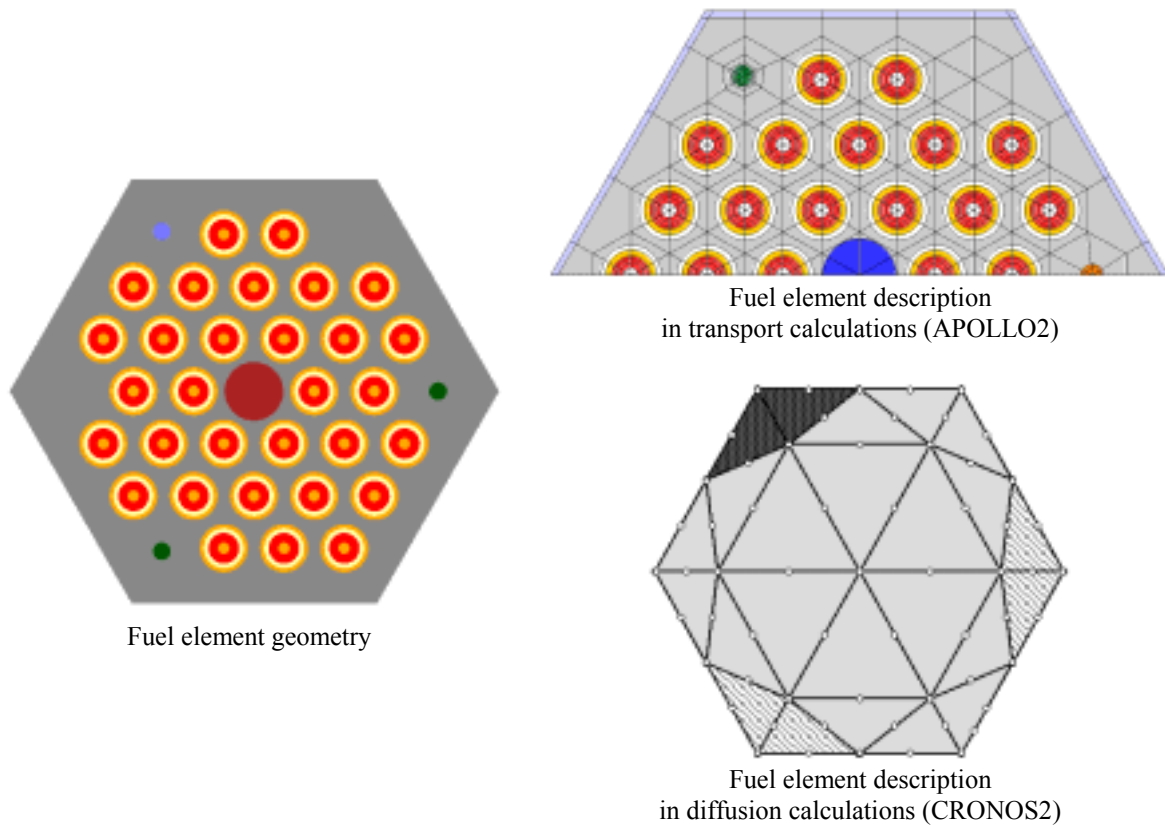


FIG. 2.46. Fuel element modelling in the improved Transport – Diffusion calculation scheme.

When such a heterogeneous fuel block geometry is used in the diffusion calculations, the impact have been evaluated for the three core configurations with 18, 24 and 30 columns, on the basis of a 2D simplified core with no axial leakage and with an average uranium enrichment. The diffusion calculations are compared to the Monte Carlo one. The results obtained for the first configuration are presented in Figure 2.47.

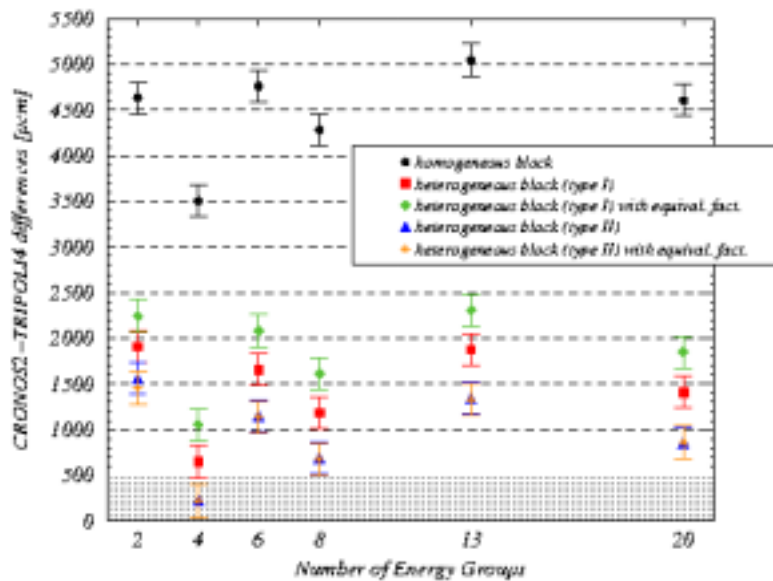


FIG. 2.47. 2D calculation comparison for different energy group structure in the diffusion calculation - 18 columns core -

As previously mentioned, the most important impact is obtained for the 18 columns core loading. As far as the **full core** is concerned, the diffusion-Monte Carlo 2D-discrepancies become **quite acceptable** with the use of the new finite elements. Moreover, the use of equivalence factors has been implemented in order to respect the global absorption rate between the APOLLO-2 transport calculations (172 groups) and the CRONOS-2 diffusion calculations with few groups. This option has not been considered afterwards because of its small impact (Figure 2.47) on the finite element of type II.

As a conclusion, a detailed description of the fuel block improves largely the results by giving a higher weight to the BP absorption in the fuel blocks. It allows getting quite acceptable values comparing to the reference TRIPOLI4 2D-calculation for the full core but a remaining discrepancy of about 1% can be observed for the annular core. This could be attributed to the cross section generation stage where the environment of the BP would not be representative of the one existing in the annular core configuration. Indeed, in this configuration the BP is surrounded by much more graphite (reflector) that thus increases the absorption flux-weighted cross-section. Besides, it would appear that there is actually no specific trend concerning the energy structure to be retained in the CRONOS-2 calculations.

### ***Streaming modelling in the transport calculations***

For improving the Transport – Diffusion calculation scheme, the streaming effect has been taken into account by using **anisotropic diffusion coefficients** in the core calculations. These diffusion coefficients have been evaluated in the fuel element transport calculation performed by APOLLO2. The Benoist method [2-29] available in APOLLO2 (called *TIBERE* model) is based on the B1 heterogeneous neutron leakage model. However, it might not be applicable in the large HTTR channels of the control rod graphite blocks (three large channels per block). Therefore, another analytical model (Benoist [2-30]) has also been tested on one control rod block alone and compared to the *TIBERE* model. With this formulation, the corrected diffusion coefficient is given by:

$$\frac{D_k}{\frac{1}{3}\lambda_m} = 1 + \frac{V_c}{V_t} \left( 1 + \frac{c}{\lambda_m} Q_k \right) \quad (1)$$

where:

- $Q_r = 1 - 1/\Delta$        $Q_z = 2 - \frac{3\pi}{4} B_z c$        $k = r, z$  (*radial or axial*)
- $c$  is the channel radius,  $\lambda_m$  is the mean free path of the moderator (graphite)
- $V_c = \pi c^2$  is the channel volume,  $V_t$  is the cell volume,  $B_z$  is the axial buckling

and:  $\Delta = \frac{\gamma}{\gamma + 1 - \frac{2b'}{1 - b' \frac{V_c}{V_t}}}$ ,  $b' = \frac{\gamma + \frac{1}{2}}{\gamma + 1}$ ,  $\gamma = \frac{c}{\lambda_m}$

In order to validate these models (*TIBERE* and analytical model), Monte-Carlo and diffusion calculations have been performed on the simplified geometry: the control rod block is surrounded by fuel elements and the axial structure of the geometry is the same as HTTR's core. The results are gathered in Table 2-28.

TABLE 2-28. Streaming Effect Calculated on the Simplified Core

	TRIPOLI4	CRONOS2	
	$k_{\text{effectif}} \pm 3\sigma$	2 gr.	8 gr.
<b>Homogeneous control block</b>	1,27958 ± 0,00090	1,26320	1,27397
<b>Heterogeneous control block</b>			
<i>TIBERE</i>	1,27247 ± 0,00100	1,26040	1,27057
Analytical formulation		1,25657	1,26672
<b>Streaming effect [pcm]</b>			
<i>TIBERE</i>		222	267
Analytical formulation	560 ± 135	526	570

The streaming effect calculated by CRONOS2 is underestimated when the anisotropic diffusion coefficients are evaluated by the *TIBERE* model. The underestimation of the streaming effect is the consequence of a non-adequate calculation of the axial diffusion coefficient (underestimation of – 20 to – 25 % on  $D_z$  calculation). On the other hand, the use of an analytical model for the axial diffusion coefficient allows obtaining results in good agreement with the reference (Monte-Carlo method).

#### 2.2.2.4 New Results

##### *New available data*

The new data benchmark (HTTR-FC2) has been defined by JAERI [2-27]. It has been a good opportunity to implement the new enhanced methods coming from the previous analyses. Besides, the impact of these new data on the reactivity has been evaluated on the basis of the Monte Carlo calculations and of the improved core diffusion calculations. These effects are listed in Table 2-29.

Table 2-29. Analysis of the Benchmark Data Impact on the Results

	Number of fuel columns	
	18	30
Residual air in the graphite porosity and re-evaluated impurities in the dummy fuel blocks	– 0.82 % <sup>(1)</sup>	– 0.3 % <sup>(1)</sup>
Worth of the CR insertion to the top of the reflector (R2 group)	– 0.22 % <sup>(1)</sup> – 0.32 % <sup>(2)</sup>	– 0.25 % <sup>(2)</sup>
Temporary neutron detector	non evaluated	non evaluated

<sup>(1)</sup> Evaluated with the Transport – Monte-Carlo calculation scheme

<sup>(2)</sup> Evaluated with the Transport - Diffusion calculation scheme

The overall effect can exceed 1 % according to the core configuration. One can note that the impact of the residual air in porosities and the impurities in graphite are much more higher in the thin annular core configuration. This could explain the decrease of the discrepancies between the experiment and the first calculation results, correlated to the number of dummy fuel blocks (graphite) discharged during the criticality approach.

## Diffusion calculation results

As far as the diffusion calculations are concerned, new developments carried out in APOLLO2 and CRONOS2 take into account:

- the exact position of the BP in the fuel block, by using new finite elements mesh in the core model
- the streaming effect, by generating anisotropic diffusion coefficients from both 2D-Pij calculations and analytical formulation.

The use of the HTTR-FC2 data associated with a complete description of the axial heterogeneity of the BP poison led to new core diffusion calculation results. This was done for several energy structures in CRONOS2 without observing a main trend which would allow to select a reference as energy mesh.

The final results are partially gathered in Figures 2.48 and 2.49. Figure 2.48 illustrates, with 8 energy groups, the impact of the different model assumptions on the reactivity as a function of the number of fuel columns loaded into the core. Figure 2.49 shows a streaming effect ranging from 2.25 % in the 18 columns core configuration to 1.8 % in the full core configuration. These results highlight also the importance of the used leakage model for evaluating the neutron streaming in the control rods graphite blocks. Indeed, the first model (*TIBERE* model) gave some values varying from 1.8 to 1.5 %.

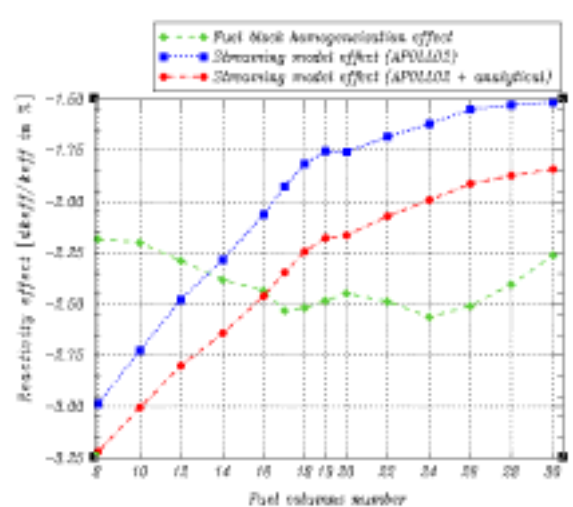
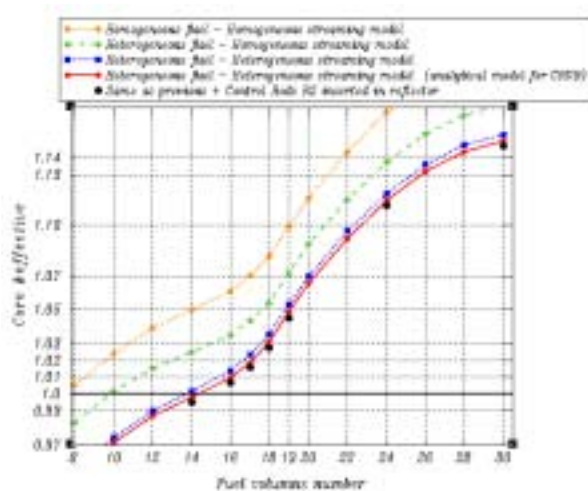


FIG. 2.48.  $k_{eff}$  values obtained with different core models. FIG. 2.49. Neutron streaming and fuel block homogenisation effect.

It is noticeable that the number of fuel columns needed to achieve criticality increases by about 7 in comparison with the first results (Table 2-25) when considering the presence of the detectors and the CR inserted in the upper reflector. However, at first criticality, a discrepancy remains between the diffusion and the Monte-Carlo calculations ( $0.9 \% < \Delta k/k < 1.7 \%$ ). This underscores the limits of a method based on a cross section homogenisation from a fundamental mode calculation (infinite medium) that is barely pertinent for the 18 columns core configuration. The actual environment (reflector blocks) should be considered and should take place instead of the white boundary condition in the 2D APOLLO2 transport calculations, before homogenising and collapsing locally the cross sections inside the fuel element.

### Final results of the HTTR-FC2 problem

All the final results are gathered in Table 2-30. In the case of the Monte Carlo code TRIPOLI4, the discrepancy between the measurement and the calculation for the 18 fuel columns configuration is reduced to  $\Delta k/k \sim 0.8 \%$ , when considering the revised data of the HTTR benchmark.

As far as diffusion calculations are concerned, the discrepancy is now reduced to  $\Delta k/k \sim 2.7$  or  $1.7 \%$  (depending on the number of energy groups), when taking into account the improved models and the revised data. After all, it must be stressed that all the calculation results obtained for the fully loaded core configuration fit the experiment, especially if one consider the experimental uncertainties.

Table 2-30. Experimental, Monte-Carlo and Final Diffusion Results

	<b>TRIPOLI4<sup>1)</sup></b> <i>3D Monte Carlo</i> <i>172 gr &amp; pointwise</i>	<b>CRONOS2<sup>1)</sup></b> <b>(Final model)</b> • 3D Diffusion • 4 groups - 8 groups	<b>EXPERIMENT</b>
<b>30 col.</b>	1.13833 $\pm$ 0.00090	1.1362 - 1.1451	1.1363 $\pm$ (> 3.6 %)
<b>24 col.</b>	*	1.1000 - 1.1096	1.0834 $\pm$ (> 2 %)
<b>19 col.</b>	1.02692 $\pm$ 0.00043	1.0351 - 1.0432	1.0152 $\pm$ ?
<b>18 col.</b>	1.00855 $\pm$ 0.00090	1.0178 - 1.0275	subcritical

<sup>1)</sup> detector impact included ( $\Delta k = 0.002$ ) but not modelised

### Excess reactivity HTTR-EX

All the excess reactivity are gathered below according to the previous values and the definition of the excess reactivity given in [2-22].

Table 2-31. Excess Reactivity for Different Core Configurations

Excess Reactivity <sup>1)</sup> ( $k_{\text{eff}} - 1$ )/ $k_{\text{eff}}$ [%]	Number of loaded fuel columns			
	<b>18</b>	<b>19</b>	<b>24</b>	<b>30</b>
<b>Diffusion</b>	1.7 < $\Delta k/k$ < 2.7	3.4 < $\Delta k/k$ < 4.1	9.1 < $\Delta k/k$ < 9,9	12,0 < $\Delta k/k$ < 12,7
<b>Monte-Carlo</b>	+ 0.85	+ 2.6	Non evaluated	+ 12,15

<sup>1)</sup> detector impact included ( $\Delta k = 0.002$ ) but not modelised

### Control rod position for criticality: HTTR-CR

The results presented in Table 2-32 correspond to the fully loaded core configuration. The control rod positions for obtaining criticality with all the inserted CR are: **178.7 cm** with the diffusion calculations and **177.9 cm** with Monte-Carlo code.

Table 2-32:  $k_{\text{EFF}}$  for Different CR Insertion Depths. Full Core

Control Rod Insertion Depth [cm]	APOLLO2 – TRIPOLI4	APOLLO2 – CRONOS2	Exp.
178,9	$1,00117 \pm 0,00024$	1,00020	critical
177,6	$0,99972 \pm 0,00038$	0,99840	*

It is noteworthy that the CR insertion depth is well evaluated with the diffusion core calculation despite of the discrepancies with the Monte Carlo results observed in this configuration without the inserted CR. This remark underscores the fact that the CR worth is overestimated with the diffusion method especially in this case where no equivalence factors have been used in order to respect either the flux or the absorption rates between the multi-group transport calculations and the broad group diffusion core calculations.

### Scram reactivity: HTTR-SC

In this section, the CR worth has been evaluated (Table 2-33) in the fully loaded core configuration and according to:

$$\rho = \frac{k_{\text{crit.}} - k_{\text{Control Rods IN}}}{k_{\text{crit.}} \times k_{\text{Control Rods IN}}}$$

As far as the CR insertion in the reflector is concerned, an unexplained result has been observed with the Monte Carlo calculation, whilst a good agreement can be observed for the overall CR worth inserted in the core. Once again, these results highlight the overestimation of the absorbant in the diffusion calculation when one compares the CRONOS2 results to those of TRIPOLI4.

Table 2-33. Control Rod Worth

	TRIPOLI4 (Monte Carlo)	CRONOS2 (diffusion)	Exp.
<i>CR inserted in the reflector</i>			
$k_{\text{critique}}$	$1,00117 \pm 0,00024$	1,00020	*
$k_{\text{RCR}}$	$0,92215 \pm 0,00040$	0,90245	*
$\rho_{\text{RCR}} [\%]$	<b>8,56</b>	<b>10,83</b>	<b>12,0 ± 1,2</b>
<i>All the CR inserted in the core</i>			
$k_{\text{critique}}$	$1,00117 \pm 0,00024$	1,00020	
$k_{\text{RCR}}$	$0,68396 \pm 0,00030$	0,63982	
$\rho_{\text{RCR}} [\%]$	<b>46,32</b>	<b>56,31</b>	<b>46,0 ± 4,6</b>



*Isothermal temperature coefficient in the core : HTTR-TC*

The temperature coefficients have been evaluated from the following expression:

$$\rho_{12} = \frac{k_{T_1} - k_{T_2}}{k_{T_1} \times k_{T_2}} \cdot \frac{1}{(T_1 - T_2)}$$

They have been estimated only from  $k_{\text{eff}}$  of the core diffusion calculations performed at different temperature (Table 2-34).

Table 2-34.  $K_{\text{EFF}}$  as Function of the Temperature

Temperature [K]	APOLLO2 – CRONOS2
300	1,00395
340	0,99724
380	0,99088
420	0,98455
460	0,97823
480	0,97525

According to the results gathered in Table 2-35, the temperature coefficients range from - 15 to - 16 pcm/°K between 300 to 420 K.

Table 2-35. Isothermal Temperature Coefficients

Temperature coefficients [pcm/°K]	APOLLO2 – CRONOS2
$\rho_{320}$	- 16,75
$\rho_{360}$	- 16,09
$\rho_{400}$	- 16,22
$\rho_{440}$	- 16,40
$\rho_{470}$	- 15,62

*2.2.2.5 Concluding Remarks*

The HTTR's core physics benchmarks have been treated with two different calculation scheme: a transport-diffusion method and a transport-Monte Carlo one. These benchmarks constitute the first opportunity in reactor physics to model and benchmark the codes and methods in thin **annular core geometry**. This important point led to the limitations of the classical two-steps core modeling based on a transport-diffusion chained calculations in order to take accurately into account the **core/reflector interface**. Moreover, one of the other characteristics of the HTTR core in its annular configuration was the presence of a large number of uncommon **big channels** offering the possibility for the neutrons to leak from the active zone (*streaming effect*). In addition to that, the **important axial and radial heterogeneities** in the core (burnable poison, many different enrichments) make the HTTR a real challenge in reactor physics.

**Quite acceptable** results are obtained in the **fully loaded core configuration**. Besides, some **discrepancies** between the calculations and the experiment appear for intermediate core configuration (**thin annular core**) close to criticality. Parts of these

discrepancies were reduced to  $\Delta k/k \sim 0.85 \%$  with the Monte-Carlo calculations taking into account the new benchmark data (air in porosity and impurities in graphite). As far as the deterministic approach is concerned, the discrepancies were analysed and tackled by different treatments. However, a difference of at least **1 %** remains between the diffusion and the Monte Carlo calculations in the **annular core** configuration when considering revised data.

#### 2.2.2.6 Addendum

Several energy structure have been tested in course of this study for the core diffusion calculations (see Table 2-36): the 2, 4, 8 and 20 group meshes, currently applied in the PWR studies and the 6 and 13 groups which have already been used in the GT-MHR related analyses.

Table 2-36. Group Structures for Core Diffusion Calculations

13 gr	6 gr	Limite inférieure en énergie [Mev]
1	1	$1,8315,10^{-1}$
2	2	$1,0104,10^{-3}$
3		$1,6745,10^{-5}$
4	3	$4,1293,10^{-6}$
5		$2,13,10^{-6}$
6	4	$1,305,10^{-6}$
7		$7,90,10^{-7}$
8		$6,2501,10^{-7}$
9	5	$3,91,10^{-7}$
10		$3,145,10^{-7}$
11		$1,15,10^{-7}$
12	6	$5,9,10^{-10}$
13		$1,1,10^{-10}$

*The 6 and 13 group structures*

8 gr	4 gr	2 gr	Limite inférieure en énergie [Mev]
1	1	1	$9,0718,10^{-1}$
2	2		$2,7324,10^{-3}$
3			$5,0045,10^{-3}$
4	3		$2,7679,10^{-6}$
5		$1,67,10^{-6}$	
6		$6,2501,10^{-7}$	
7	4	2	$1,6,10^{-7}$
8			$1,1,10^{-10}$

*The 2, 4 and 8 group structures*

20 gr	Limite inférieure en énergie [Mev]	20 gr	Limite inférieure en énergie [Mev]
1	4,493	11	$5,560.10^{-5}$
2	2,231	12	$4,000.10^{-6}$
3	1,353	13	$6,250.10^{-7}$
4	$4,979.10^{-1}$	14	$3,500.10^{-7}$
5	$1,832.10^{-1}$	15	$2,200.10^{-7}$
6	$6,738.10^{-2}$	16	$1,340.10^{-7}$
7	$2,479.10^{-2}$	17	$7,700.10^{-8}$
8	$9,119.10^{-3}$	18	$3,000.10^{-8}$
9	$2,035.10^{-3}$	19	$1,000.10^{-8}$
10	$4,540.10^{-4}$	20	$1,1.10^{-10}$

*The 20 group structure*

### 2.2.3. Germany [2-44]

#### 2.2.3.1. Introduction

The calculations described in this report were performed within the framework of the IAEA Co-ordinated Research Programme (CRP) on "Evaluation of the High Temperature Gas Cooled Reactor Performance". The benchmark problems of the HTTR's start-up core physics experiments have been defined by the Oarai Research Establishment, JAERI. The ISR of Research Centre Jülich joined this CRP with the aim of testing its reactor code system. The intention is to use the HTTR's start-up core experiments in order to find validation criteria for the calculational methods hitherto used and the scope within which they can be applied. Pre-test and post-test results have been obtained for the benchmark problems HTTR-FC and HTTR-EX described in detail in [2-34]: the number of fuel columns has been evaluated for the approach to first criticality when the fuel columns are charged from the core periphery, and the excess reactivities for 18, 24, and 30 fuel columns have been determined.

#### 2.2.3.2. Computational Methods and Nuclear Data

An overview of the code system used in the criticality calculations is given in Figure 2.50. All calculations performed are based on a 123-group cross section library which was generated from the JEF-2.2 nuclear data files [2-35]. The NITAWL module of the AMPX-77 system [2-36] was used to calculate shielded cross sections in the resonance region where the Nordheim integral treatment is employed. The double heterogeneity of the fuel due to the coated particles (cp) and the fuel rod assembly was taken into account by Dancoff factors which were calculated by the ZUT code [2-37] and supplied as input data for NITAWL. The 1-d cell calculations were performed by the TOTMOS code [2-38], a one-dimensional transport-corrected integral spectrum code for eigenvalue calculations and the generation of homogenized broad group constants. In order to evaluate the effect of the inhomogeneous distribution of the burnable poison (BP) in the axial direction, additional cell calculations in two-dimensional r-z geometry were performed by the discrete ordinates transport code DORT [2-39]. The eigenvalues and flux distributions of the whole reactor were calculated by the diffusion code CITATION [2-40] in 3-d triangular-z geometry. The OCTAGN module was used to transform the cell-weighted broad group cross section data into a format such that they can be used in the CITATION code. To take into account the effect of increased neutron streaming in the coolant channels and in the large holes of the core and the reflector, anisotropic diffusion coefficients were used. These were calculated by the MARCOPOLO code on the basis of the multigroup integral transport theory [2-41]. In order to estimate the influence of this effect the calculations for the whole reactor were performed with and without streaming corrections.

The geometric dimensions of the fuel and core components and the atom number densities of the materials given in Ref. [2-34] were used in the pre-test calculations. In the post-test calculations residual air in the pores of the graphite, aluminium in the temporary neutron detector holders and a revised boron impurity in some dummy fuel blocks were taken into account [2-42].

#### ***Dancoff Factors***

The NITAWL code considers the double heterogeneity of a system by Dancoff factors. The Dancoff factor is defined as the probability that a neutron emitted from the surface of the fuel region of the fuel element under consideration will have its next collision

in the fuel region of any other fuel element. Applying this definition to a fuel rod lattice filled with fuel in the form of coated particles the ZUT code calculates the Dancoff factor as a sum of the single rod Dancoff factor and the probability that a neutron leaving the first rod will reach another fuel rod and be absorbed there. The Dancoff factors were calculated for all the different types of compacts and are listed in Table 1 together with the corresponding  $U^{238}$  resonance integrals. The Dancoff factors remain nearly constant because they are only dependent on the slightly varying geometric dimensions of the fuel compacts and rods, and on the density of the graphite matrix. In pre-test and post-test prediction the Dancoff factors are the same.

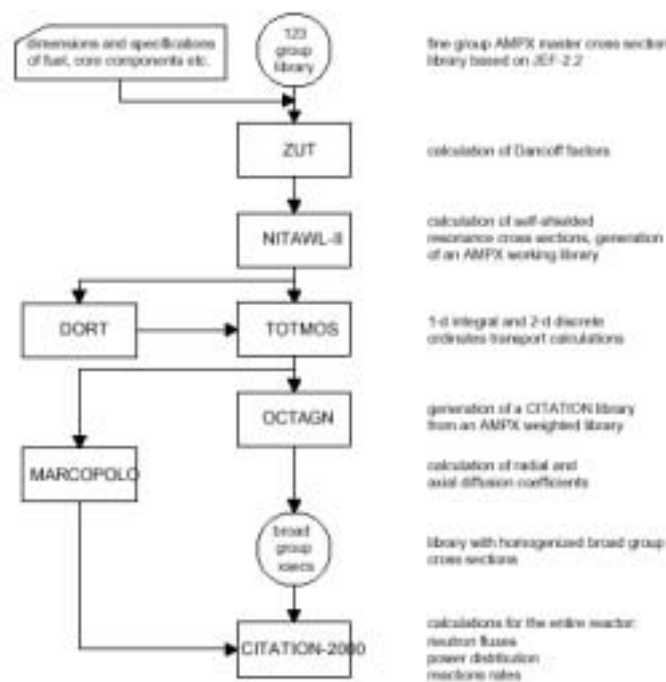


FIG. 2.50. Programme Structure of HTTR Calculations.

Table 2-37. Dancoff Factors and  $U^{238}$  Resonance Integrals for the Different Uranium Enrichments

Enrichment (wt.%)	Packing Frac.	Vol. of Fuel Comp. (cm <sup>3</sup> )	Boron Imp. (ppm)	Dancoff Factor	$U^{238}$ Res.Int.
3.301	29.6	17.63	0.95	0.7225	43.82
3.864	30.4	17.69	0.91	0.7283	44.12
4.290	30.5	17.70	0.90	0.7284	44.17
4.794	30.3	17.72	0.88	0.7296	44.20
5.162	30.5	17.65	0.90	0.7300	44.36
5.914	30.3	17.70	0.51	0.7280	44.67
6.254	29.9	17.69	0.54	0.7270	44.85
6.681	30.3	17.65	0.50	0.7280	44.91
7.189	30.8	17.69	0.85	0.7258	45.09
7.820	28.8	17.67	0.87	0.7214	44.97
9.358	29.8	17.72	0.89	0.7223	45.91
9.810	29.3	17.71	0.90	0.7245	45.89

### 2.2.3.3. Pre-Test Calculations

#### 1-d Cell Calculations

For the 15 types of BP-fuel combinations given in Appendix B of [2-34], one-dimensional cell calculations were performed by the TOTMOS code using the following scheme:

1. 123 cell-weighted group constants of the cp-cell-model were calculated in spherical geometry using a 123 group cross section library. The cell model consists of 3 zones: the kernel, the coatings, and the corresponding matrix zone. A white boundary condition was used at the outer surface of the cell.
2. Subsequently, 123 cell-weighted group constants of the fuel rod cell were calculated in cylindrical geometry and with the same group structure as used for the cp cell. The zones of the cylindrical fuel rod cell were: the central hole, the fuel compact, the graphite sleeve, the coolant channel, and the corresponding graphite block. The fuel rod cell is shown in Figure 2.51. The group constants of the materials in the fuel zone were the cell-weighted cross sections resulting from the cp cell calculation. The cross section of the fuel cell was equivalent to the cross section of the fuel block divided by the number of fuel rods. A white boundary condition was used at the outer surface of the cell.
3. In a subsequent third cell calculation, the cylindrical cell model shown in Figure 2.52 consists of a BP rod surrounded by a second zone representing the remaining fuel block. The group constants of the materials in this second zone were the cell-averaged 123 group constants resulting from the fuel rod cell calculation. The cross section of the BP cell was the same as the cross section of the fuel block divided by the number of BP rods. Again a white boundary condition was used at the outer cell surface. In this last step of the cell calculations, cell-weighted condensation was performed to four broad energy groups used in the calculations of the whole reactor. The energy group structure of the four groups is given in Table 2-38.

The  $k_{\infty}$ -values of the cp-, fuel- and BP-cell calculations are given in Table 2-39 for all 15 types of BP-fuel combinations.

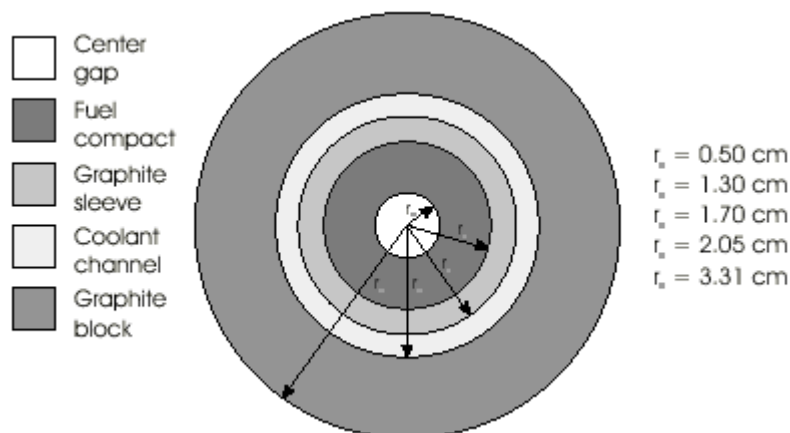


FIG. 2.51. Pre-Test 1-d Cylindrical Model of the Fuel Rod Cell for TOTMOS.

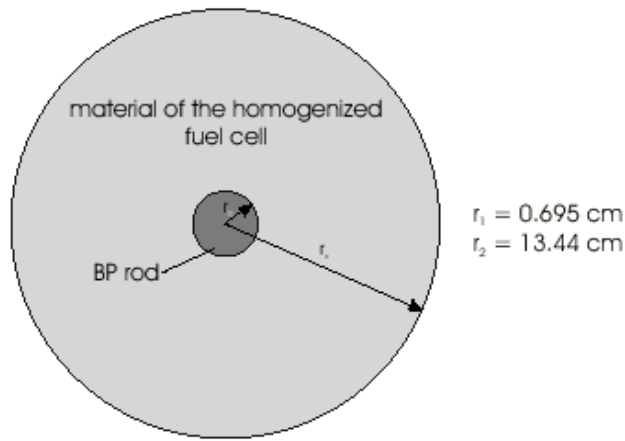


FIG. 2.52. Pre-Test 1-d Cylindrical Model of the BP Cell for TOTMOS

Table 2-38. Group Structure in the Diffusion Calculation

4 Group Set (Pre-Test)		26 Group Set (Post-Test)	
Group	Upper Energy Boundaries (eV)	Group	Upper Energy Boundaries (eV)
1	$1.492 \times 10^7$	1	$1.492 \times 10^7$
		2	$7.408 \times 10^6$
		3	$3.679 \times 10^6$
		4	$6.721 \times 10^5$
2	$1.111 \times 10^5$	5	$1.111 \times 10^5$
		6	$1.931 \times 10^4$
		7	$3.355 \times 10^3$
		8	$1.585 \times 10^3$
		9	$7.485 \times 10^2$
		10	$2.754 \times 10^2$
		11	$1.301 \times 10^2$
		12	$6.144 \times 10^1$
3	29.0	13	29.0
		14	13.7
		15	8.32
		16	5.04
4	1.86	17	2.38
		18	1.29
		19	0.65
		20	0.35
		21	0.20
		22	0.12
		23	0.08
		24	0.05
		25	0.02
		26	0.01

Table 2-39. Pre-Test Results of the TOTMOS Cell Calculations; No BP Adjustment

ID. No.	Enr. (wt.%)	$k_{\infty}$ -Values of the		
		Cp Cell	Fuel Cell	BP Cell
343320	3.4	0.6282	1.4285	1.1309
393320	3.9	0.6562	1.4604	1.1819
673320	6.7	0.7782	1.5457	1.3355
793320	7.9	0.8208	1.5610	1.3716
433120	4.3	0.6771	1.4957	1.2167
483120	4.8	0.6995	1.5142	1.2514
943120	9.4	0.8713	1.5996	1.4161
993120	9.9	0.8854	1.6021	1.4254
433325	4.3	0.6771	1.4790	1.1887
523325	5.2	0.7163	1.5095	1.2437
633325	6.3	0.7619	1.5376	1.2955
593125	5.9	0.7484	1.5476	1.2855
633125	6.3	0.7619	1.5559	1.3002
723125	7.2	0.7969	1.5726	1.3351
793125	7.9	0.8208	1.5802	1.3565

***2-d cell calculations and BP adjustment***

In the 1-d TOTMOS cell calculations, the axial heterogeneity of the BP is not explicitly taken into account. Thus, the efficiency of the BP is overestimated since the axial self-shielding of the BP cannot be considered. In order to find out the influence of the axial heterogeneity of the BP, 2-d cell calculations were performed by the discrete ordinates transport code DORT. For this purpose two series of 2-d cell calculations were performed:

1. In a first series of DORT calculations, the BP region is homogenized as in the 1-d cell. The results are given in Table 2-40 and compared with the corresponding results obtained by the TOTMOS code. It is seen that there is quite good agreement between the two computational methods.
2. In a second series of calculations the axial heterogeneity of the BP is explicitly considered in the 2-d DORT calculations. The geometric model is shown in Figure 2.53 and the results are presented in Table 2-40. It can be seen that the heterogeneity has quite a large influence on the infinite multiplication constant of the BP cell and considerably reduces the efficiency of the BP in comparison to the 1-d cell calculations.

In order to transfer this information into the 1-d cell calculation the  $B^{10}$  concentration was reduced in such a way that the resulting  $\Delta k$  was the same as that obtained in the two series of DORT calculations. In all BP-fuel combinations, the  $B^{10}$  concentration had to be reduced by 22% to 30 %.

Table 2-40. Pre-Test Infinite Multiplication Factors of the BP Cell Obtained by Different Methods

Case	$k_{\infty}$ -Values of the BP Cell				
	TOTMOS Homog.	DORT Homog.	$\Delta k$ (TOTM.-DORT)	DORT Heterog.	$\Delta k$ (DORT <sub>Het.</sub> -DORT <sub>Hom.</sub> )
343320	1.1309	1.1347	0.0038	1.1741	0.0394
393320	1.1819	1.1855	0.0036	1.2225	0.0370
673320	1.3355	1.3376	0.0023	1.3654	0.0275
793320	1.3716	1.3737	0.0021	1.3981	0.0244
433120	1.2167	1.2204	0.0037	1.2575	0.0371
483120	1.2514	1.2548	0.0034	1.2896	0.0348
943120	1.4161	1.4181	0.0020	1.4417	0.0236
993120	1.4254	1.4274	0.0019	1.4499	0.0225
433325	1.1887	1.1929	0.0041	1.2343	0.0414
523325	1.2437	1.2475	0.0037	1.2852	0.0378
633325	1.2955	1.2988	0.0033	1.3329	0.0342
593125	1.2855	1.2892	0.0037	1.3264	0.0373
633125	1.3002	1.3037	0.0036	1.3401	0.0363
723125	1.3351	1.3383	0.0032	1.3719	0.0335
793125	1.3565	1.3594	0.0029	1.3908	0.0313



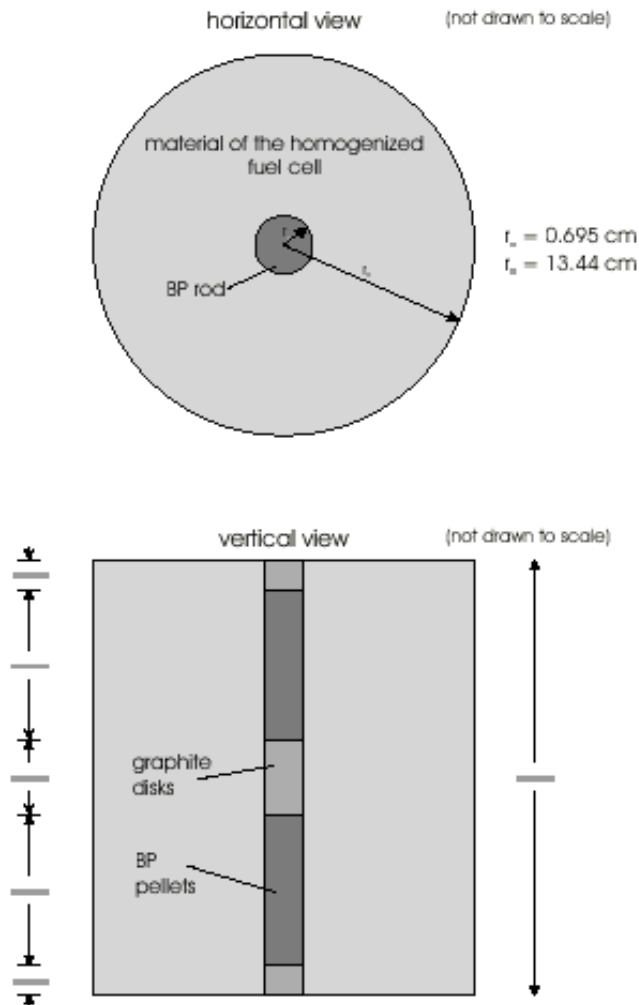


FIG. 2.53. Pre-Test 2-d Cylindrical Model of the BP Cell for DORT.

### ***Streaming correction***

The presence of the insertion holes in the control rod guide blocks and of the coolant channels in the fuel and reflector blocks leads to an increased neutron streaming in the axial direction. A possibility of treating this effect within the diffusion theory is the use of anisotropic diffusion coefficients. These are determined by the MARCOPOLO code which calculates anisotropic multigroup diffusion coefficients from the leakages of a heterogeneous lattice cell. For all types of blocks with insertion holes or coolant channels, 1-d cylindrical cells are defined which, in most situations, consist of a central hole representative of the block and surrounded by an associated material zone. As typical examples, the cell models of the dummy fuel block MB-2 and of the control rod guide block CB-1 are shown in Figures 2.54 and 2.55. The MARCOPOLO code allows the calculation of the buckling-independent diffusion constants  $D_k^g$  ( $k=r,z$ ) in these cells taking into account linear anisotropic scattering. The group constants used in the MARCOPOLO calculations were provided by the TOTMOS code. For the control rod guide blocks CB-1 and CB-2 and the reflector block RB-1, the ratio of the anisotropic diffusion coefficients to the homogeneous diffusion coefficients are given in Table 2-41. The resulting values for all block types of interest are input into the CITATION code in the form of these streaming correction factors.

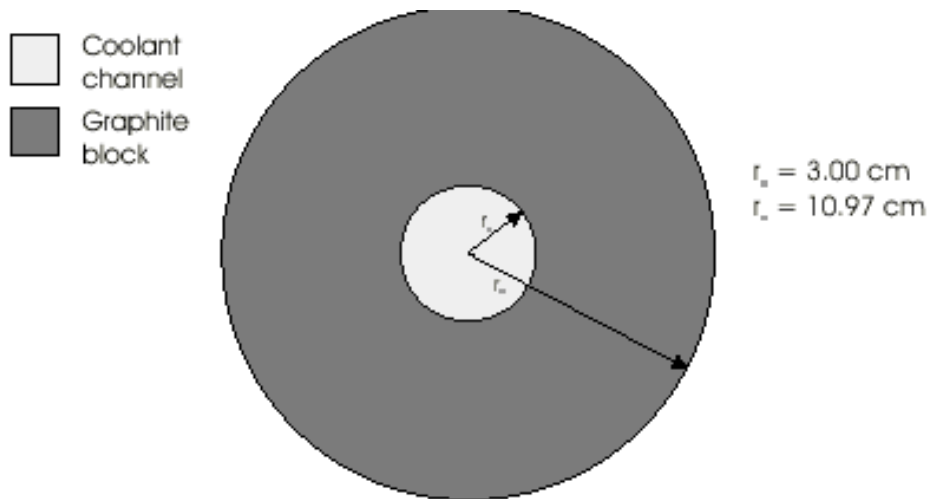


FIG. 2.54. 1-d Cylindrical Model of the Simplified Dummy Fuel Block B (MB-2).

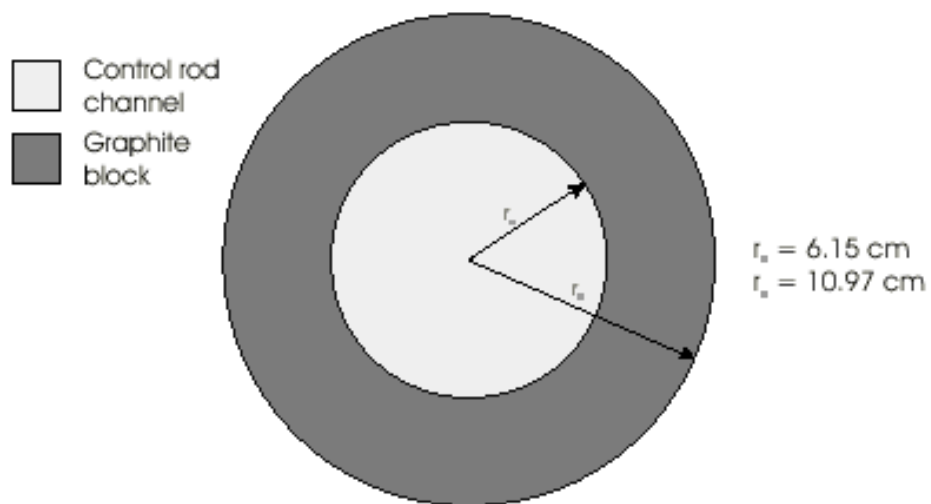


FIG. 2.55. 1-d Cylindrical Model of the Control Rod Guide Block CB-1.

Table 2-41. Pre-Test Streaming Correction Factors Obtained by the MARCOPOLO Code

Group	Streaming Correction Factors					
	CR Guide Block CB-1		CR Guide Block CB-3		Repl.Refl.Block RB-1	
	$D_r/D_{hom}$	$D_z/D_{hom}$	$D_r/D_{hom}$	$D_z/D_{hom}$	$D_r/D_{hom}$	$D_z/D_{hom}$
1	1.1403	1.5740	1.1562	1.6317	1.0199	1.0513
2	1.1761	1.9333	1.1963	2.0358	1.0246	1.0786
3	1.1812	1.9497	1.2016	2.0537	1.0286	1.0836
4	1.1877	2.0243	1.2090	2.1369	1.0307	1.0908

### ***Whole reactor calculations***

Using the 4-group cross sections from the NITAWL-TOTMOS cell calculations the whole HTTR reactor was modelled with the CITATION diffusion code. A 3-dimensional triangular-z model was chosen. Each block was divided horizontally into 6 meshes and vertically into 4 meshes. The assembly was modelled by dividing the volume into spectral zones related to the material compositions. There are 45 different material zones.

Six pairs of control rods in the side reflector of the HTTR cannot be fully withdrawn to the top of the reflector. The effect of this CR insertion on reactivity is given as  $\Delta k = 0.004$  in Ref. [2-34] and is subtracted from the calculated  $k_{\text{eff}}$ -values. Fuel columns are loaded clockwise from the periphery to the centre according to the loading order given in Ref. [2-34]. A thin annular core is formed at 18 fuel column loading, and a thick annular core is achieved when 24 fuel columns are loaded. A full core contains 30 fuel columns. According to this loading scheme four series of diffusion calculations were performed for 9 up to 30 fuel columns in the core:

- without streaming correction of the diffusion constant and without BP adjustment,
- with streaming correction, but without BP adjustment,
- with BP adjustment, but without streaming correction,
- with both corrections.

The  $k_{\text{eff}}$ -values of these four series are shown in Figures 2.56 and 2.57, respectively. As can be seen, the influence of the streaming correction is nearly independent of the BP adjustment. With and without BP adjustment the streaming correction causes a difference in  $k_{\text{eff}}$  from  $\Delta k = 0.02$  at a loading with 9 fuel columns down to  $\Delta k = 0.015$  for the case of a fully loaded core. This decrease in  $\Delta k$  can be explained by the fact that dummy fuel blocks with large holes and a great neutron streaming effect are subsequently replaced by fuel blocks which exhibit nearly no streaming effect. Moreover, it is found that the neutron streaming in the coolant channels of the top and bottom replaceable reflector can be neglected, because the decrease in  $k_{\text{eff}}$  caused by this effect was only  $\Delta k = 0.08\%$ .

On the other hand, the effective multiplication constants are increased by the BP adjustment: the "boron adjusted"  $k_{\text{eff}}$ -values are greater than the uncorrected ones, and the difference in  $k_{\text{eff}}$  increases with the increasing number of fuel rods. But it is evident that the increase of the effective multiplication constant due to the BP adjustment is not compensated by the effect of neutron streaming.

When taking into account the neutron streaming in the channels and holes of core and reflector, the neutron shielding in the BP rods, and when the reactivity of the CR insertion is subtracted, the first criticality will be achieved at 16 fuel columns loading. The excess reactivity amounts to  $\Delta k/k = 0.42\%$ . The excess reactivity of the thin, thick, and the fully loaded core is 2.48 %, 10.27 %, and 13.85 %, respectively. All results are shown in Tables 2-42 and 2-43.

These results agree very well with the average results of the deterministic calculations of all CRP-5 participants presented at the 1st Research Coordination Meeting [2-22].

Table 2-42. Pre-Test Effective Multiplication Factor and Excess Reactivity at the First Criticality

No. of Fuel Columns	$k_{eff}$	$\rho$ [% $\Delta k/k$ ]
15	0.9991	-0.90
16	1.0042	+0.42

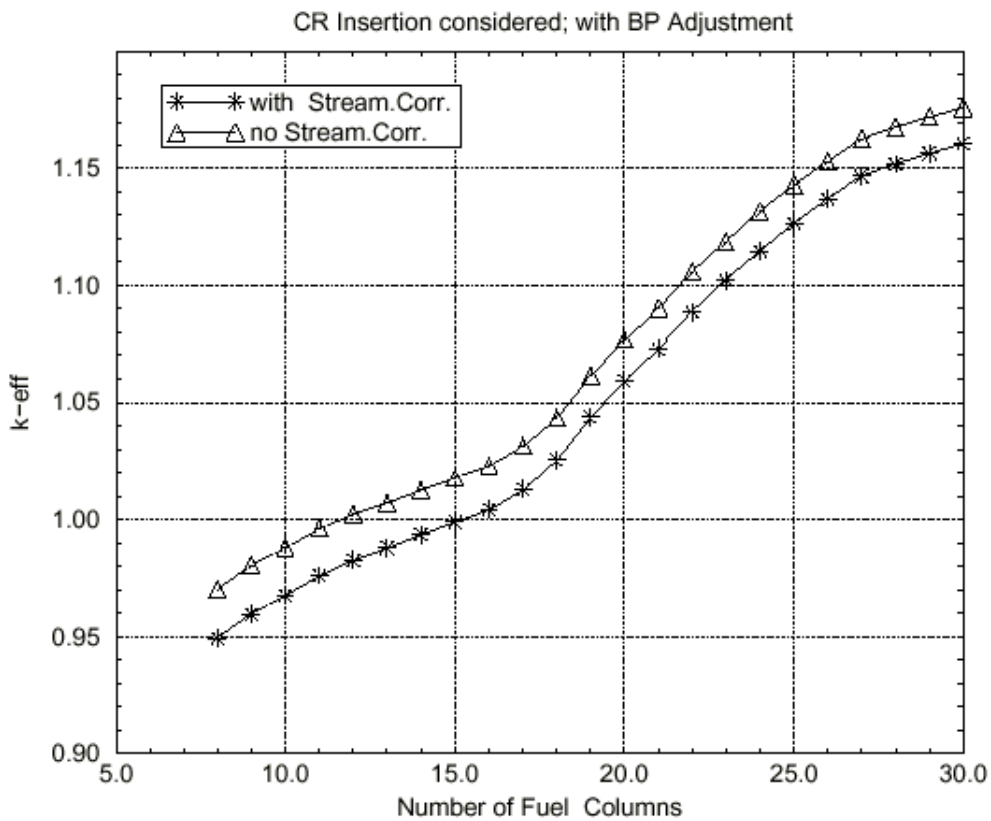


FIG. 2.56. Pre-Test Effective Multiplication Factor as a Function of the Number of Fuel Columns.

Table 2-43. Pre-Test Effective Multiplication Factor and Excess Reactivity at 18, 24, and 30 Fuel Columns Loading

No. of Fuel Columns	$k_{eff}$	$\rho$ [% $\Delta k/k$ ]
18	1.0254	2.48
24	1.1145	10.27
30	1.1607	13.85

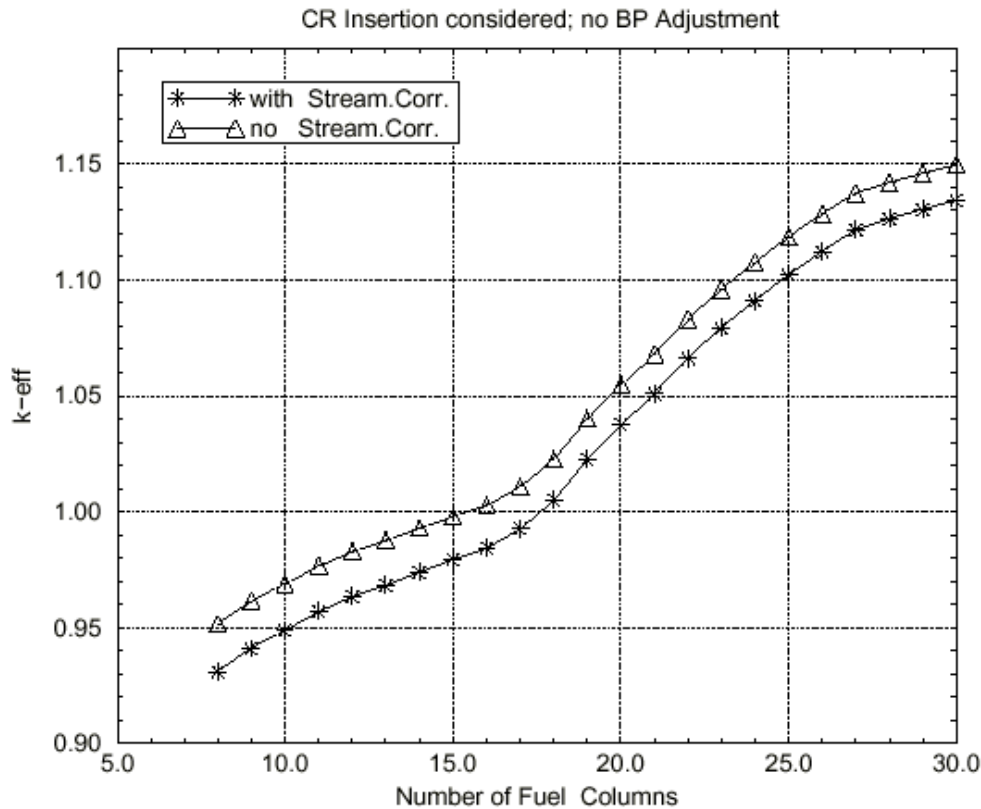


FIG. 2.57. Pre-Test Effective Multiplication Factor as a Function of the Number of Fuel Columns.

### Discussion of the pre-test results

In the pre-test diffusion calculation the first criticality of the HTTR was achieved with 16 fuel columns in the case of fuel columns loading from the core periphery. This corresponds to a critical mass of 29.6 kg  $U^{235}$ . It turned out that the burnable poison in the fuel blocks and the neutron streaming in the holes of the CR guide columns had a great influence on the neutronic characteristics of the HTTR and that an accurate treatment of these effects is required.

The experiment showed that the HTTR got critical with 19 fuel columns with an excess reactivity of about  $\Delta k/k=1.5\%$ . The discrepancy between measurement and the FZJ pre-test calculation amounted to  $\Delta k=0.0287$  at the first criticality for a 4 group diffusion calculation with no leakage iteration; at 30 fuel columns in the core this difference was with  $\Delta k=0.0261$  of the same order as can be seen on Table 2-44. Only at 27 and 30 fuel columns loading the calculated excess reactivities were in the experimental error intervals, showing some error compensating effects in the calculations. Six possible reasons for these discrepancies are given in the following:

- a non-adequate modelling of the fuel and BP unit cells,
- the neglect of the detailed structure of the HTTR fuel block in the whole core calculations,
- the use of few group homogenized cross sections in the whole core diffusion calculation without leakage iteration,

- the consideration of another than the actual boron impurity in some dummy fuel blocks and of helium instead of air in the graphite pores,
- a not yet adequate treatment of the self-shielding in the BP rods,
- and an underestimation of the neutron streaming.

Table 2-44. Results of the Pre-Test Diffusion Calculations together with the Experimental Results

No. of Fuel Columns	$k_{\text{eff}}$ Calc.	$k_{\text{eff}}$ Exper.	$\Delta k$ Calc.-Exp.	$\rho$ [% $\Delta k/k$ ] Calc.	$\rho$ [% $\Delta k/k$ ] Exper.
9	0.9596	0.9282	0.0314	-4.21	-7.7 <sup>a</sup>
12	0.9827	0.9481	0.0346	-1.76	-5.5 <sup>a</sup>
15	0.9991	0.9652	0.0339	-0.09	-3.6 <sup>a</sup>
16	1.0042	0.9701	0.0341	0.42	-3.1 <sup>a</sup>
17	1.0129	0.9785	0.0344	1.27	-2.2 <sup>a</sup>
18	1.0254	0.9913	0.0341	2.48	-1.0 <sup>a</sup>
19	1.0439	1.0152	0.0287	4.21	1.5
21	1.0731	1.0417	0.0314	6.81	4.0±1.1 <sup>b</sup>
24	1.1145	1.0834	0.0311	10.27	7.7±2.1 <sup>b</sup>
27	1.1469	1.1198	0.0271	12.81	10.7±3.0 <sup>b</sup>
30	1.1607	1.1346	0.0261	13.85	12.0±3.3 <sup>b</sup>

(CR Insertion considered  $\Delta k=0.004$ )

<sup>a</sup> from 1/M Measurement [2-43]

<sup>b</sup> from IK Method

#### 2.2.3.4. Post-Test Calculations

##### **Assessment of differences**

The first four possible reasons of the differences between pre-test prediction and experiment were analysed in more detail using simplified (1-d and 2-d) models of a thin annular core with 18 fuel columns:

- improved modelling of the fuel and burnable poison (BP) unit cells,
- consideration of the exact position of the BP rods in the fuel blocks,
- use of many group homogenized cross sections or accurate leakage iteration in the whole core diffusion calculation,
- and consideration of the revised boron impurity in some dummy fuel blocks and of residual air in the pores of the graphite, as proposed by JAERI [2-42].

In the course of these studies it turned out that the change in the unit cell models, the increase of the boron impurity in some dummy fuel blocks and the replacement of the helium in the pores of the graphite blocks by air reduced the multiplication constant of the simplified core models. However, this reduction was more than compensated by the increase of  $k_{\text{eff}}$  in a 26 groups diffusion calculation or a 4 groups diffusion calculation with leakage feedback used to describe the core/reflector coupling accurately.

In all pre-test calculations, the HTTR fuel blocks were homogenized with only six triangular meshes per fuel block and the **exact position** of the BP rods in the fuel blocks as mentioned in the 2<sup>nd</sup> item was not considered. When taking into account the detailed BP positions in the whole core diffusion calculations with 24 horizontal meshes per fuel block the multiplication constant decreased significantly, as can be seen on Table 2-45. Furthermore, it can be seen that the increase of the group number from 26 up to 52 in the whole core diffusion calculation has only a small influence on the  $k_{\text{eff}}$ -values. Thus, a group structure of 26 energy groups seems to be sufficient to get reliable results. The 26 energy group structures is given in Table 2-38.

Table 2-45. Effective Multiplication Constants and  $\Delta k$  Values for a Simplified 2-d, Triangular HTTR Core Model (18 Fuel Columns) Using Different Unit Cell Models and Group Structures (No Streaming Correction and BP Adjustment; Helium in Graphite Pores and Old Boron Impurity)

	$k_{\text{eff}}$		$\Delta k$
	Old Unit Cells 24 Mesh Hom. Model	New Unit Cells 24 Mesh Het. Model	New-Old Model
4 Groups	1.0484	0.9857	-0.0637
26 Groups	1.0840	1.0419	-0.0421
$\Delta k$ 26-4 Groups	+0.0355	+0.0563	—
52 Groups	1.0842	1.0430	-0.0412
$\Delta k$ 52-26 Groups	+0.0004	+0.0011	—

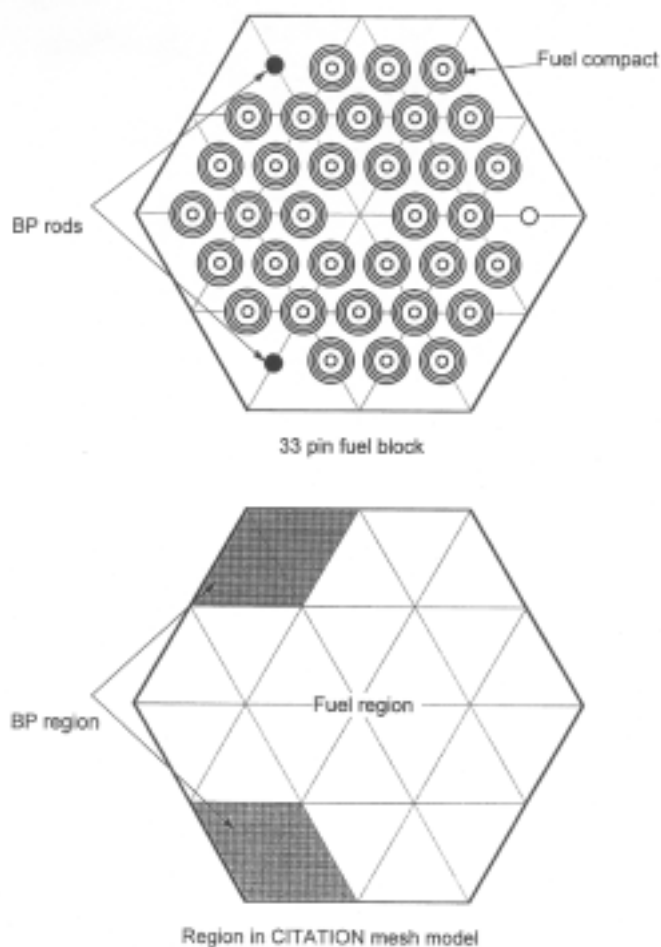
Concerning item 5 of the possible reasons for the discrepancies, the axial self-shielding in the BP rod was considered by a  $B^{10}$  adjustment in the 1-d cell calculations of the pre-test prediction. Now, an improved treatment of the axially heterogeneous distribution of the BP was performed by 2-d calculations.

As to item 6 of the possible reasons, the comparison with the Japanese Monte Carlo calculations shows, that the neutron streaming has been underestimated in the previous calculations. Therefore, modified diffusion constants for treating this effect were determined on the basis of the Japanese results.

### ***Heterogeneity of the HTTR fuel block***

In order to consider the exact position of the BP rods in the whole core diffusion calculation 24 horizontal meshes per fuel block have to be chosen and each fuel block has to be divided into three different regions:

- one fuel region, corresponding to 18/24 of the whole block,
- two BP regions, each corresponding to 2/24 of the whole block,
- one "empty BP" region, according to 2/24 of the whole block.



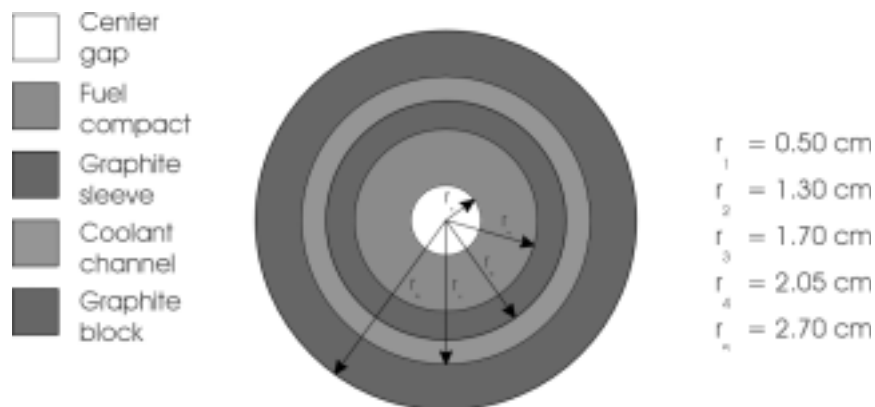
***FIG. 2.58. HTTR Fuel Block and Post-Test Model Used in the Diffusion Calculation with 24 Horizontal Meshes per Block.***



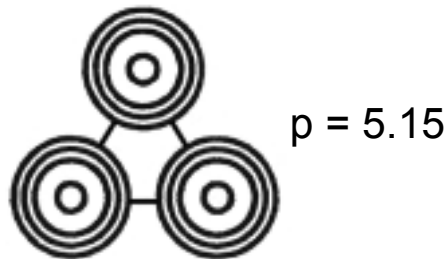
The fuel block with 33 fuel pins is shown in Figure 2.58 together with the modelling of the different regions in the whole core diffusion calculation with 24 horizontal meshes per block. Basing on this subdivision of the fuel block it is necessary to create new unit cell models adapted to this situation.

***New 1-d fuel cell model***

The new fuel cell presented in Fig 2.59 together with the basic geometry has an outer radius of 2.70 cm corresponding to the pitch of the fuel rod lattice in the fuel block. Explicitly modelled are: the inner helium channel, the compact, the sleeve, and the outer helium gap. The cell-weighted 26 group cross sections generated on the basis of this new fuel cell model are used in the fuel region of each fuel block in the whole core diffusion calculation.



1-d cylindrical model of the fuel cell for TOTMOS



Dimension of the Fuel Lattice Used as Basis in the New Fuel Cell Model

FIG. 2.59. Post-Test 1-d Cylinder Model of the Fuel Cell.

***New 2-d BP Cell Model***

In the pre-test prediction, the axial self-shielding of the BP was taken into account by a reduction of the  $B^{10}$  concentration in the 1-d TOTMOS cell calculations. Now, in the post-test calculations the axial self-shielding was considered by another method: it was provided that the absorption rate in the BP-regions of the whole core diffusion calculation is the same as in the detailed 2-d DORT-cell calculation for these regions.

Thus, the axial self-shielding of the BP was taken into account by a more accurate method compared to the method of reducing the  $B^{10}$  concentration. This reduction overestimated the efficiency of the BP compared to the use of group constants obtained in a 2-d DORT-cell calculation by about  $\Delta k \approx 0.0065-0.0077$ .

The revised 2-d cell model used in the DORT calculations is shown in Fig 2.60. This BP unit cell consists of three radial zones: the BP rod, surrounded by a graphite zone, corresponding to the area of 2/24 block, and a third radial zone representing the remaining fuel block. The group constants in this third radial zone are the cell-averaged group constants resulting from the fuel cell calculation. The first radial zone, the BP rod, is divided axially into five zones, representing the BP pellets, the graphite disks, and the upper and lower plugs of the graphite block. In the 2-d cell calculation the group constants are homogenized over the two inner radial zones and over the five axial zones of the BP rod and condensed to 26 energy groups. These 26 zone-weighted group constants obtained by the 2-d DORT cell calculation are used in the two BP-regions of each fuel block in the whole core diffusion computation.

The cross section of this BP cell corresponds to the cross section of the fuel block divided by the number of BP rods.

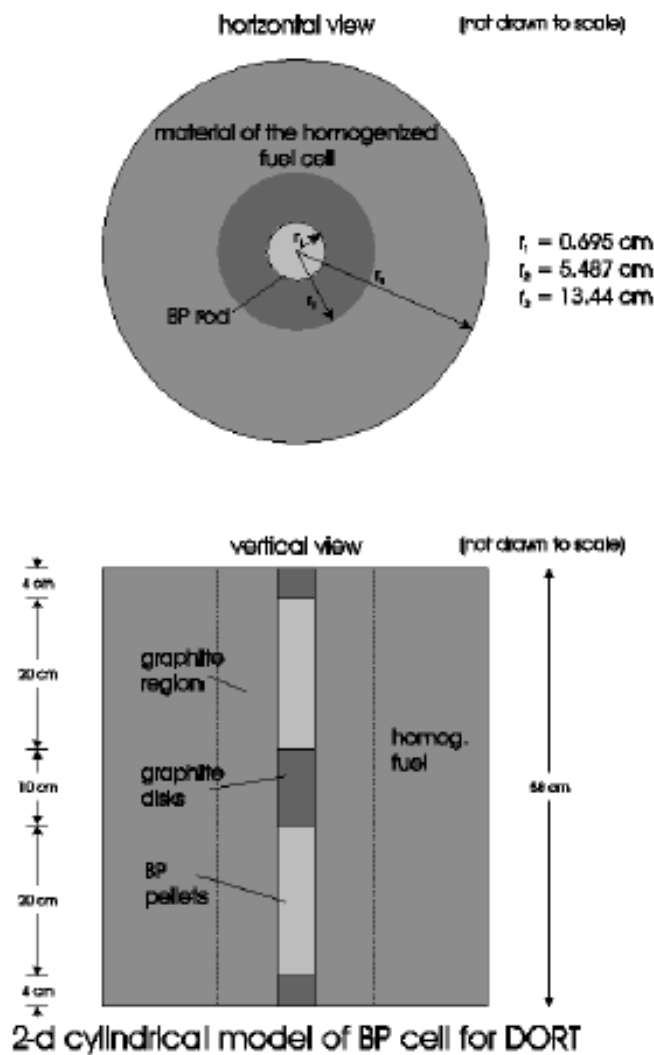


FIG. 2.60. Post-Test 2-d Cyl. Model of the BP Cell for DORT

### *New graphite cell model*

The 26 group constants used in the third region of the HTTR fuel block are the zone-weighted cross sections supplied by the 1-d "empty" BP unit cell calculation. This "empty BP" cell has the same area as the cross section of the fuel block. It consists of three zones: the inner hole filled with helium instead of BP, surrounded by a graphite zone, corresponding to the area of 2/24 block, and a third zone, in which the cross sections of the homogenized fuel cell and of the homogenized replaceable reflector cell RB-2 are taken into account, because all empty BP holes in the 18 fuel columns assembly are situated at the outer surface of the annular core. Thus, the neutron spectrum in this "empty BP" unit cell is determined by the outer reflector and the corresponding fuel block.

In the "empty" BP cell calculation, the group constants are homogenized over the two inner zones and condensed to 26 energy groups. The  $k_{\infty}$ -values obtained by the three different cell calculations are given on Table 2-46.

Table 2-46. Post-Test Results of the three Different Cell Calculations

Case	Enr. (wt.%)	$k_{\infty}$ -Values of the		
		New Fuel Cell 1-d TOTMOS	New BP Cell 2-d DORT	Empty BP Cell 1-d TOTMOS
343320	3.4	1.2808	1.1598	1.4955
393320	3.9	1.3075	1.2068	1.5419
673320	6.7	1.3814	1.3464	1.6900
793320	7.9	1.3943	1.3782	1.6983
433120	4.3	1.3230	1.2415	1.5693
483120	4.8	1.3374	1.2728	1.5987
943120	9.4	1.4140	1.4216	1.7288
993120	9.9	1.4157	1.4297	1.7356
433325	4.3	1.3230	1.2181	1.5707
523325	5.2	1.3490	1.2677	1.6164
633325	6.3	1.3742	1.3143	1.6581
593125	5.9	1.3665	1.3083	1.6477
633125	6.3	1.3742	1.3217	1.6594
723125	7.2	1.3893	1.3529	1.6808
793125	7.9	1.3943	1.3712	1.7011

### *Modified streaming correction*

In the course of the pre-test calculations, it turned out: the streaming effect calculated by the CITATION code using anisotropic diffusion coefficients determined by the MARCOPOLO code [2-41] was about 33% smaller than the effect calculated by the Monte

Carlo code MVP of JAERI [2-22], in the case of 18 fuel columns in the core. In the case of the fully loaded core, the streaming effect was smaller and the difference between the diffusion and Monte Carlo calculation was reduced to about 15%, as can be seen on Table 2-47.

Table 2-47. Streaming Effects in Pre-Test and Post-Test Calculations

No. of Fuel Col.	Streaming Effect [% $\Delta k/k$ ]				
	Monte Carlo Code MVP <sup>a</sup> for:	Pre-Test Diffusion Code with Stream. Corr. for:	Post-Test Diffusion Code with <b>Modified</b> Stream. Corr. for:	Pre-Test Diffusion Code with Stream. Corr. for:	Post-Test Diffusion Code with <b>Modified</b> Stream. Corr. for:
	CR Guide Columns Irrad. Columns			all Coolant Channels and Holes of Core and Reflector	
18	2.30	1.53	2.32	1.69	2.46
24	—	1.27	1.92	1.38	2.02
30	1.30	1.10	1.68	1.18	1.75

<sup>a</sup> Results of JAERI-HTTR, presented at the 1st RCM of the CRP-5.

One possibility of getting more accurate diffusion constants is to compare the corresponding streaming effects obtained by diffusion and by Monte Carlo calculations and to adapt the anisotropic diffusion coefficients to the results of the Monte Carlo calculations.

Therefore, the neutron streaming coefficients calculated with the MARCOPOLO code had to be modified. When increasing the neutron streaming correction factors of the CR-guide and irradiation columns by about 20% (in r- and in z-direction) the streaming effect calculated by the diffusion code was the same as the reactivity effect resulting from the Monte Carlo calculation, as can be seen on Table 2-47.

For the control rod guide blocks CB-1 and CB-2, and the irradiation block IB-1 the ratios of the anisotropic diffusion coefficients to the homogeneous diffusion coefficients are given in Table 2-48 together with the modified streaming correction factors. They are listed in the 4 group structure. But in the core calculation 26 group constants were used, which were interpolated from the 4 group values.

Thus, in the recalculation of the first criticality and of the excess reactivities, these modified anisotropic diffusion coefficients for all CR-guide and irradiation columns were used. The anisotropic diffusion constants of the upper and lower replacable reflector, of the fuel and the dummy fuel blocks remained unchanged because there was no comparable Monte Carlo calculation.

Table 2-48. Streaming Correction Factors Obtained by the MARCOPOLO Code together with the Modified Factors Deduced from MVP Monte Carlo Calculations

Group	CR Guide Block CB-1		CR Guide Block CB-3		Irrad. Block IB-1	
	$D_r/D_{hom}$	$D_z/D_{hom}$	$D_r/D_{hom}$	$D_z/D_{hom}$	$D_r/D_{hom}$	$D_z/D_{hom}$
<b>Streaming Correction Factors (Pre-Test)</b>						
1	1.140	1.574	1.156	1.632	1.139	1.567
2	1.176	1.933	1.196	2.036	1.175	1.923
3	1.181	1.950	1.202	2.054	1.180	1.939
4	1.188	2.024	1.209	2.137	1.186	2.013
<b>Modified Streaming Correction Factors (Post-Test)</b>						
1	1.368	1.889	1.387	1.958	1.367	1.880
2	1.411	2.320	1.435	2.443	1.410	2.308
3	1.417	2.340	1.442	2.465	1.416	2.327
4	1.424	2.429	1.451	2.564	1.423	2.416

Table 2-49. Streaming Effects in the Pre-Test and Post-Test Detailed Core Calculations

No. of Fuel Columns	Pre-Test <b>Benoist</b> Stream. Correction $\Delta k_{eff}$	Post-Test <b>Modified</b> Stream. Correction $\Delta k_{eff}$
9	-0.0203	-0.0301
12	-0.0194	-0.0284
15	-0.0187	-0.0275
16	-0.0185	-0.0264
17	-0.0181	-0.0257
18	-0.0176	-0.0253
19	-0.0175	-0.0249
24	-0.0166	-0.0245
27	-0.0159	-0.0237
30	-0.0153	-0.0229

The use of the modified anisotropic diffusion constants decreased the  $k_{\text{eff}}$ -values in the whole core diffusion calculations by about  $\Delta k \approx 0.01$ - $0.007$  compared to the pre-test results, as can be seen on the Table 2-49.

### Whole reactor calculations

In the post-test calculations the whole core was modelled with the CITATION diffusion code using the **26** group cross sections from the cell calculations. A 3-dimensional triangular-z model was chosen, as in the pre-test-prediction, but each block was divided vertically into 4 meshes and horizontally into **24** meshes, in order to arrange the BP rods and the empty BP-hole at their exact positions in the fuel block.

As in the pre-test calculations, four series of diffusion calculations were performed in order to show the influence of the modified streaming correction and of the axial BP heterogeneity on the  $k_{\text{eff}}$ -values. Both effects are augmented compared to the pre-test prediction, as can be noticed on Tables 2-49 and 2-50. The modified streaming correction reduces the multiplication constants by about  $\Delta k = 0.03$  at 9 fuel columns loading to  $\Delta k = 0.023$  at 30 fuel columns in the core. On the other hand, the more accurate treatment of the axial heterogeneity of the BP increases the multiplication factor by about  $\Delta k = 0.024$  at 9 fuel columns in the core and by about  $\Delta k = 0.034$  at fully loaded core. This stronger effect of the BP when using the more accurate treatment of the axial heterogeneity is mainly due to the fact, that the  $\Sigma_{\text{abs}}$ -values obtained by the 2-d cell calculation are significantly smaller in the thermal energy range than those macroscopic absorption cross sections obtained by the 1-d cell calculation with  $B^{10}$  reduction, as can be seen from Fig 2.61.

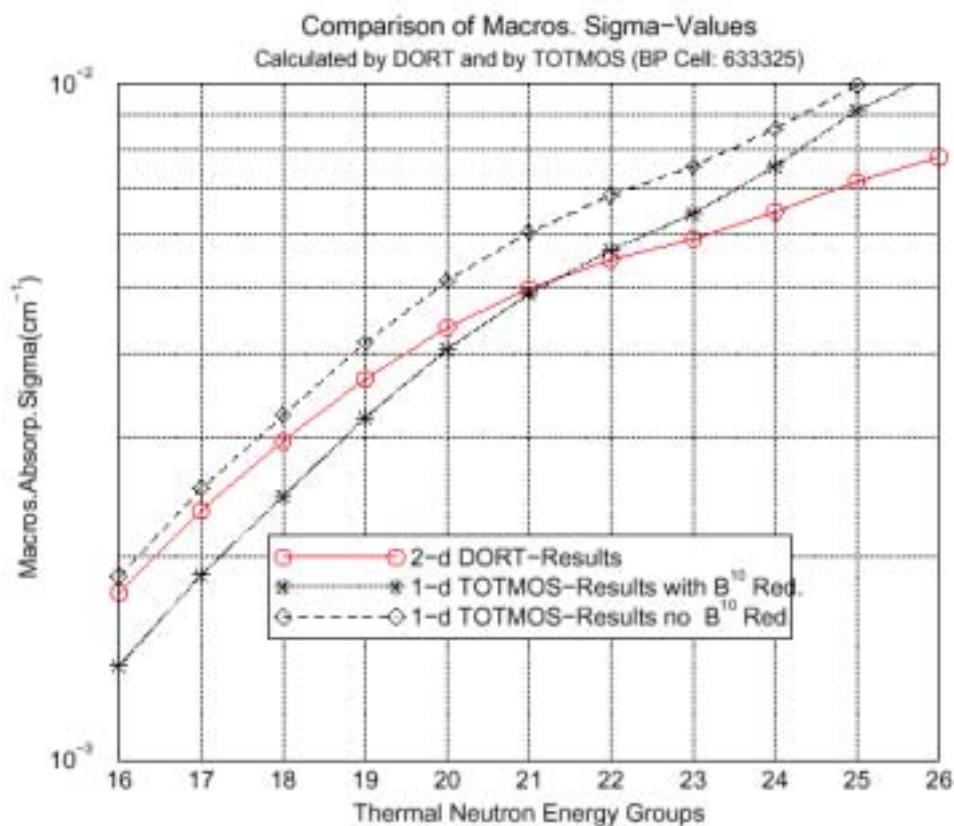


FIG. 2.61. Comparison of Macroscopic Absorption Cross Sections Obtained by Different Methods.

Table 2-50. Axial Self-Shielding Effect of the BP in the Detailed Core Calculations Obtained by Different Methods

No. of Fuel Columns	Pre-Test $B^{10}$ -Reduction $\Delta k_{\text{eff}}$	Post-Test 2-d BP Cell Calculation $\Delta k_{\text{eff}}$
9	+0.0189	+0.0245
12	+0.0196	+0.0255
15	+0.0200	+0.0257
16	+0.0201	+0.0265
17	+0.0203	+0.0269
18	+0.0207	+0.0273
19	+0.0213	+0.0281
24	+0.0242	+0.0314
27	+0.0252	+0.0333
30	+0.0263	+0.0336

Table 2-51. Results of the Post-Test Diffusion Calculations together with the Experimental Results

No. of Fuel Columns	$k_{\text{eff}}$ Calc.	$k_{\text{eff}}$ Exper.	$\Delta k$ Calc.-Exp.	$\rho$ [% $\Delta k/k$ ] Calc.	$\rho$ [% $\Delta k/k$ ] Exper.
9	0.9408	0.9282	0.0126	-6.30	-7.7 <sup>a</sup>
12	0.9642	0.9481	0.0161	-3.70	-5.5 <sup>a</sup>
15	0.9811	0.9652	0.0159	-1.90	-3.6 <sup>a</sup>
16	0.9866	0.9701	0.0165	-1.40	-3.1 <sup>a</sup>
17	0.9959	0.9785	0.0174	-0.41	-2.2 <sup>a</sup>
18	1.0080	0.9913	0.0167	0.79	-1.0 <sup>a</sup>
19	1.0263	1.0152	<b>0.0111</b>	2.60	1.5
21	1.0556	1.0417	0.0139	5.30	4.0±1.1 <sup>b</sup>
24	1.0944	1.0834	0.0110	8.60	7.7±2.1 <sup>b</sup>
27	1.1261	1.1198	0.0063	11.20	10.7±3.0 <sup>b</sup>
30	1.1336	1.1346	-0.0010	11.80	12.0±3.3 <sup>b</sup>

(CR Insertion and AI in the Neutron Detector Holders considered:  $\Delta k=0.004+0.002$ )

<sup>a</sup> from 1/M Measurement [2-43]

<sup>b</sup> from IK Method

Table 2-52. Post-Test Effective Multiplication Factor and Excess Reactivity at the First Criticality

No. of Fuel Columns	$k_{eff}$	$\rho$ [% $\Delta k/k$ ]
17	0.9959	-0.41
18	1.0080	+0.79

Table 2-53. Post-Test Effective Multiplication Factor and Excess Reactivity at 18, 24, and 30 Fuel Columns Loading

No. of Fuel Columns	$k_{eff}$	$\rho$ [% $\Delta k/k$ ]
18	1.0080	0.79
24	1.0944	8.60
30	1.1336	11.80

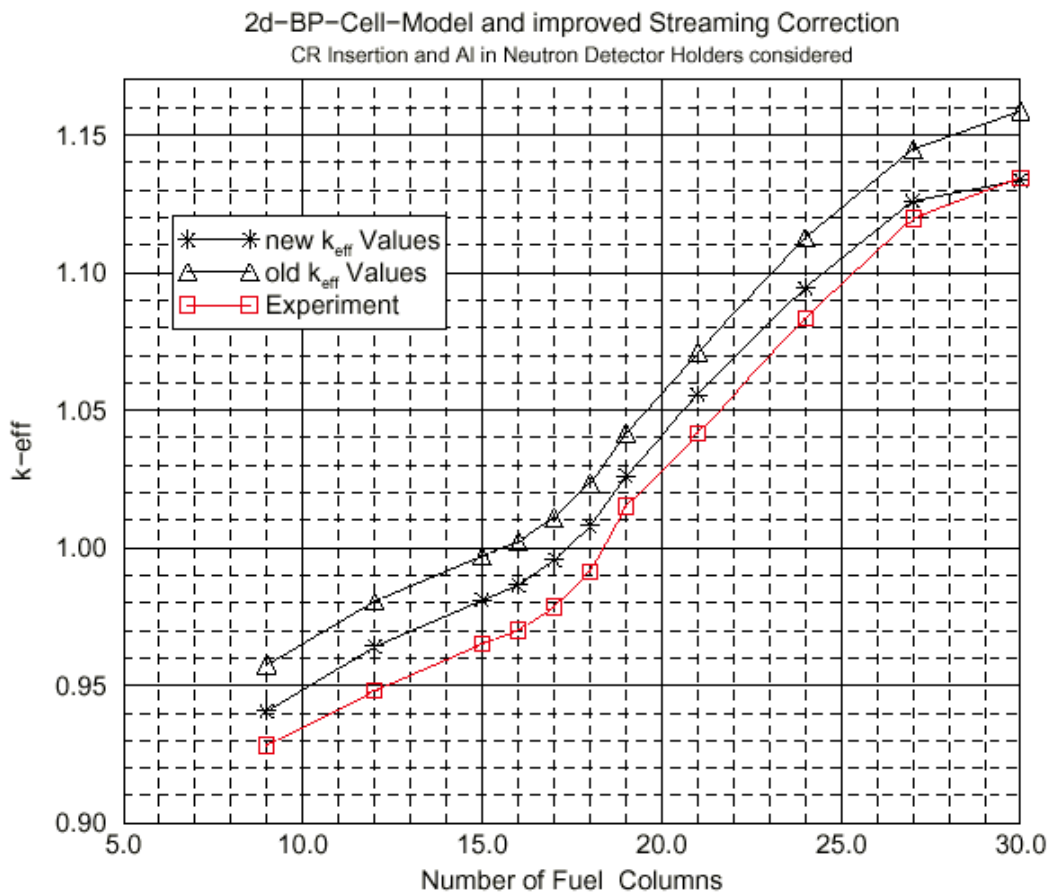


FIG. 2.62. New and Old  $k_{eff}$ -Values of the Diffusion Calculations in Comparison with the Experimental Results.



When taking into account the modified neutron streaming in the big holes of the core, the neutron axial self-shielding in the BP rods by the more accurate method, and when not only the reactivity of the CR insertion of  $\Delta k=0.0040$ , but also the reactivity of the aluminium in the temporary neutron detector holders of  $\Delta k=0.0020$  is subtracted, the first criticality will now be achieved at 18 fuel columns loading. The excess reactivity amounts to  $\Delta k/k=0.79\%$ . The excess reactivity of the thin, thick, and the fully loaded core is 0.79%, 8.63%, and 11.80%, respectively. The results of the post-test calculations are given on Tables 2-51 to 2-53. In Figure 2.62 the new and the old  $k_{\text{eff}}$ -values of the diffusion calculations are presented together with the experimental results.

### ***Discussion of the post-test results***

When summing up all post-test studies, the intermediate calculations using simplified whole core models and the final calculations for the detailed core, the analysis yields the following effects at 18/19 fuel columns in the core compared to the pre-test results:

- when considering the detailed structure of the HTTR fuel block in the whole core calculation, the multiplication factor decreases by about  $\Delta k \approx 0.043$ ,
- the description of the detailed energy-dependence of the neutron flux adequately by a fine energy group structure increases the  $k_{\text{eff}}$ -value by about  $\Delta k \approx 0.035$ ,
- when taking into account the residual air in the graphite pores, the higher boron impurity in some dummy fuel blocks, and the reactivity effect of the aluminium in the neutron detector holders, the multiplication factor is reduced by about  $\Delta k \approx 0.009$ ,
- when considering the axial heterogeneity of the BP by 2-d cell calculations,  $k_{\text{eff}}$  of the whole core calculation increases by about  $\Delta k \approx 0.0068$ ,
- and when treating the neutron streaming effect by modified diffusion constants on the basis of the Japanese Monte Carlo results,  $k_{\text{eff}}$  is reduced by about  $\Delta k \approx 0.0075$ .

The number of fuel columns to achieve first criticality increases by about 2 fuel columns in comparison to the pre-test results presented at the 1.RCM [2-22]. Now, the discrepancy between measurement and FZJ diffusion calculation amounts to  $\Delta k=0.0111$  at the first criticality (i.e. at 19 fuel columns), as can be seen on Table 2-51.

#### *2.2.3.5. Conclusion*

Altogether it turns out that the following procedures seem to be necessary for a better approach to the experimental results:

- detailed heterogeneity of the BP- and fuel-region in the whole core calculation,
- use of fine group constants or of broad group constants including detailed leakage information in the diffusion calculations of the whole core in order to describe the core/reflector coupling accurately,
- treatment of the axial self-shielding in the BP rods by 2-d cell calculations,
- consideration of an enhanced neutron streaming, brought about here by an adaption of the diffusion constants to results of Monte Carlo calculations.

When applying these improvements and regarding the actual boron impurity in some dummy fuel blocks together with air in the graphite pores the first criticality was recalculated for 18 fuel columns in the case of fuel loading from the core periphery. This corresponds to a

critical mass of 33.1 kg U<sup>235</sup>. Moreover, it turned out that a 26 energy group structure in the whole core diffusion calculation seems already to be sufficient to get reliable results when no detailed leakage information is used in generating the group constants for the whole core diffusion calculations.

#### *2.2.3.6. Acknowledgement*

The authors would like to thank Dr. W. Scherer for very useful discussions and comments, and Ms. A. Kuhr for supporting all the plots.

#### 2.2.4. Indonesia [2-45]

Initial fuel loading and first criticality of the HTTR was accomplished in an annular core configuration. This core type was chosen because of high inherent safety characteristics for loss of coolant accidents [2-46]. The vessel cooling system (VCS) surrounding the reactor pressure vessel is designed to maintain the fuel temperature at less than the limit of 1600°C in a loss of coolant accident. The decay heat will be transferred radially through the regions of fuel, side removable and permanent reflector blocks and reactor pressure vessel to the cooling panel of the VCS through heat conduction, radiation and convection without any necessity for active cooling. Introduction of the annular core type is expected to enhance the heat removal mechanism, because the heat transfer pass is shortened by thinning of the fuel region.

In the fuel loading for the start up core physics experiment, three different types of core were considered; a thin and thick annular core with 18 and 24 fuel columns, respectively, and a fully loaded core that consists of 30 fuel columns. The core size of the HTTR is about half the size of a commercial HTGR. The high excess reactivity in HTTR is similar to that of HTGR, which is needed to compensate for the effects of temperature, xenon, burn up, etc. during power operations.

Data for the core physics calculation is applied by JAERI [2-47]. The data include specifications for core and its internal components, such as fuel, replaceable reflectors, permanent reflectors, control rods and dummy fuel elements. Also provided are HTTR details (see Section 2.1.1) including specifications of the fuel rods, fuel compacts and coated fuel particles. Generation of cell cross-sections in this study was performed using CELL Module of the SRAC-EWS Code system [2-48], while the whole core calculation was performed using CITATION Module of the SRAC-EWS.

The purpose of this study was to examine when the first criticality can be achieved by using annular thin core loading mechanism. In this loading type, fuel columns are loaded in the outermost fuel region (18 columns, i.e. annular thin core). Upon completion of the initial 18 columns, fuel loading continued with another 6 fuel columns in the inner fuel region (24 columns, i.e. thick annular core). Finally, loading continued with another 6 fuel columns in the innermost fuel region, so that the full core of 30 fuel columns was achieved.

##### 2.2.4.1. Calculation Model and Procedures

Calculation for the start up benchmark test was developed and all calculations were performed for a core temperature of 300K and helium pressure of 1 atm. Figure 2.63 depicts the flow diagram of the benchmark calculation.

The SRAC-EWS Code system [2-48] was applied throughout the calculations, and the nuclear data in 107 groups were obtained mainly from JENDL-2 and ENDF/B-V. The ULIBMAKE module of SRAC-EWS was applied to generate the user library files from the public library, and to collapse these files into 54 energy groups. The CELL module was then applied to further condense these into 6 groups (i.e. 3 groups in the fast region and 3 groups in the thermal region), and to perform the cell calculation in collision probability method. In order to save computing time, the cell calculations were performed by using the equivalent cylindrical geometry (i.e. Wigner-Seitz's approximation) with white boundary conditions, rather than the more exact hexagonal geometry, because the effect on  $k_{\infty}$  was minimal, as pointed out by Jeong et al.[2-49]. The CITATION module was applied for the whole core diffusion calculations.

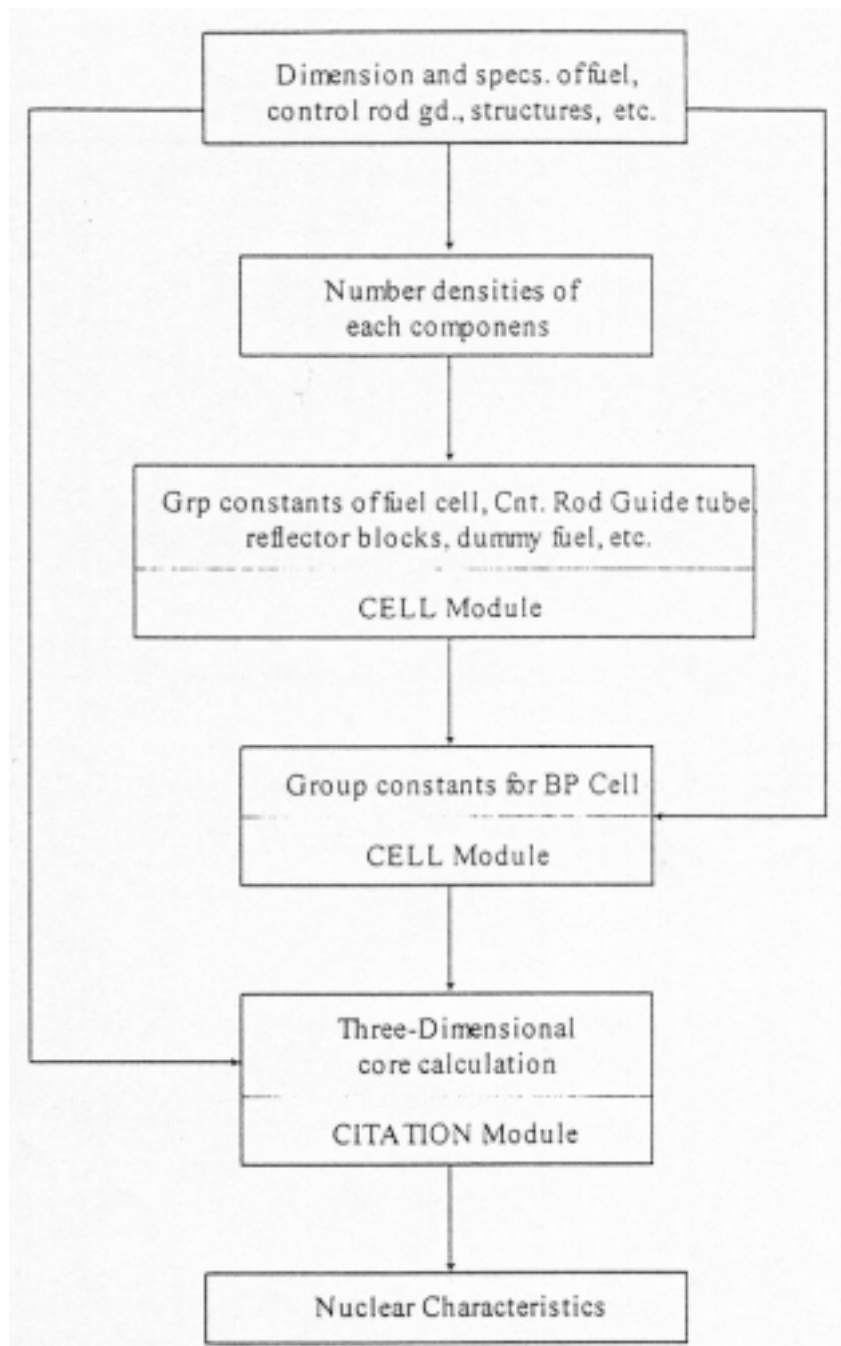


FIG. 2.63. Flow diagram of cell and core calculation for the HTTR benchmark test.

Cell Module was used to generate group constants of fuel and graphite blocks for succeeding core calculations. In the fuel region, the double-heterogeneity effects originating from the structure of fuel rods, in which the coated fuel particles are dispersed within fuel compacts, is considered by the use of collision probability method. The outer radius  $R$  for 33 pin block was 3.249 cm, and for 31 pin fuel block was 3.352 cm.

For the control rod guide tube with three large holes, the collision probability method was also utilized. Radial and axial directional diffusion coefficients for the unit cell were calculated by Benoist's formula. Similar to a coolant channel in fuel block, the content of the holes was helium gas at 1 atm at 300K. An isotropic (white) reflection boundary condition

was applied in the outer boundary of the unit cell. The unit cell in the control rod guide consists of a void region and a graphite region. The inner and outer radius of this region was 6.15 and 10.91 cm, respectively.

Average group constants of the fuel rods were calculated by using a one-dimensional cylindrical fuel cell model as depicted in Figure 2.64. The neutron flux was calculated by the collision probability method, which was used to average the group constants in the fuel cell geometry. The average group constants of the fuel block with BPs were also calculated in one-dimensional cylindrical BP cell model shown in Figure 2.64. The outer radius of the BP region was 0.7 cm, while the homogenized fuel region outer radius was 13.44 cm. Variation in axial composition of material was taken into consideration by averaging the number density in the axial region, taking into account their volume ratio. Table 2-54 shows the infinite multiplication factor ( $k_{\infty}$ ) for each fuel cell and its corresponding BP cell.

The whole core calculation was performed using CITATION Module of SRAC-EWS in  $\theta$ -R-Z geometry. The angular distribution was made such that there were nine angular-divisions in one-sixth of the horizontal cross-section of the core.

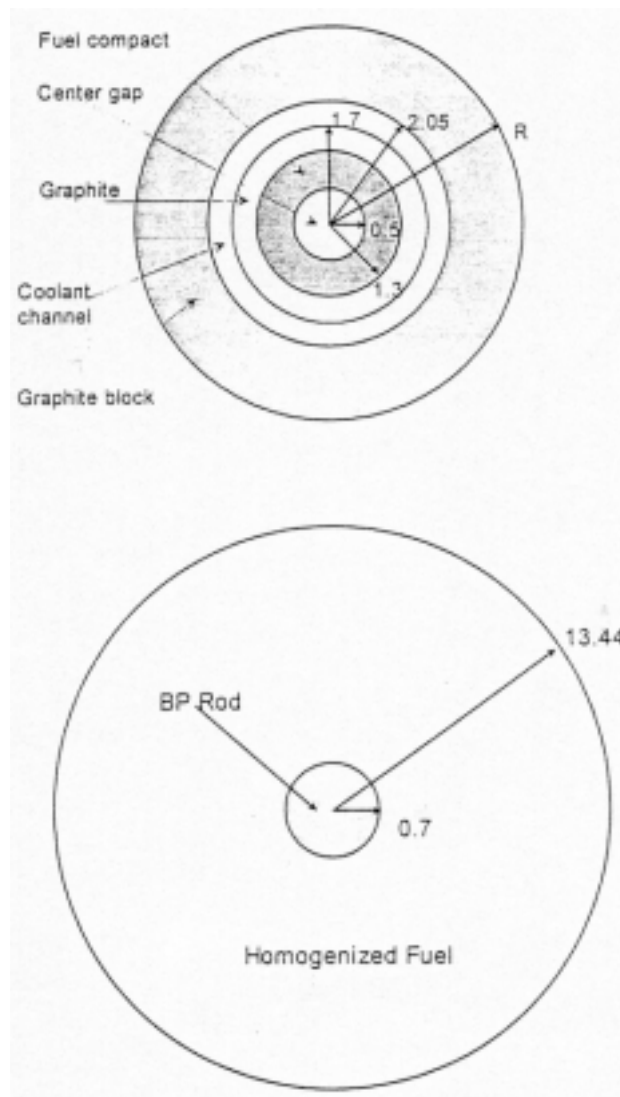


FIG. 2.64. Fuel and BP Cell model for the SRAC code calculation (all units in cm.).

Table 2-54. Calculated infinite multiplication factors ( $k_{\infty}$ ) for Fuel Cell and BP Cells

Fuel Layer	Fuel block ID	Infinite mult. factor ( $k_{\infty}$ )	
		Fuel Cell	BP Cell
1 <sup>st</sup> layer	f673320	1.5446	1.3519
	f793320	1.5594	1.3872
	f943120	1.5996	1.4309
	f993120	1.6013	1.4397
2 <sup>nd</sup> layer	f523325	1.5082	1.2566
	f633325	1.5375	1.3092
	f723125	1.5721	1.3469
	f793125	1.5782	1.3674
3 <sup>rd</sup> layer	f433325	1.4786	1.2012
	f523325	1.5082	1.2566
	f593125	1.5470	1.2966
	f633125	1.5553	1.3116
4 <sup>th</sup> and 5 <sup>th</sup> layer	f343320	1.4259	1.1491
	f393320	1.4593	1.1979
	f433120	1.4932	1.2340
	f483120	1.5129	1.2653

#### 2.2.4.2. Results of HTTR-FC Calculations

Multiplication factors ( $k_{\text{eff}}$ ) and excess reactivity  $\rho$  for first criticality where all control rods are assumed to be fully withdrawn, are provided in Table 2-55. It indicates that the first criticality is to be achieved when 18 columns of fuel are inserted. The excess reactivity at first criticality was determined be 0.577 %  $\Delta k/k$ . At 17 columns, the  $k_{\text{eff}}$  was determined to be exactly one with zero excess reactivity, therefore the first criticality was estimated at 18<sup>th</sup> column.

Table 2-55. Effective multiplication factor and excess reactivity at first criticality.

Number of fuel column	Effective mult. factor [ $k_{eff}$ ]	$\rho$ [% $\Delta k/k$ ]
16	0.9979	-0.002
17	1.0000	0.000
18	1.0058	0.577

#### 2.2.4.3. Results of HTTR-EX Calculation

Multiplication factors ( $k_{eff}$ ) and excess reactivity  $\rho$  for 30 column, 24 column and 18 columns of fuel loaded into the core are shown in Table 2-56. The control rods were assumed to be in the fully withdrawn condition for all calculations.

Table 2-56. Effective multiplication factor and excess reactivity at 18, 24 and 30 column loaded in the core.

Number of fuel column	Effective mult. factor [ $k_{eff}$ ]	$\rho$ [% $\Delta k/k$ ]
18	1.0058	0.577
24	1.0692	6.472
30	1.0931	8.517

## 2.2.5. Netherlands

Both NRG and IRI are taking part in the benchmark of start-up core physics of the HTTR. To compare the performance of the SCALE based IRI code package with that of the WIMS/PANTHER code package of NRG, a calculational intercomparison has been performed. This report first describes the NRG and IRI efforts in the first Benchmark (Phase 1) using the Monte Carlo code KENO Va (3-D) and the diffusion theory codes BOLD VENTURE (2-D) and PANTER (3-D). In the second phase of the Benchmark, only KENO calculations are performed, for the scram reactivities of the core and reflector control rods and the isothermal temperature coefficients are given for phase 2. NRG/IRI also participated in the start-up measurement of reactivity and reactor noise measurements.

### 2.2.5.1. Computational Methods and Associated Data

The computational tools used at IRI for the cross section generation have been described previously [2-51]. In short it contains, as a branch, the SCALE-4 code system with master libraries produced by NJOY from the JEF-2.2 basic nuclear data files. The used reactor codes for this study are: KENO-Va and BOLD VENTURE. At ECN, the WIMS-7B code system has been used for this study which has libraries also based on JEF-2.2. The reactor code used at ECN is PANTHER-5.0.

#### *Cross sections for use in KENO [2-50]*

In KENO, only the coated fuel particles (CFP's) in the fuel compacts are homogenised with the graphite matrix of the fuel compacts; all other reactor components can be modelled explicitly. As the fuel also contains the only two resonant nuclides ( $^{235}\text{U}$  and  $^{238}\text{U}$ ) present in the core model, the only problem is the generation of cross sections for the homogenised fuel compacts. The general CFP and compact data are given in Table 2-57.

Table 2-57. General CFP and Compact Data

	radius ( $\mu\text{m}$ )	density ( $\text{g}/\text{cm}^3$ )	material
fuel kernel	298.5	10.79	$\text{UO}_2$
1st coating	358.5	1.14	PyC (low dens.)*
2nd coating	389.5	1.89	PyC
3rd coating	418.5	3.20	SiC
4th coating	464.5	1.87	PyC

\*PyC: Pyrolytic graphite

Compact dimensions: i/o diameter = 1.00/2.60 cm, height = 3.91 cm.

Since the problem is similar to the generation of cross sections for the fuel pebbles of a pebble-bed type HTR, the following scheme was adopted from the analysis work for HTR-PROTEUS [2-51]:

1. First only the coated fuel particles inside a fuel rod are considered. An infinite close-packed hexagonal CFP lattice is calculated by BONAMI, NITAWL and XSDRNPM. XSDRNPM is run in spherical geometry for a white boundary elementary cell of the CFP lattice. This elementary cell contains two regions: a sphere of 0.0597 cm diameter which contains the fuel kernel of  $\text{UO}_2$  surrounded by the homogenised mixture of the



coating layers and graphite matrix in the fuel compact. The matrix graphite contains some natural boron to represent impurities in the graphite. A cell-averaged weighted library, WGH(1), is produced which takes the self-shielding of the fuel in the Caps into account.

2. An infinite fuel-rod lattice is treated by BONAMI and NITAWL to obtain working library WRK(1). The unit cell with cylinder geometry has three regions. The innermost region is a channel filled with helium (0.5 cm radius). This region is surrounded by a cylinder of 1.3 cm radius with the fuel. The outermost region surrounding the fuel contains fuel block graphite ( $r = 3.29$  cm). A triangular lattice is assumed with a pitch of 6.2668 cm, consistent with 1/33rd block for the 33-rods fuel block. This step is required because it provides the unweighted data for the materials outside the fuel region. The overall Dancoff factor for the core has been deduced from the Dancoff factors for a lattice of CFP's in a fuel compact and for a lattice of fuel rods in a fuel block [2-51].
3. The library WRK(1) cannot be used for the fuel-rod lattice cell calculation as it would not take into account the self-shielding in the CFP's. Therefore the WGH(1) and WRK(1) libraries are merged. All fuel-region materials are taken from the weighted library WGH(1), the other materials from WRK(1). The resulting library is called WRK(2).
4. XSDRNPM is run with working library WRK(2) for the unit cell of the infinite fuel-rod lattice. This unit cell of cylindrical geometry has five radial zones: 1. Channel with helium ( $r \leq 0.5$  cm). 2. Fuel zone ( $r \leq 1.3$  cm). 3. Graphite sleeve of fuel rod ( $r \leq 1.7$  cm). 4. Fuel hole in fuel block filled with helium ( $r \leq 2.05$  cm). 5. Fuel block graphite with reduced density to take the fuel handling hole into account. The radius is of this zone is 3.2903 cm (1/33rd fuel block). If no axial dimensions are used, this run yields the  $k_{\infty}$  of the fuel rod lattice. XSDRNPM is run with a buckling search option to get a critical system (by the addition of a leakage term in the form of  $DB^2\phi$ ). The weighted library WGH(2) with zone-averaged cross sections is produced.
5. In order to obtain a working library for KENO, WGH(2) and WRK(1) are merged. The cross sections for the nuclides inside the fuel compact are taken from WGH(2), and the cross sections for all nuclides in the other components (He, C,  $^{10}\text{B}$ , and  $^{11}\text{B}$ ) are taken from WRK(1). The resulting library is denoted as WRK(3).

No group collapsing is done in any of these steps. All libraries contain cross section data for 172 energy groups! A simpler scheme would have been possible if no comparison had to be made for two-groups cross sections.

#### Two-group cross sections

In total, five two-group cross-section libraries have been generated for comparison purposes:

1. GRLAT2GR: XSDRNPM output of step 1 but with condensation
2. RODLAT\_K: output of XSDRNPM k-calculation using WRK(1) as input (step 2)
3. HTTR\_K: output of step 4 XSDRNPM, k-calculation and using WRK(2) as input
4. HTTR\_B2: output of step 4 XSDRNPM, buckling search using WRK(2) as input

5. RODLAT\_B2: output of XSDRNPM, buckling search and using WRK(1) as input (step 2)

The order of the numbers in Tables 2-58 and 2-59 correspond to this order. These five sets enable the assessment of the effects of step 1, the separate treatment for the coated particles, and of the spectrum used for weighing (buckling search versus k-calculation). Table 2-58 lists the microscopic total (MT = 1), absorption (MT = 27), and transport (MT = 1000) cross section for the nuclides in the fuel compact. For the uranium isotopes <sup>235</sup>U and <sup>238</sup>U also the total number of fission neutrons (MT = 452), and the fission (MT = 18) and capture (MT = 101) cross-section are specified in Table 2-59.

Table 2-58. Two-group cross sections for nuclides in the fuel compact (5.2 w% enrichment)

nuclide	$\sigma_{\text{tot}}$ (b)		$\sigma_{\text{abs}}$ (b)		$\sigma_{\text{tr}}$ (b)	
	group 1	group 2	group 1	group 2	group 1	group 2
<sup>10</sup> B	26.86	1132.4	24.49	1130.3	39.62	1446.0
	47.49	2147.2	45.16	2145.0	34.45	3074.8
	49.40	2157.8	47.07	2155.6	29.67	3080.0
	45.26	2095.3	42.92	2093.2	34.56	2214.5
	44.19	2096.0	41.85	2093.8	34.59	2246.6
<sup>11</sup> B	4.239	4.871	6.414 E-5	1.620 E-3	2.796	3.970
	4.267	4.941	8.961 E-5	3.073 E-3	2.686	4.757
	4.274	4.945	9.209 E-5	3.088 E-3	2.608	4.762
	4.223	4.941	8.706 E-5	2.999 E-3	3.441	4.666
	4.225	4.937	8.562 E-5	3.000 E-3	3.436	4.666
C	4.128	4.752	1.295 E-4	1.000 E-3	2.808	3.988
	4.157	4.805	1.567 E-4	1.891 E-3	2.693	4.857
	4.164	4.808	1.576 E-4	1.900 E-3	2.617	4.862
	4.114	4.805	1.673 E-4	1.845 E-3	3.373	4.727
	4.116	4.803	1.647 E-4	1.846 E-3	3.369	4.729
O	3.754	3.889	1.231 E-3	5.561 E-5	7.760	9.647
	3.752	3.943	1.293 E-3	1.062 E-4	2.701	3.840
	3.751	3.912	1.295 E-3	1.054 E-4	2.654	3.794
	3.736	3.910	1.437 E-3	1.023 E-4	3.189	3.757
	3.740	2.941	1.407 E-3	1.036 E-4	3.185	3.790
Si	2.669	2.113	2.644 E-3	5.057 E-2	2.050	1.807
	2.612	2.176	3.548 E-3	9.593 E-2	2.382	2.187
	2.605	2.178	3.621 E-3	9.640 E-2	2.399	2.190
	2.652	2.174	3.570 E-3	9.361 E-2	2.356	2.133
	2.651	2.173	3.511 E-3	9.363 E-2	2.353	2.134
<sup>235</sup> U	22.93	177.5	12.52	163.2	88.03	733.5
	28.80	366.3	18.20	351.5	22.22	543.2
	29.00	363.6	18.39	349.0	19.18	534.0
	27.53	352.2	17.08	337.6	21.45	374.3
	27.60	356.8	17.12	342.1	21.96	385.8
<sup>238</sup> U	14.77	10.08	1.775	0.853	64.94	26.97
	17.71	10.87	3.779	1.545	22.11	11.52
	15.86	10.79	2.879	1.534	14.31	11.38
	15.47	10.75	2.670	1.492	12.65	10.79
	17.27	10.84	3.541	1.510	15.24	10.92

Table 2-59. Total fission neutrons and fission and capture cross section of uranium isotopes

nuclide	ν		σ <sub>fis</sub> (b)		σ <sub>capt</sub> (b)	
	group 1	group 2	group 1	group 2	group 1	group 2
<sup>235</sup> U	2.440	2.439	8.457	137.8	4.066	25.42
	2.437	2.438	11.82	299.9	6.378	51.63
	2.437	2.438	11.92	297.7	6.462	51.25
	2.438	2.438	11.12	288.0	5.962	49.67
	2.438	2.438	11.16	291.8	5.967	50.31
<sup>238</sup> U	2.736	2.489	4.581 E-2	3.577 E-6	1.729	0.853
	2.740	2.489	4.628 E-2	6.640 E-6	3.732	1.545
	2.741	2.489	4.610 E-2	6.591 E-6	2.833	1.534
	2.742	2.489	5.093 E-2	6.407 E-6	2.619	1.492
	2.741	2.489	5.023 E-2	6.486 E-6	3.491	1.510

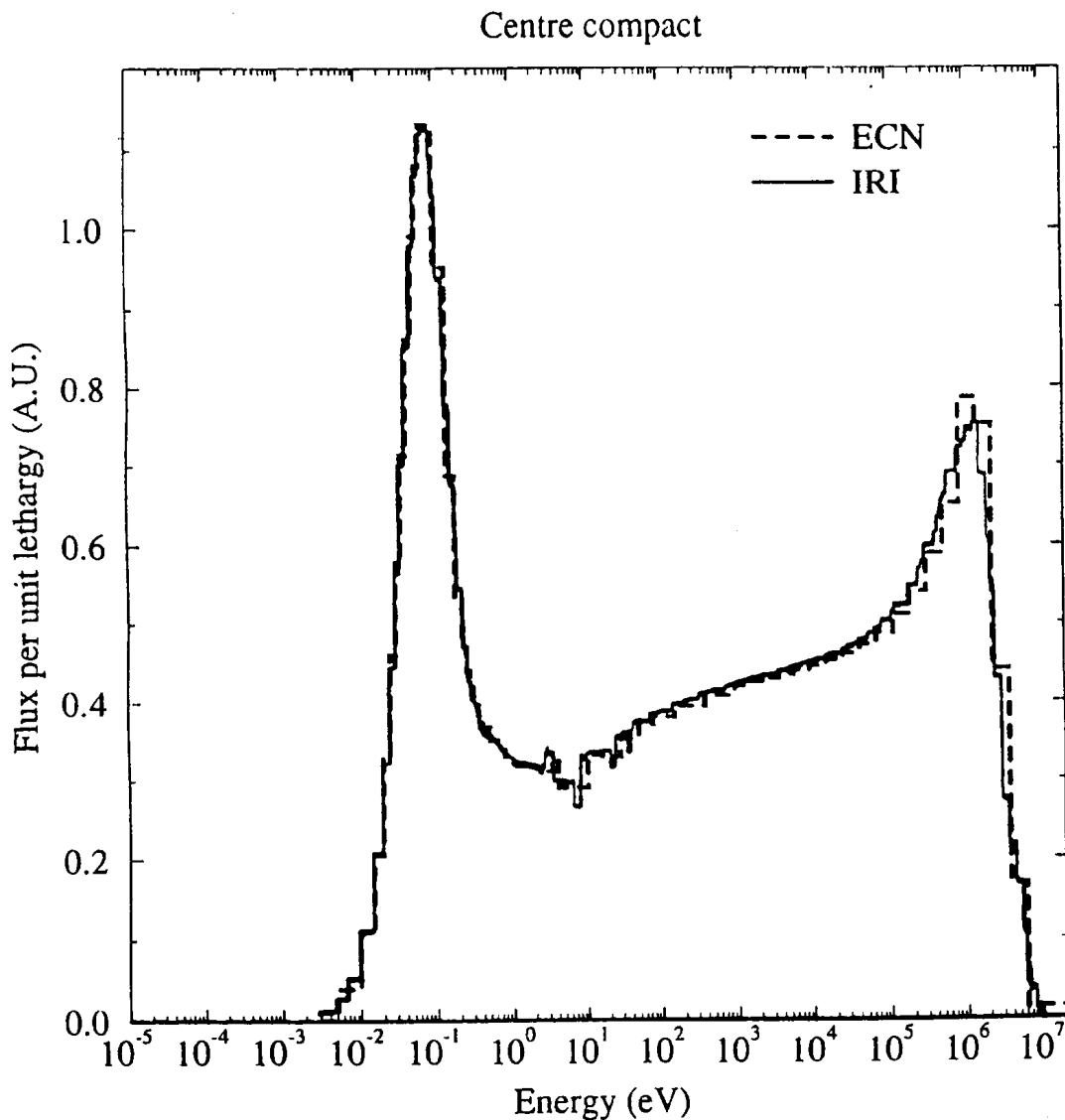


FIG. 2.65. Neutron spectrum in the centre of the compact

The spectrum in the grain lattice is much harder than in the fuel-rod lattice, which explains the lower values of the cross sections. The extra leakage term in the buckling search is seen to slightly reduce the cross sections, because of the greater leakage of low energy neutrons compared to high energy neutrons. Figure 2.65 shows the spectrum in the centre of the fuel-rod lattice, as calculated with step 4 of the cross-section generation procedure.

#### *Cross sections for BOLD-VENTURE [2-50]*

In BOLD VENTURE the core region is represented by five rings, containing the A, B, C, D, and E labelled columns, respectively. The material in each ring is completely homogenised. In order to maintain the reaction rates, the cross-section generation procedure for KENO was extended.

The first three steps are identical to the procedure for KENO. The fourth step is similar, but now a cell weighting is performed instead of a zone weighting. Subsequent steps are new.

4. XSDRNPM is run with working library WRK(2) for the unit cell of the infinite fuel-rod lattice. This unit cell of cylindrical geometry has five radial zones: 1. Channel with helium ( $r \leq 0.5$  cm). 2. Fuel zone ( $r \leq 1.3$  cm). 3. Graphite sleeve of fuel rod ( $r \leq 1.7$  cm). 4. Fuel hole in fuel block filled with helium ( $r \leq 2.05$  cm). 5. Fuel block graphite with reduced density to take the fuel handling hole into account. The burnable poison rods are not taken into account. The radius of this zone is 3.2903 cm (1/33rd fuel block). XSDRNPM is run with a buckling search option to get a critical system (by the addition of a leakage term in the form of  $DB^2\phi$ ). The weighted library WGH(2) with cell-averaged cross sections is produced.
5. Unweighted cross sections for the materials outside the fuel blocks (i.e. inside the control rod guide blocks and reflector) have to be added to WGH(2). These unweighted cross sections of C,  $^{10}\text{B}$ , and  $^{11}\text{B}$ , were taken from WRK(1). The resulting library is called WRK(3).
6. XSDRNPM is run with library WRK(3) for a 1D-model of the reactor. This model contains six radial zones. The first five represent the five rings of the core region, the outermost zone represents the permanent reflector. The radii of the zones were calculated to be 19.0064 cm, 50.2861 cm, 82.8468 cm, 115.6112 cm, 148.4444 cm, and 214.9814 cm. With these radii, the area of the rings is identical to the true area of the columns (the pitch in the core region is taken to be 18.1 cm, hence the space between the blocks is taken into account). The material within each zone is completely homogenised. The atomic densities in the homogenised zones can be found in the appendix. Note that the burnable poison rods are not taken into account. XSDRNPM is run with a buckling search option and with zone weighting, producing weighted library WGH(3). For the homogenised KENO model the 172 groups were not condensed, for BOLD VENTURE the groups were condensed to 13 broad groups, like for HTR-PROTEUS [2-52].

#### Two-group cross sections

Two-group cross section data is obtained by condensing the 172 fine groups to 2 broad groups in step 6 of the procedure in section 2.2.1. The results are summarised in Tables 2-60 thru 2-63.

Table 2-60. Two-group cross sections for the uranium isotopes in the fuel compact (5.2 w% enrichment) in the radial zones B, C, and D.

nuclide	$\sigma_{\text{tot}}$ (b)		$\sigma_{\text{abs}}$ (b)		$\sigma_{\text{tr}}$ (b)		
	group 1	group 2	group 1	group 2	group 1	group 2	
<sup>235</sup> U	B	27.79	368.3	17.18	354.2	21.30	372.4
	C	29.13	378.6	18.39	364.5	23.47	372.8
	D	27.34	380.8	16.78	366.7	21.04	385.3
<sup>238</sup> U	B	15.71	10.47	2.716	1.544	12.34	10.45
	C	16.08	10.50	2.929	1.581	12.11	10.45
	D	15.59	10.50	2.651	1.589	12.26	10.48

Table 2-61. Two-group cross sections for the uranium isotopes in the fuel compact (5.2 w% enrichment) in the radial zones B, C, and D.

nuclide	$\nu$		$\sigma_{\text{fis}}$ (b)		$\sigma_{\text{capt}}$ (b)		
	group 1	group 2	group 1	group 2	group 1	group 2	
<sup>235</sup> U	B	2.438	2.438	11.18	302.3	6.000	51.94
	C	2.438	2.438	11.92	311.1	6.469	53.44
	D	2.439	2.438	10.93	313.0	5.846	53.72
<sup>238</sup> U	B	2.742	2.489	5.486 E-2	6.640 E-6	2.661	1.544
	C	2.741	2.489	4.884 E-2	6.805 E-6	2.880	1.581
	D	2.742	2.489	5.667 E-2	6.841 E-6	2.595	1.589

Table 2-62. Two-group cross sections for the non-fissionable nuclides in the fuel compact (5.2 w% enrichment) in radial zones B, C, and D.

nuclide	$\sigma_{\text{tot}}$ (b)		$\sigma_{\text{abs}}$ (b)		$\sigma_{\text{fis}}$ (b)		
	group 1	group 2	group 1	group 2	group 1	group 2	
O	B	3.807	3.775	1.543 E-3	1.063 E-4	3.350	3.620
	C	3.814	3.773	1.371 E-3	1.090 E-4	3.394	3.620
	D	3.805	3.773	1.590 E-3	1.096 E-4	3.351	3.619
Si	B	2.730	2.106	3.694 E-3	9.728 E-2	2.500	2.059
	C	2.672	2.108	3.707 E-3	9.978 E-2	2.455	2.059
	D	2.749	2.108	3.683 E-3	1.003 E-1	2.501	2.061
C	B	4.177	4.639	1.779 E-4	1.917 E-3	3.515	4.566
	C	4.220	4.636	1.647 E-4	1.966 E-3	3.609	4.568
	D	4.163	4.635	1.813 E-4	1.977 E-3	3.510	4.564
<sup>10</sup> B	B	45.86	2177.5	43.47	2175.3	34.01	2199.1
	C	49.71	2233.4	47.33	2231.3	39.40	2201.1
	D	44.63	2245.6	42.23	2243.5	33.07	2266.9
<sup>11</sup> B	B	4.287	4.778	8.849 E-5	3.116 E-3	3.593	4.498
	C	4.332	4.778	9.315 E-5	3.197 E-3	3.690	4.497
	D	4.273	4.778	8.700 E-5	3.214 E-3	3.589	4.501

Table 2-63. Two-group cross sections for the nuclides in the graphite of the blocks in all radial zones

		$\sigma_{\text{tot}}$ (b)		$\sigma_{\text{abs}}$ (b)		$\sigma_{\text{fis}}$ (b)	
nuclide		group 1	group 2	group 1	group 2	group 1	group 2
C	A	4.290	4.810	1.178 E-4	2.185 E-3	3.442	4.772
	B	4.130	4.844	1.440 E-4	2.039 E-3	3.418	4.774
	C	4.183	4.846	1.351 E-4	2.095 E-3	3.528	4.776
	D	4.113	4.847	1.463 E-4	2.106 E-3	3.412	4.779
	E	4.347	4.815	1.161 E-4	2.580 E-3	3.500	4.786
	reflector	4.524	4.818	1.274 E-4	2.858 E-3	3.971	4.838
<sup>10</sup> B	A	57.31	2483.0	54.99	2480.7	38.79	2600.2
	B	46.77	2316.6	44.43	2314.4	34.60	2344.3
	C	50.72	2380.7	48.40	2378.5	40.15	2345.5
	D	45.50	2393.3	43.16	2391.1	33.63	2424.5
	E	65.21	2933.7	62.91	2931.5	38.77	2788.0
	reflector	97.35	3251.3	95.12	3249.1	79.34	3168.6
<sup>11</sup> B	A	4.404	4.976	1.017 E-4	3.554 E-3	3.515	4.701
	B	4.239	4.995	8.824 E-5	3.316 E-3	3.495	4.705
	C	4.294	5.003	9.325 E-5	3.407 E-3	3.609	4.705
	D	4.221	5.004	8.666 E-5	3.425 E-3	3.490	4.718
	E	4.462	5.020	1.116 E-4	4.199 E-3	3.583	4.721
	reflector	4.640	5.052	1.519 E-4	4.654 E-3	4.066	4.755

### ***Cross sections for PANTHER***

Cross sections for the reactor code PANTHER have been generated by means of the code suite WIMS-7B. Apart from service modules for group condensing and material homogenisation, two collision probability modules were used to calculate the flux weighted cross sections of the fuel cell (PROCOL) and for the fuel blocks or assemblies, control guide blocks and reflector blocks (PIJ).

In order to avoid much extra work, densities, impurities and sizes of graphites, CFPs and coatings, weighted means of these parameters where appropriate, have been derived to be used all over the reactor. This leads to the following standardised parameters for the CFPs (Table 2-64).

Table 2-64. Parameters for CFPs

	radius ( $\mu\text{m}$ )	density ( $\text{g}/\text{cm}^3$ )	material
fuel kernel	297.95	10.774	UO <sub>2</sub>
1st coating	358.80	1.127	PyC (low density)
2nd coating	389.45	1.896	PyC
3rd coating	418.35	3.225	SiC
4th coating	464.20	1.866	PyC

Compact dimensions: i/o diameter = 1.00/2.60 cm, height = 3.91 cm.

Table 2-65. Parameters for Graphites

	density (g/cm <sup>3</sup> )	impurity (ppm B <sub>nat</sub> )
matrix	1.690	0.82
sleeve	1.770	0.37
fuel / control block	1.770	0.40
repl. reflector	1.760	0.37
perm. reflector	1.732	1.91

**PROCOL [2-50]**

In the WIMS-suite a cell module PROCOL, based on collision probabilities, exists to calculate fluxes in systems with spherical grains packed in a matrix with an annular geometry.

A cell radius of 3.29 cm has been used, consistent with a lattice of a 1/33rd part of a fuel block or assembly, in which explicitly modelled: the inner gas channel (r = 0.50 cm), the compact (r = 1.30 cm), the gas gap (r = 1.3125 cm), sleeve (r = 1.70 cm) and the fuel hole drilling in the fuel block (r = 2.05 cm). Using this model, flux weighted cross sections are obtained for homogenised CFP's + matrix + gas gap, to form the compact material with cross sections in the 69 neutron energy groups structure of the library.

The spectrum in the centre of the inner gas channel in the compact with 5.2 w% enrichment is shown in Figure 2.65. Comparison with the spectrum as obtained with the KENO cross sections is very good. Differences are only due to the resolution of the spectrum with the number of energy groups used in the calculations (KENO: 172 vs. WIMS: 69).

Accordingly obtained cross sections were condensed to 16 neutron energy groups for subsequent use in the WIMS assembly module PIJ, which calculates collision probabilities in multi-pin assembly systems.

For comparison purposes microscopic cross sections for the nuclei present in the compacts were condensed to two group cross sections. In WIMS only microscopic absorption and fission cross sections are easily available, but for some elements transport and total cross sections could be deduced from macroscopic cross sections. Values for an enrichment of 5.2 w% are given in the Tables 2-66 and 2-67 and can be compared with those values given in Tables 2-58 and 2-59. Agreement is in general rather good which can be confirmed by the spectrum comparison of Figure 2.65 and the calculated neutron multiplication factors:  $k_{inf} = 1.499$  for the 'KENO'-cell and  $k_{inf} = 1.493$  for the 'PROCOL'-cell.

Table 2-66. Comparative two group cross-sections (5.2 w% enrichment)

nuclide	$\sigma_{tot}(b)$		$\sigma_{abs}(b)$		$\sigma_{tr}(b)$	
	group 1	group 2	group 1	group 2	group 1	group 2
<sup>10</sup> B			42.20	2095		
C	4.086	4.805	1.882E-4	1.856E-3	3.348	4.702
O			1.527E-3	1.029E-4		
Si			3.623E-3	9.397E-2		
<sup>235</sup> U			1.692E+1	3.397E+2		
<sup>238</sup> U			2.747	1.499		

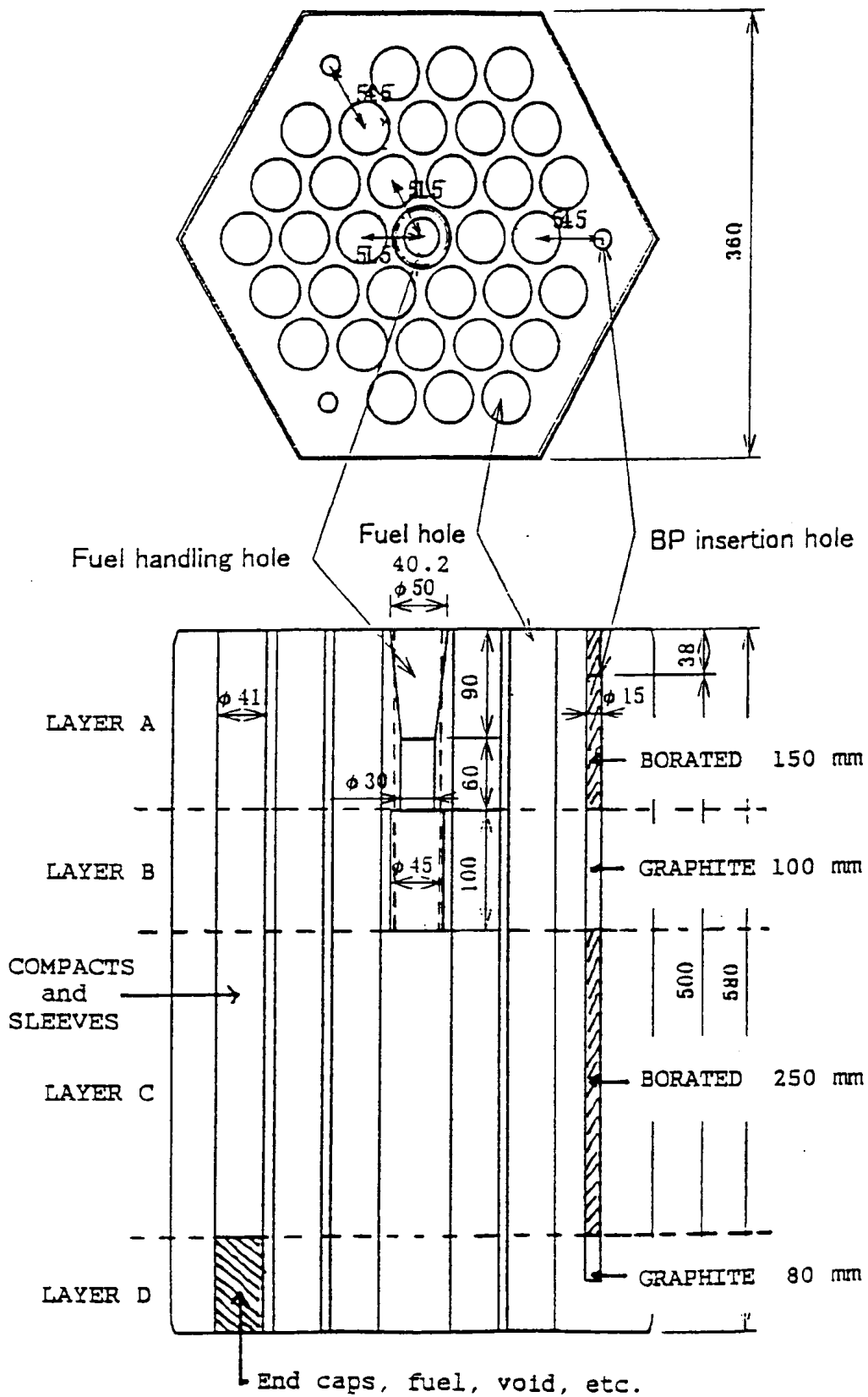


FIG. 2.66. Axial composition of the fuel block in PIJ.



Table 2-67. Uranium isotope fission neutrons and fission and capture cross-sections (5.2 w% enrichment)

	$\nu$		$\sigma_{\text{fis}}(\text{b})$		$\sigma_{\text{capt}}(\text{b})$	
	group 1	group 2	group 1	group 2	group 1	group 2
$^{235}\text{U}$	2.438	2.438	1.096E+1	2.897E+2	5.968	49.91
$^{238}\text{U}$	2.742	2.489	5.768E-2	6.437E-6	2.689	1.499

### *PIJ [2-50]*

For modelling in PIJ the fuel assembly has been adapted in the following way:

1. The stack of compacts has been moved to the top such that the upper rim of the upper compact is flush with the fuel block,
2. The upper graphite plug and buffer plate has been moved to the bottom,
3. The stack with the burnable poison (BP) pellets and graphite disks has been moved to the top as in 1, while the length of the upper section of the BP stack has been changed from 20 cm to 15 cm and the lower section to 25 cm
4. The fuel handling hole has been simplified by taking an effective diameter of 4.017 cm and a length of 25.0 cm.

This way four layers in the assembly can be created (See Figure 2.66):

1. First layer of 15 cm height with compacts, fuel handling hole (FHH) and BP pellets,
2. Second layer of 10 cm with compacts, FHH and graphite disks,
3. Third layer of 25 cm with compacts, graphite for FHH and with BP pellets,
4. Fourth layer of remaining 8 cm with a mix of 4.75 cm of compact, 2.35 cm of graphite and 0.9 cm of void at the fuel positions and graphite at the FHH and BP positions.

The void has been modelled in the empty BP insertion leg.

For each layer a model of the fuel assembly has been laid out in which the hexagonal perimeter has been replaced by an equivalent circle (radius 19.01 cm). Within this circle the fuel positions (comprising: inner gas space, compact, sleeve and outer gas space), FHH and BP insertion holes are modelled at the exact positions and filled with the materials in conformance. This circle in turn is surrounded by another circle (radius 38.01 cm), divided into 12 segments, to accommodate the matching surrounding materials for the fuel assembly under study (Figure 2.67). It makes a total of 206 material regions per assembly layer.

The coolant bearing reflector blocks in the 1st, 2nd and 8th reactor layer are modelled in the same way but with empty fuel holes and of reduced diameter.

To reduce the number of materials, the PIJ model is finally divided into seven regions: one central region comprising the FHH position and the six inner fuel positions, and the six surrounding segments (Figure 2.67). Materials within a region are homogenised or smeared to one material. Finally the seven materials for the four layers are smeared, according to their height, to seven final materials for one assembly having flux weighted cross sections in 16 neutron energy groups.

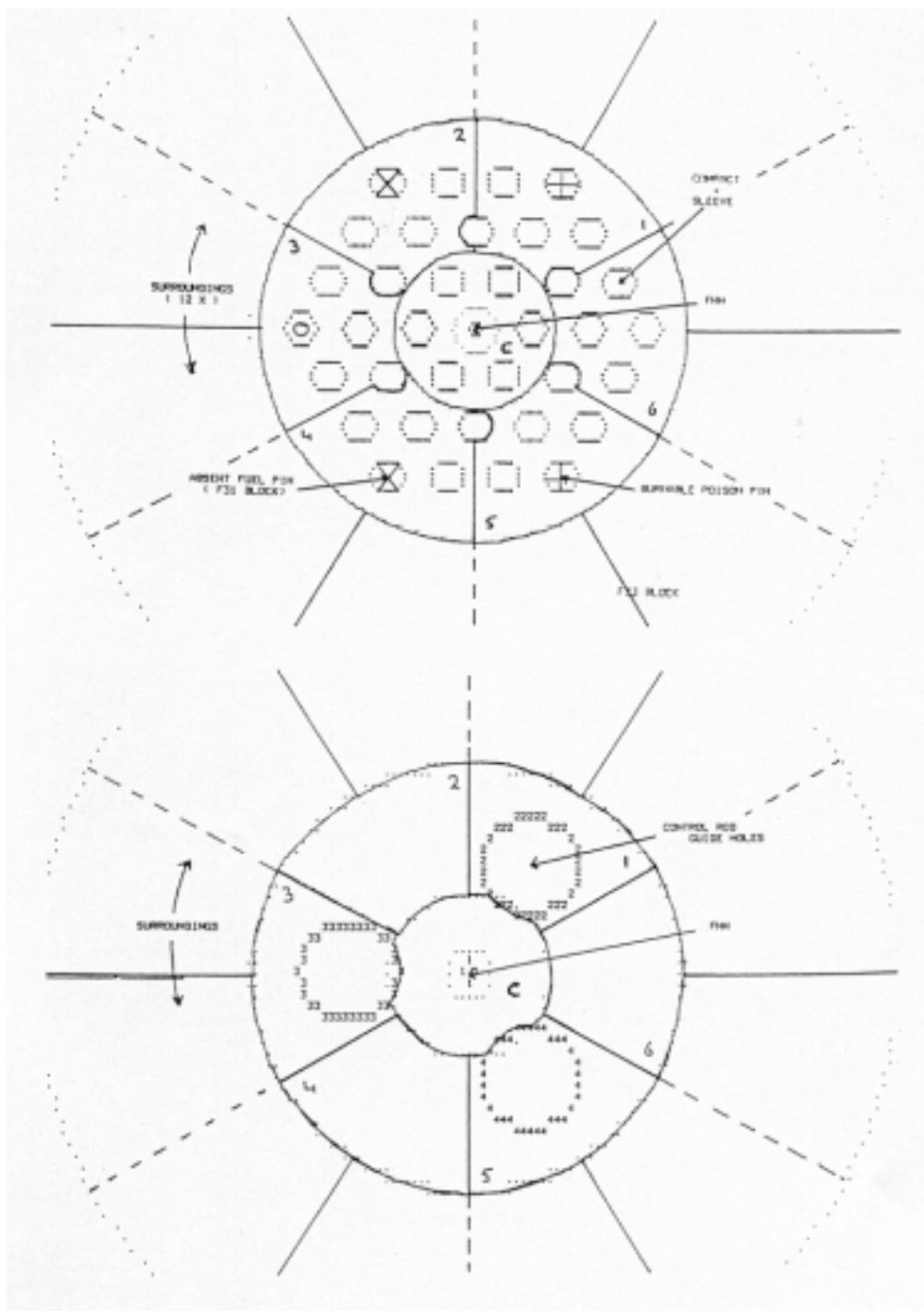


FIG. 2.67. Radial composition of the fuel and control guide block in PIJ.

The procedure for the control guide blocks (Figure 2.67) and reflector blocks is similar; also divided into seven regions but with only two layers, with and without FHH.

The advantage of the sub-division in seven regions is that the anomalies in a block, like BP stacks, absent fuel pins, control guide holes, control rods, etc. are confined to only one region a piece and are not smeared over the entire block. This allows for more pronounced local absorption and/or streaming, which form major problems for modelling this kind of reactor cores.

For all 48 different block configurations (enrichments, block types, surroundings, etc.) two runs with PIJ were done; first a run without control rods (unrodded) and a second run with control rod material modelled in the control guide holes and using rodded material in that sector of the surrounding where present (rodded). Afterward all cross sections were condensed to two energy groups ( $E_{th} = 2.1$  eV) and organised in such a way that it can be used in the reactor code PANTHER, leading to 336 different materials in as well a rodded state as an unrodded state.

By making use of the modules PROCOL and PIJ the double heterogeneity formed by the CFP's and the fuel rods has been modelled explicitly and therefore no Dancoff factor has to be introduced.

#### 2.2.5.2. Results of HTTR first criticality (Phase 1) [2-53]

The KENO model is a very detailed model of the HTTR in which practically all components are modelled explicitly, with the following exceptions:

- 1 As mentioned before, the coated fuel particles were homogenised with the graphite matrix of the fuel compacts.
- 2 It is not possible to model hexagonal blocks in KENO-Va. Therefore, the permanent reflector was approximated by a cylinder of 214.98 cm radius which preserves the volume of the actual reflector. Furthermore, the hexagonal blocks in the core and in the replaceable reflector were represented by cylinders of 36 cm diameter (the distance between the parallel faces of the blocks). These cylinders (which contain all fuel rods and the two burnable poison rods or all coolant channels) were placed in a large cylinder of graphite with a radius of 162.9 cm (Figure 2.68).

In the BOLD VENTURE model the HTTR is represented by an R-Z model. It contains six zones in the radial direction, and nine in the axial direction, one for each layer. The six radial zones are the:

1. Central control rod guide column (column A)
2. First fuel zone (the six B columns)
3. Second fuel zone (the 12 C columns: 6 fuel columns and 6 control rod columns)
4. Third and fourth fuel zone (the 18 D columns)
5. Replaceable reflector (the 24 E columns)
6. Permanent reflector

The height of each layer is 58 cm, except layer 9 (42.9 cm). The radii of the zones are: 19.01, 50.29, 82.85, 115.61, 148.44 and 214.98 cm. Calculations were performed with a 2 cm mesh, both in axial as radial direction and the BP rods were simulated by adding boron to the radial zones B, C and D to such an extent that a reactivity change, as determined by auxiliary KENO calculations, was reached.

For PANTHER a 3-D model has been developed in a hexagonal representation, taking a cluster of seven sub hexes (size: 13.68 cm flat-to-flat) per hexagonal reactor assembly position in the radial direction and 5 layers per assembly in the axial direction. This leads to 937 radial reactor channels with an equivalent radius of 220 cm and 45 axial layers of 11.6 cm.

Control rods, those left partially inserted in the E-column ring, reached only till the bottom level of the upper block (464 cm level).

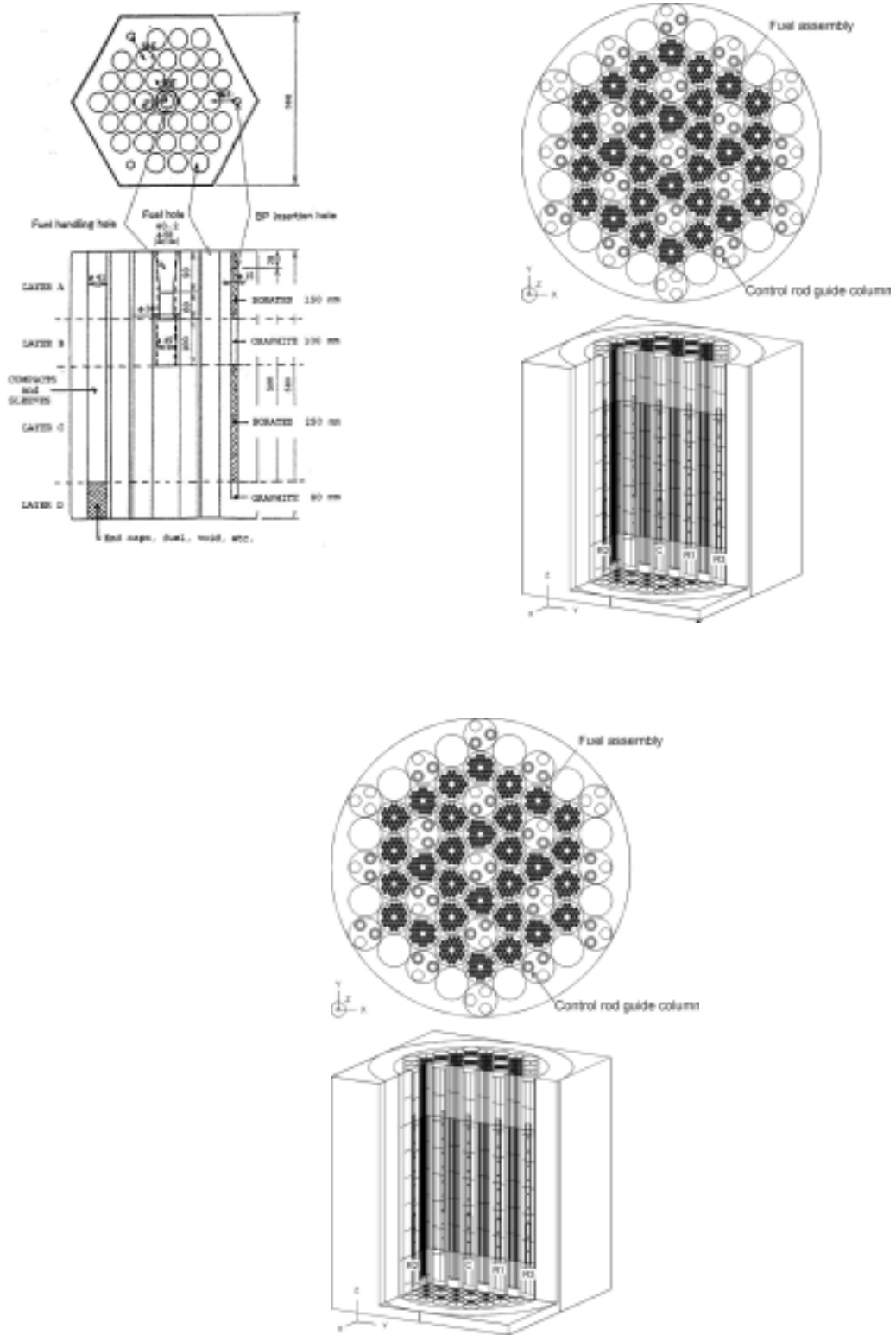


FIG. 2.68. Fuel assembly (upper left), the KENO model for the critical reactor (upper right, and the KENO model for scram with all control rods (C, R1, R2 and R3) (lower).

Materials defined and prepared in the WIMS data generation phase has been laid down according to proper compositions and orientations of the reactor assembly blocks in the reactor. For the simple core all enrichments were set at 5.2 w%.

In PANTHER the assemblies which carry control rods are represented by two sets of nuclear data: one set for the part where is no control rod inserted (unrodded) and a set for the rodded part. The control rod insertion depth for a certain control rod bank determines whether PANTHER uses the set for the rodded material or for the unrodded material in a particular mesh, thus enabling to drive a control rod.

The results of the calculations by the different codes are presented in Table 2-68. Good agreement can be found between the KENO and PANTHER results, the higher  $k_{eff}$  values for BOLD VENTURE can be attributed to neutron streaming in the control rod guiding holes

Table 2-68. Comparison of the results

	KENO	BOLD-VENTURE	PANTHER	Measured
$k_{eff}$ simple core	$1.1278 \pm 0.0005$	1.1592	1.1251	
$k_{eff}$ fully loaded core				
- rods withdrawn	$1.1584 \pm 0.0005$	1.1974	1.1595	
- rods inserted	$0.6983 \pm 0.0005$		0.7510	$0.685 \pm 0.010$
critical insertion				
- above bottom core	170.5 cm		161.5 cm	178.9 cm

### 2.2.5.3. Results HTTR-CR and HTTR-TC for Phase 2 [2-53]

The second benchmark of the HTTR core physics (Phase 2), defined by JAERI (Yamashita et.al.,1999b), were also calculated by using the Monte-Carlo code KENO-Va (V4.3) for a fully loaded core with 30 fuel elements and are presented in the Working Material of the IAEA meeting (CRP-5, 1999 and Türkcan et.al.,1999).

In Phase 2 the following answers were requested by the organisers:

- Scram reactivities (control rod worth) for the control rods at critical position and after a scram of the reflector control rods (HTTR-SR) and after a scram with all the control rods (HTTR-SA), both at a temperature of 300K.
- Isothermal temperature coefficients for a fully loaded core from 280K to 480K in six temperature steps (HTTR-TC). Where the control rod settings (C, R1 and R2) have slightly different settings due to a temperature elevation and the critical insertion of the control rods C, R1 and R2, (R3 stays fully out) at 480K.

For the new benchmark calculations, new cross-sections were prepared for seven different temperatures using the aforementioned procedure. The resulting cross section libraries contain data for 172 neutron energy groups. The geometry input for KENO is revised to be able to calculate the questions of the benchmark. The KENO model for a critical reactor and with all control-rods inserted is shown in Figure 2.69. The results of the benchmark-phase 2 will be summarised.

The scram reactivity of control rods ( $\Delta k/k$ ) is defined by:

$$\rho_R = \frac{k_{Crit} - k_{RCR-in}}{k_{Crit} \cdot k_{RCR-in}}$$

With:  $k_{Crit}$  : Effective multiplication factor at critical CR position  
and  $k_{RCR-in}$ : Effective multiplication factor at CR position after scram.

Table 2-69 provides the scram reactivity for two different scram conditions. Also given are the calculated and measured scram reactivities (Fujimoto et.al.,1999).

Table 2-69. Scram Reactivities (Using KENO)

CR Group	Critical position (mm) HTTR-Crit	Position after scram (mm) HTTR-SR	Position after scram (mm) HTTR-SA
C	1789	1789	-41
R1	1789	1789	-41
R2	1789	-41	-41
R3	Full out	-41	-41
$k_{eff}$ (average)	$1.0093 \pm 0.000$	$0.9178 \pm 0.0005$	$0.6809 \pm 0.0005$
$\rho_{calc}$		$0.0988 \pm 0.0007$	$0.4778 \pm 0.0007$
$\rho_{meas}$		$0.120 \pm 0.012$	$0.46 \pm 0.04$

The results using PANTHER calculations include the following:

$$K_{eff} \text{ at critical position} = 1.0088; \quad K_{eff} \text{ after scram HTTR-SA} = 0.7317; \quad \rho = 0.375$$

This scram reactivity is too low but can be explained by neutron streaming in the control rod holes and is to be recalculated by means of anisotropic cross section

In the second question of the benchmark, the isothermal temperature coefficients (HTTR-TC) for a fully loaded core between temperatures 280K to 480K (in six steps) were asked, where the control rods C, R1, and R2 have slightly different settings due to temperature elevation (13 mm).

The effective multiplication factors should be calculated for the following temperatures: 280, 300, 340, 380, 420, 460 and 480 Kelvin and the isothermal temperature coefficients should be calculated at: 290, 320, 360, 400, 440 and 470 Kelvin. The insertion depth of C, R1, R2 is the same at level =1776 mm and R3 again is fully withdrawn. Also the critical position for those control rods at 480K, with R3 fully out, is requested for the benchmark.

The following relation should evaluate the isothermal temperature coefficients for a fully loaded core from the effective multiplication factors:

$$\rho_n = \frac{k_{n+1} - k_n}{k_{n+1} \cdot k_n} \cdot \frac{1}{(T_{n+1} - T_n)}$$

$\rho_n$  : Temperature coefficient between  $T_n$  and  $T_{n+1}$  ( $\Delta k/k/K$ )

$T_n$  : Core temperature at  $n^{\text{th}}$  measurement (K)

$T_{n+1}$  : Core temperature at  $n+1^{\text{th}}$  measurement (K)

$k_n$  : Effective multiplication factor at  $T_n$

$k_{n+1}$  : Effective multiplication factor at  $T_{n+1}$ .

Results of the calculations are shown in Figure 2-69. The calculated isothermal temperature coefficient (average between 320K and 440K) is  $-14.7$  (pcm/°C), while the measured value equals  $-14.2$  (pcm/°C) on the average. The calculated critical control rod position at 480K is 1879 mm, while for the measurements at  $T=395$ K; 1873 mm and at  $T=418$ K; 1903 mm are found.

The PANTHER value for  $\rho_{iso} = -15.2$  (pcm/K), and the control rod position at 480K is calculated as 1934 mm.

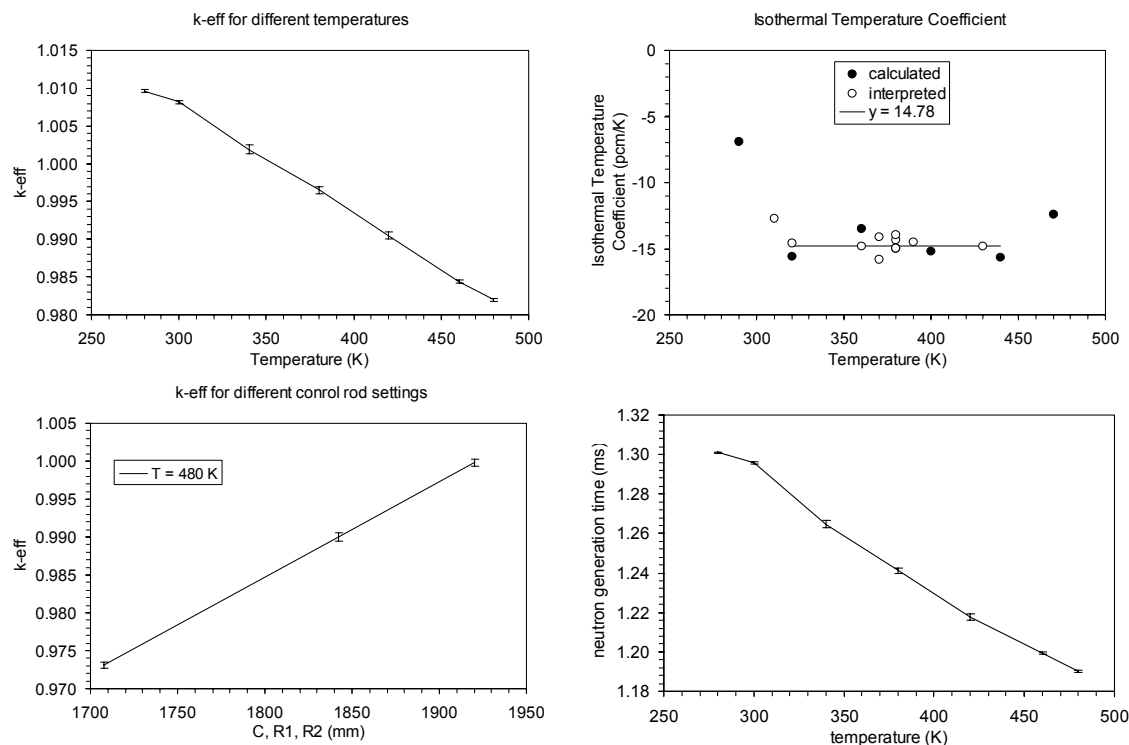


Figure 2-69. The effective multiplication factor (upper left) and isothermal temperature coefficient at different core temperatures, the multiplication factor for the different control rod settings of C, R1, R2 while R3 fully out (lower left) and the neutron generation time versus core temperatures (lower right).

#### 2.2.5.4. Reactivity and reactor noise measurements of the HTTR during the start-up Cores [2-53]

During the start-up phase of the HTTR at different core configurations, reactivity and the reactor noise measurements were carried out in parallel with measurements of the HTTR Physics group. For these measurements two temporary compensated ionization chambers CIC-A and B were used. Figure 2.70 shows the horizontal cross-section of the core and the positions of the detectors. Measurements were carried out by using the signal processing system DSA-2 (Türkcan, 1993) in real-time. During the on-line reactivity experiments, the measured DC signals were digitised and the reactivity is calculated by using the Inverse Kinetics Method (IK).

During the first critical approach after the loading of the 19th fuel assembly, the source criticality at very low power is achieved. For criticality, first the neutron source is removed and then by moving the central control rod (C) to compensate for the reactivity until the first criticality of the reactor is reached on Nov. 10 1998. Figure 2.71 shows the result of reactivity measurements during this approach to criticality.

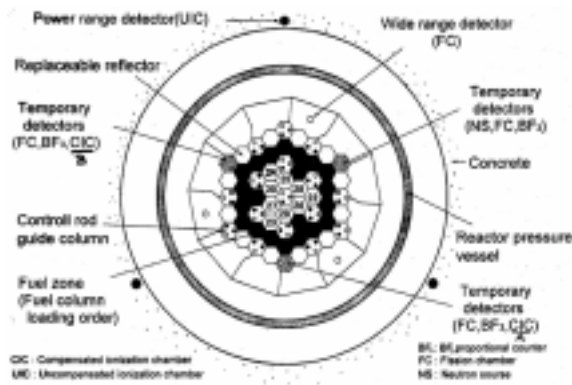


FIG. 2.70. HTTR horizontal neutron detector positions.

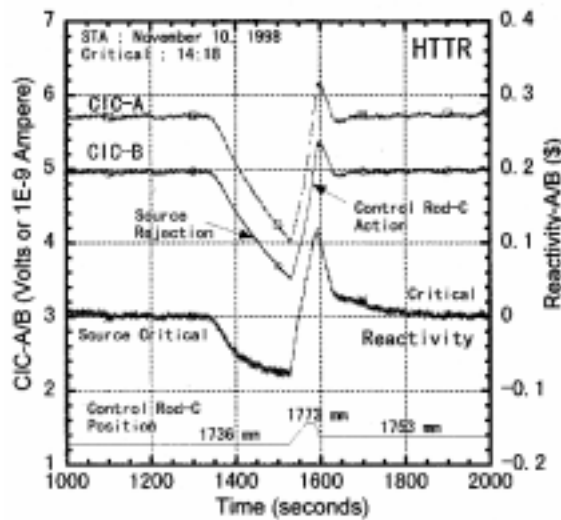


FIG. 2.71. Approach to criticality.

After each new fuel loading, the reactivity value of the control rods were measured successively with the IK-method by the HTTR physics group and by DSA-2 system. As an illustration the experimental result is given in Figure 2.72 for full core with 30 fuel assemblies. By the move of a control rod with a small step, the neutron flux is increased while the reactor power is kept in the same power range by compensating the reactivity effect by another control rod. For each action where the reactivity is constant over about 80 seconds the calculated reactivity is averaged. This way the average reactivity worth of the control rod is determined for this stepwise change.

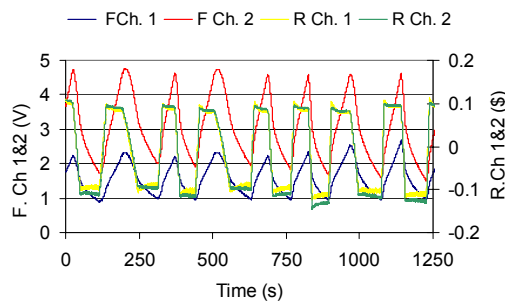


Fig. 2.72. Measured reactivity at HTTR full core. Two measured signals of Ch1 (A) and Ch2 (B) and the computed reactivity (blocks match each other, scale on right).



For the reactor noise measurements, the same neutron detector signals were used and the reactor at very low power is kept as stable as possible especially for this measurement. Signals were conditioned for the noise measurements by using high- and low-pass filters with a gain amplifier.

Inherent to this type of reactor, the neutron generation time is long and was calculated to be:  $1.173 \pm 0.001$  ms. The prompt neutron decay constant is quite close to the decay of the fastest delayed neutrons, therefore no intermediate plateau can be recognized in the measured spectral functions such as the Normalized Auto and Cross Power Spectral Density (NAPSD and CPSD).

Figure 2.73 is an example of the measurements on the 21 fuel element critical core at very low power. Our investigations indicated a shortcoming of the bandwidth of the used current amplifiers due to the large cable capacity of about 100-m of cable between the detectors and the amplifiers. This situation was not possible to change during the measurements.

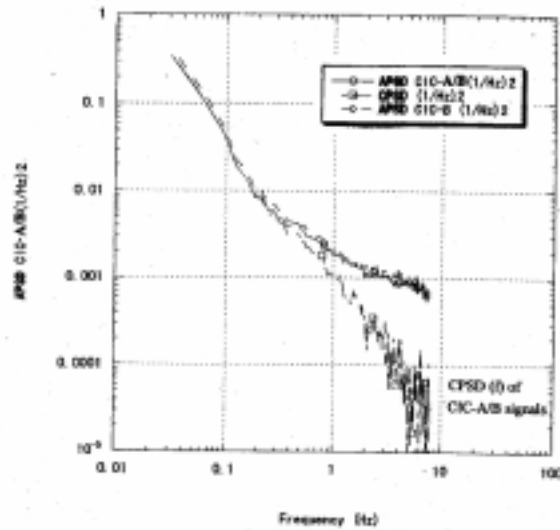


FIG. 2.73. The NAPSD( $f$ ) and the NCPSD( $f$ ) functions measured 21<sup>st</sup> fuel loading. The cross spectra do not give clear break frequency.

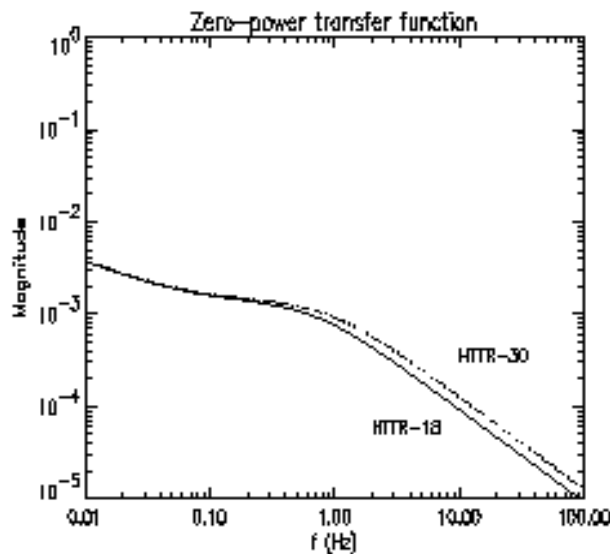


FIG. 2.74. Calculated Transfer functions for 18 and 30 fuel elements loading.

The measured coherence between the two neutron detector signals was 0.8 at 0.1 Hz and gradually decreased to 0.2 at 1 Hz and the phase between them practically zero. The calculation of the zero-power transfer function (Figure 2.74) for different number of fuel loading, the transfer function shape did not changed considerably.

#### *2.2.5.5. Summary of results [2-53]*

On the level of cell calculations a good agreement has been obtained between the cross sections and the spectra as prepared by the SCALE-system and as prepared by WIMS. Calculations with detailed geometry converged to very good agreement between the results of PANTER and the results of KENO with an exact geometrical model. In the second phase, KENO results gave very well the measured values of the scram reactivities as well as the estimation of the isothermal temperature coefficient within the requested temperature interval. Therefore, we can conclude that the Benchmark calculations of the start-up physics calculations were successful and that the results of the reactivity measurements and the reactor noise analysis done at the HTTR, using the DSA-2 system, resulted with a good agreement with the results of the HTTR Physics Group.

## 2.2.6 Russian Federation

### 2.2.6.1. General analysis method and model description

For benchmark calculations in the diffusion approximation, the computer codes WIMS-D/4 and JAR [2-54, 2-55] were used. MCU [2-56] and MCNP [2-57] codes, implementing the Monte Carlo method were used as reference.

WIMS-D/4 code [2-54] has been used for computation of few-group constants for fuel and the reflector blocks. The code is provided by 69 groups system of nuclear constants in neutrons retardation and thermalization range on the basis of estimated nuclear data in ENDF/B6 format. The characteristic feature of WIMS is a two-stage approach to solution of a task on spatial-energy neutron distribution in a reactor cell. In the first place, a detailed spectrum in 69 groups for every zone, typical for the cell is calculated: fuel, shell, coolant and moderator. Then cross-sections are convoluted to a specified few-group approximation, where detailed distribution of neutrons in geometric zones of the unit cell is calculated. The  $S_4$ -method was used in the first and second stages of the calculation. In calculations of spatial resonance screening the equivalence theorem is employed, with the help of which the heterogeneous problem comes to an effective homogeneous. Detailed matrices of scattering cross-sections for main moderators are employed in the neutron thermalization calculations.

For calculating of fuel cells with double heterogeneity of fuel allocation at the fuel particles and fuel compacts level in the cell, containing burnable poison rods, Segev algorithm [2-57] was used to calculate macro-sections for a cell, associated with a fuel compact. When using the Segev formalism to calculate parameters of fuel compact cell containing fuel particles, it is necessary to carry out calculation of three cell variants in cluster PIJ or PIJ-PERSEUS option:

a) a cell associated with fuel particle is calculated, whereby a kernel is represented as a rod with radius  $r_c = 9/16 r_{sph}$ , on which matrix moderator and shells, related to one kernel is "spread". Radii of the rod and cell are chosen so that collision probability for neutron, born in fuel, would be the same in cylindrical of module and spherical geometry. At cell calculation by WIMS at the given stage a value of Dankoff factor for inner region of the cell ( $\gamma$ ) is found;

b) a cell associated with annular fuel compact, in which the fuel and the compact matrix graphite are homogenized, is calculated. Panniculus equivalent of serves the shell from graphite, taking into account of gas, related to one compact, serves a layer of external moderator. At this stage, a value of Dankoff factor for outer region of the cell ( $\Gamma$ ) is determined;

c) then  $\gamma_{input}$  is calculated:

$$\gamma_{input} = \frac{1}{1 + (1/\gamma - 1) \cdot \frac{Vm \cdot \Sigma_m}{Vm \cdot \Sigma_m + \frac{1/L}{1/A + 1/\Gamma - 1}}} \quad (1)$$

where  $V_m$ - volumetric fraction;

$\Sigma_m$ - moderator scattering cross-section;

$L = M^{1/3} \cdot l_{cell}$ ; (2)

$M$  - number of fuel particles;

$l_{cell}$  - mean chord;

$A$  - Bell factor.

The value of  $\gamma_{\text{input}}$  is set in WIMS and calculation as per point “b” is repeated. Resulting from the last calculation macroscopic cross sections and multiplication factor for a fuel compact cell with double heterogeneity are acquired. It is worth noting that value of  $\gamma_{\text{input}}$  weakly depends on fuel temperature and isotopic content in case of fixed geometrical parameters of a cell.

Basic results of diffusion calculations were obtained in two-group approximation with thermal and epithermal neutrons cut-off energy of 0.625 eV, selected from the condition of obtaining values of cross-sections for transition to higher energy close to zero, that is defined by the calculation model of scattering in the reactor code, that takes account of energy transitions only in direction with energy decrease.

In the case of the cell with burnable poison rods calculation, the same option of WIMS code was used, but poison rods in the cell was surrounded by effective fuel composition accounting shielding factors.

The macro-constants received at WIMS calculation stage were then accepted as input data for reactor calculations by JAR code. The JAR code [2-55] is based on the nodalization method and makes it possible to carry out three-dimensional analysis of reactor in few-group diffusion approximation. The reactor core and the reflectors in a calculations model are represented as hexagonal prisms, broken down on triangles and divided into axial zones.

For the diffusion calculations the effect of nonuniformity of poison in burnable rods through its height was not accounted. Burnable pellets and graphite pellets were homogenized in the vertical direction. The construction of control rods absorber through its height is considered as uninterrupted.

The MCU code [2-56] based on Monte-Carlo methodology employs a combination of DLC/MCUDAT-1 data libraries. MCU is used to compute a continuous spectrum of neutrons moderation within energy range  $10^{-5}$  eV to 20 MeV, and to solve the neutron transport problem with an external source or criticality problems. To describe cross-sections in the field of unresolved resonance the subgroup dividing method is used, while for resolved resonance range the detailed description of cross-sections is possible on the basis of special data libraries.

MCNP code (Version 4A) [2-57] has been used to check results of the diffusion calculations. Nowadays the MCNP code is one of the most efficient computer codes, which employ the Monte Carlo methodology for analysis of high temperature gas-cooled nuclear reactors. Possibility of the MCNP code application for analysis of HTGRs with double heterogeneity of fuel allocation on the basis of comparison with other codes, was studied before. The heterogeneous model by MCNP was described so that each coated fuel particle was located in unit lattice by lattice geometry in fuel compact (not randomly).

Among essential features of MCNP, a capability to compute continuous spectrum of neutrons moderation within energy interval 0 to 20 MeV in approximation of point-by-point representation of nuclear data, should be pointed out. Point cross sections of neutrons interaction with nucleus were prepared by using of NJOY code in ENDF /B6 format. In criticality analysis a multiplication factor  $k_{\text{eff}}$  was estimated by three ways for larger reliability: viz.: estimation on collisions (col), estimation on absorption (abs), estimation on free track length (trk). Moreover, a combined estimation of  $k_{\text{eff}}$  based on the first three estimates was made.

For the Monte-Carlo calculations the detailed geometry was taken into account, but the presence of spacers on the fuel rod surface was not into account. Beside in the MCNP-4A code version the location of coated particles in graphite matrix is described in the form of cubic lattice.

Design model of the reactor (Figure 2.75) consists of nine layers of hexagonal fuel blocks and graphite blocks of reflector, divided into four subzones each. The reactor core includes reactivity compensation rods viz.: control rods and channels for the reserve shutdown system. It is supposed in the analysis, that the poison in the reactivity compensation rods is distributed axially without gaps, which actually are available in the rods and are about 6.45 % from the overall length of rod.

To calculate few-group cross sections and multiplication factor, the fuel block is modeled by an annular cell, area of which corresponds to 1/3 area of a fuel block; a burnable poison rod, a hole for it or a graphite disk are set in the center and surrounded by fuel, medium content of which corresponds to 1/3 of the block. The fuel content of the surrounding zone is homogenized taking into account a double heterogeneity of fuel arrangement and from condition to preserve multiplication factor value and migration parameters in the zone compared with a cell of the annular fuel compact (Figure 2.76).

Macro-cross sections of graphite reflector cells, including absorber rods or gas cavities for their location, were also calculated using the WIMS code. A calculation model of the cell is the following: in the center of the cell a cavity or an absorber rod (with detailed description of its geometry) is set, which is surrounded by a graphite ring with outer of radius, determined from the condition of surrounding graphite volume conservation. A layer, containing homogenized fuel content is used as an outer layer of the cell (for shaping spectrum).

Information about the codes, calculation models and nuclear data are summarized in Tables 2-70, 2-71 and 2-72.

Table 2-70. Codes, model and library of nuclear data for Diffusion calculations

Items	Name of Country	Russia
	Name of Institute	OKBM
Nuc. data file	ENDF/B6	
Fuel cell code	WIMS-D/4	
Theory	S4	
Model	Cylindrical cell	
Cut - off energy	0,625 eV	
No. of groups	69	
BP cell code	WIMS-D/4	
Theory	S4	
Model	Cylindrical cell	
No. of groups	69	
Control rod cell cal.	WIMS-D/4	
The theory	S4	
Model	Cylindrical cell with the central absorbing zone	
Number of groups	69	
Core cal. Code	JAR-3D	
Model	Triangular lattice (6 points / blocks)	
No. of groups (Fast + Thermal)	1+1	

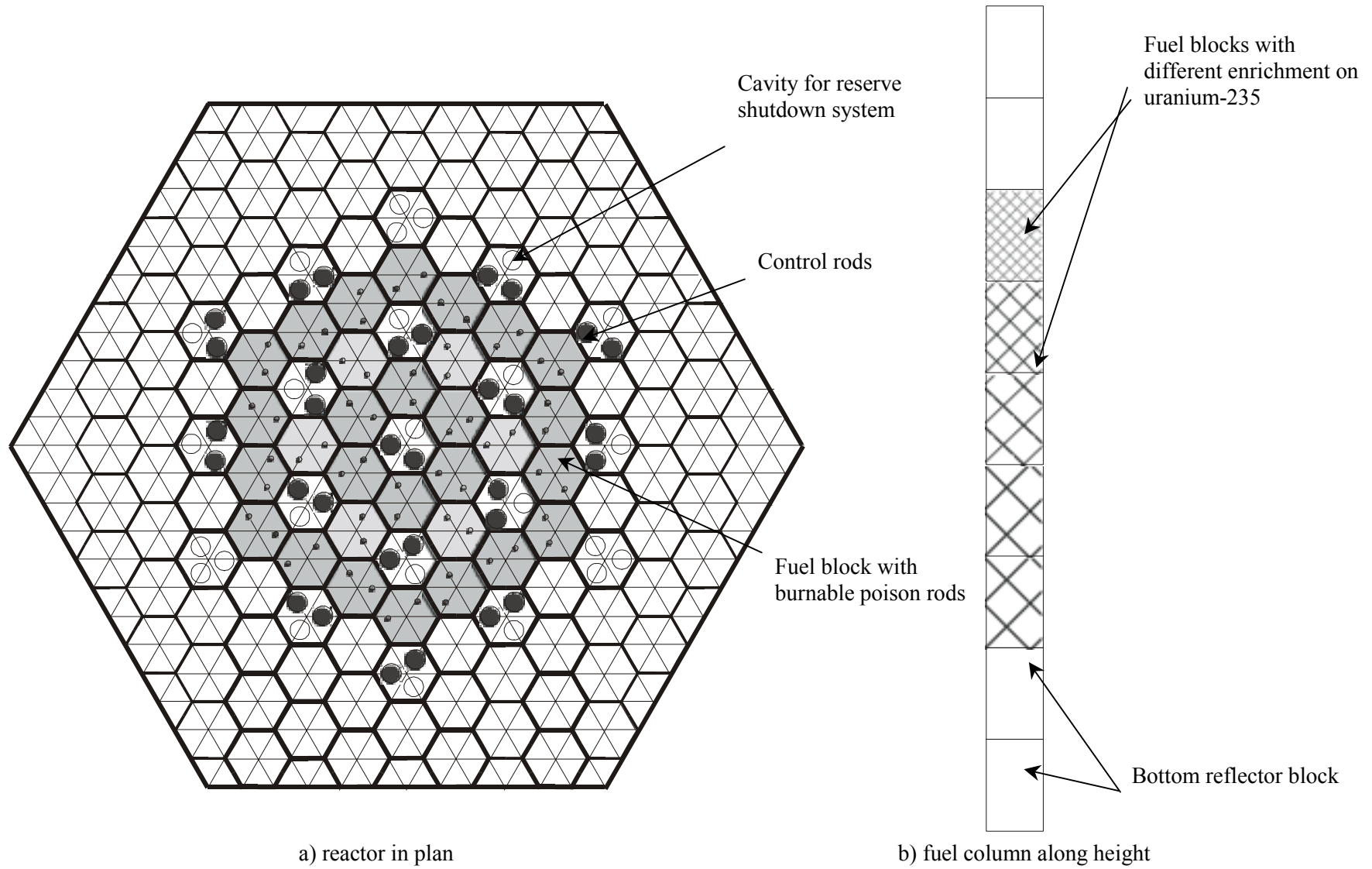


FIG. 2.75. Reactor calculation model.

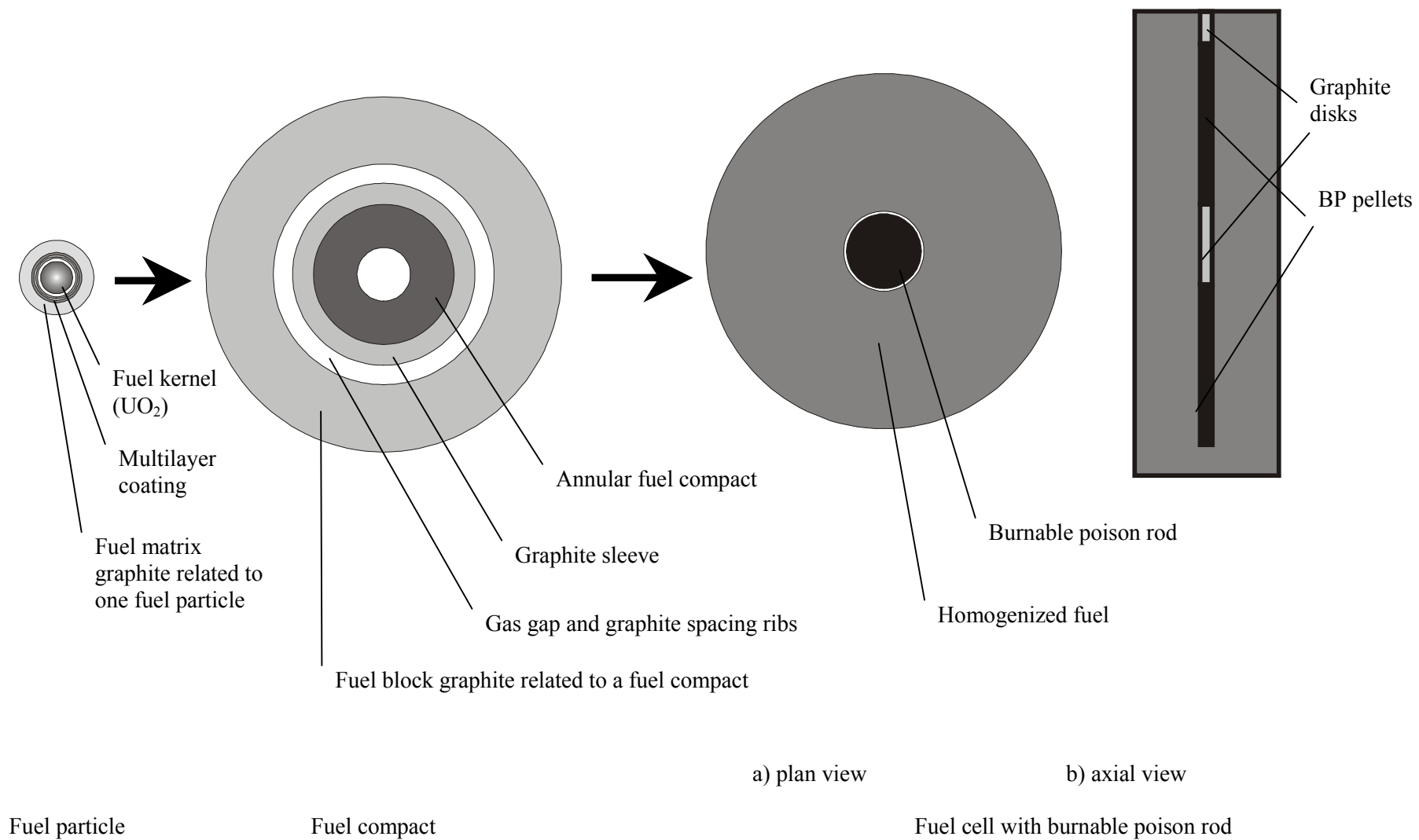


FIG. 2.76. Fuel cell calculation mode.

Table 2-71. Codes, models and nuclear data library for Monte-Carlo calculations

Items	Name of Country: Russia	Name of Country: Russia
	Name of Institute: RRC KI	Name of Institute: IBRAE
Code	MCU	MCNP 4A
Nuc. data file	DLC/MCUDAT-1.0	ENDF/B6 NJOY
Energy structure	Continuous	Continuous
Coated fuel particles	Fuel kernels are not smeared	Detailed account
History	200	2000 (up to 16000)
Batches	5000	1000
Skipped batches	1	10

Table 2-72. Methodical errors

Items	Calculation refining	Taking into account	
		Monte-Carlo calculation (MCNP)	Diffusion calculation
Forms and dimensions			
1 Graphite sleeve	Detailed description of spacer of sleeve	homogenized	homogenized
	Detailed description of gaps between sleeve and fuel compact	+	+
2 Graphite block	Detailed description of coolant holes, gaps between dowel pin and dowel socket, chamfering and spot facing	+	homogenized
3 Handling hole of graphite block	Detailed description of handling hole in all types of blocks	+	-
4 Position of burnable absorber	Horizontal locations of burnable absorbers are displaced on 3 mm to outside in fuel blocks	+	+
5 Shock absorber of control rod (CR)	Detailed description of shock absorber	+	-
6 Heterogeneity of coated fuel particle (CFP)	Statistical treatment of CFP location in fuel compact	cubic cell	determined location
Composition and density			
7 Fuel compact	Fuel compact load drift from nominal	-	-
8 Air in graphite block	Residual air in graphite block	-	-
9 Impurity in dummy block	Impurity revising from 2,5ppm to 3,1ppm	+	+
10 Partial insertion of R2-CR	Partial insertion of R2-CR	+	+
11 Composition of graphite block	Different composition of a part of permanent reflector and dummy block	+	+



### 2.2.6.2. First Criticality [HTTR-FC]

Analysis of  $k_{eff}$  dependence on sequence of fuel columns loading was conducted, the results of which are depicted in Figure 2.77. In connection with this a consecutive loading, considered accordingly the specification JAERI-memo 10-005 and symmetrical loading, is illustrated in Figures 2.78, 2.79 and 2.80. First results of HTTR-FC calculations were presented at IAEA Research Coordination Meeting [2-59]. The results on phase 2 of Benchmark Problems of HTTR were presented at IAEA CRP-5 meeting in China, October 18-22, 1999 [2-60, 2-61].

Characteristic	Value		
	Diffusion calculations	Monte Carlo calculations	
Organization	OKBM	IBRAE	RRC KI
Number of fuel columns	16	16	17
$k_{eff}$	1.005	$1.006 \pm 0.0016$	$1.004 \pm 0.0012$
Reactivity excess, $\% \Delta k/k$	0.498	0.596	0.398

In the annular reactor core of small diameter with high neutron leakage value of multiplication factor is very sensitive to disposition of fuel columns, and to ensure a power distribution symmetry, while control rods in upper most position it makes sense to load the fuel columns symmetrically throughout the core.

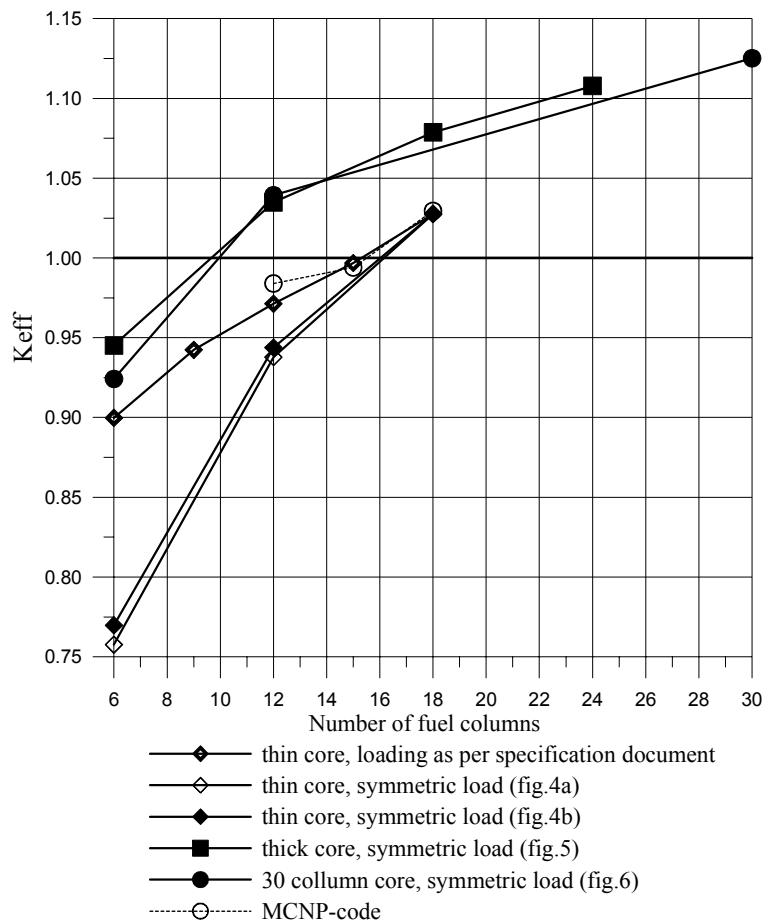


FIG. 2.77.  $k_{eff}$  versus number and arrangement of fuel columns.

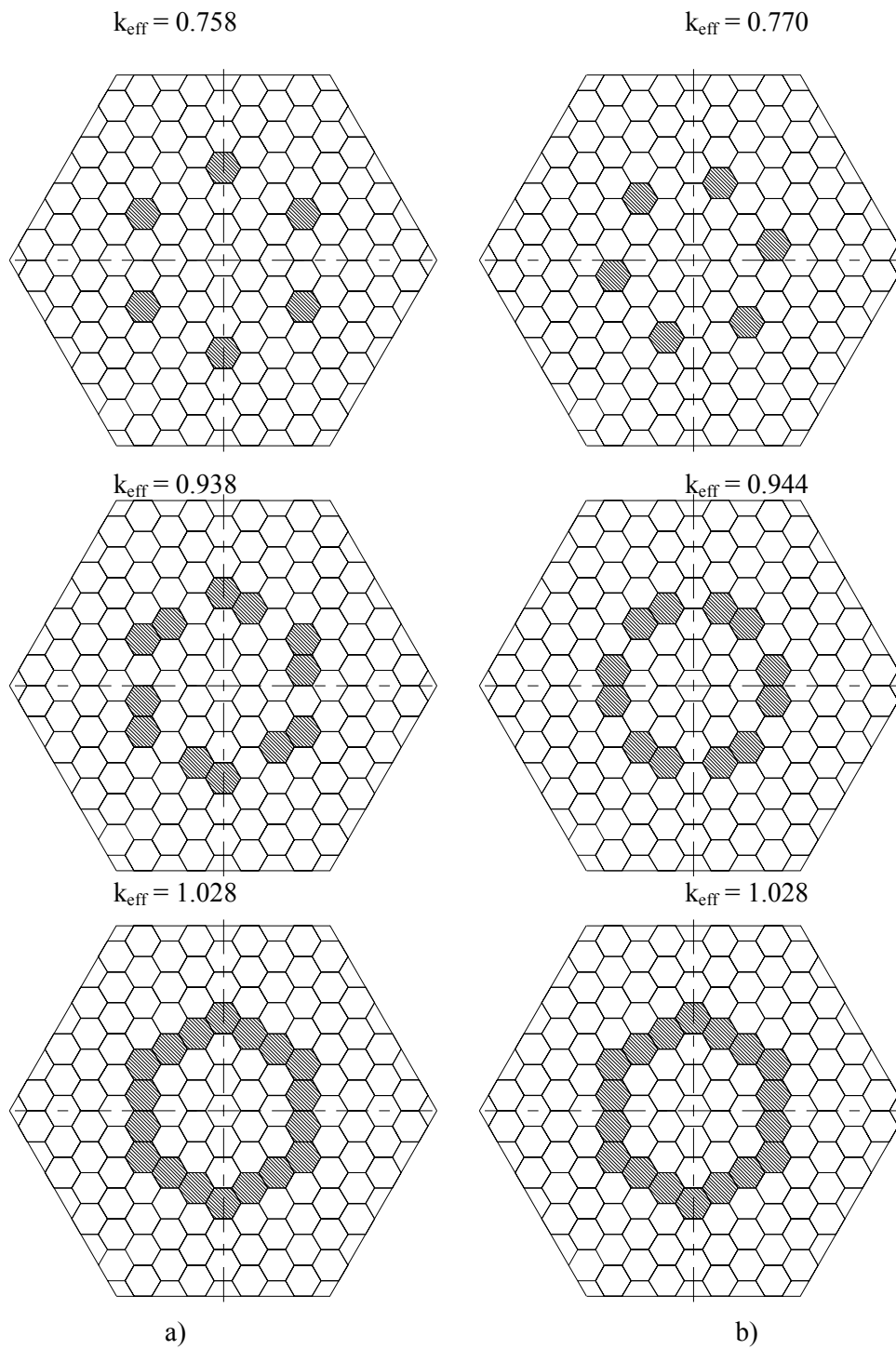


FIG. 2.78.  $k_{\text{eff}}$  versus fuel columns arrangement (thin annular core – 18 fuel columns).

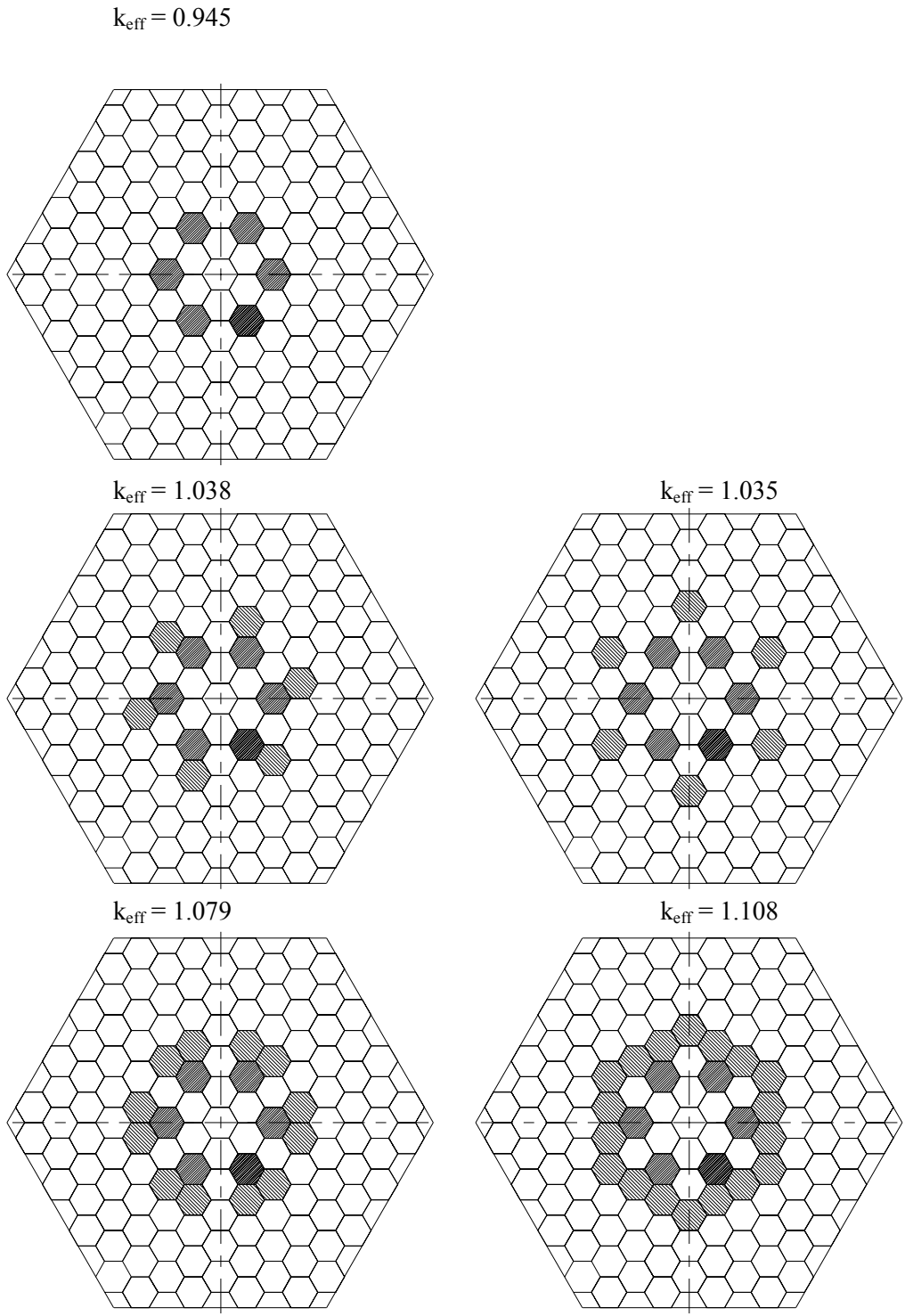
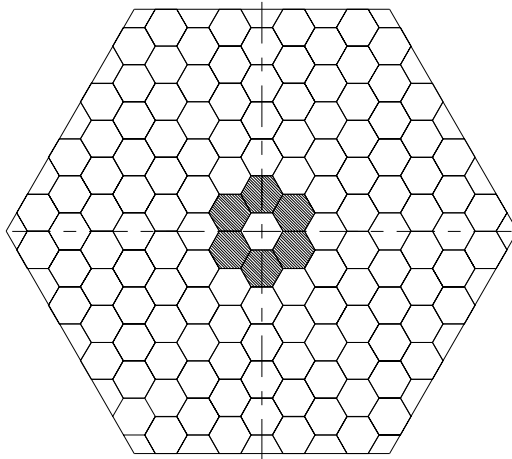
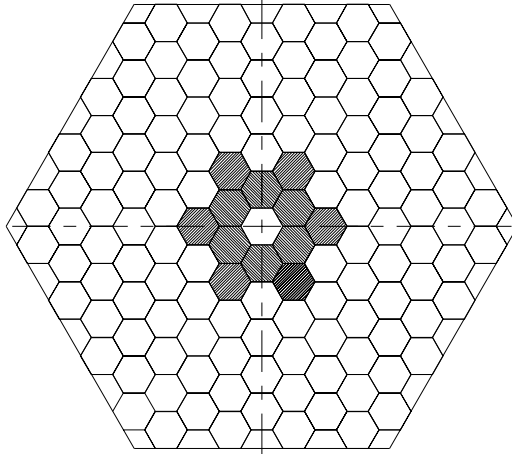


FIG. 2.79.  $k_{eff}$  versus fuel columns arrangement (thick annular core – 24 fuel columns).

$k_{\text{eff}} = 0.924$



$k_{\text{eff}} = 1.039$



$k_{\text{eff}} = 1.125$

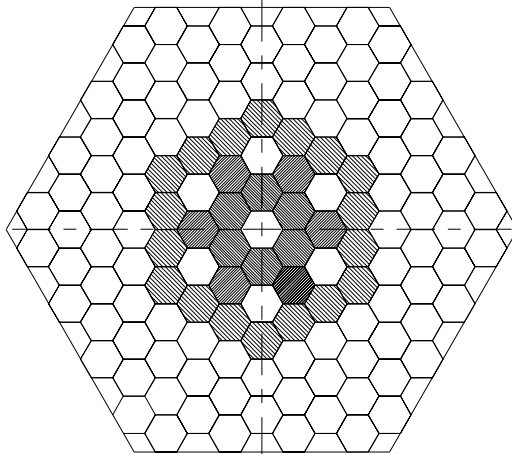


FIG. 2.80.  $k_{\text{eff}}$  versus fuel columns arrangement (completely loaded core – 30 fuel columns).

The Diffusion and Monte-Carlo calculations for the benchmarks HTTR-CR, HTTR-EX, HTTR-SC and HTTR-TC are provided in Tables 2-73, 2-74, 2-75 and 2-76, respectively.

### 2.2.6.3. Control Rod Position at Criticality [HTTR-CR]

Table 2-73. Results of Diffusion and Monte-Carlo Calculations for the HTTR-CR Benchmark Problem

Characteristic	Value		
	Diffusion calculations	Monte Carlo calculations	
Organization	OKBM	IBRAE	RRC KI
Depth of control rods insertion, cm			
18 columns	271	259	306
24 columns	196	195	201
30 columns	166	170	154

### 2.2.6.4. Excess Reactivity [HTTR-EX]

Table 2-74. Results of Diffusion and Monte-Carlo Calculations for the HTTR-EX Benchmark Problem

Characteristic	Value		
	Diffusion calculations	Monte Carlo calculations	
Organization	OKBM	IBRAE	RRC KI
Reactivity excess, % $\Delta k/k$			
18 columns	2.68	2.70	1.70
24 columns	9.73	10.83	9.80
30 columns	11.14	13.55	13.40

### 2.2.6.5. Scram Reactivity [HTTR-SC]

Table 2-75. Results of Diffusion and Monte-Carlo Calculations for the HTTR-SC Benchmark Problem

Characteristic	Value		
	Diffusion calculations	Monte Carlo calculations	
Organization	OKBM	IBRAE	RRC KI <sup>b)</sup>
Worth of reflector rods			
$k_{\text{eff}}^{\text{krit}}$ <sup>a)</sup>	1.0023	1.0098±0.0015	1.0265±0.0010
$k_{\text{eff}}^{\text{after drop}}$	0.9242	0.9205±0.0017	0.9349±0.0010
$\rho$ ( $\Delta k/k$ )	0.0843	0.0961±0.0020	0.0955±0.0014
Worth of all rods			
$k_{\text{eff}}^{\text{krit}}$	1.0023	1.0098±0.0015	1.0265±0.0010
$k_{\text{eff}}^{\text{after drop}}$	0.6573	0.7172±0.0015	0.6746±0.0010
$\rho$ ( $\Delta k/k$ )	0.5237	0.4040±0.0019	0.5081±0.0014

<sup>a</sup> Calculated value at critical position of rods obtained by experiment.

<sup>b</sup> Results were obtained accordingly the JAERI-memo 10-005 including graphite impurity data.

### 2.6.6.6. Isothermal Temperature Coefficient [HTTR-TC]

Table 2-76. Results of Diffusion and Monte-Carlo Calculations for the HTTR-TC Benchmark Problem

Characteristic	Value		
	Diffusion calculations	Monte Carlo calculations	
Organization	OKBM	IBRAE	RRC KI
Multiplication factor of reactor for design temperatures			
$k_{280}$	-	-	-
$k_{300}$	1.0023	1.0032±0.0002	1.0279±0.0010
$k_{340}$	0.9930	0.9954±0.0002	1.0232±0.0010
$k_{380}$	0.9844	0.9886±0.0002	1.0160±0.0010
$k_{420}$	0.9768	0.9822±0.0002	1.0124±0.0010
$k_{460}$	0.9699	0.9754±0.0004	1.0052±0.0010
$k_{480}$	0.9665	0.9726±0.0002	1.0026±0.0010
Temperature coefficients ( $\Delta k/k/K$ )			
$\rho_{290}$	-	-	-
$\rho_{320}$	$-(2.33 \cdot 10^{-4})$	$-(1.95 \pm 0.10) \cdot 10^{-4}$	$-(1.1 \pm 0,4) 10^{-4}$
$\rho_{360}$	$-(2.19 \cdot 10^{-4})$	$-(1.73 \pm 0.10) \cdot 10^{-4}$	$-(1.7 \pm 0,4) 10^{-4}$
$\rho_{400}$	$-(1.97 \cdot 10^{-4})$	$-(1.65 \pm 0.10) \cdot 10^{-4}$	$-(0.9 \pm 0,4) 10^{-4}$
$\rho_{440}$	$-(1.82 \cdot 10^{-4})$	$-(1.77 \pm 0.16) \cdot 10^{-4}$	$-(1.8 \pm 0,4) 10^{-4}$
$\rho_{470}$	$-(1.81 \cdot 10^{-4})$	$-(1.48 \pm 0.32) \cdot 10^{-4}$	$-(1.3 \pm 0,7) 10^{-4}$

### 2.6.6.7 Accuracy analysis

The accuracy analysis of obtained results is presented as follows:

Table 2-77. Excess reactivity [HTTR-EX]

	18 columns	24 columns	30 columns
Diffusion calculations			
Average value	1.6	8.8	12.4
(in accordance with RCM4 results), % $\Delta k/k$			
Average calculated value / experimental value, %	<sup>a)</sup>	+ 14	+ 3.3
WIMS-JAR value / average calculated value, %	+ 67	+ 10	- 10
WIMS-JAR value / experimental value, %	<sup>a)</sup>	+ 26	- 7
Monte - Carlo calculations			
Average value	1.87	10.1	13.0
(in accordance with RCM4 results), % $\Delta k/k$			
Average calculated value / experimental value, %	<sup>a)</sup>	+ 31	+ 8.3
MCNP value / average calculated value, %	+ 44	+ 7	+ 4.2
MCU value / average calculated value, %	- 9	- 3	+ 3
MCNP value / experimental value, %	<sup>a)</sup>	+ 40	+ 13
MCU value / experimental value, %	<sup>a)</sup>	- 27	+ 12

<sup>a)</sup> Experimental result is absent.

Table 2-78. Control rods worth [HTTR-SC]

	Reflector control rods	All control rods
Diffusion calculations		
Average value (in accordance with RCM4 results), % $\Delta k/k$	9.2	48.0
Average calculated value / experimental value, %	- 23	+ 4.3
WIMS-JAR value / average calculated value, %	- 8	+ 9
WIMS-JAR value / experimental value, %	- 30	+ 14
Monte - Carlo calculations		
Average value (in accordance with RCM4 results), % $\Delta k/k$	9.2	44.8
Average calculated value / experimental value, %	- 23	-2.6
MCNP value / average calculated value, %	+ 4.4	- 10
MCU value / average calculated value, %	+ 3.8	+ 13
MCNP value / experimental value, %	- 20	- 12
MCU value / experimental value, %	- 20	+ 10

Table 2-79. Isothermal Temperature Coefficient at the temperature range 345 – 407 K [HTTR-TC]

Diffusion calculations	
Average value (in accordance with RCM4 results), % $\Delta k/k \cdot K$	- 1.63·10 <sup>-4</sup>
Average calculated value / experimental value, %	+ 21
WIMS-JAR value / average calculated value, %	+ 28
WIMS-JAR value / experimental value, %	+ 54
Monte - Carlo calculations	
Average value (in accordance with RCM4 results), % $\Delta k/k \cdot K$	- 1.32·10 <sup>-4</sup>
Average calculated value / experimental value, %	- 2
MCNP value / average calculated value, %	+ 28
MCU value / average calculated value, %	- 4
MCNP value / experimental value, %	+ 25
MCU value / experimental value, %	- 4

Calculated accuracy analysis, presented above, demonstrates that 2-groups diffusion approximation gives significant error at the excess reactivity calculation for annular core with one ring of fuel assemblies. Multigroups approximation is necessary for calculation of this type core.

Large spread in excess reactivity values, obtained by MCNP and MCU, from average calculated and experimental values demonstrates the necessity to revise calculational initial data (it may be moisture content in graphite), on the one side, and also to verify nuclei data library, on the other side.

Errors analysis of control rods worth estimation shows the necessity to use multigroups diffusion approximation for calculations of the side reflector control rods worth that is estimated with the most errors. Deviation of Monte-Carlo calculated control rods worth from experimental one in the range about 20 % should be also referred to the effect of initial data because deviation in calculated values is not large.

To estimate reactivity temperature coefficients, it is necessary additional analysis characterized by accurate modeling of geometry and fuel particle distribution, detail modeling of burnable poison position by Monte-Carlo method, and for diffusion calculations it is necessary to model axial distribution of burnable poison, neutrons streaming for adequate accounting of leakage and to use multigroups approximation.

Significant dependence of the results versus chosen nuclei data libraries and its compiled programs defines the necessity to choice basic library for HTGRs calculation especially for description of thermalization effects.



## **2.2.7 Turkey [2-63]**

### *2.2.7.1 Introduction*

This study was performed by the Nuclear Engineering Department of Hacettepe University as a contribution to the IAEA CRP on Evaluation of HTGR Performance. MCNP-4B has been utilized throughout this study. This is a multi-purpose Monte Carlo n-particle (neutron, photon, and electron) transport code. This code also performs criticality analysis.

#### ***Reactor configuration***

The HTTR core consists of vertically arranged prismatic hexagonal blocks with a height of 580 mm and a width of 360 mm across flats. Active core height and effective diameter are 390 cm and 230 cm, respectively. Reactor core is made of 30 fuel columns and 7 control rod guide columns. The core is enveloped by 24 prismatic hexagonal reflector blocks. Among them, there are 9 control rod guide columns, 12 replaceable reflector columns, and 3 irradiation columns. The whole structure is surrounded by a permanent graphite side reflector.

Each fuel column consists of 2 top reflector blocks, 5 fuel assemblies, and 2 bottom reflector blocks. There are two types of fuel assemblies; containing 31 and 33 fuel rods. Each fuel block contains three burnable poison (BP) insertion holes, but only two of them are filled with B<sub>4</sub>C-C BP materials (containing 2.22 or 2.74 wt. % natural boron). Fuel rods contain 14 annular fuel compacts. These compacts are manufactured with inner and outer diameters of 10 and 26 mm, respectively, and a height of 39 mm. UO<sub>2</sub> coated particles are embedded into graphite matrix to form the fuel compact.

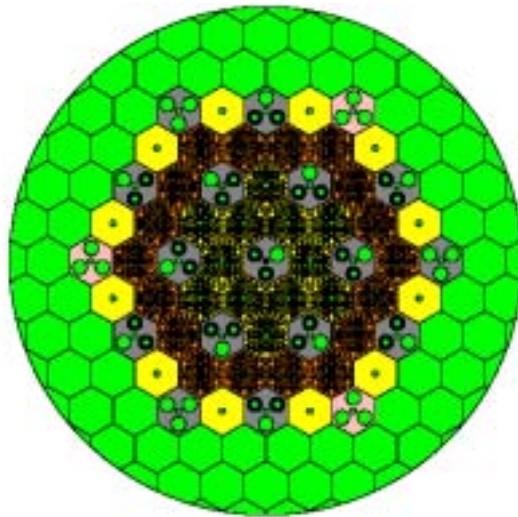
Each assembly contains only one type of fuel. However, there are 12 different fuel types with enrichments between 3 to 10 wt. % of U-235.

Control rods are made of B<sub>4</sub>C pellets surrounded by a 3.5 mm thick sleeve of Alloy 800H. The total height of control rods is 3094 mm. Inner and outer diameters are 65 mm and 113 mm, respectively. The detailed geometry and composition of all components are provided by a JAERI report [2-5].

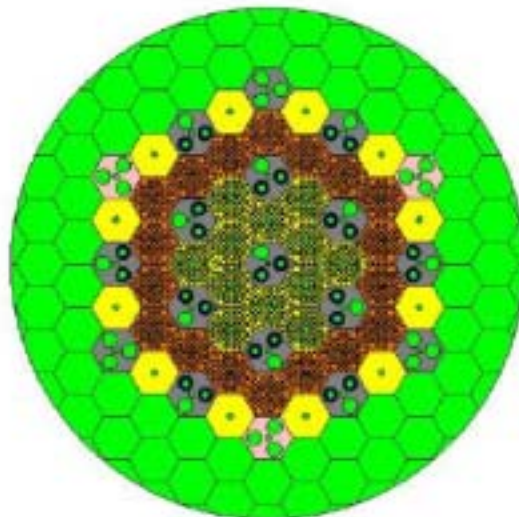
#### ***MCNP-4B model and simulations***

MCNP-4B is a very versatile computer code for particle transport simulation purposes [2-62]. It has extended features in geometrical modeling in three dimensional space. Every single geometrical detail is defined to generate full HTTR core model. The complexity of the reactor core structure makes this detailed modeling necessary. Horizontal cross sectional views of the HTTR core model with MCNP-4B are given in Figures 2.81 and 2.82.

The primitive element of the model is a coated particle. It is generated with five concentric spherical volumes. Then, these coated particles are uniformly distributed into a hexagonal prismatic lattice which is enclosed by an annular cylindrical graphite. Thus, a fuel compact model containing 13 000 CFP's is obtained. 14 fuel compacts are then stacked into a structure made of graphite. Hence, a fuel assembly is constructed by placing this structure in a hexagonal lattice. Two types of fuel assemblies with 31 and 33 fuel rods are generated in the model as mentioned earlier.



*Figure 2.81. Top cross sectional view of HHTR core in MCNP model.*



*Figure 2.82. Cross sectional view of HHTR core in MCNP model with fuel and control assemblies.*

Control rod guide columns are also modeled by leaving appropriate space for two control rod locations and a location for reserve shut down rod system.

MCNP provides a facility to visualize the geometry described in the model. Hence, the model is geometrically verified by taking views from different vertical and horizontal cross sections.

Criticality calculations are performed with 5 000 source particles generated in each cycle. 150 cycles are considered in each run. ENDF/B-VI cross section library with continuous energy is utilized throughout the calculations. Cross sections are evaluated at respective temperatures as necessary for graphite provided by thermal neutron libraries TMCCS.

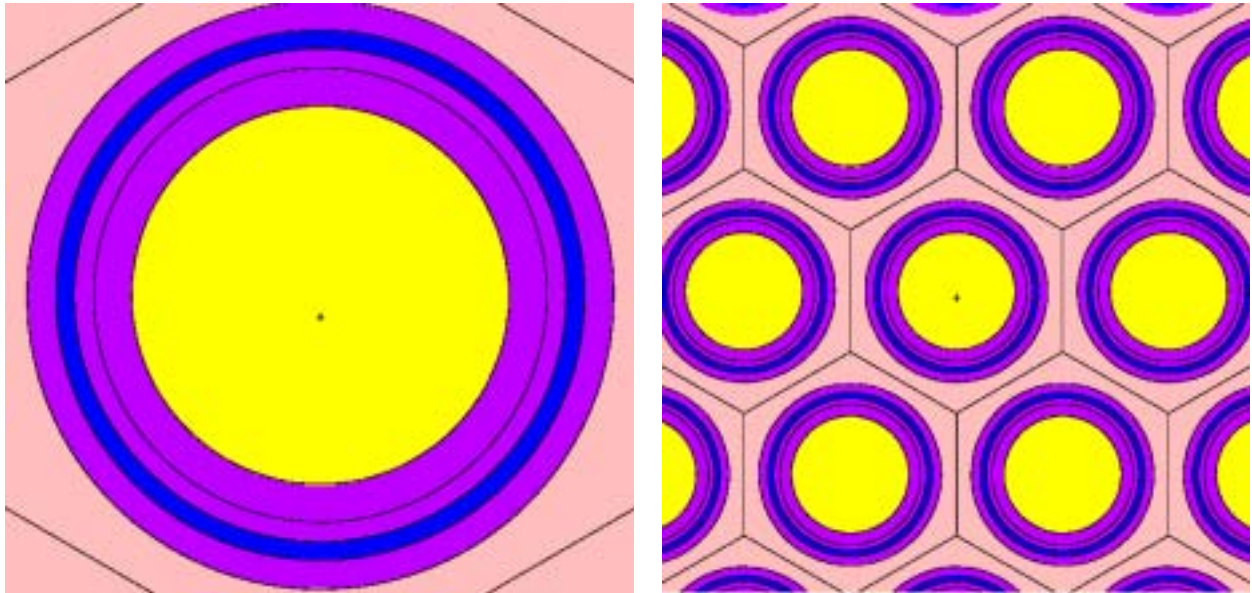


Figure 2.83. MCNP model for Coated Fuel Particles and their arrangement in fuel compact.

#### 2.2.7.2. Results of benchmark problems

In this work, all benchmark problems except HTTR-FC (Phase 2) are studied.

##### **Benchmark Problem 1: HTTR-FC (Phase 1)**

The number of fuel columns necessary to make the reactor critical is calculated in this problem. Results of criticality calculations with different number of fuel columns are shown in Table 2-80. The reactor is estimated to be critical with 15 fuel columns.

Table 2-80. Variation of effective multiplication factor with loaded fuel columns

Fuel Columns	k-eff	Error
12	0.98703	0.00102
13	0.99252	0.00102
14	0.99923	0,00103
<b>15</b>	<b>1.00503</b>	<b>0.00108</b>
16	1.01226	0.00103
17	1.02079	0.00105
18	1.03073	0.00097
19	1.04698	0.00091

##### **Benchmark Problem 2: HTTR-CR**

Control rod insertion depths for three different core loading configurations are evaluated for criticality condition. Control rod positions are evaluated from the top of the 8th layer (the top of the bottom reflector).

Table 2-81. Control rod depths for criticality and calculated k-eff

Fuel Columns	Control Rod Depth (cm)	k-eff	Error
18	285	0.99859	0.0016
24	210	1.00864	0.0013
30	164	0.99901	0.00152

**Benchmark Problem 3: HTTR-EX**

Excess reactivities are calculated for three different cases mentioned in Benchmark Problem 2 assuming 300K for moderator and fuel temperatures.

Table 2-82. Excess reactivities for three different fuel loading conditions

Fuel Columns	k-eff	Error	Excess Reactivity (%)
18	1.03073	0.00097	2.981
24	1.11969	0.00109	10.689
30	1.15641	0.00095	13.525

**Benchmark Problem 4: HTTR-SC**

There are two cases defined in this problem for evaluating scram reactivity. The first one is specified with all reflector control rods are inserted when the reactor is critical. The second one is evaluated with all control rods, reflector and core, are inserted again with the critical configuration for fully loaded core.

Table 2-83. Scram reactivities with two different cases

	Scram Reactivity (%)
All reflector CRs in	-7.75
All reflector & core CRs in	-37.96

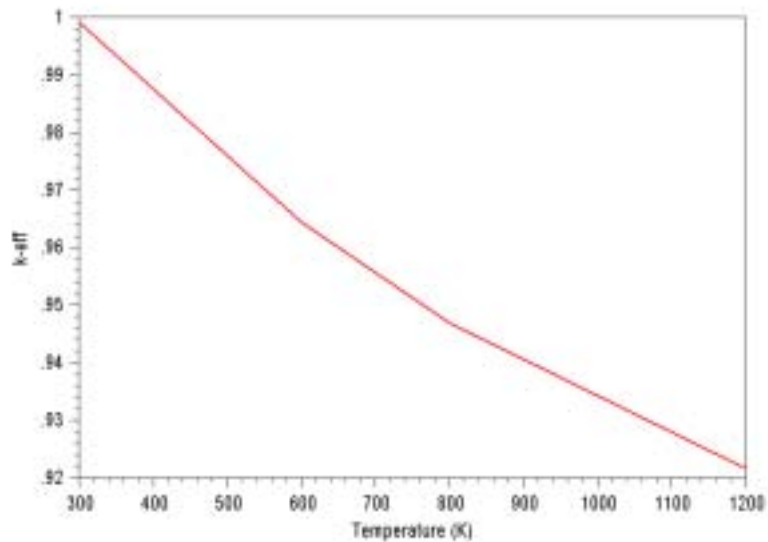
**Benchmark Problem 5: HTTR-TC**

Isothermal temperature coefficients are calculated for fully loaded core at the critical condition. Critical control position is also evaluated at a temperature of 480K to be 190 cm..

Table 2-84. Critical control rod position at 480K and corresponding k-eff

CR Position (cm)	k-eff	Error
190	0.99838	0.00220

As mentioned earlier cross sections for graphite at different temperatures are evaluated by means of thermal neutron libraries TMCCS. There are only 4 sets of data at temperatures of 300K, 600K, 800K, and 1200K available in our institution. Therefore, k-eff evaluations are performed for these temperatures and the results are shown in Figure 2.84. As it can be seen from the figure, the slope of the curve change with temperature. Therefore, it is rather difficult to make an accurate prediction of isothermal temperature coefficient. However, it is estimated for the first two datum points to be  $1.20 \cdot 10^{-4} \text{ 1/K}$  around the midpoint of 450K.



*Figure 2.84. Effective multiplication factor as a function of temperature.*

### 2.2.7.3 Conclusions

Results of HTTR start-up core physics benchmark problems are presented. Calculations are performed by MCNP-4B code and with continuous energy cross sections. Phase 2 of the first problem (HTTR-FC) is not taken into consideration. Therefore, all problems are performed with the original data. Geometrical modeling is done such a way that geometrical details are considered as much as possible. Isothermal reactivity calculation presented in this study is not very reliable due to the lack of relevant cross section data at rather close temperatures.

## 2.2.8. United States of America

### 2.2.8.1. Computational Methodology [2-64]

The Monte Carlo code MCNP4a was used because the geometry of the coated fuel particles, fuel compacts, burnable poisons, fuel element and reflector blocks, and the control rods and guide blocks can be modeled explicitly without any approximations.

MCNP4a uses nuclear data from the ENDF/IBV Version 0 cross-section library. The thermal neutron scattering matrix for the graphite is based on  $S(\alpha,\beta)$  of ENDF/BV. A core temperature of 300 K is assumed. MCNP4a uses track length, collision and analog estimators in determining the most probable value of  $k_{\text{eff}}$ . The number of histories per batch is 10,000. The number of batches is 30 with the first 5 batches being discarded for statistical reasons.

The core is modeled with hexagonal lattices consisting of control rod guide blocks, fuel blocks, and permanent reflector blocks. The fuel rods and the burnable poison rods are modeled using cylindrical body descriptions. The fuel block consists of several smaller hexagonal lattices containing one fuel rod or burnable poison rod. This is shown in Figure 2.85. In the geometry description, all hexagonal lattices were assigned to a universe number. The universe numbers were then input into an array to describe the location of the hexagonal blocks in the reactor core. The cross sectional model of the 30 column reactor core is shown in Figure 2.86. The outer boundary of the permanent reflector was modeled as a cylinder.

The fuel compacts are modeled with cylinders and the coated fuel particles (CFPS) are modeled with spherical body descriptions. In the control rod insertion heterogeneous excess reactivity calculations all five layers of the CFP are modeled. There are about 13,000 fuel particles defined for each fuel compact corresponding to a packing density of 30 percent with each fuel rod containing 14 fuel compacts. Therefore, for plotting simplicity the fuel particles are not shown in the cross sectional view of the fuel hexagonal lattice. The fuel particle representation is smeared in the homogeneous excess reactivity calculation.

The burnable poison rods are modeled as two different types. The geometry does not change but the material composition of the burnable poison changes. There are graphite disks that separate the upper and lower parts of the burnable poison rods. There are only two of the three burnable poison rod positions filled for each fuel hexagonal lattice. The third hole contains helium coolant. The configurations of the burnable poison rods change throughout the reactor core and are shown in Figure 2.86.

There are three pairs of control rods in the outer ring that are not used in the approach to criticality. All other control rod insertion depths change as one bank. The control rods in the outer ring can only be removed to 72.5 cm from the top of the active core because a sufficient negative reactivity addition rate must be available for reactor scrams. The control rods in the inner rings can be fully withdrawn. Therefore, in the excess reactivity model, the control rod holes contain coolant in the inner rings.

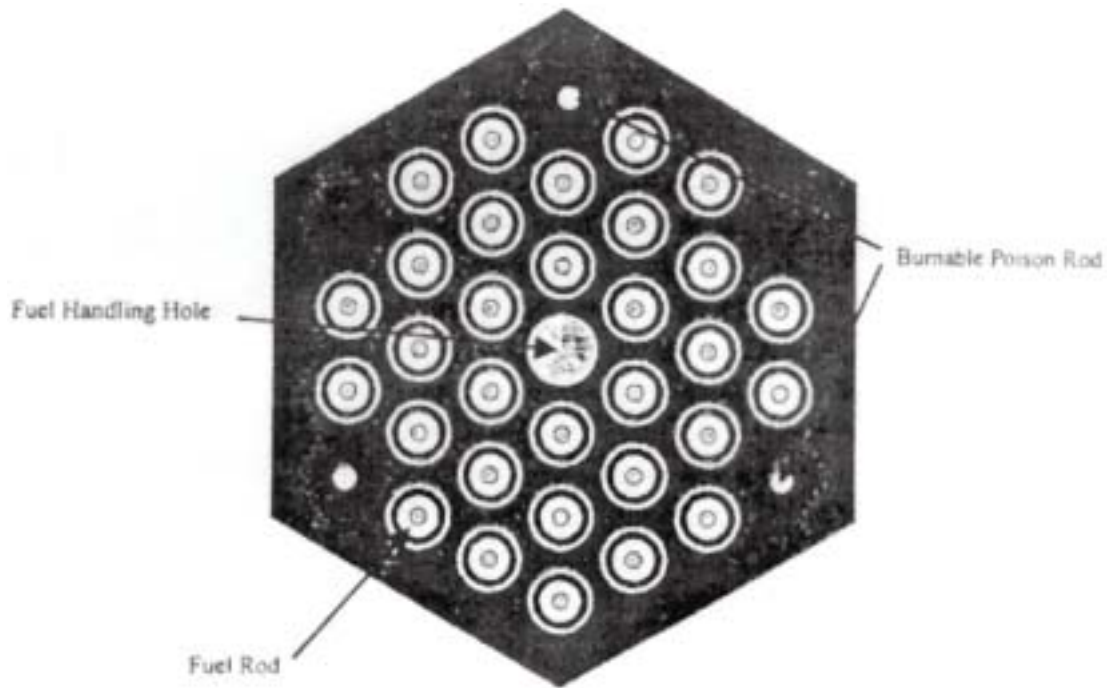


FIG. 2.85. Cross section of fuel assembly.

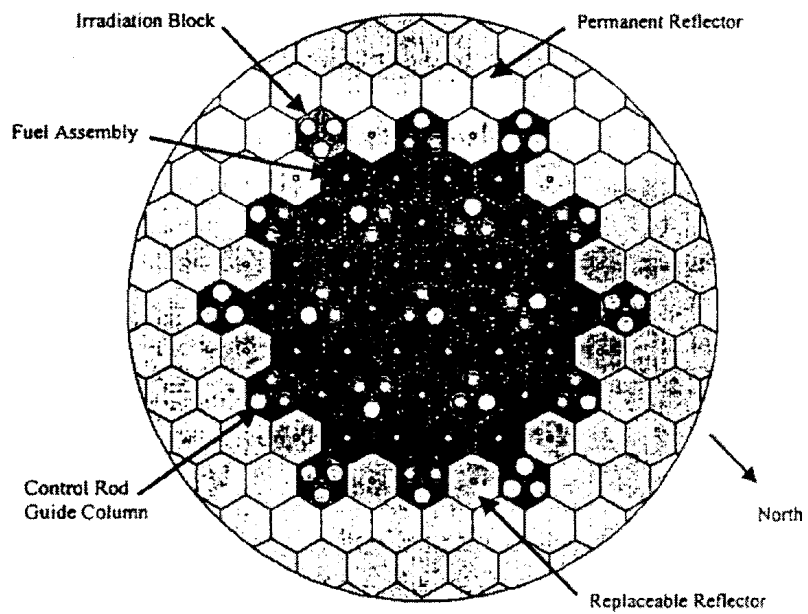


FIG. 2.86. Cross section of fully loaded core.

2.2.8.2. Results [2-65]

Monte Carlo methodology was used by the US in the initial HTTR criticality benchmark problem (MCNP-4a), and the calculations were limited to the case of the full core (30 elements loaded). The calculations made for the initial problem consisted of k-effective calculations vs. control rod (C/R) position in cm from the bottom of the active core, evaluated at two different uniform core temperatures (300K and 800K), and are shown in Figure 2.87 and detailed in Table 2.85.

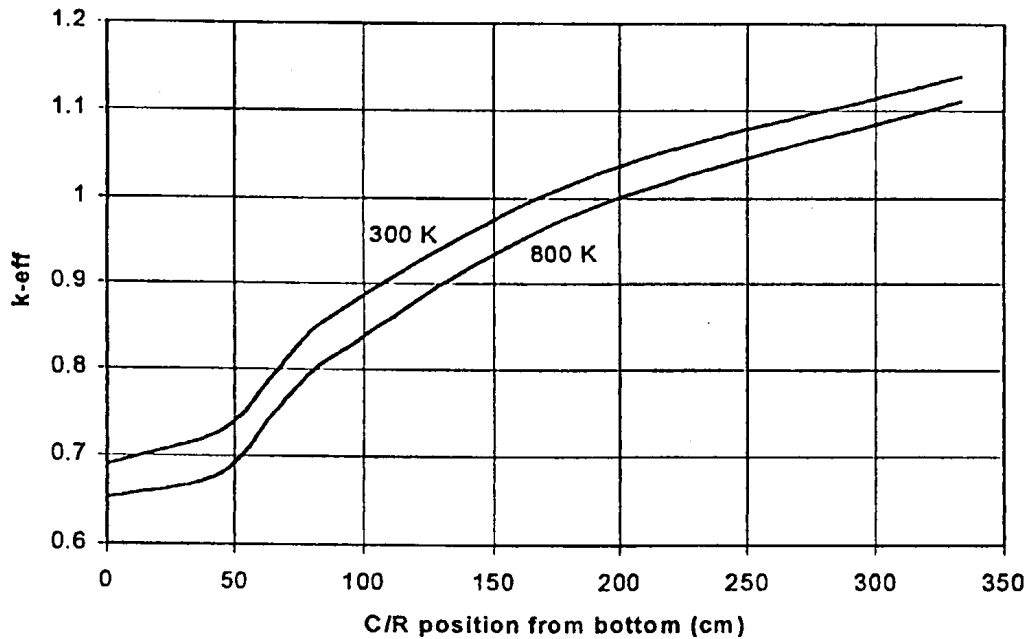


FIG. 2.87: HTTR K-eff at 300 and 800K.

Table 2-85. MCNP4a Summary of results

Core Temperature, K	Critical Rod Position (cm)*	K-eff, (Rods Full In)	K-eff, (Rods Full Out)	Homogeneous Model, K-eff, (Rods Full Out)
300	159	0.6899 (0.005)	1.1400 (0.004)	1.1336 (0.005)
800	194	0.6525 (0.006)	1.1118 (0.005)	1.10598 (0.004)

\* Distance withdrawn (from bottom of the active core)

As specified in JAERI's initial benchmark problem description, the outer ring of reflector C/Rs were modeled as fully withdrawn, and all other rods were moved in tandem, and their position with respect to the bottom of the active core denoted as "C/R position."

There were no additional MCNP calculations done for Part 2 of the benchmark; hence only simplified and approximate versions of the new benchmark problem calculations were done using the original MCNP results. Hence for the scram reactivity estimates, "scram from critical" results are for the case of all but the outer ring reflector rods inserting, rather than the configurations specified in the new benchmark. The calculations of temperature coefficient of reactivity also use just the original k-effective calculations at 300K and 800K rather than the more detailed calculations specified in the new benchmark.



## HTTR-SC

For the HTTR-SC benchmark problem, the scram reactivity from critical was calculated from the expression noted in the benchmark definition (difference between  $k$ -effective at critical [1.0] and  $k$ -effective with rods fully inserted divided by the product of the two  $k$ -effective values). The results are shown in Figure 2.88 as a function of core temperature (curve labeled "Scram from Critical."). Hence the scram reactivity for the core temperature specified in the benchmark (300K) is 0.45. The scram reactivity for the case where all rods drop would be somewhat greater. Note that at higher temperatures, the scram reactivity would be greater, as expected. Figure 2.88 shows a scram reactivity of 0.53 for a uniform core temperature of 800K. Also shown in Figure 2.88 is the reactivity worth of the rods (again excluding those rods in the outermost outer reflector ring) for full travel.

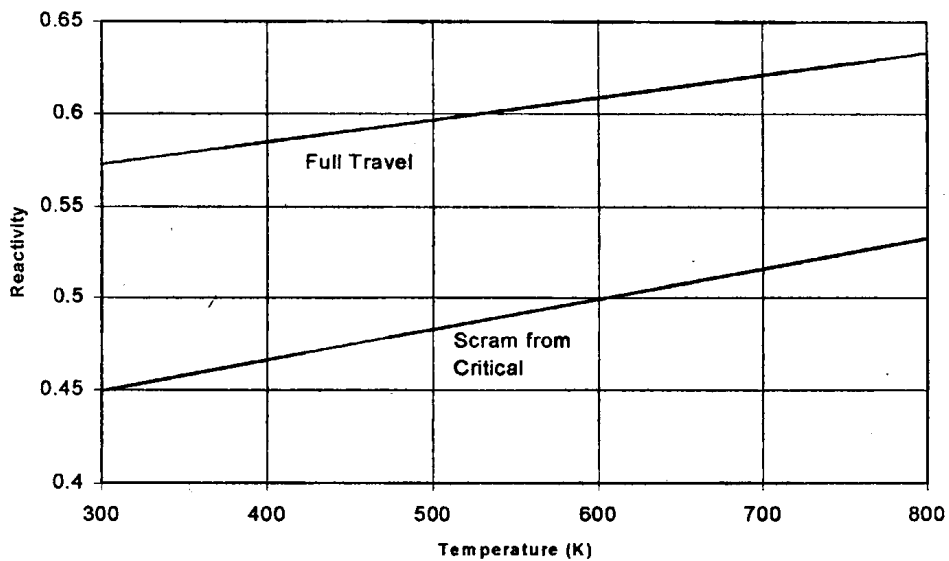


FIG. 2.88. Scram reactivity

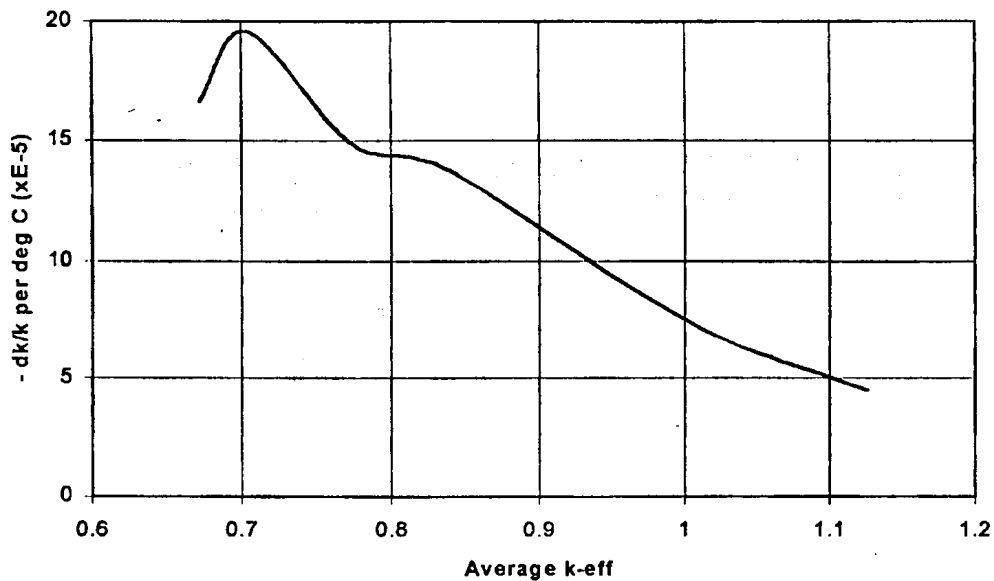


FIG. 2.89. Temperature coefficient of reactivity vs.  $k$ -eff

## *HTTR-TC*

The calculated temperature coefficient of reactivity (for uniform core temperatures) is plotted vs. average k-effective in Figure 2.89. The point at which the curve intersects  $k\text{-eff} = 1.0$  is taken as the value of interest. That value is  $-7.5\text{E-}5$  dk/k per degree C, evaluated at the average of the two temperatures at which k-eff was calculated (300K and 800K), i.e. 550K. It is interesting to note that Figure 2.89 shows that the coefficient becomes considerably more negative at smaller values of k-eff. One interpretation of this result would suggest that if the shutdown reactivity is calculated using the conventional estimated temperature coefficient of reactivity (at  $k\text{-eff} = 1.0$ ), a non-conservative (low) value of shutdown reactivity would result.

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## Chapter 3

### HIGH TEMPERATURE ENGINEERING TEST REACTOR THERMAL HYDRAULIC BENCHMARKS

#### 3.1. THERMAL HYDRAULIC BENCHMARK INFORMATION

This chapter includes a review of Member State investigation into the HTTR related thermal hydraulic benchmark problems of vessel cooling (VC) at 30 MW and the transient behaviour of loss of off-site electric power (LP) at 15 and 30 MW.

##### 3.1.1. Description of HTTR Systems Related to Vessel Cooling and Loss of Power

Included in this section are specific descriptions of the HTTR systems associated with the vessel cooling and loss of off-site electric power benchmark problems. A description of the general HTTR facility has been provided herein as Section 2.1.1.

###### 3.1.1.1 HTTR Vessel Cooling [3-1]

This benchmark problem includes predicting the heat removal capacity of the vessel cooling system (VCS) at 30 MW power operation and the temperature profile on the surface of the side panel. These predictions are to be compared with the actual measured heat removal capacity of the VCS and the associated temperature profile. Table 3-1 shows reference operation conditions at 30MW power operation.

Table 3-1. 30MW operating conditions of the HTTR

Coolant temperature at reactor inlet/outlet	395°C/850°C
Mass flow rate	12.4kg/s
Thermal power	30MW
Primary coolant pressure	4MPa
Water inlet temperature of VCS	25°C

###### *Vessel cooling system description*

The reactor and VCS are installed in the reactor cavity surrounded by the primary radiation shielding. The VCS consists of upper, side and lower panels. It has two completely independent sets of cooling systems to assure minimum heat removal to keep fuel and RPV integrity during depressurization accident. They are designated as “A” and “B” systems. Cooling panels are located on the inner surface of the reactor cavity as shown in Figure 3.1. Figures 3.2 and 3.3 represent the heat transfer schematic and expected cooling panel flow path, respectively. Table 3-2 includes VCS specifications.

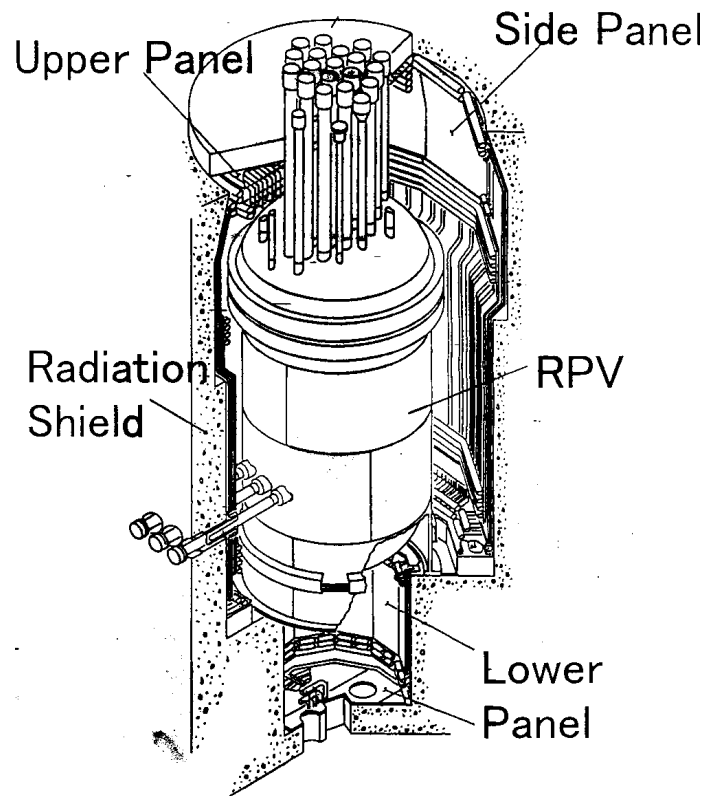


FIG. 3.1. HTTR cooling panel schematic.

Table 3-2. Major Specifications of the VCS

Number of set	2 set (A & B systems)
Minimum requirement for heat removal	0.3MW
Nominal water flow rate	
Total	86 ton/hour/each set
Lower panel	6.5 ton/hour/each set
Side panel	72 ton/hour/each set
Bottom panel	6.5 ton/hour/each set
Allowable temperature for components	90□

The upper panel is on the lower surface of the upper radiation shielding as shown in Figure 3.1. It consists of 96 cooling tubes in a steel casing. Each system has 48 cooling tubes. Water is distributed to six inlet headers from the inlet ring header and flows to the outlet ring header through cooling tubes and outlet headers. As an example, the flow diagram is depicted in Figure 3.2. Flow rate of the upper panel is 6.5 t/h and average flow rate per cooling tube is  $3.76 \times 10^{-2}$  kg/s as shown in Table 3-3. There are 31 penetrations for the standpipes in the upper panel.



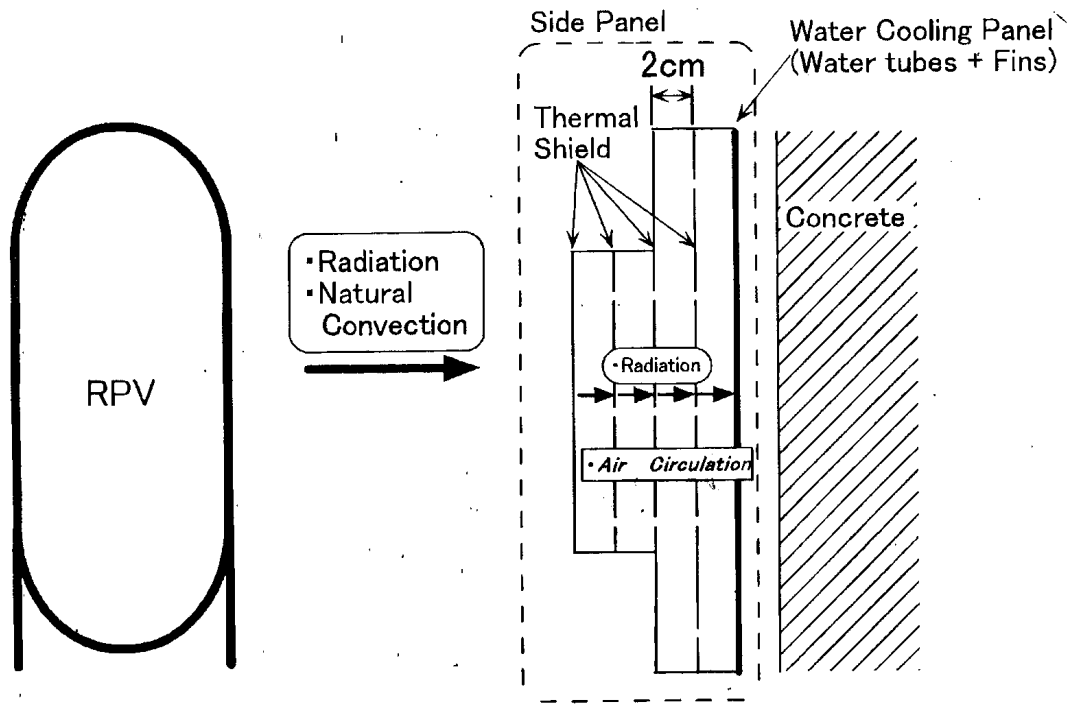


FIG. 3.2: Schematic of heat transfer from RPV to side panel

Table 3-3. Water flow rate for each independent cooling system

	Flow rate (Total)	Number of tubes	Outer diameter Thickness	Flow rate (Each tube)
Side panel	72 t/h	108	25.4mm 3.5 mm	1.85e-1 kg/s
Lower panel (side)	6.5 t/h	96	25.4mm 3.5 mm	3.76e-2 kg/s
Upper panel	6.5 t/h	48	15.9mm 3.2 mm	3.76e-2 kg/s
Lower panel (bottom)	6.5 t/h	8 or 3	25.4mm 3.2 mm	(See Table 4)

The side panel consists of twelve units. Water flows into each unit through a ring header, removes heat and goes out through another ring header. A reference unit of the side panel has a total of 18 cooling tubes, 9 for the "A" system and 9 for the "B" system. Flow rate for the side panel is 72 t/h and average flow rate for each tube is 1.85e-1kg/s. A steel plate connects adjacent two tubes. There are two thermal reflector plates between EL.17.35M and EL.30.99M in the side panel (Table 3-4). They are stainless steel and carbon steel plates. Two more carbon steel plates exist between EL.19.025M and EL.27.175M. There are nine horizontal gaps to absorb axial thermal expansion of the thermal reflector plates. These provide flow passages for air natural circulation in the side panel. The skirt panel on the RPV skirt removes heat from the RPV skirt directly to prevent concrete heat up under the RPV skirt.

The lower panel is on the bottom and side surface of a space under the RPV as shown in Figure 3.1. Water flows through the lower panel (bottom) and lower panel (side) in turn and returns to the water cooler. The bottom part is similar to the upper panel and its cooling tubes are installed between two plates. It has four headers and that are connected with 3 or 8 cooling tubes. In the “A” system, water comes from the water cooler flows through cooling tubes from header 1 to header 4 and goes out to the inlet ring header of the lower panel (side). Average flow rates are shown in Table 3-4. In the “B” system, water flows from the header 4 to header 1. The lower panel (side) is similar to the side panel and consists of twelve units. Each unit has 16 cooling tubes, 8 for the “A” system and 8 for the “B” system. Water is distributed to each unit from the inlet ring header. The water then goes upward through 4 cooling tubes, and returns to the outlet ring header through the other 4 cooling tubes. Average flow rates are shown in Table 3-4.

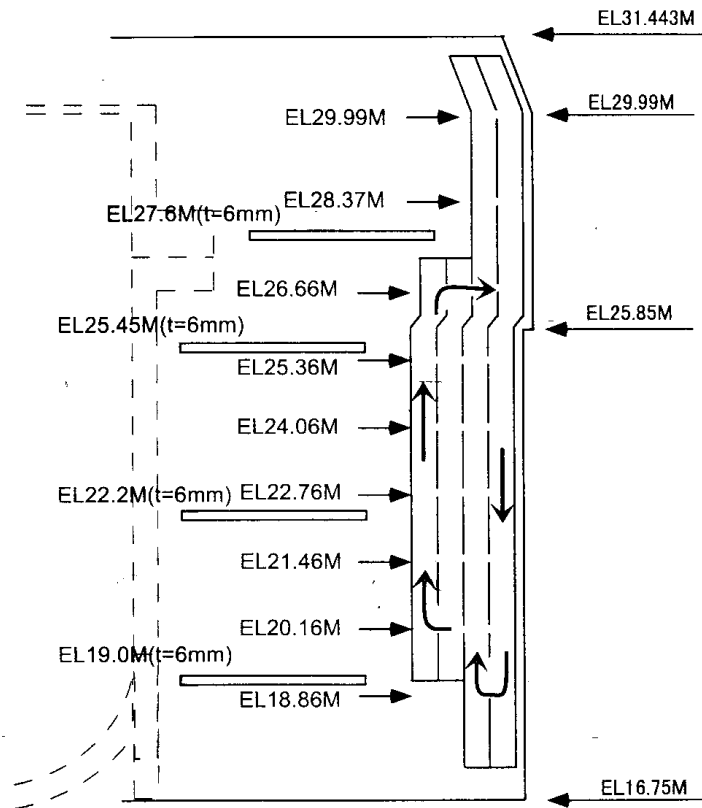


FIG. 3.3. Assumed flow passage in side panel

Table 3-4. Number of cooling tubes of lower cooling panel (bottom)

	Total	A system	B system	Average flow rate of one tube
Between Header 1 and Header 2	16	8	8	2.25e-1 kg/s
Between Header 2 and Header 3	6	3	3	6.02e-1 kg/s
Between Header 3 and Header 4	16	8	8	2.25e-1 kg/s

Table 3-5. Thermal radiation reflector plate

Panel	Numbers	Elevation
Side panel	3 (Carbon steel) 1 (Stainless steel)	From EL.19.025M to EL.27.175M
	1 (Carbon steel) 1 (Stainless steel)	From EL.17.35M to EL.19.025M From EL.27.175M to EL.30.99M
Lower panel (side)	1 (Carbon steel) 1 (Stainless steel)	From EL.13.40M to EL.16.71M

There are four ring plates of 6 mm thickness in reactor cavity. Their elevations are EL.27.6M, EL.25.45M, EL.22.2M and EL.19.0M. They separate the reactor cavity into five spaces. A ring plate is also in the space under the RPV. Its elevation is 15.15M.

The material properties, thermal conductivity and emissivity of components, the correlation of heat transfer coefficient of helium and power distribution in the core are as follows:

(1) 2.25Cr-1Mo steel (RPV, standpipe, core support gird, core support plate)

Thermal conductivity  $T < 282.22(C)$  32.0(kcal/mhC)  
 $282.22(C) < T < 676.67(C)$   $-7.861e-3T + 36.245$  (kcal/mhC)  
 $676.67(C) < T$   $-1.459e-2T + 44.652$  (kcal/mhC)  
 Emissivity 0.8

(2) Stainless steel (Casing of shielding block, thermal shield, thermal reflector)

Thermal conductivity  $0.0107T + 9.888$  (kcal/mhC)  
 Emissivity 0.48

(3) Carbon steel (Thermal reflector, ring plate)

Table 3-6. Thermal conductivity of Carbon Steel

Temperature (C)	Thermal conductivity (kcal/mhC)
0	44.6
100	43.6
200	41.4
300	42.8
400	36.0

Emissivity 0.8

(4) PGX blocks (Permanent reflector, hot plenum, lower plenum, bottom)

Table 3-7. Thermal conductivity of PGX graphite

Temperature (C)	Thermal Conductivity (kcal/mhC)	
	Radial	Axial
100	100.74	74.52
200	92.28	67.54
300	84.62	61.34
400	77.71	55.87
500	71.54	51.05
600	66.01	46.87
700	61.18	43.31
800	56.89	40.25
900	53.22	37.66
1000	50.06	35.5
1100	47.39	33.73
1200	45.14	32.33
1300	43.37	31.18
1400	41.95	30.28
1500	40.82	29.59
1600	40.04	29.02

Emissivity 0.8

(5) IG-110 (Core components)

Table 3-8. Thermal conductivity of IG-110 graphite

Temperature (C)	Thermal conductivity (kcal/mhC)
400	68.15
600	56.1
800	47.75
1000	42.15
1200	38.35
1400	35.75

Emissivity 0.8

(6) ASR-0RB (Carbon block)

Table 3-9. Thermal conductivity of Carbon block

Temperature (C)	Thermal Conductivity (kcal/mhC)
100	7.452
200	7.884
300	8.244
400	8.532
500	8.82
600	9.036
700	9.216
800	9.36
900	9.468
1000	9.54
1100	9.612

(7) B<sub>4</sub>C/C (Neutron absorber)

Thermal conductivity            16.2 (kcal/mhC)

(8) Correlation of heat transfer coefficient in the annual flow passage in the reactor

$$Nu = 0.023 Re^{0.8} Pr^{0.4} (Di/Do)^{-0.16} (Tw/Tb)^{-0.5}$$

Where;

- Nu    Nusselt number
- Re    Reynolds number
- Pr    Prandtl number
- Di    Inner diameter of annual flow passage
- Do    Outer diameter of annual flow passage
- Tw    Wall temperature
- Tb    Bulk temperature

(9) Power distribution in the core

Table 3-10. Power distribution

Axial fuel layer	Fuel zone number		
	1	2	3+4
1	0.031944	0.033058	0.096679
2	0.060489	0.062328	0.170170
3	0.060698	0.057048	0.158590
4	0.033081	0.034271	0.095860
5	0.021392	0.022164	0.062224

### 3.1.1.2 HTTR loss of off-site power event description [3-3]

The benchmark problems concerning loss of off-site electric power simulation are designated as HTTR-LP (15MW, 30MW). During normal HTTR operation (referred as parallel loaded operation), the intermediate heat exchanger, primary and secondary pressurized water coolers are operated simultaneously. In HTTR-LP (15MW), analytical simulation of transient behavior of the reactor and plant during the loss of off-site electric power from normal operation under 15MW thermal power is evaluated. Similar evaluation of transient behavior of the reactor and plant during the loss of off-site electric power from the normal operation under 30MW thermal power is conducted as HTTR-LP (30MW). In both simulation cases, estimation items are as follows; transition of (1) hot plenum block temperature, (2) reactor inlet coolant temperature, (3) reactor outlet coolant temperature, (4) primary coolant pressure, (5) reactor power, (6) heat removal of auxiliary heat exchanger. Estimation duration is for 10hr. from the beginning of the loss of off-site electric power.

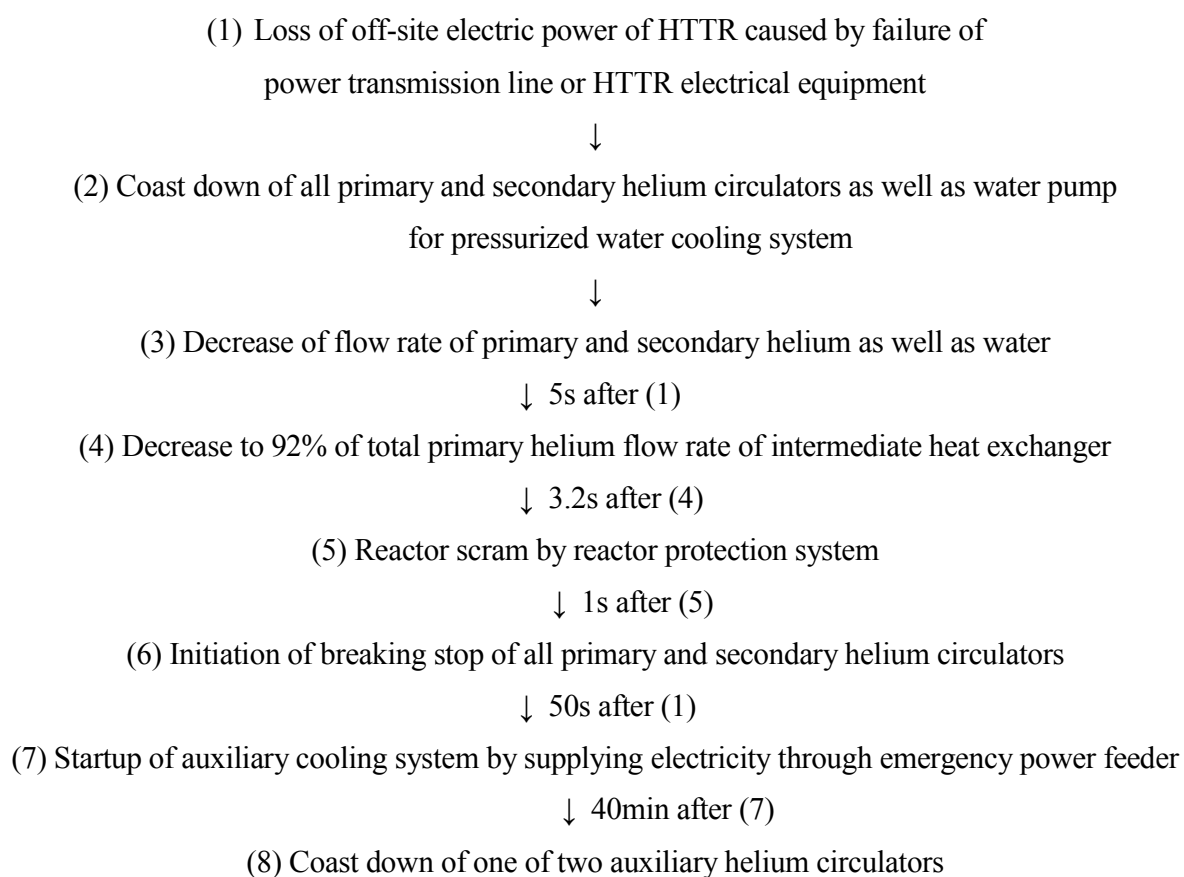


Fig. 3.4. Event scenario for the analyzed transient of loss of off-site electric power from 30MW.

The loss of off-site electric power is caused by failure of the power transmission line or the HTTR electrical equipment. All primary and secondary helium circulators as well as the water pump for the pressurized water cooling system coast down immediately after the loss of off-site electric power. Accordingly, flow rates of primary and secondary helium as well as water reduce.

Figure 3.4 depicts the event scenario for the analyzed transient during the loss of off-site electric power from 30MW. The flow rate of primary helium, which is deflected at a hot header and discharged around the heat transfer tubes of the intermediate heat exchanger, decreases to 92% of normal flow rate in 5s after the loss of off-site electric power. In 3.2s after 92% flow rate is reached, the reactor is scrammed by the reactor protection system. In 1s after the reactor scram, breaking stop of all the primary and secondary helium circulators is initiated. The auxiliary cooling system starts up in 50s after the loss of off-site electric power by electricity supplied through the emergency power feeder. The auxiliary cooling system mainly consists of an auxiliary heat exchanger, two auxiliary helium circulators, air cooler and two water pumps. Helium flow rate of the auxiliary cooling system reaches about 1.2kg/s in 20s after the startup of the auxiliary cooling system. One of the two auxiliary helium circulators is coasted down in 40min after the startup of the auxiliary cooling system to prevent the graphite blocks, composing the reactor core, from overcooling. Then helium flow rate of the auxiliary cooling system decreases to about 0.8kg/s.

### ***HTTR plant conditions at 15MW and 30MW***

Initial conditions at 15MW and 30MW power before startup of auxiliary cooling system are shown in Table 3-11. These include the reactor core (thermal power, reactor inlet and outlet temperatures, primary coolant pressure and flow rate as well as hot plenum block temperature). Tables 3-12, 3-13, 3-14 and 3-15 show the major process conditions (temperature, pressure and flow rate) of the intermediate heat exchanger, primary and secondary pressurized water coolers as well as the air cooler for the pressurized water cooling system, respectively. Helium and water flow rates of the auxiliary cooling system in standby are fixed to 0.036kg/s and 5.5kg/s, respectively, to prevent the auxiliary heat exchanger from the thermal shock. During the standby of the auxiliary cooling system, the two auxiliary helium circulators are not operated and one of the two water pumps for the auxiliary cooling system is driven. Tables 3-16 and 3-17 show the major process conditions (temperature, pressure and flow rate) of the auxiliary heat exchanger and air cooler for the auxiliary cooling system, respectively.

Table 3-11. Major process conditions of reactor core

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Thermal power	15MW	30MW
Reactor inlet coolant temperature	About 241°C	395°C
Reactor outlet coolant temperature	About 470°C	850°C
Primary coolant pressure	About 3MPa(abs)	4MPa(abs)
Primary coolant flow rate	12.4kg/s	12.4kg/s
Hot plenum block temperature	About 490°C	About 890°C

Table 3-12. Process conditions of intermediate heat exchanger

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Primary helium inlet temperature	About 468°C	850°C
Primary helium outlet temperature	About 238°C	395°C
Primary helium pressure	About 3MPa(abs)	4MPa(abs)
Primary helium flow rate	4.1kg/s	4.1kg/s
Secondary helium inlet temperature	About 154°C	About 241°C
Secondary helium outlet temperature	About 431°C	About 783°C
Secondary helium pressure	About 3.1MPa(abs)	4.1MPa(abs)
Secondary helium flow rate	3.6kg/s	3.6kg/s

Table 3-13. Process conditions of primary pressurized water cooler

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Helium inlet temperature	About 468°C	850°C
Helium outlet temperature	About 242°C	395°C
Helium pressure	About 3MPa(abs)	4MPa(abs)
Helium flow rate	8.3kg/s	8.3kg/s
Water inlet temperature	About 89°C	About 135°C
Water outlet temperature	About 110°C	About 175°C
Water pressure	About 2.6MPa(abs)	3.5MPa(abs)
Water flow rate	115kg/s	115kg/s

Table 3-14. Process conditions of secondary pressurized water cooler

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Helium inlet temperature	About 430°C	About 782°C
Helium outlet temperature	About 154°C	About 240°C
Helium pressure	About 3.1MPa(abs)	4.1MPa(abs)
Helium flow rate	3.6kg/s	3.6kg/s
Water inlet temperature	About 89°C	About 135°C
Water outlet temperature	About 110°C	About 175°C
Water pressure	About 2.5MPa(abs)	3.4MPa(abs)
Water flow rate	60kg/s	60kg/s



Table 3-15. Process conditions of air cooler for pressurized water cooling system

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Water inlet temperature	About 110°C	About 175°C
Water outlet temperature	About 58°C	About 80°C
Water pressure	About 2MPa(abs)	3MPa(abs)
Water flow rate	70kg/s	70kg/s
Air inlet temperature	About 33°C	About 33°C
Air outlet temperature	About 58°C	About 81°C
Air pressure	0.1MPa(abs)	0.1 MPa(abs)
Air flow rate	605kg/s	605kg/s

Table 3-16. Process conditions of auxiliary heat exchanger

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Helium inlet temperature	About 452°C	About 812°C
Helium outlet temperature	About 50°C	About 62°C
Helium pressure	About 3MPa(abs)	4MPa(abs)
Helium flow rate	0.036kg/s	0.036kg/s
Water inlet temperature	About 35°C	About 36°C
Water outlet temperature	About 39°C	About 45°C
Water pressure	About 2MPa(abs)	About 3MPa(abs)
Water flow rate	5.5kg/s	5.5kg/s

Table 3-17. Process conditions of air cooler for auxiliary cooling system

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Water inlet temperature	About 39°C	About 45°C
Water outlet temperature	About 35°C	About 36°C
Water pressure	About 1MPa(abs)	About 2MPa(abs)
Water flow rate	5.5kg/s	5.5kg/s
Air inlet temperature	About 33°C	About 33°C
Air outlet temperature	About 34°C	About 42°C
Air pressure	0.1MPa(abs)	0.1 MPa(abs)
Air flow rate	106.5kg/s	106.5kg/s

Conditions after startup of auxiliary cooling system include helium flow rate of the auxiliary cooling system reaches about 1.2kg/s in 20s after the startup of the auxiliary cooling system. Helium flow rate of the auxiliary cooling system decreases to about 0.8kg/s in 40min after the startup of the auxiliary cooling system. Water flow rate of the auxiliary cooling system is fixed to about 18.3kg/s all the time after the startup of the auxiliary cooling system.

### **3.1.2. HTTR Thermal Hydraulic Benchmark Problem Descriptions**

Two sets of thermal hydraulic benchmark problems associated with the HTTR are investigated in this section. These include:

- The prediction of the amount of heat removed by the Vessel Cooling System (VCS) at 30 MW power and the associated temperature profile on the surface of the side panel, and
- The analytical simulation on transient behavior of the reactor and plant during the loss of off-site electric power for HTTR operation at 15 and 30 MW.

The calculations formulated by the participating Member States will then be compared with actual experimental values achieved during testing on the HTTR.

## 3.2 THERMAL HYDRAULIC BENCHMARK ANALYSIS/RESULTS

### 3.2.1. Japan

#### 3.2.1.1. HTTR-VC Test and analytical results of heat removal capability [3-1]

Heat removal of the Vessel Cooling System (VCS) was calculated using the SSPHEAT code which was developed to analyze the temperature distribution in the in-core structure test section (T2) of HENDEL with complicated passages of helium flow [3-2]. The helium flow was simulated by a thermal-flow element, which is a uniaxial element in three-dimensional space with the ability to conduct heat and transmit fluid between its nodal points. The element has two parameters, temperature and pressure, at each nodal point. In the computer code, the model is solved by the finite element method (FEM).

Analytical conditions for full power operation are shown in Table 2-18. Reactor power, reactor inlet temperature and flow rate are 30MW, 395°C and 12.4kg/s, respectively. Heat removal of 0.3MW is minimum requirement for the VCS to keep fuel and RPV integrity at the accident conditions.

Table 3-18. Analytical conditions and result at full power

Reactor power	30 MW
Reactor inlet temperature	395 degree C
Coolant flow rate in reactor	12.4kg/s
Cooling water temperature of VCS	25 degree C
Reactor outlet temperature	850 degree C
VCS heat removal	0.77MW

The calculated VCS heat removal at full power of 30MW is 0.77MW. The analysis shows the VCS satisfies this requirement. The calculated VCS heat removal is larger than the maximum design value of 0.6MW. It is because air natural circulation occurs in the side panel and the air transfers heat to cooling tubes through thermal reflector plates which are provided to reduce radiant heat from the reactor.

Figure 3.5 shows the temperature profile for the HTTR-VC analysis and actual measurements at 9MW power. In a comparison of test and analytical results at 9 MW operation of HTTR the VCS heat removal at steady state was ~ 0.22MW. Table 3-19 shows the test and analytical conditions. Table 3-20 and Figure 3.5 show the comparison of test and analytical results at 9MW power.

Table 3-19. Test and analytical condition at 9MW operation

Reactor power	9 MW
Reactor inlet temperature	175 degree C
Coolant flow rate in reactor	12.4kg/s
Cooling water temperature of VCS	28.5 degree C
Reactor outlet temperature	312 degree C

Calculated heat removal with no air circulation in the side panel was 0.13MW and calculated temperatures on the side panel do not agree with the measured temperatures. Calculated heat removal increases to 0.2MW when taking into consideration the air circulation in the side panel and temperature distributions agree well with the measured temperatures as shown in Figure 3.5.

Table 3-20. Comparison of test and analytical results at 9MW operation

Test	0.22 MW
Calculation (No air circulation in side panel)	0.13 MW
Calculation (Air circulation in side panel)	0.20 MW

Table 3-21 shows the test condition and result at 30MW operation. Reactor inlet and outlet temperatures and coolant flow rate are 392°C, 844°C and 12.3kg/s, respectively. The VCS heat removal reached about 0.81MW which compares favorably with the analytical estimate of 0.77MW. Figure 3.6 shows the temperature profile on the RPV and side panel. Test results showed that RPV temperatures at EL.19-27m are 340~360°C and about 20°C lower than the analytical results. Side panel temperatures also showed that test results are lower than analytical results.

Table 3-21. Test condition and result at 30MW operation

Reactor power	30MW
Reactor inlet temperature	392°C
Reactor outlet temperature	844°C
Coolant flow rate	12.3kg/s
Cooling water temperature of VCS	33.2°C
Test	0.81 MW

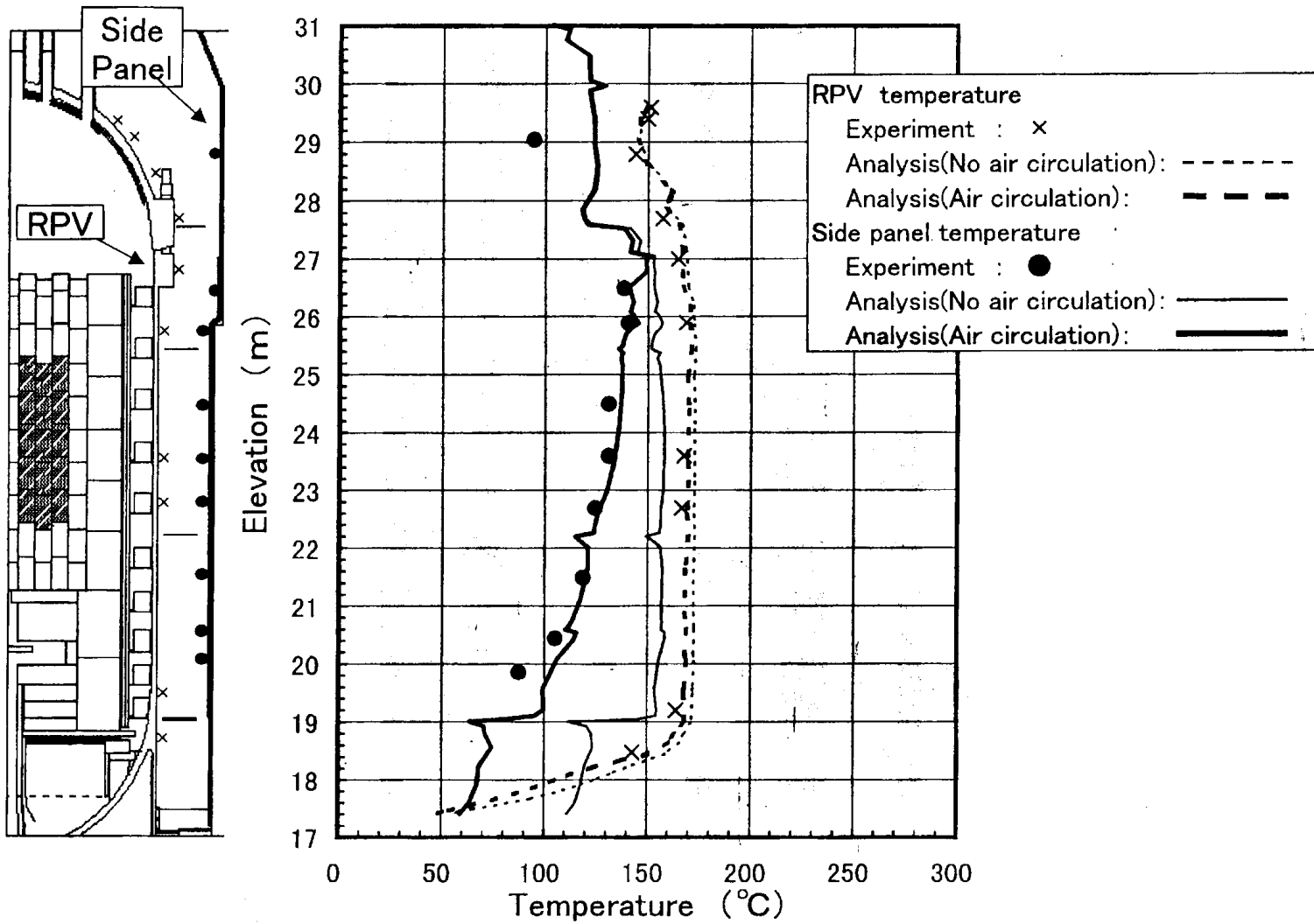


FIG 3.5. Temperature profile for the HTTR-VC analysis and actual measurements at 9MW power.

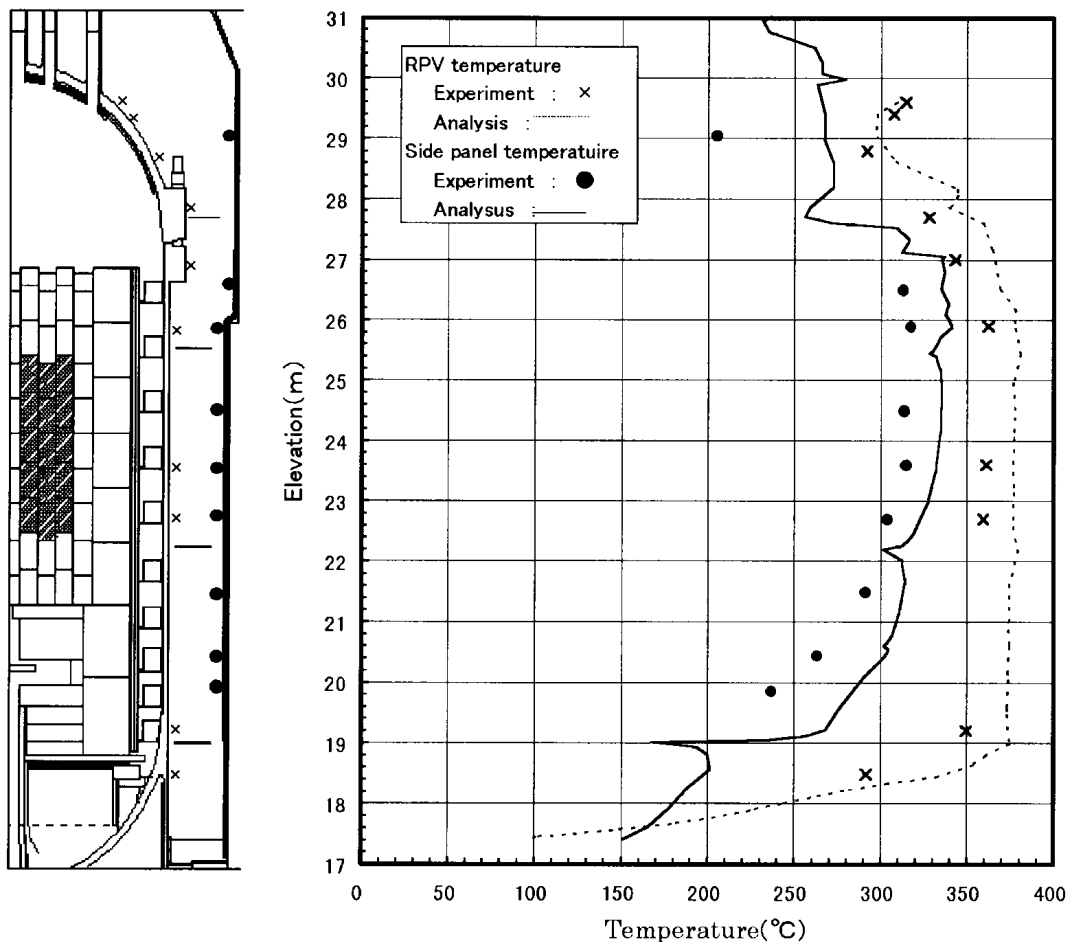


FIG. 3.6. Temperature Profiles on the RPV and Side Panel at 30 MW.

### 3.2.1.2. HTTR-LP Test Results

#### *Pre-estimation results by the “ACCORD” code [3-4]*

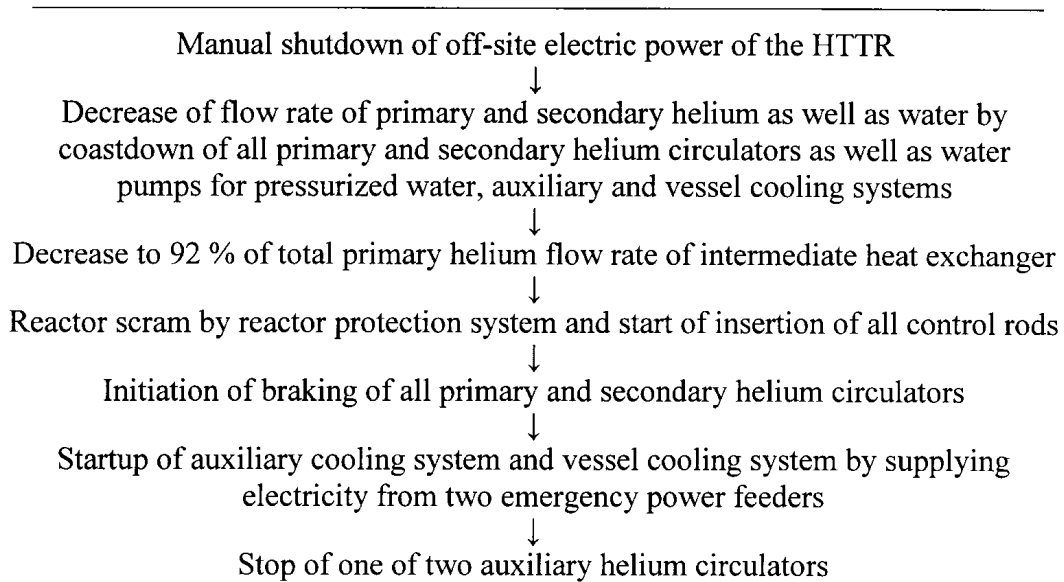
The pre-estimation results of the benchmark problems concerning the loss of off-site electric power simulation of the HTTR by the ‘ACCORD’ code include the transition of the hot plenum block temperature, reactor inlet and outlet coolant temperatures, primary coolant pressure, reactor power and heat removal of the auxiliary heat exchanger. The estimation duration is for 3600s from the beginning of the loss of off-site electric power.

The analyzed transient of loss of off-site power from 30MW has been described previously in Section 3.1.1.2. The maximum heat removal of the auxiliary heat exchanger during the operation of the two auxiliary helium circulators under the loss of off-site electric power from 15 and 30MW will be approximately 1.5 and 3MW, respectively.

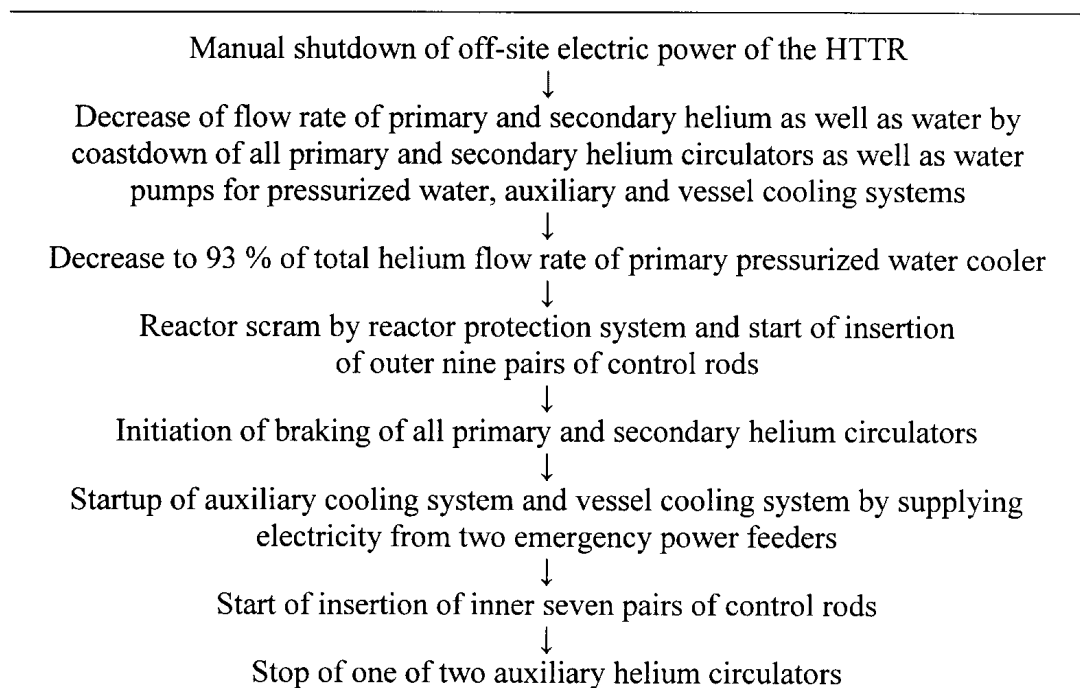
#### *Results of loss of off-site electric power tests from 15 and 30 MW operations [3-4]*

The transient experienced upon tests of the loss of off-site electric power from 15 and 30 MW operations are shown in Figures 3.7 and 3.8, respectively. All of the primary and secondary helium circulators as well as the water pumps for the pressurized water, and the auxiliary and the vessel cooling systems coasted down immediately after the loss of off-site electric power. In 1.6 s after the loss of off-site electric power, the reactor scammed by the

reactor protection system. At the test from 15 MW operation, the reactor scram occurred by decreasing flow rate of primary helium of the intermediate heat exchanger to the scram point of 3.8 kg/s (92 % of the total). At the test from 30 MW operation, the reactor scram happened by decreasing flow rate of helium of the primary pressurized water cooler to the scram point of 7.7 kg/s (93 % of the total).



*FIG. 3.7. Transient during loss of off-site electric power test from 15 MW operation.*



*FIG. 3.8. Transient during loss of off-site electric power test from 30 MW operation.*

At the test from 15 MW operation, all the control rods were inserted simultaneously into the reactor core by gravity within the design criterion of 12 s. At the test from 30 MW operation, the outer nine pairs of the control rods were inserted into the replaceable reflector region of the core within 12 s at first, and 40 min later the other inner seven pairs fell into the fuel region within 12 s.

In about 2 s after the reactor scram, braking of all the primary and secondary helium circulators was initiated by flowing electric current from their associated batteries. The braking stop times, when the rotations of the helium circulators became to zero, were 10 s or less after the open of the reactor scram breaker.

In about 50 s after the loss of off-site electric power, the auxiliary cooling system and the vessel cooling system started up by supplying electricity from two emergency power feeders. In 40 min after the startup of the auxiliary cooling system, one of the two auxiliary helium circulators stopped by design to reduce thermal stresses of the core graphite components.

Table 3-22 shows the thermal power, the reactor inlet and outlet coolant temperatures, the primary coolant pressure and flow rate as well as the hot plenum block temperature before the loss of off-site electric power tests from 15 and 30 MW operations. Temperatures of the hot plenum blocks are measured at representative three points. Because no significant temperature difference among the hot plenum blocks appeared, mean temperature among the hot plenum blocks was used.

Tables 3-23, 3-24, 3-25 and 3-26 show the temperature, pressure and flow rate of the intermediate heat exchanger (IHX), the primary and secondary pressurized water coolers as well as the air cooler for the pressurized water cooling system before the loss of off-site electric power tests from 15 and 30 MW operations, respectively.

Table 3-22. Conditions of reactor before LP tests

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Thermal power	15MW	30MW
Reactor inlet coolant temperature	242°C	392°C
Reactor outlet coolant temperature	468°C	828°C
Primary coolant pressure	2.9MPa(abs)	4.0MPa(abs)
Primary coolant flow rate	12.4kg/s	12.4kg/s
Hot plenum block temperature	391°C	677°C

Table 3-23. Conditions of IHX before LP tests

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Primary helium inlet temperature	468°C	828°C
Primary helium outlet temperature	235°C	383°C
Primary helium pressure	2.8MPa(abs)	4.0MPa(abs)
Primary helium flow rate	4.1kg/s	4.1kg/s
Secondary helium inlet temperature	158°C	248°C
Secondary helium outlet temperature	420°C	751°C
Secondary helium pressure	2.9MPa(abs)	4.0MPa(abs)
Secondary helium flow rate	3.6kg/s	3.6kg/s



Table 3-24. Conditions of primary pressurized water cooler before LP tests

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Helium inlet temperature	468°C	828°C
Helium outlet temperature	235°C	390°C
Helium pressure	2.8MPa(abs)	4.0MPa(abs)
Helium flow rate	8.2kg/s	8.2kg/s
Water inlet temperature	79°C	120°C
Water outlet temperature	100°C	160°C
Water pressure	2.3MPa(abs)	3.5MPa(abs)
Water flow rate	111kg/s	113kg/s

Table 3-25. Conditions of secondary pressurized water cooler before LP tests

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Helium inlet temperature	420°C	751°C
Helium outlet temperature	152°C	243°C
Helium pressure	2.9MPa(abs)	4.0MPa(abs)
Helium flow rate	3.6kg/s	3.6kg/s
Water inlet temperature	79°C	120°C
Water outlet temperature	100°C	158°C
Water pressure	2.3MPa(abs)	3.5MPa(abs)
Water flow rate	57kg/s	59kg/s

Table 3-26. Air cooler conditions for pressurized water cooling system before LP tests

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Water inlet temperature	100°C	160°C
Water outlet temperature	45°C	66°C
Water pressure	2.2MPa(abs)	3.3MPa(abs)
Water flow rate	65kg/s	76kg/s
Air inlet temperature	7°C	8°C
Air outlet temperature	62°C	88°C
Air pressure	0.1MPa(abs)	0.1MPa(abs)
Air flow rate	270kg/s	370kg/s

Tables 3-27 and 3-28 show the temperature, pressure and flow rate of the auxiliary heat exchanger and the air cooler for the auxiliary cooling system before the loss of off-site electric power tests from 15 and 30 MW operations, respectively. The auxiliary cooling system is in standby during the normal operation. During the standby, the two auxiliary helium circulators were not operated, however, a small amount of helium of 0.031 kg/s from the primary helium purification system was provided into the auxiliary cooling system to keep the auxiliary heat exchanger from thermal shock at the startup of the auxiliary cooling system. Furthermore, water pressure of the auxiliary cooling system was regulated through its controlling system, and one of the two water pumps for the auxiliary cooling system was driven. The water flow rate in standby was fixed at 5.6 kg/s.

Table 3-27. Conditions of auxiliary heat exchanger before LP tests

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Helium inlet temperature	426°C	754°C
Helium outlet temperature	74°C	97°C
Helium pressure	2.8MPa(abs)	4.0MPa(abs)
Helium flow rate	0.031kg/s	0.031kg/s
Water inlet temperature	59°C	63°C
Water outlet temperature	62°C	71°C
Water pressure	1.3MPa(abs)	2.3MPa(abs)
Water flow rate	5.6kg/s	5.6kg/s

Table 3-28. Conditions of air cooler for auxiliary cooling system before LP tests

Items	HTTR-LP (15MW)	HTTR-LP (30MW)
Water inlet temperature	62°C	71°C
Water outlet temperature	59°C	63°C
Water pressure	0.5MPa(abs)	1.5MPa(abs)
Water flow rate	5.6kg/s	5.7kg/s
Air inlet temperature	7°C	8°C
Air outlet temperature	8°C	9°C
Air pressure	0.1MPa(abs)	0.1MPa(abs)
Air flow rate	41kg/s	118kg/s

Figures 3.9 and 3.10 show transient behaviors of the rotation and the flow rate of the auxiliary helium circulators during the loss of off-site electric power tests from 15 and 30 MW operations, respectively. In about 1 s after the startup of the auxiliary cooling system, the rotations of the two auxiliary helium circulators reached about  $3000 \text{ min}^{-1}$ . This rotation was kept for about 2 s. After that, the rotation of the two auxiliary helium circulators rose up to about  $8500 \text{ min}^{-1}$  at the constant rate by the automatic frequency control. In about 14 s after the startup of the auxiliary cooling system, helium flow rates of the auxiliary helium circulators (A) and (B) reached about 0.73 kg/s and about 0.75 kg/s respectively at the test from 15 MW operation, while they achieved about 0.85 kg/s and about 0.89 kg/s at the test from 30 MW operation. In 40 min after the startup of the auxiliary cooling system, helium flow rates of the auxiliary helium circulator decreased to about 0.78 and about 0.87 kg/s at the tests from 15 and 30 MW operations, respectively. On the contrary, all the time after startup of the auxiliary cooling system, water flow rate was fixed at about 20.0 kg/s by operating the two water pumps.

Estimation items on the benchmark problems are as follows; transition of (1) hot plenum block temperature, (2) reactor inlet coolant temperature, (3) reactor outlet coolant temperature, (4) primary coolant pressure, (5) reactor power, (6) heat removal of auxiliary heat exchanger. Estimation duration is for 10 hr from the beginning of the loss of off-site electric power.

Figures 3.11 and 3.12 show transient behaviors of the temperatures of the hot plenum block, the reactor inlet and outlet coolant during the loss of off-site electric power tests from 15 and 30 MW operations, respectively. The hot plenum block temperature decreased continuously after the startup of the auxiliary cooling system. The reactor inlet coolant

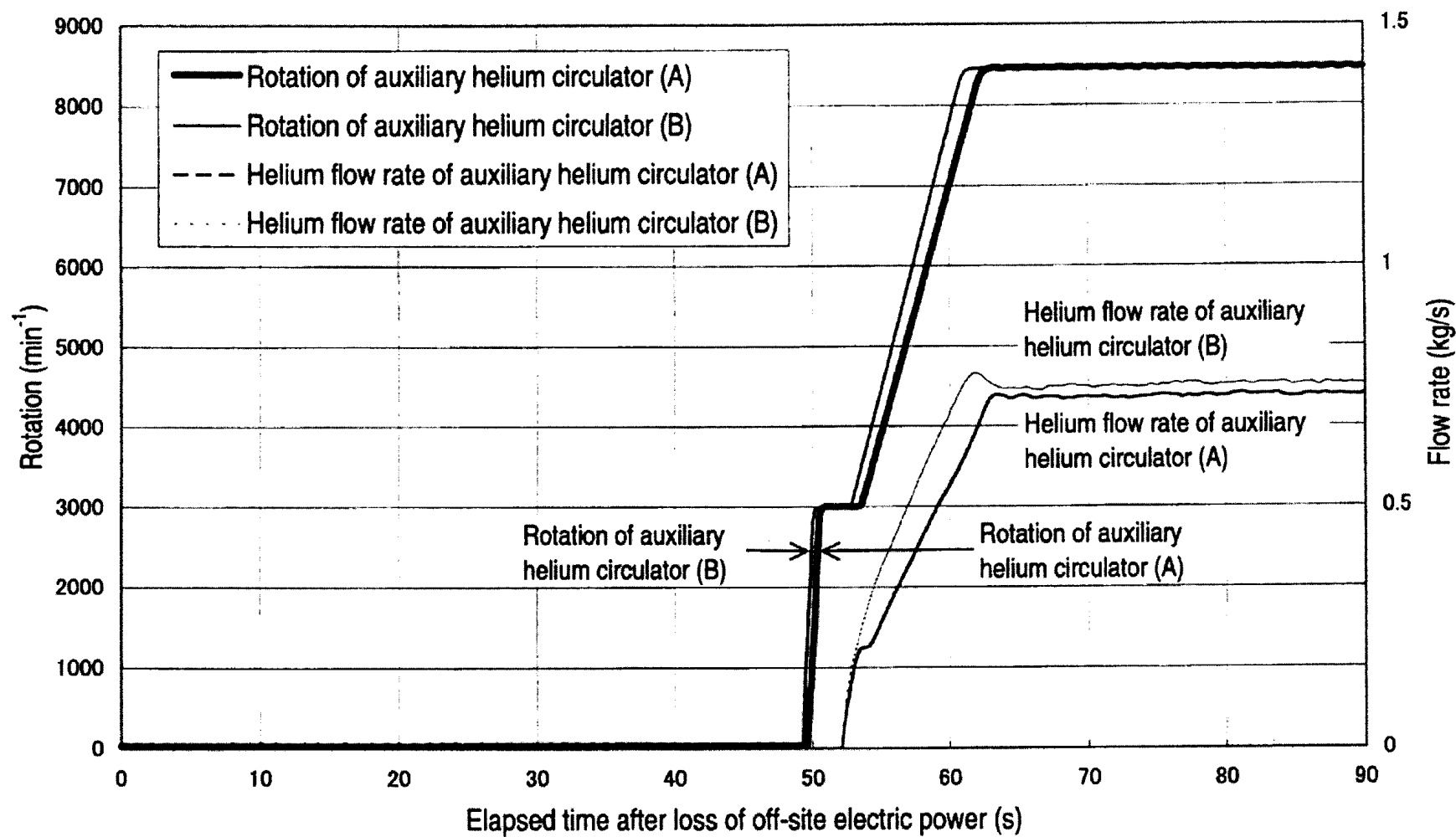


FIG.3.9. Transient behaviors of rotation and flow rate of auxiliary helium circulators during loss of off-site electric power test from 15 MW operation.

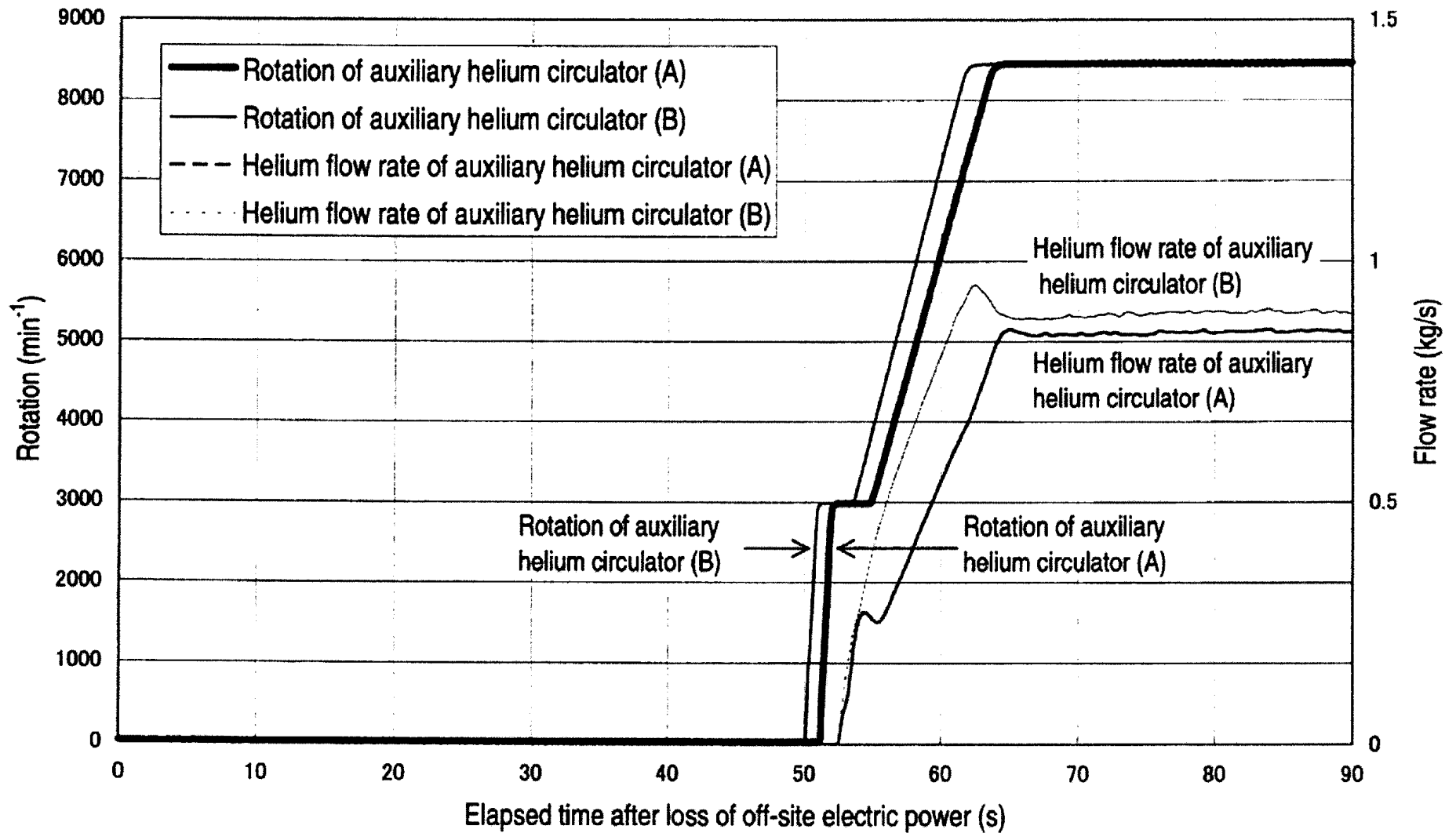


FIG. 3.10. Transient behaviors of rotation and flow rate of auxiliary helium circulators during loss of off-site electric power test from 30 MW operation.

temperature reduced rapidly by the startup of the auxiliary cooling system in about 50 s after the loss of off-site electric power. The reactor inlet coolant temperature decreased suddenly by the stop of one of the two auxiliary helium circulators in 40 min after the startup of the auxiliary cooling system, and since then it dropped gradually. The reactor outlet coolant temperature reduced continuously due to the insertion of the control rods and the startup of the auxiliary cooling system.

Figures 3.13 and 3.14 show transient behaviors of the primary coolant pressure and the reactor power during the loss of off-site electric power tests from 15 and 30 MW operations, respectively. The primary coolant pressure decreased gradually with reduction of the coolant temperature. The reactor power dropped drastically by the insertion of the control rods, and then the subcriticality of the reactor was maintained.

Figures 3.15a and 3.15b show transient behaviors of the heat removal of the auxiliary heat exchanger during the loss of off-site electric power tests from 15 and 30 MW operations, respectively. In about 50 s after the loss of off-site electric power the auxiliary cooling system started up, and subsequently about 10 min later flow rates and temperatures of helium and water became steady. Then the heat removal of the auxiliary heat exchanger was approximately 1.8 MW at the test from 15 MW, while it was approximately 4.0 MW at the test from 30 MW operation. The heat removal decreased suddenly by the stop of one of the two auxiliary helium circulators in 40 min after the startup of the auxiliary cooling system, and since then it dropped gradually.

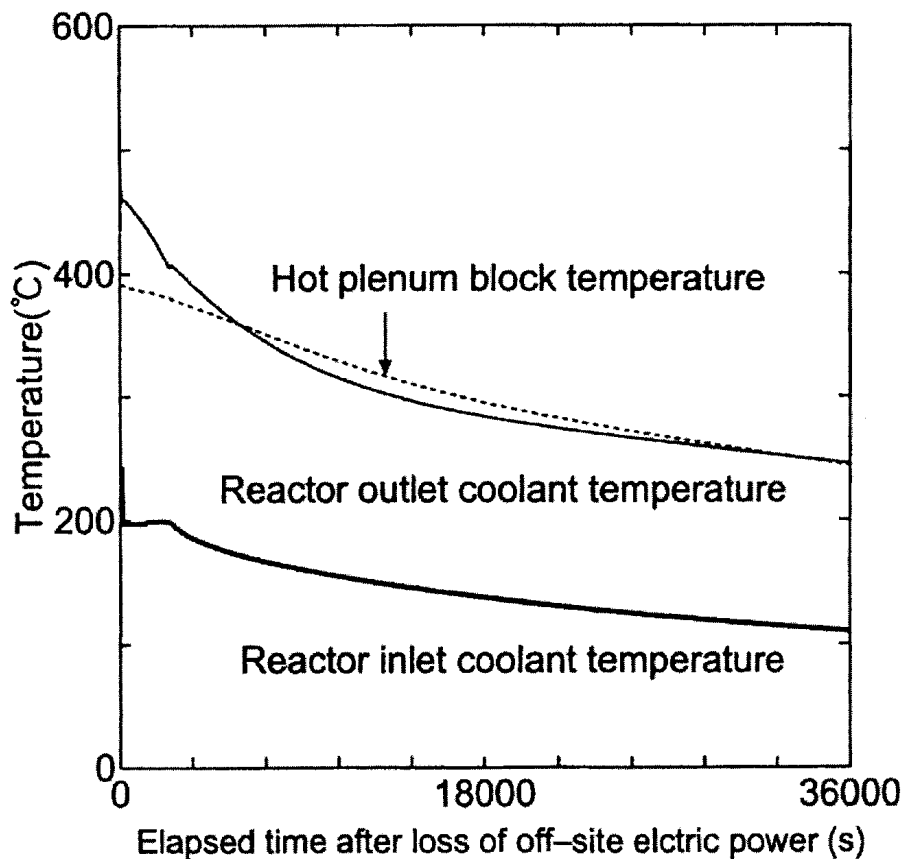


FIG. 3.11. Transient behaviors of temperatures of hot plenum block, reactor inlet and outlet coolant during loss of off-site electric power test from 15 MW operation.

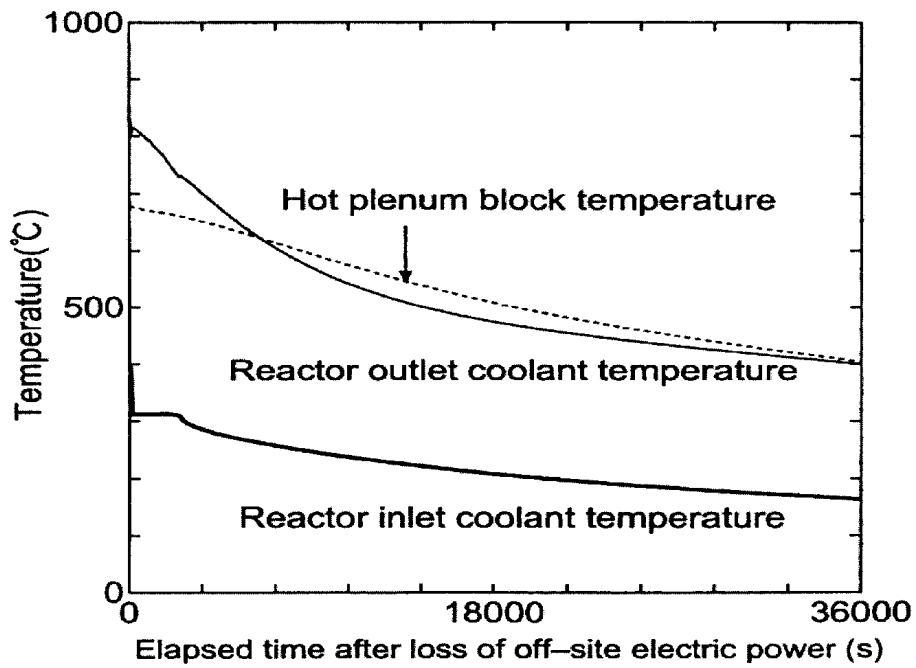


FIG. 3.12. Transient behaviors of temperatures of hot plenum block, reactor inlet and outlet coolant during loss of off-site electric power test from 30 MW operation.

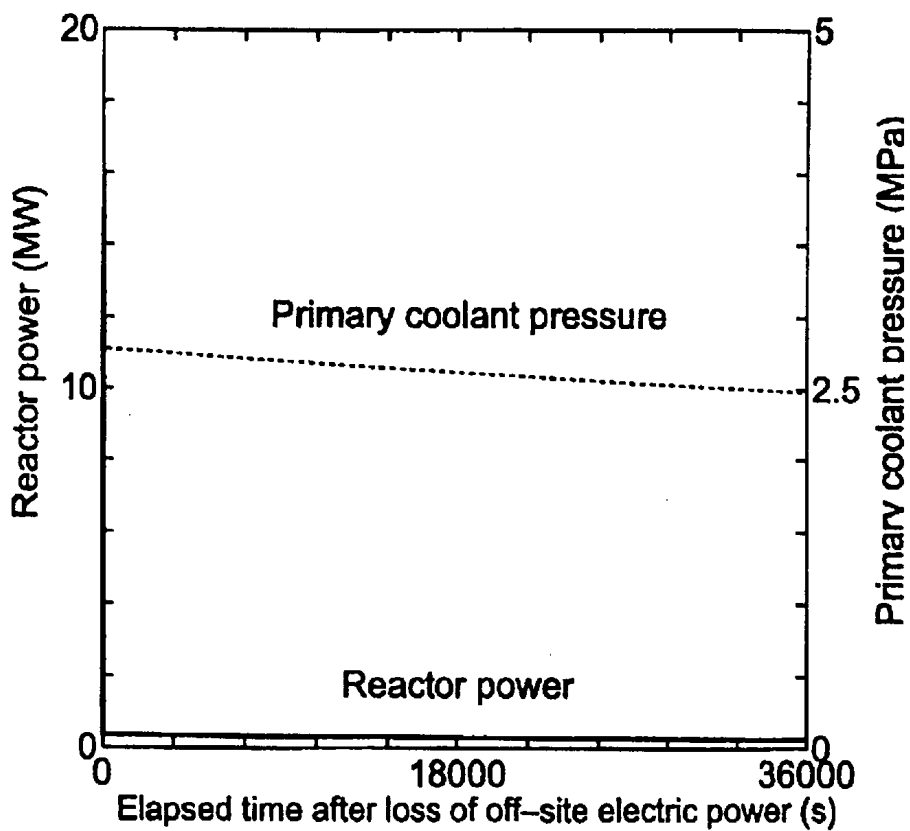


FIG. 3.13. Transient behaviors of primary coolant pressure and reactor power during loss of off-site electric power test from 15 MW operation.

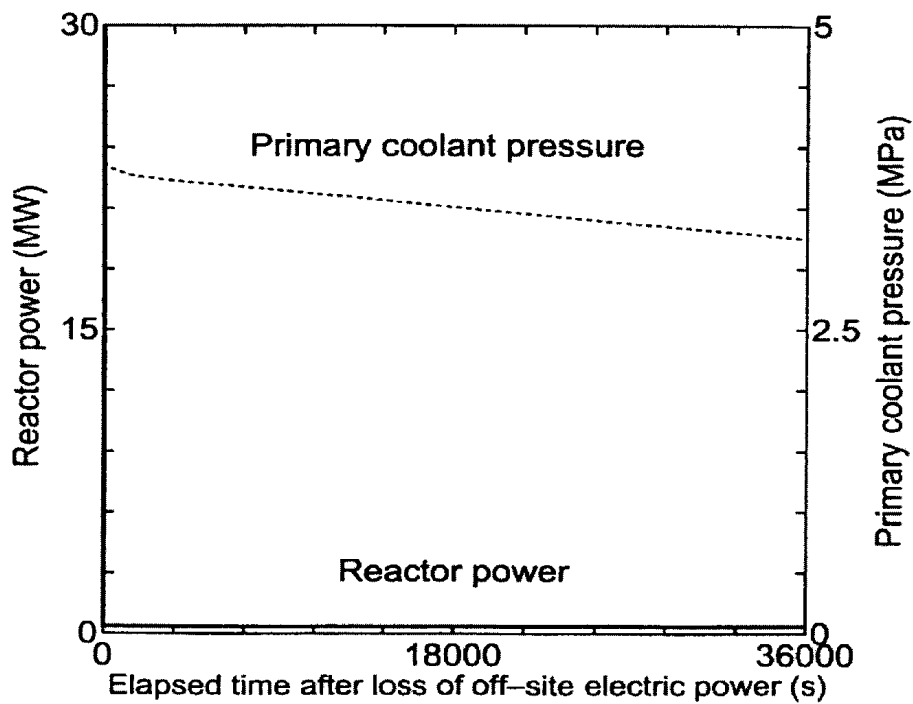


FIG. 3.14. Transient behaviors of primary coolant pressure and reactor power during loss of off-site electric power test from 30 MW operation.

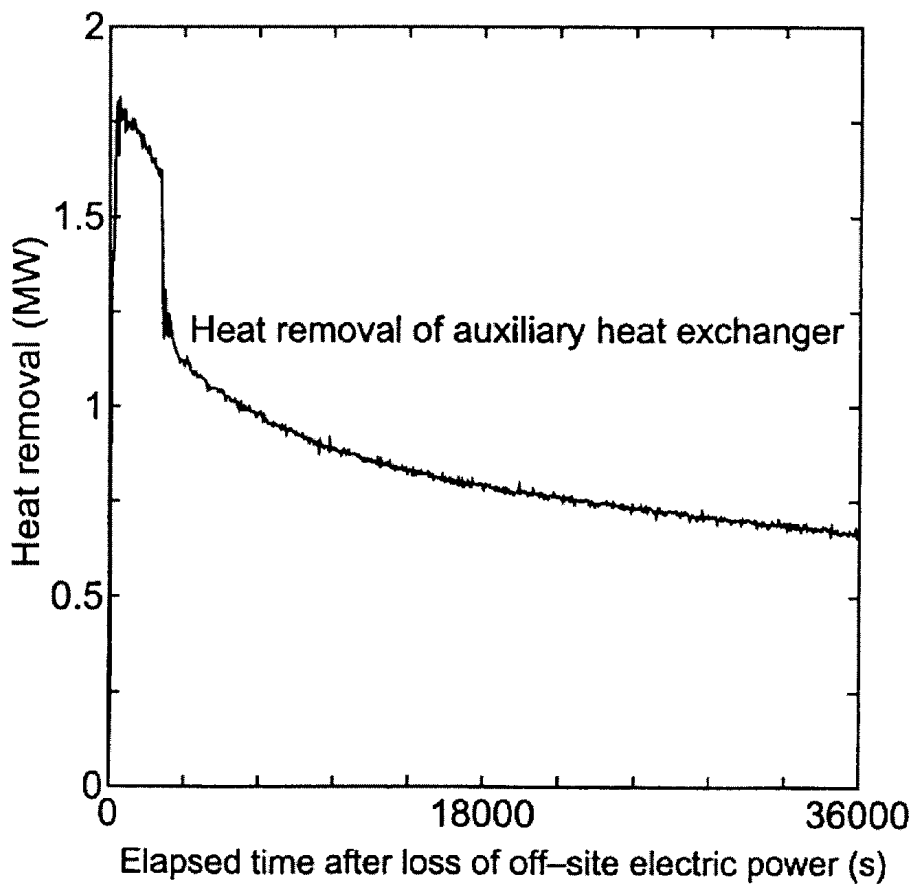


FIG. 3.15a. Transient behavior of heat removal of auxiliary heat exchanger during loss of off-site electric power test from 15 MW operation.

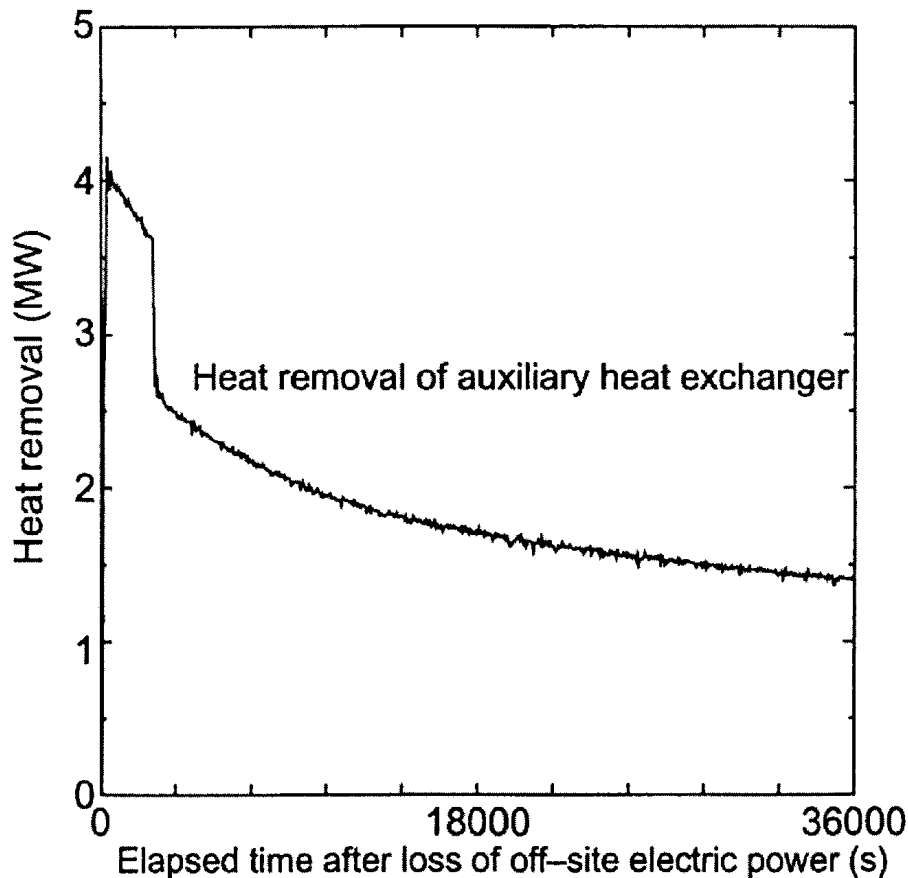


FIG. 3.15b. Transient behavior of heat removal of auxiliary heat exchanger during loss of off-site electric power test from 30 MW operation.

The following major results were determined from the loss of off-site electric power tests.

- (1) Because helium circulators and water pumps coasted down immediately after the loss of off-site electric power, flow rates of helium and water decreased to the scram points. In about 50 s after the loss of off-site electric power, the auxiliary cooling system and the vessel cooling system started up by supplying electricity from two emergency power feeders.
- (2) At the test from 15 MW operation, all the control rods were inserted simultaneously into the reactor core by gravity. At the test from 30 MW operation, the outer nine pairs of the control rods were inserted into the replaceable reflector region of the core at first, and 40 min later the other inner seven pairs fell into the fuel region. The reactor power dropped drastically by the insertion of the control rods, and then the subcriticality of the reactor was kept.
- (3) In about 10 min after the startup of the auxiliary cooling system, flow rates and temperatures of helium and water of the auxiliary heat exchanger became steady. Then the heat removal of the auxiliary heat exchanger was approximately 1.8 MW at the test from 15 MW operation, while it was approximately 4.0 MW at the test from 30 MW operation. In 40 min after the startup of the auxiliary cooling system, one of two auxiliary helium circulators stopped. Temperature of hot plenum block decreased continuously after the startup of the auxiliary cooling system.



### 3.2.2. Russian Federation

#### 3.2.2.1. HTTR Vessel Cooling (HTTR-VC) [3-6]

##### *Description of the codes*

SM-1, GTAS-M and DUPT codes were used for computation of power transferred from the reactor vessel to the reactor cavity cooling system and temperature distribution on the side panel. The SM-1 code is intended for computation of transient temperatures in structures with arbitrary geometry and based on solution of the heat conduction equation by heat balance method.

Thermal conductivity, gas convection and radiation are taken into account while determining effective conductivity factor in gas gaps. The code doesn't include solution of fluid motion and energy equations.

The GTAS-M code is intended to analyze transient temperature of HTGR cores and structures. The code allows the capability to obtain 3-D temperature distribution for a given geometry at designed "regime parameters - time" relations. The code is based on solution of the energy equation for solids in 3-D approximation. The three dimensions are restricted by steady geometry in any section along the height. The fluid motion equation is solved in stationary approximation. Cooling of a region to be calculated is allowed by several groups of flow rates which are united by collectors. Distribution of the coolant between channels of each group is performed on a condition of equal pressure difference between high and low pressure collectors. The code doesn't allow calculation of connective coolant flow in 2-D and 3-D approximations.

DUPT code is intended for computation of coolant connective flow in 2-D geometry. The code solves Navier-Stokes equations in Buossinesq approximation.

More detailed description of the codes is presented in [3-5].

##### *Methodology*

A simplified methodology was developed for the benchmark taking into account complex conditions for heat transfer between the reactor vessel and the reactor cavity cooling system and lack of a code to solve the problem as a whole. The space between the reactor vessel and the cavity is divided on three sections:

- upper (EL.27.22 m and EL.31/44.3 m);
- side (EL.19.65 m and EL.27.22 m);
- lower (EL.16.75 m and EL.19.65 m).

The schematic diagram for the benchmark problem is shown on Figure 3.16.

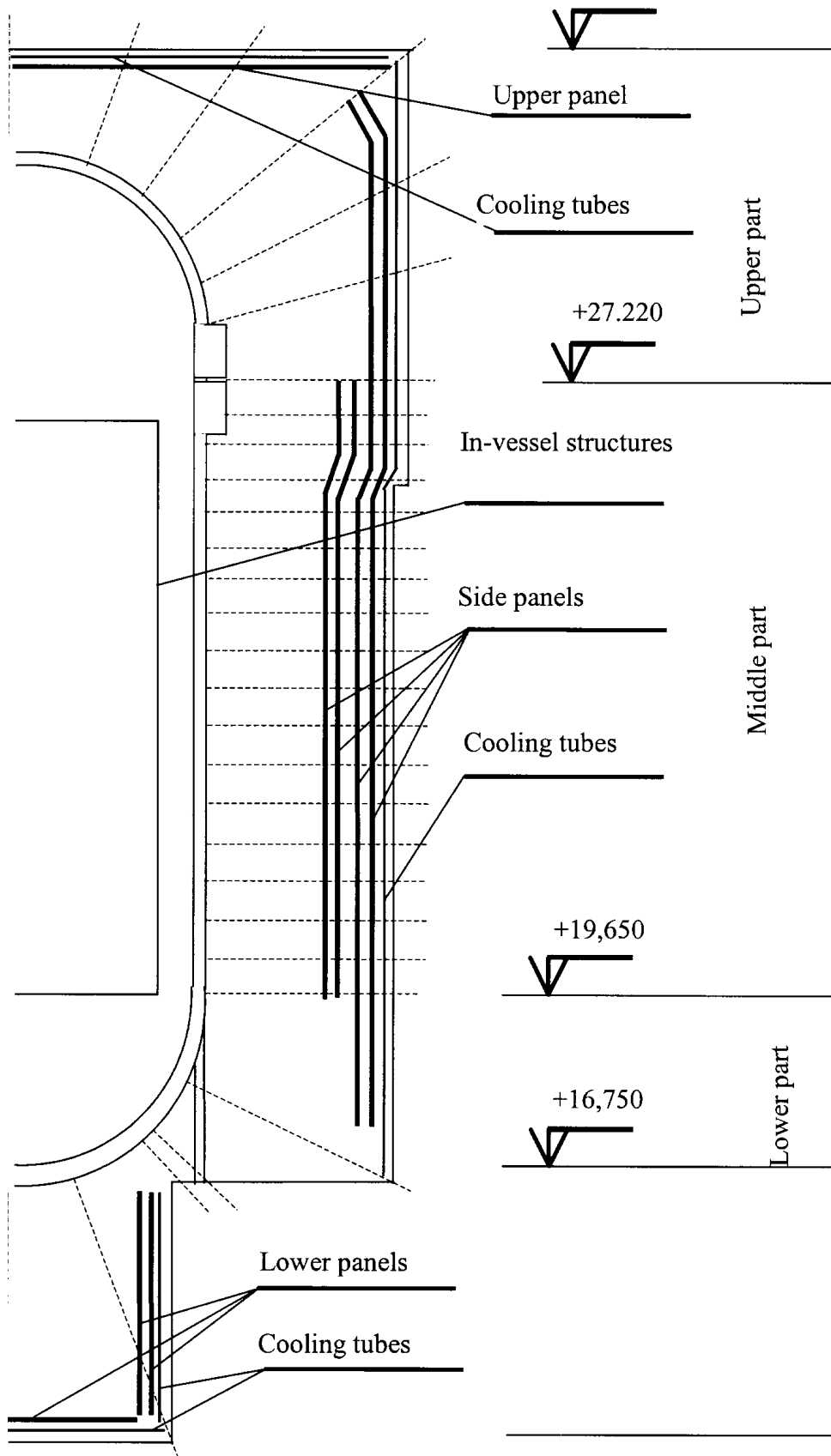


FIG. 3.16. Computing range diagram.

The GTAS-M code is used for estimation of heat removal from the reactor vessel only for its cylindrical section. The collector model of the code does not allow taking into consideration the real geometry of the upper and lower sections.

Heat transfer from the reactor vessel for the upper and lower sections is performed with using SM1 and DUPT codes. The temperature of the in-vessel structures and water inlet temperature of the reactor cavity cooling system was adopted as boundary conditions. Preliminary calculations to estimate the temperature of the in-vessel structures and sensitivity study showed that its value may be taken equal to 425 °C.

Use of SM1 code for the upper and lower sections enables to account for the head and bottom shape in calculations of heat transport by radiation. The heat transfer factor of the air in the cavity was derived from velocity profiles in the upper that were calculated by DUPT code. The air temperature was assumed to be equal to the air temperature at outlet of the middle section. Heat removal by the standpipes was taken for increase of surface of heat transport by convection. ISI plates were not taken into consideration in the analysis.

### **Results**

The results of analyses are presented in Figures 3.17 and 3.18 and in Tables 3-28 and 3-29. Figure 3.17 presents temperature distribution over the reactor vessel height and side panel surface of the vessel cooling system during the reactor operation at 9 MW power. Figure 3.18 depicts the temperature distribution during reactor operation at 30 MW power.

Table 3-28. Computed values of heat removal during the reactor operation at 9 MW power

Computing range	Amount of removed heat, kW
Upper part	3.19
Middle part	119
Lower part	11.3
Total	133.5

The analysis results demonstrated that during the reactor operation at 9 MW power 133.5 kW of heat is removed to the reactor vessel cooling system panels, 77 % of them in the middle part of the computing range. The portion of heat transferred by radiation amounts to 55 %.

Table 3-29. Computed values of heat removal during the reactor operation at 30 MW power

Computing range	Amount of removed heat, kW
Upper part	16.5
Middle part	419
Lower part	58.7
Total	494.1

In accordance with performed analyses, during the reactor operation at 30 MW power about 494 kW of heat is removed to the reactor vessel cooling system panels. 74.2 % of this is removed in the middle part of the computing range. This portion of heat being transferred by radiation amounts to 64 %.

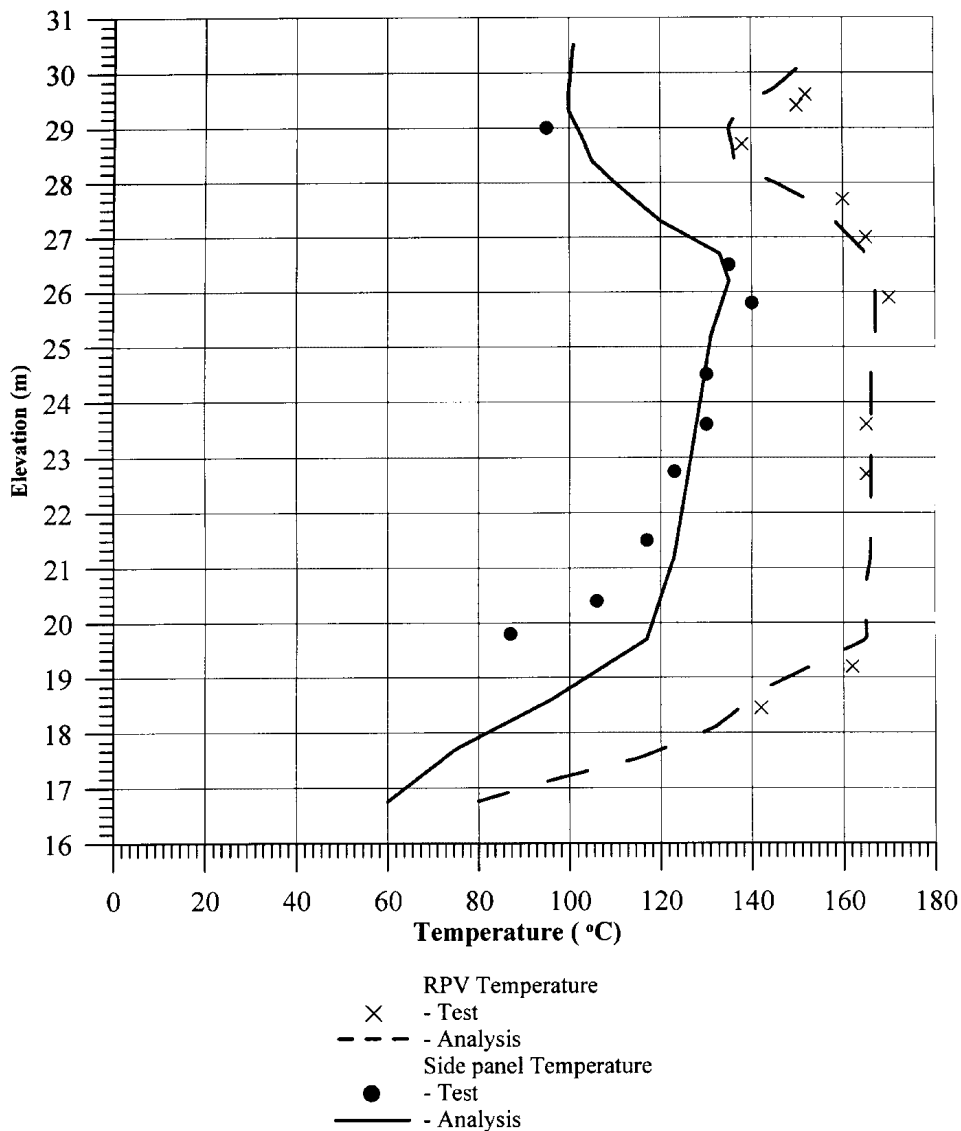


FIG. 3.17. Distribution of reactor vessel and side panel temperatures over height during the reactor operation at 9 MW power.

### 3.2.2.2. HTTR Loss of Electric Power (HTTR-LP) [3-10]

#### Description of the VGM code

The VGM-code is intended for calculating normal and emergency transients in nuclear power plants cooled by water or helium. Simulating regimes include:

- normal operation,
- urgent drop in power due to failure of equipment,
- emergency regimes (inadvertent withdrawal of control rods, loss of load, loss of power, etc.).

Circuit and system simulation includes main equipment having an effect on the transients. The mathematical simulation comprises neutronic process, thermal-hydraulic process in the primary, secondary circuits and emergency decay heat removal system, heat transfer in fuel elements, a model of reactivity control system.

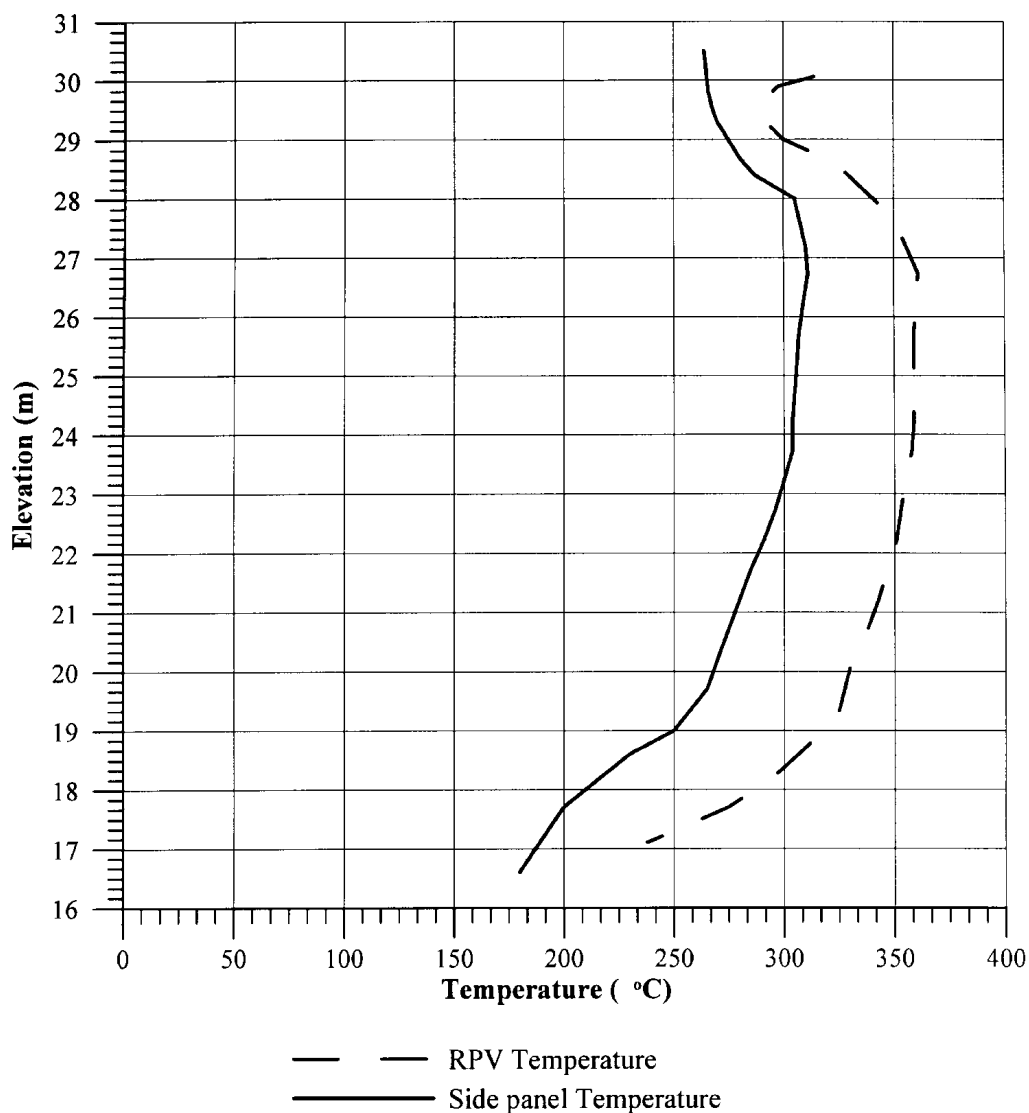


FIG. 3.18. Distribution of reactor vessel and side panel temperatures over height during the reactor operation at 30 MW power.

Reactor power is determined by the point kinetics method in approximation of six groups of delayed neutron sources. Reactivity dependence on coolant temperature and density, fuel and moderator temperatures is taken into account at calculation of feedback reactivity effects. The feedback effects are described taking into consideration their non-linearity and non-uniformity of multiplying characteristics in the core space.

The thermal-hydraulic process is simulated in the frame of 1-D model with account of natural convection of the coolant and possibility to change a direction of the circulation. Mixing of flow rates with different temperatures is assumed to be ideal. The core thermal model allows computation of coolant and moderator temperatures in axial and radial directions versus core inlet temperature, flow rate distribution and power.

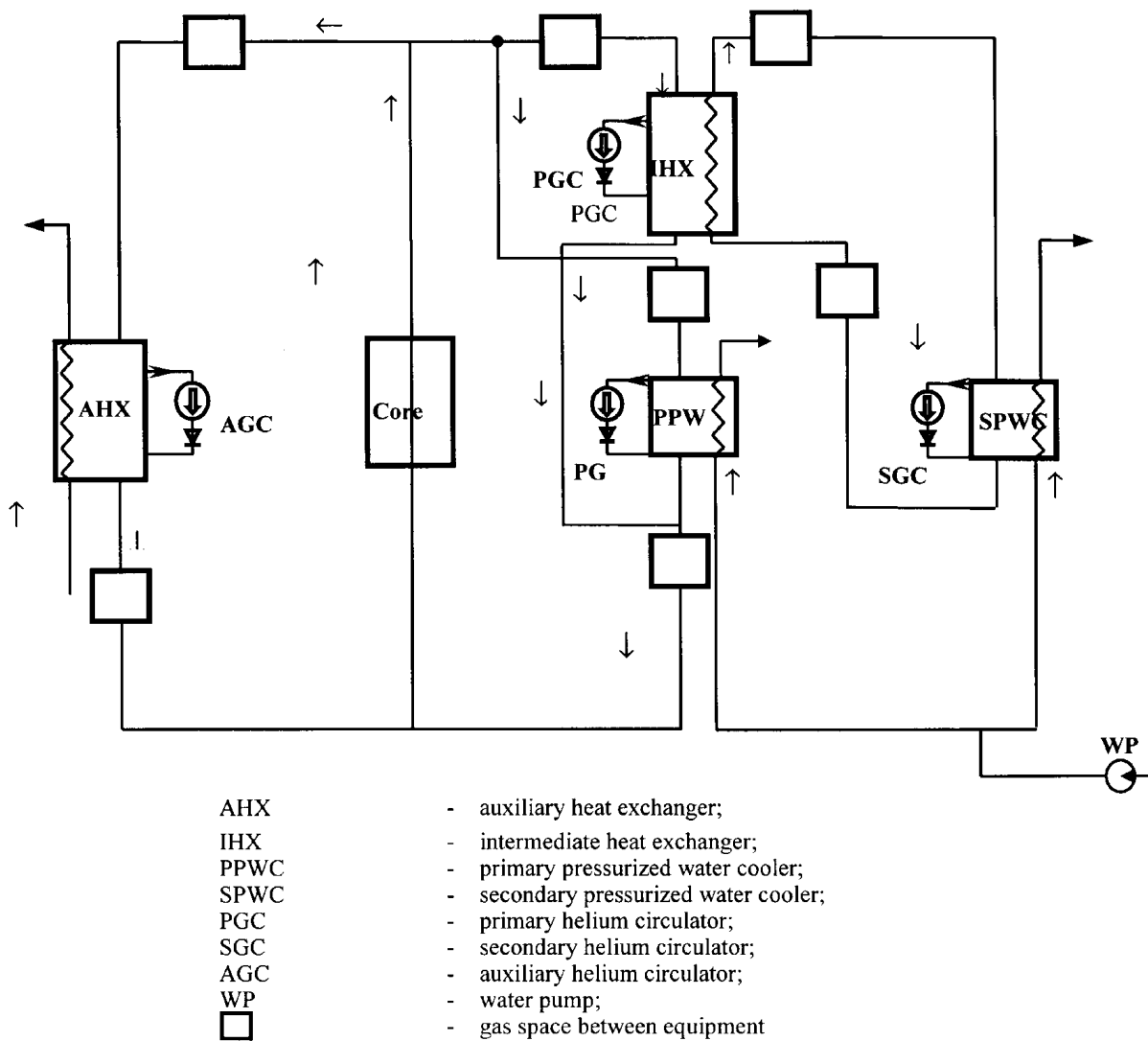


FIG. 3.19. HTTR plant calculation diagram.

The mathematical model of reactor control and protection system simulates operation of automatic regulator on relay principle, account of possible time delay during formation of shutdown signals, acceleration time of absorber rods and balls.

The VGM code provides the capability to calculate core power density, temperature distribution in fuel elements, coolant temperatures and flow-rates in primary, secondary and emergency cooling circuits.

The text of the code is written on the following:

FORTRAN-77

Operational system-MS DOS 6.22

Type of language translator-MICROSOFT FORTRAN VERSION 5.00

A more detailed description of the code is presented in [3-7].

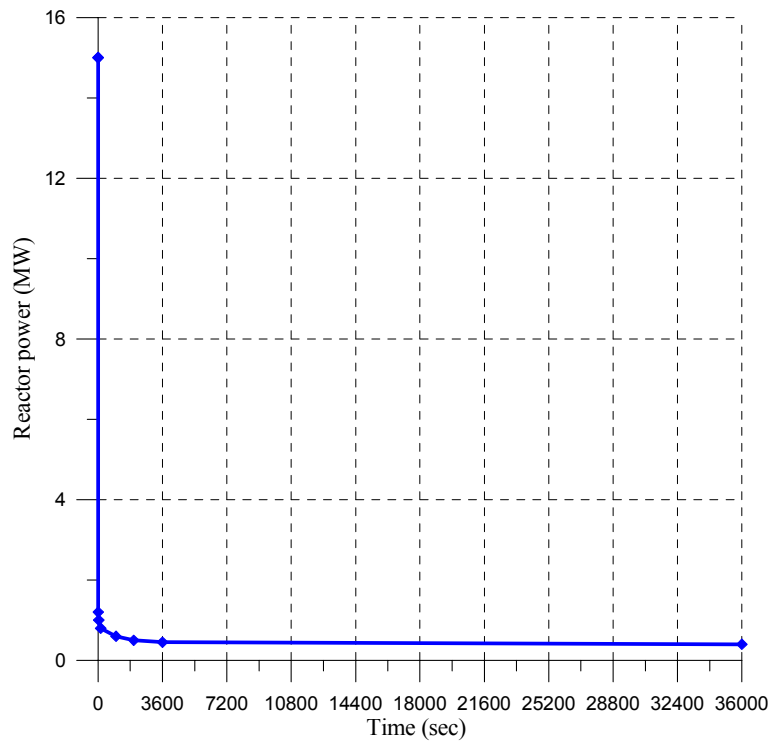


FIG. 3.20. Transient behavior of reactor power during loss of off-site electric power test (15 MW).

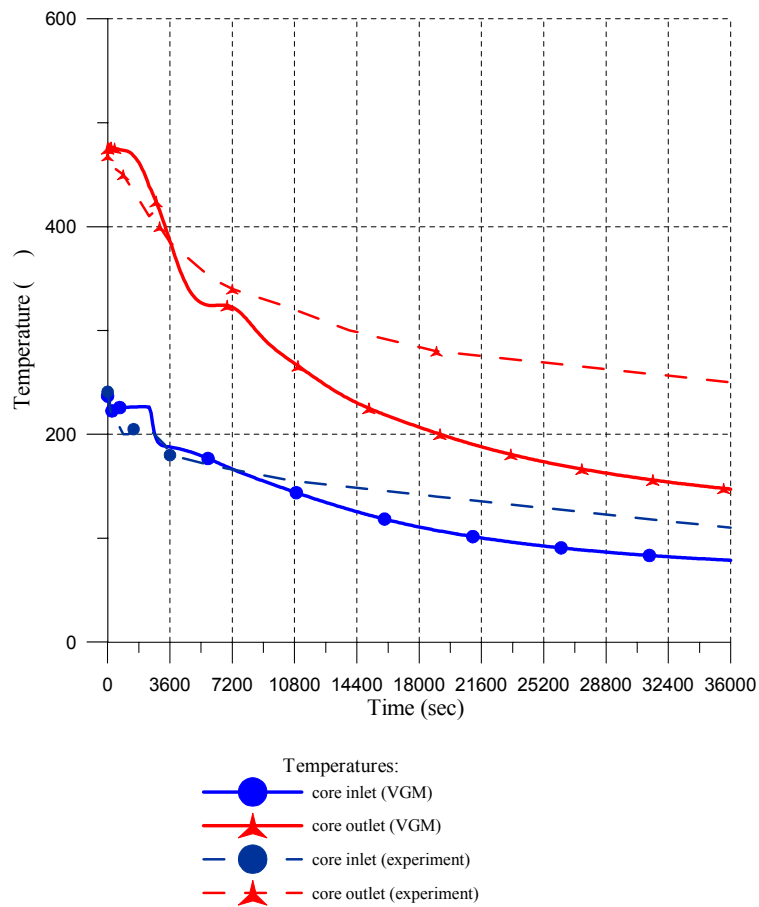


FIG. 3.21. Transient behavior of temperatures during loss of off-site electric power test (15 MW).

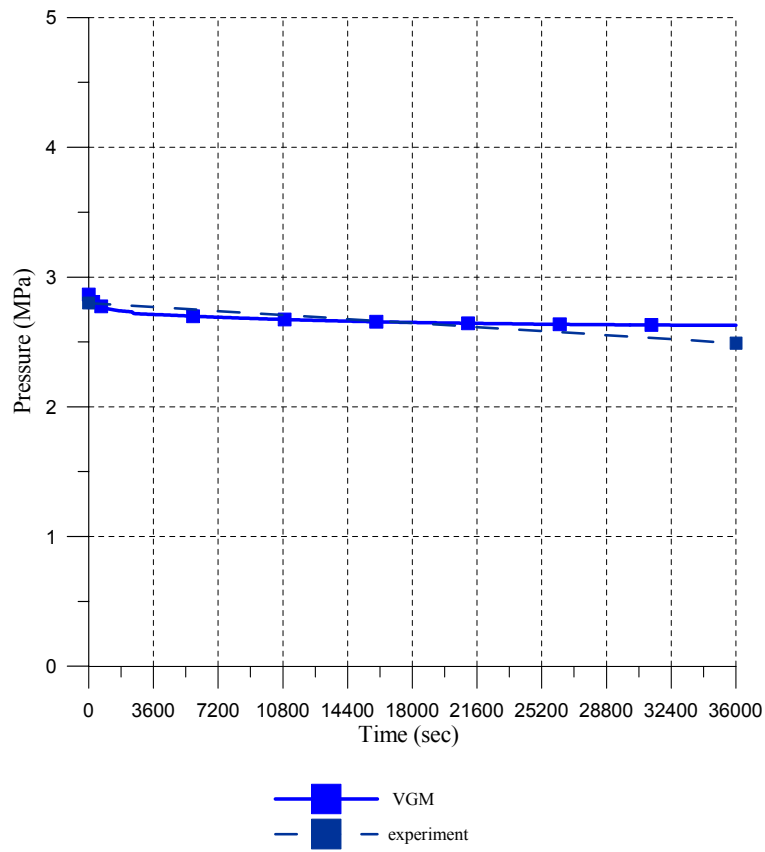


FIG. 3.22. Transient behavior of pressure during loss of off-site electric power test (15 MW).

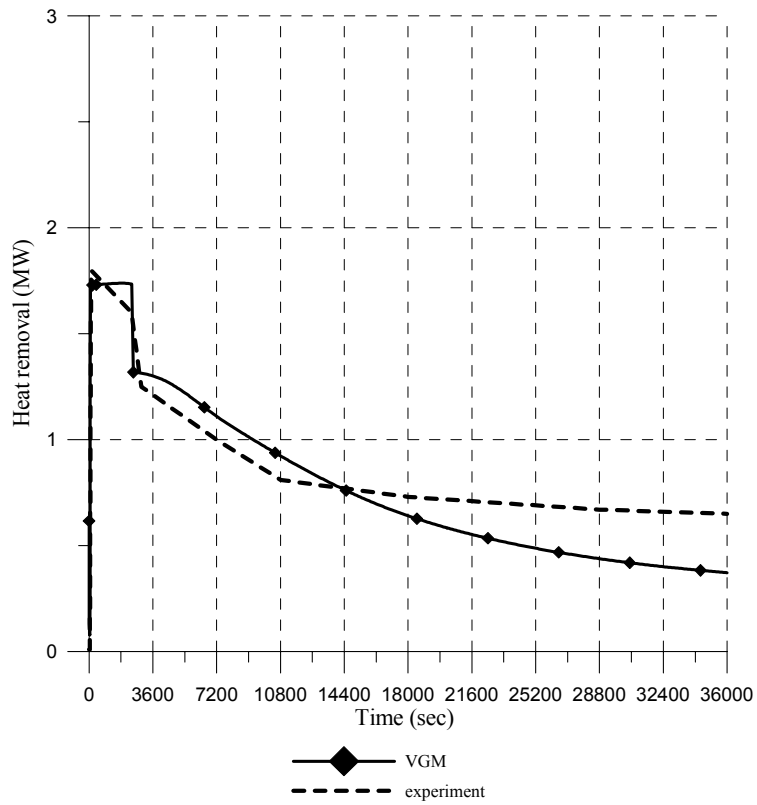


FIG. 3.23. Transient behavior of heat removal during loss of off-site electric power test (15 MW).



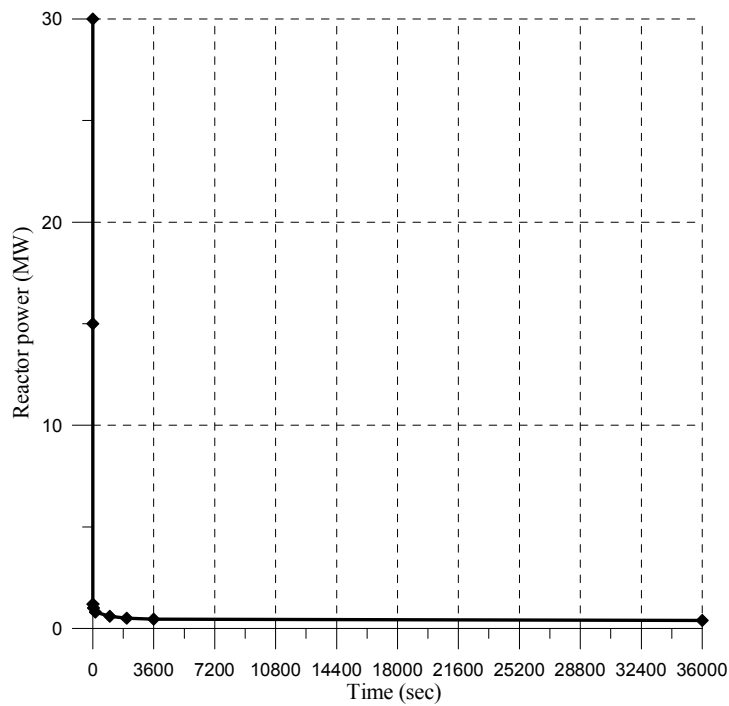


FIG. 3.24. Transient behavior of reactor power during loss of off-site electric power test (30 MW).

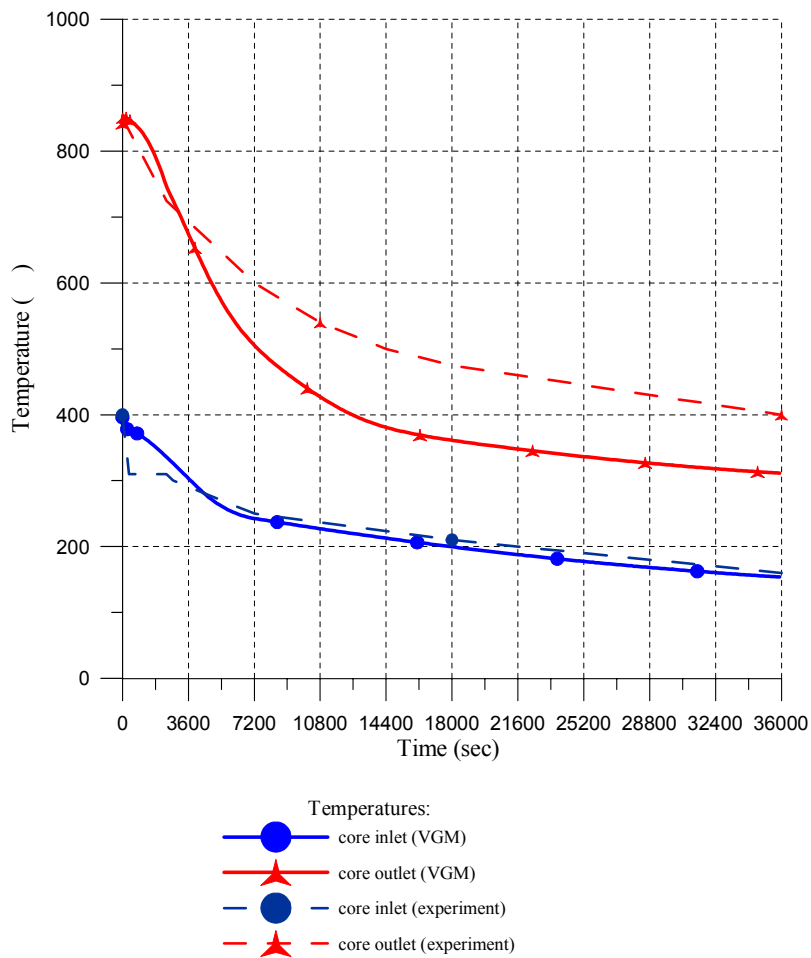


FIG. 3.25. Transient behavior of temperatures during loss of off-site electric power test (30 MW).

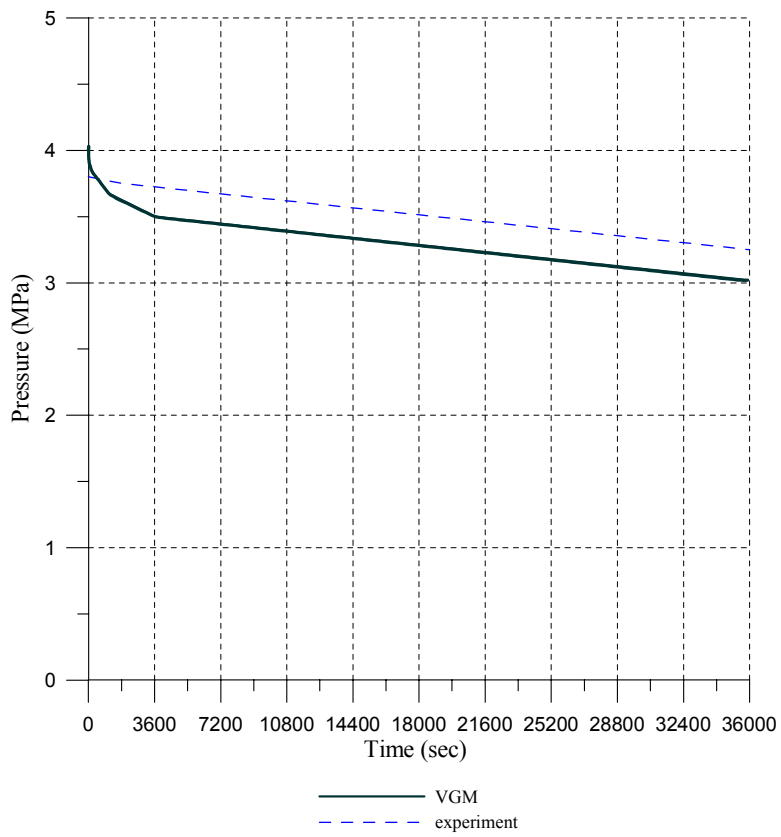


FIG. 3.26. Transient behavior of pressure during loss of off-site electric power test (30 MW).

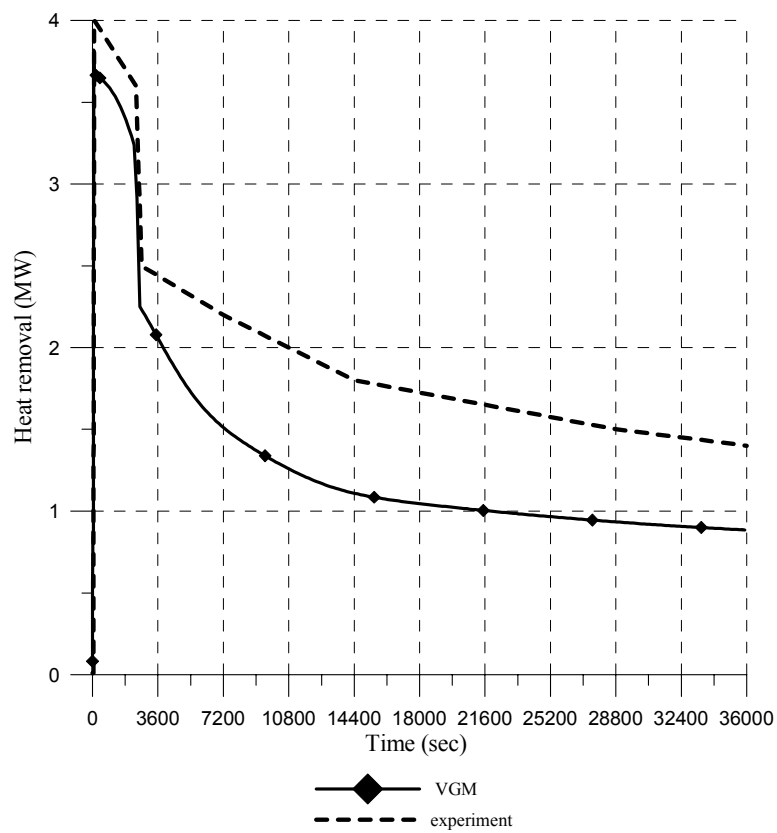


FIG. 3.27. Transient behavior of heat removal during loss of off-site electric power test (30 MW).

## ***Methodology***

The HTTR calculational model was developed (Figure 3.19) on the basis of data presented in [3-8, 3-9]. The number of breakdown sections that were adopted included the following:

- core height	42
- IHX and PPWC length	10
- heat exchange tube thickness	4
- gas spaces	2.

For a description of the transient heat transfer in the core, two types of cells with average and peak power are chosen. Every cell includes a fuel rod, a gap with cooling helium and equivalent mass of graphite in one block per one cell. The heat transfer in the axial direction by conductivity is not taken into account.

The heat exchange equipment is simulated as a single equivalent tube. Axial heat transfer is ignored.

## ***Results***

Results of the calculation of HTTR behavior during the loss of power are shown on Figures 3.20, 3.21, 3.22 and 3.23 for 15 MW thermal power and on Figures 3.24, 3.25, 3.26 and 3.27 for 30 MW thermal power.

Flow rates of coolant through the primary gas circulators are reduced to zero in 10 s after the “loss of off-site electric power” signal.

Emergency protection actuates in 1.6 s after loss of off-site electric power. Neutron power decreases down to close to decay heat for 7 s.

The comparison of the calculated and experimental data shows the following.

In the beginning of the transient (0-2 hr) one can see a good agreement between calculated and experimental values of coolant temperatures. Ratios of experimental and calculated heating of the coolant are more than 1.1 and 1.3 for 15 and 30 MW tests, respectively. Further, the discrepancy increases up to 2 for 15 MW and 1.5 for 30 MW at the transient end-point (10 hrs).

The calculated values of heat removal for 15 MW are in a good agreement with the experimental data during first five hours of the transient. Further, the experimental values are well higher than calculated ones. Their ratio is 1.75 for the transient end-point (10 hrs).

For 30 MW an acceptable agreement between calculated and experimental results is only during first hour of the transient. The further transient behavior is also characterized by excess of the experimental heat removal as against the experimental one which reaches ~ 1.7 for the end-point.

The discrepancies can be explained by the following reasons:

1) The decay heat error has a strong effect on the calculated results of temperatures and heat removal, since the decay heat several times more than accumulated in the core and

reflectors. This specially affects on the discrepancies in the interval of 5-10 hrs. Increase of decay heat leads to reduction of discrepancies. For example, decay heat 1.5 as large gives full agreement between calculated and experimental results.

2) Another reason may be connected with influence of the finite break-down of the core in axial and radial directions on the results. Further analysis is needed to be carried out for this case.

### 3.2.3 South Africa [3-11]

This report details the benchmarking of the Flownet Nuclear simulation package as used by Pebble Bed Modular Reactor in South Africa. The benchmark is of a loss of off-site power event that was tested at the HTTR facility. The event involves the electric power supply to the circulators being cut off, reactor SCRAM and the activation of the Auxiliary Cooling system to remove decay heat.

#### 3.2.3.1 System Overview

The HTTR system was modelled and analysed with Flownet Nuclear and is comprised of a graphite-moderated prismatic block reactor, a primary cooling circuit, a secondary cooling circuit and an auxiliary cooling circuit. Helium is used as the working fluid. During normal operation the helium is heated in the reactor and is then circulated via concentric ducts to two parallel heat exchangers. The first is the Primary Pressurised Water Cooler (PPWC) that is a shell-and-tube water-cooled exchanger. The second is the Intermediate Heat Exchanger (IHX) that is a spirally wound shell and tube type. Its function is to transfer heat to the secondary helium circuit. Both heat exchangers' exits are connected to helium circulators providing the mass flow rate needed in the circuit. The HTTR system can either be operated using only the PPWC as cooler or both the PPWC and the secondary circuit (the latter is called parallel cooling operation). A schematic of the HTTR cycle is shown in Figure 3.28.

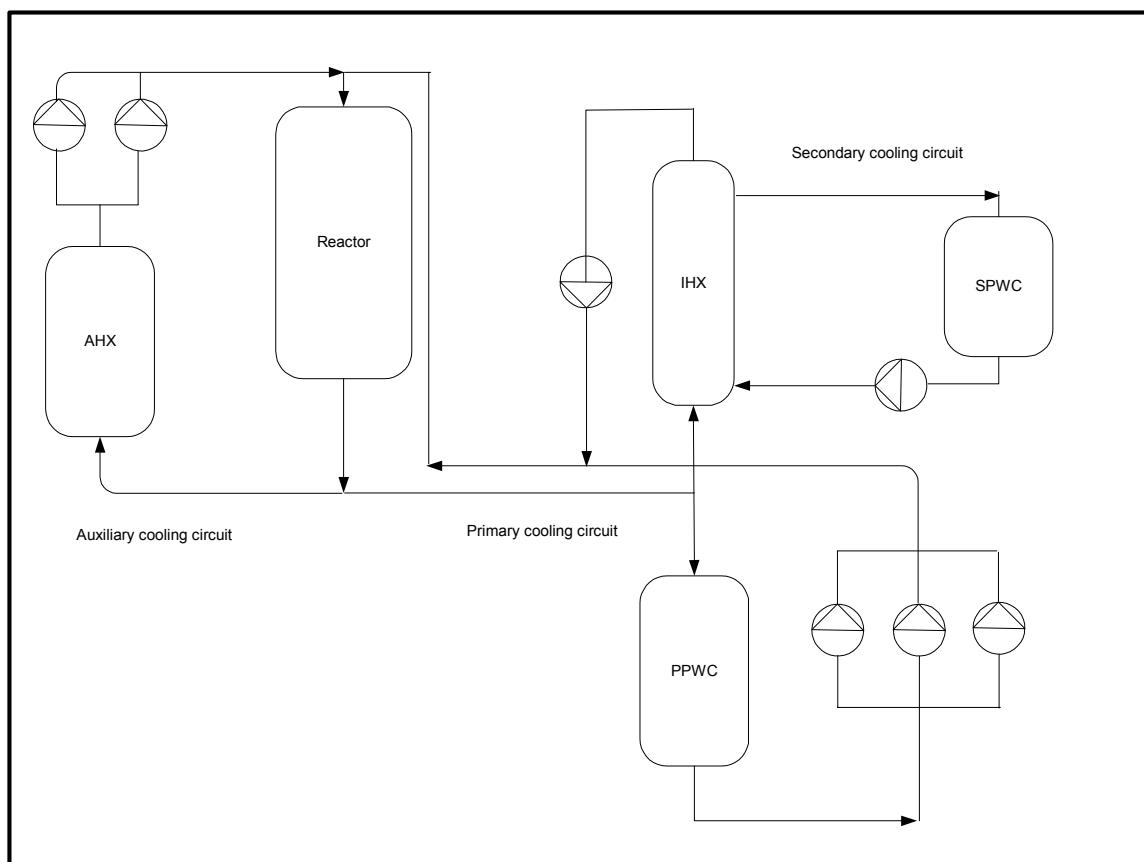


FIG. 3.28. Schematic diagram of the High Temperature engineering Test Reactor.

In the secondary helium circuit a helium circulator is used to circulate flow between the IHX and the Secondary Pressurised Water Cooler (SPWC), the latter being of similar but smaller design to the PPWC. Both water coolers' watersides are on water circuits that exchange heat with the atmosphere via water-to-air heat exchangers.

During an accident condition such as the loss of off-site electrical power the Auxiliary cooling circuit is used. This circuit uses two gas circulators to force a low mass flow rate of helium between the reactor and the Auxiliary Heat Exchanger (AHX). The AHX is a water cooler of similar but smaller design to the PPWC. The function of this system is to remove decay heat from the reactor core after a reactor shutdown or SCRAM has been initiated.

### 3.2.3.2 *Flownet Nuclear*

Flownet Nuclear is a pipe-network simulation code used for solving thermodynamic and fluidic problems in pipe networks and energy systems. M-Tech Industrial in collaboration with the School for Mechanical and Materials Engineering at Potchefstroom University in South Africa has developed the code over the past ten years. Initially the code was used for large gas pipe networks but has since been extended to handle most types of thermal-hydraulic systems. The following points highlight the capabilities of the program:

- The ability to deal with compressible as well as incompressible fluids.
- Full heat transfer by means of conduction, convection and radiation.
- High solution speed. The typical solution time of a 200-element network is less than 30 seconds on a standard PC.
- Various types of network components are available in the program
  - Constant and variable area pipes and ducts
  - Restrictors with loss or discharge coefficients.
  - Cooling orifices for which NASA empirical correlations are used to calculate the discharge coefficient.
  - Non-circular ducts with area changes. With this element the user has the option of taking the effect of the injected momentum and variations in total temperature on the change in total pressure into account.
  - Various types of heat exchangers.
  - Compressible flow through long pipes.
  - Compressors and turbines
  - Pebble-bed and solid fuel HTGR reactors as special models.
  - Heat transfer paths
  - The ability to specify a fixed mass source or a sink in junctions and volumes between flow elements
  - Variable fluid properties.
  - Ability to automatically subdivide long pipes into smaller increments.
  - Gravitational effects in compressible flows

Flownet Nuclear has also been extensively benchmarked and it was found that results are in close agreement with experimental data and analytical calculations. An example is a test involving compressible gas flow through a long pipe. At time zero a valve is suddenly closed at the end of the pipe. The inertia of the fluid causes auto-compression at the end of the pipe that in turn results in a pressure wave travelling back through the pipe. Flownet Nuclear data is benchmarked against data obtained using the Lax-Wendroff method. Figure 3.29 shows a plot of total pressure in the middle of the pipe verses time.

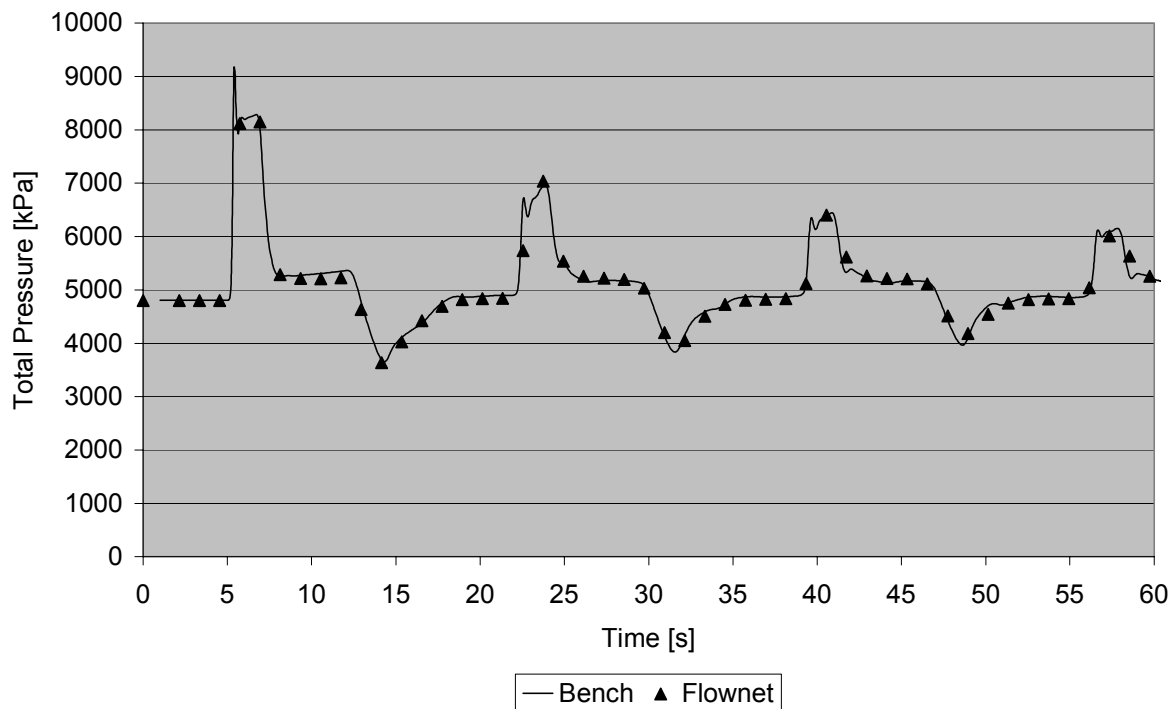


FIG. 3.29. Total pressure versus time for the sudden closure of a valve at the end of the pipe.

Applications of the code include:

- Ventilation systems.
- Compressed air systems of mines and factories.
- Air distribution system of a conveyor belt system that uses an air cushion to suspend the belt.
- Water reticulation systems.
- Heat recovery systems.
- Flow through the combustion chamber of an aircraft engine.
- Flow through the re-heater of an aircraft engine.
- City gas pipe networks.
- Brayton power cycle as in the PBMR.
- Complete aircraft air-conditioning system.

The detail concerning the theory behind Flownet Nuclear falls beyond the scope of this document.

### 3.2.3.3 Modeling of the HTTR in Flownet

#### **Cycle description**

Flownet uses *elements* and *nodes* to represent system components. An element refers to the actual components such as pipes, turbo-machines or heat exchangers. Nodes are junction points for elements but can also represent regions of large volume and low flow velocity – tanks and reservoirs.

Figure 3.30 is a diagram of the HTTR system as implemented in terms of Flownet elements and nodes. Elements are shown as numbered circles while the numbered squares are nodes. The diagram shows the use of pipe elements (DW and DG), compressors (CM), heat exchangers (HX and RX) and conductive heat transfer paths (CHT). Only the main, auxiliary and secondary cooling systems are modelled for simplicity. Cooling water flows are modelled as constant-mass flow cold-water streams. The vessel cooling system is not modelled in this investigation.

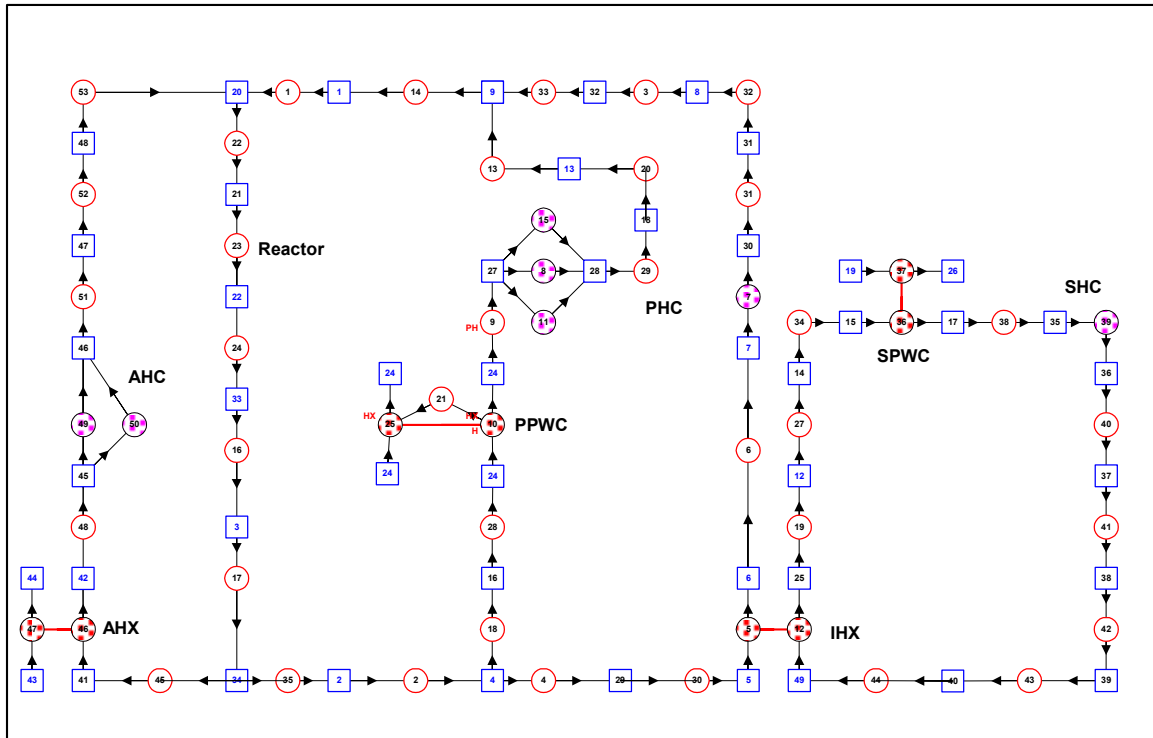


FIG. 3.30. HTTR Flownet Network

The descriptions of the elements used in the network are shown in Table 3-30.

Table 3-30. Elements used in Flownet network

Element no.	Description	Type
1	Annular duct to top of reactor	Gas duct
2	Reactor to T-junction	Gas duct
3	Contraction to gas duct	Gas duct
4	Contraction from T-junction	Gas duct
Element no.	Description	Type
5	Intermediate heat exchanger	Heat exchanger
6	IHX to PHC	Gas duct
7	SPWC circulator	Helium circulator
8	PPWC circulator B	Helium circulator
9	PPWC to PHC	Gas duct
10	Primary Pressurized Water Cooler	Heat exchanger



11	PPWC circulator C	Helium circulator
12	IHX secondary side	Heat exchanger
13	PPWC to T-junction	Gas duct
14	T-junction to reactor	Gas duct
15	PPWC circulator A	Helium circulator
16	9th reflector layer	Gas duct
17	Hot plenum block	Gas duct
18	T-junction to PPWC	Gas duct
19	Secondary IHX to T-junction A	Gas duct
20	Annulus around PPWC	Gas duct
22	Upper reflector and shield layers	Gas duct
23	Fuel assembly (Reactor core)	HTTR
24	Lower reflector layer	Gas duct
25	PPWC water side	Heat exchanger
26	HTTR	HTTR
27	Secondary IHX to T-junction B	Gas duct
28	Distribution plate	Gas duct
29	PHC to PPWC	Gas duct
30	Contraction to IHX	Gas duct
31	PHC to IHX	Gas duct
32	Annulus around IHX	Gas duct
33	IHX to T-junction	Gas duct
34	Distribution plate	Gas duct
35	Hot plenum outlet channel	Gas duct
36	Secondary Pressurized Water Cooler	Heat exchanger
37	SPWC water side	Heat exchanger
38	SPWC to SHC	Gas duct
39	Secondary Helium Circulator	Helium circulator
40	SHC to SPWC	Gas duct
41	Annulus around SPWC	Gas duct
42	SPWC to T-junction	Gas duct
43	T-junction to tube header feed pipes	Gas duct
44	Header feed pipes	Gas duct
45	Reactor to AHX	Gas duct
46	Auxiliary Heat Exchanger	Heat exchanger
47	AHX water side	Heat exchanger
48	AHX to AHC	Gas duct
49	Auxiliary Helium Circulator A	Helium circulator
50	Auxiliary Helium Circulator B	Helium circulator
51	AHC to AHX	Gas duct
52	Annulus around AHX	Gas duct
53	AHX to Reactor	Gas duct

The detailed modelling falls beyond the scope of this document, however some comments can be made on the modelling of generic components

### *Piping and duct elements*

Most of the piping in the HTTR plant consists of double-walled pipes with hot flow from the reactor in the inner pipe and cooled return gas in the outer annulus of the pipe. Figure 3.31 shows a lateral cross-section through the primary cooling circuit piping. The

diagram clearly shows the double-walled piping and the lagging material surrounding the pipe for insulation. Flow from the reactor goes via the inner pipe to a T-junction where the flow is split up to a ratio of 3:1 between the PPWC and the IHX. Upon returning to the reactor the helium flows through the outer annular pipe.

The pipes are modelled as Darcy-Weisbach (DW) elements in Flownet. These pipe elements use the following formula to calculate the pressure drop through a pipe for compressible flow.

$$\Delta p_0 = \left( \frac{fL}{D} + \frac{\Delta T_0}{T_0} \right) \frac{p_0 \gamma M^2}{2} \quad (1)$$

where  $f$  – friction factor

$L$  – Pipe length [m]

$D$  – Pipe diameter [m]

$T_0$  – Total temperature [K]

$p_0$  – Total pressure [Pa]

$\gamma$  – ratio of specific heats of the gas

$M$  – Mach number

The friction factor is calculated using a correlation fitted to the Moody diagram with the transition between laminar and turbulent flow beginning at Reynolds numbers greater than 2300.

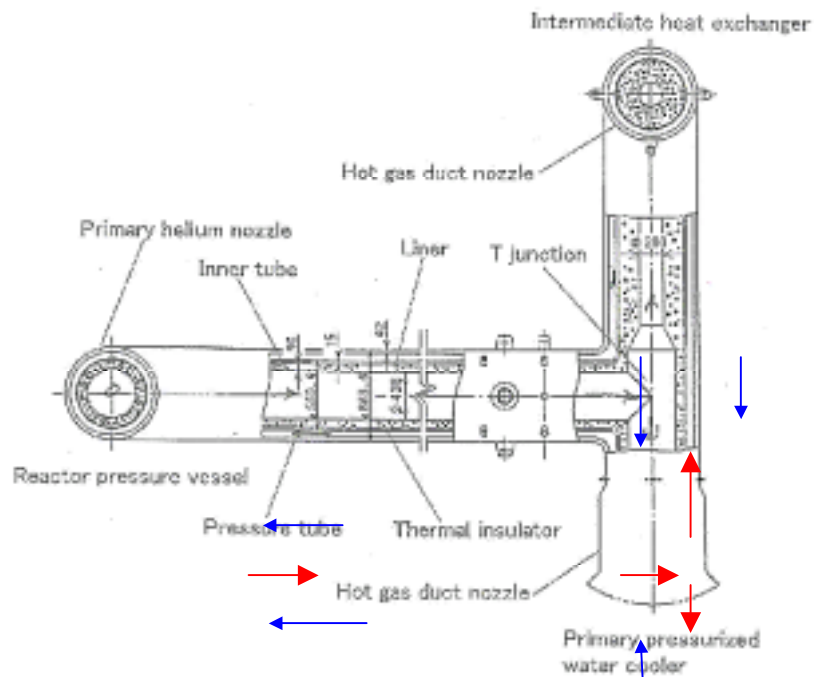


FIG. 3.31. Primary circuit piping.

In the case of the annular flows the cross-sectional flow area and wetted perimeter are inputs to Flownet that will then calculate a hydraulic diameter for use in (1). The flow area is used along with mass flow rate to calculate the velocity component of (1). All the other terms are inputs in the pipe element model.

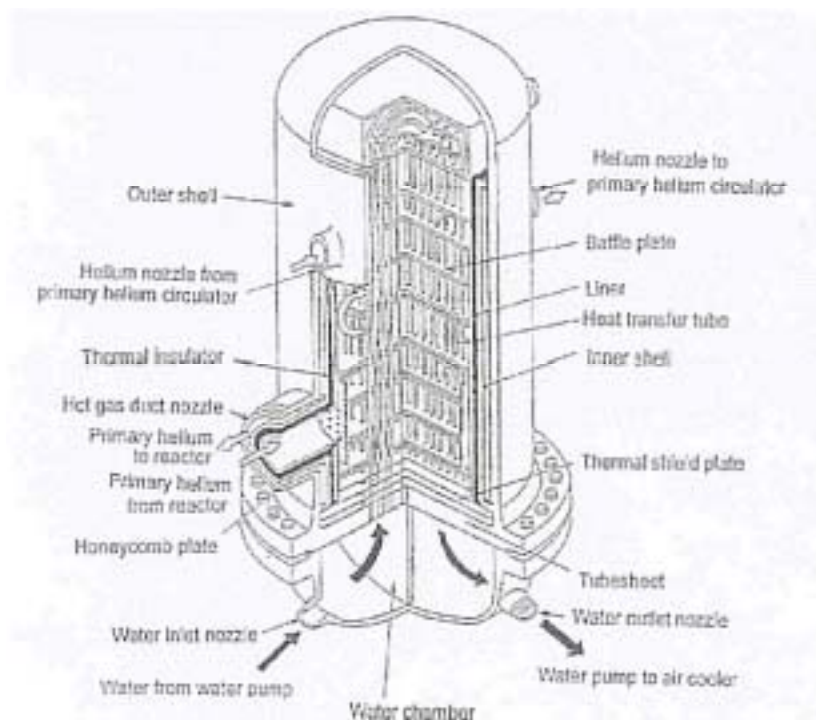
### *Heat exchanger elements*

#### Shell-and-tube heat exchangers

The PPWC, SPWC and AHX are three shell-and-tube heat exchangers of similar design. Figure 3.32 shows a bird's eye view of the inside of the HTTR's PPWC. The tubes used are un-finned. There are 5 gas passes and 2 tube passes in the heat exchanger. A distribution plate is placed at the inlet to spread the gas flow evenly through the first stage.

Flownet Nuclear has a Shell-and-Tube Heat exchanger or STX type element specifically catering for this type of heat exchanger. In all Flownet's heat exchanger models provision is made for both steady-state heat transfer parameters and parameters describing the heat exchanger's thermal inertia. Thermal inertia is very important when one wants simulating transient changes in the heat exchanger's inlet and outlet conditions.

In Flownet's STX model the tube and baffle layout and dimensions are required as inputs along with parameters such as total heat transfer area, primary/secondary area ratio and tube conduction coefficients. For transient simulations the mass capacitance and gas volume are inputs. The heat transfer coefficient and friction factor on the tube side are calculated using the Dittus-Boelter equation and the standard Darcy-Weisbach pipe model, respectively. On the shell side the Reynolds number is calculated and a corresponding Colburn  $j$  factor and friction factor is read off an input graph. These two graphs are specifically set-up for the tube layout used in the heat exchanger. The two values are then used to calculate the heat transfer and pressure drop through the heat exchanger using standard correlations.



*FIG. 3.32. Bird's eye view of the PPWC.*

## Intermediate Heat Exchanger

This heat exchanger functions as an energy interface between the primary and secondary cooling circuits. The IHX is a helically coiled shell-and-tube gas-to-gas heat exchanger. Figure 3.33 shows a diagram of the actual IHX heat exchanger. Hot gas from the reactor enters from below and flows up over the helical tube bundles to a manifold volume at the top of the heat exchanger. From there it goes to a circulator and back again through the outer annulus of the heat exchanger pressure vessel.

The tubes contain cool gas from the secondary cooling circuit. This gas flows down from the top of the heat exchanger making the configuration counter flow.

The gas-to-gas type heat exchanger is modelled as a recuperator heat exchanger (RX) in Flownet. This type of element requires the hydraulic diameter, total free flow area and flow length of both the hot and cold sides. It calculates the heat transfer and friction losses by means of Nusselt number and friction factor correlations respectively. The thermal inertia is accounted for by means of a mass-capacitance or MC value.

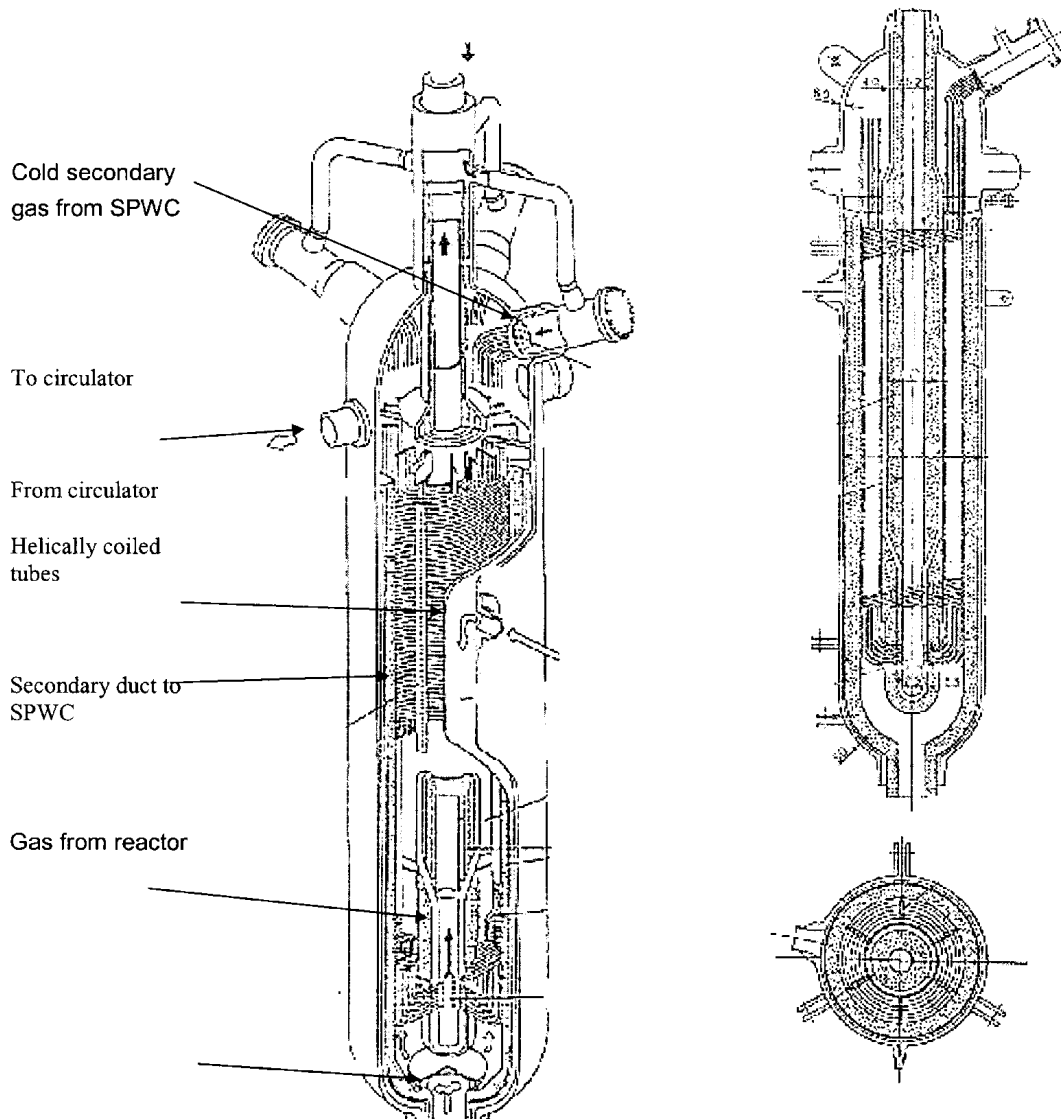


FIG. 3.33. Bird's eye view and sections of the Intermediate Heat Exchanger.

The RX model is mostly suited to simple plate-fin recuperators where the hot and cold side are of similar length and cross-section. As such the helical-coil configuration of the IHX, that has a much longer flow path on the secondary side, had to be tuned to get correct heat transfer in steady-state operation. This was done by altering the Nusselt and friction drop factors that are inputs in the RX model.

## Helium circulators

The helium circulators used in the gas circuits are single-stage centrifugal compressors. Helium from the associated cooler enters the circulator inlet and goes through a filter before being compressed. A variable speed motor is used to control the helium flow rate delivered by the circulator.

In Flownet turbines, blowers and compressors are modelled by means of characteristic curve data. The pressure ratio verses corrected mass flow ( $m\sqrt{T_0}/P_0$ ) and efficiency verses corrected mass flow characteristic curves for different corrected shaft speeds ( $N/\sqrt{T_0}$ ) are used. An input shaft speed is used along with the inlet gas conditions to find the operating point on the curves.

The graphs of pressure ratio versus invariant mass flow rate and invariant power versus invariant mass flow rate were supplied by JAERI. Graphs were supplied for the primary, secondary and auxiliary circuit machines. These graphs had to be converted into the Flownet compressor curve file format before it could be used in the circuit.

Conversion of invariant mass flow to corrected mass flow:

$$m_{cor} = m_{inv} / 3600 \quad (2)$$

$$m = \frac{m_{cor} P_{01}}{\sqrt{T_{01}}} \quad (3)$$

Since Flownet requires a graph of efficiency versus corrected mass flow rate the invariant power data had to be modified to obtain efficiency. This was done as follows:

$$P = P_{inv} P_{01} \sqrt{T_{01}} \quad (4)$$

$$P_{fluidic} = (P_{ratio}^{(\gamma-1)/\gamma} - 1) c_p T_{01} m / 1000 \quad (5)$$

$$\eta = P_{fluidic} / P \quad (6)$$

Where  $P$  – Power input to the compressor

$P_{fluidic}$  – Power delivered to fluid

$p_{ratio}$  – pressure ratio (JAERI)

$\eta$  - compressor efficiency

$T_{01}$  – Inlet temperature

$p_{01}$  – Inlet pressure

The characteristic maps for the primary and secondary circulators are the same and are shown in Figure 3.34 and 3.35. The primary circulator runs at a speed of 9400 rpm while the secondary machine is slightly faster at 9900rpm at 30MW. For 15MW they operate at 8900 rpm and 9500 rpm respectively.

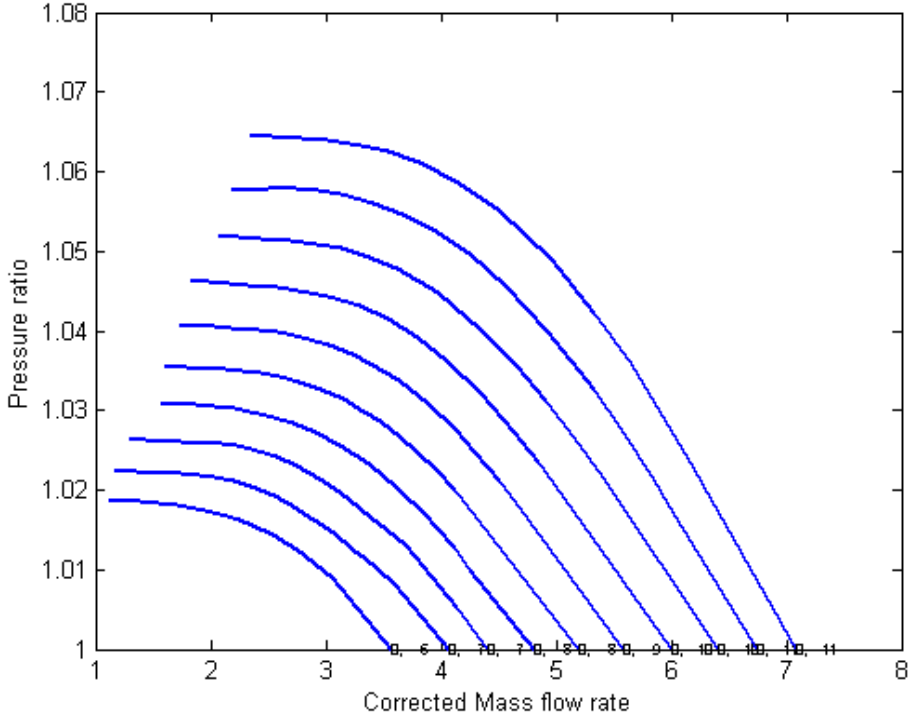


FIG. 3.34. Primary and Secondary circuit Pressure ratio map.

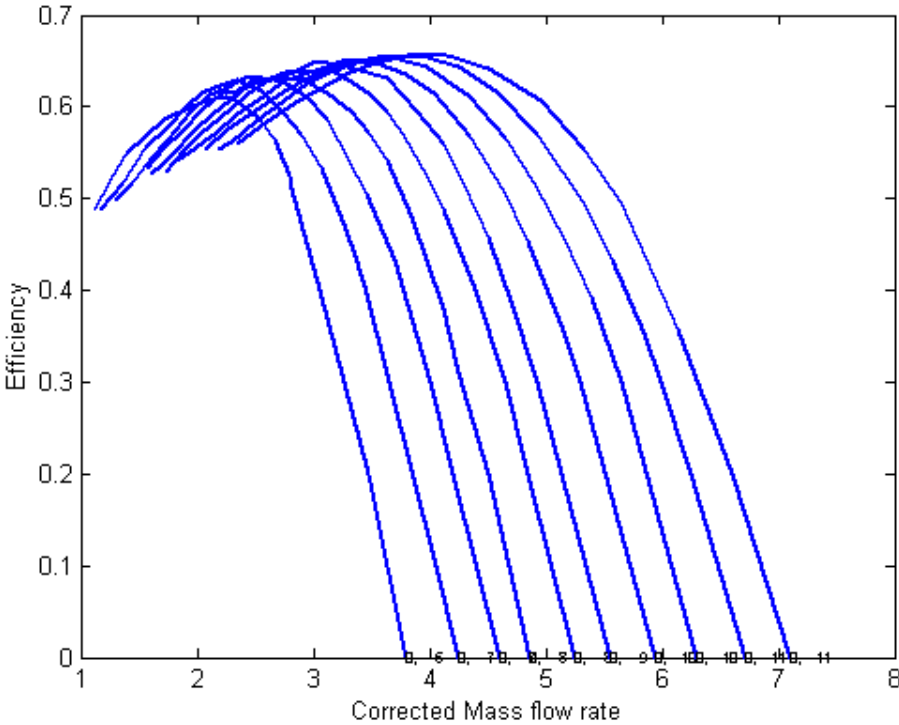


FIG. 3.35. Primary and Secondary circuit Efficiency map.

## ***HTTR Reactor***

The reactor is modeled by means of a specially developed HTTR element in Flownet Nuclear. This model encompasses all the core internals starting from the channel entry at the upper reflector through to the hot plenum beneath the core. The core nuclear dynamics are modeled with a point kinetics model that uses the core dynamics parameters as prescribed by JAERI. During the transient event the decay heat production of the core is input by means of the prescribed decay heat equation given by JAERI.

### ***3.2.3.4 Steady State Results***

#### ***Steady-state results comparison – 15MW operation***

This section discusses the results for the 15MW loss of power benchmark. The results tables contain three columns: the first column tabulates the predicted HTTR data as sent out in JAERI's first document on the benchmark [3-12], the second column contains the test data as obtained in the 15MW loss of power experiment while the last column gives the Flownet Nuclear data.

The data shows that the Flownet Nuclear results are closer to the experiment than to the predicted values. This is possibly because, for the experiment, the speeds of the circulators were greatly reduced from predicted values. It was found that if the predicted speeds are used in the simulation the mass flow becomes much higher than what is needed for 15MW power delivery in the reactor. It is assumed that JAERI was conservative in their calculation of the system resistance and hence the circulators were over sized.

Table 3-31. Reactor core results comparison

<b>Parameter</b>	<b>HTTR Estimate</b>	<b>HTTR Experimental</b>	<b>Flownet Nuclear</b>
Thermal power [MW]	15	15	15
Helium inlet temp. [°C]	241 (About)	242	240.5
Helium outlet temp. [°C]	470 (About)	468	472.7
Helium pressure [kPa]	3000 (About)	2900	2795
Helium mass flow rate [kg/s]	12.4	12.4	12.44
Hot plenum block temp. [°C]	490 (About)	391	-

Table 3-32. Intermediate Heat Exchanger results comparison

<b>Parameter</b>	<b>HTTR Estimate</b>	<b>HTTR Experimental</b>	<b>Flownet Nuclear</b>
Primary helium inlet temp. [°C]	468 (About)	468	472.7
Primary helium outlet temp. [°C]	238 (About)	235	240.77
Primary helium pressure [kPa]	3000 (About)	2800	2784
Primary helium mass flow rate [kg/s]	4.1	4.1	4.078
Secondary helium inlet temp. [°C]	154 (About)	158	160.6
Secondary helium outlet temp. [°C]	431 (About)	420	423.7
Circulator speed [rpm]	9500	8770	8770
Secondary helium pressure [kPa]	3100 (About)	2900	2900
Secondary helium mass flow rate [kg/s]	3.6	3.6	3.59

Otherwise, there are small differences in pressure and temperatures but until more detailed data becomes available on pressure drops through the HTTR these differences are considered negligible. Tables 3-31, 3-32, 3-33 and 3-34 provide a comparison of the Flownet results against the computed results supplied by JAERI for the various parts of the HTTR system.

***Steady-state results comparison – 30MW operation***

This section gives the results for the 30MW simulations. Tables 3-35, 3-36, 3-37 and 3-38 each contain the following: HTTR predicted results, Flownet Nuclear results using the prescribed shaft speeds and lastly Flownet results based on the same shaft speeds as used in the 15MW test. As can be seen the first column of Flownet results contains large differences to the predicted HTTR results. After careful analysis of the 15MW experimental data it was found that the system resistance was overestimated and so the 15MW shaft speeds were reduced.



Table 3-33. Primary Pressurized Water Cooler results comparison

<b>Parameter</b>	<b>HTTR Estimate</b>	<b>HTTR Experimental</b>	<b>Flownet Nuclear</b>
Helium inlet temp. [°C]	468 (About)	468	472.7
Helium outlet temp. [°C]	242 (About)	235	240.77
Helium pressure [kPa]	3000 (About)	2800	2784
Helium mass flow rate [kg/s]	8.3	8.2	8.308
Circulator Speed [rpm]	8900	8000	8000
Water inlet temp. [°C]	89 (About)	79	79
<b>Parameter</b>	<b>HTTR Estimate</b>	<b>HTTR Experimental</b>	<b>Flownet Nuclear</b>
Water outlet temp. [°C]	110 (About)	100	101.29
Water pressure [kPa]	2600 (About)	2300	2300
Water mass flow rate [kg/s]	115	111	111

For the 30MW case the mass flow required is the same as for 15MW, the difference in power is due to differences in inlet and outlet temperatures of the core. However, the mass flow requirement is dominant in sizing the circulators' rotational speeds and since the mass flow is the same as for 15MW the rotational speeds should be roughly the same. Column three clearly shows much better correlation between the Flownet simulation and the HTTR predicted data. These conclusions will only be clarified once experimental data is received for the 30MW case.

### 3.2.3.5 Transient Results

This section presents the transient results of the loss of power benchmark for both the 30MW and 15MW cases.

#### **Simulation setup**

This transient involves a coasting down of the main loop circulators of the HTTR. A reactor SCRAM and loss of heat exchanger cooling mass flow are also included in the transient. Later in the analysis the auxiliary circulators are started and the cooling of the reactor's decay heat is commenced.

Table 3-34. Secondary Pressurized Water Cooler results comparison

<b>Parameter</b>	<b>HTTR Estimate</b>	<b>HTTR Experimental</b>	<b>Flownet Nuclear</b>
Helium inlet temp. [°C]	430 (About)	420	423.6
Helium outlet temp. [°C]	154 (About)	152	155.86
Helium pressure [kPa]	3100 (About)	2900	2861
Helium mass flow rate [kg/s]	3.6	3.6	3.59
Circulator speed [rpm]	8100	8130	8130
Water inlet temp. [°C]	89 (About)	79	79
Water outlet temp. [°C]	110 (About)	100	99.9
Water pressure [kPa]	2500 (About)	2300	2300
Water mass flow rate [kg/s]	60	57	57

Table 3-35. Reactor core results comparison

<b>Parameter</b>	<b>HTTR Estimate</b>	<b>Flownet Nuclear – Pre-scribed Shaft speeds</b>	<b>Flownet Nuclear – 15 MW Shaft speeds</b>
Thermal power [MW]	30	30	30
Helium inlet temp. [°C]	395	412	402.7
Helium outlet temp. [°C]	850	789.2	858.9
Helium pressure [kPa]	4000	3993	3995.7
Helium mass flow rate [kg/s]	12.4	15.31	12.66
Hot plenum block temp. [°C]	890	789.2	858.9

Table 3-36. Intermediate Heat Exchanger results comparison

<b>Parameter</b>	<b>HTTR Estimate</b>	<b>Flownet Nuclear – Pre-scribed Shaft speeds</b>	<b>Flownet Nuclear – 15 MW Shaft speeds</b>
Circulator speed [rpm]	9900	9900	8770
Primary helium inlet temp. [°C]	850	789.2	858.9
Primary helium outlet temp. [°C]	395	419.53	421
<b>Parameter</b>	<b>HTTR Estimate</b>	<b>Flownet Nuclear – Pre-scribed Shaft speeds</b>	<b>Flownet Nuclear – 15 MW Shaft speeds</b>
Primary helium pressure [kPa]	4000	3977.4	3983.67
Primary helium mass flow rate [kg/s]	4.1	4.8	4.08
Secondary helium inlet temp. [°C]	241	260.7	262.74
Secondary helium outlet temp. [°C]	783	724.4	721.1
Secondary helium pressure [kPa]	4100	4092.5	4092
Secondary helium mass flow rate [kg/s]	3.6	3.87	3.86

The loss of power benchmark is modelled as a transient in Flownet. The code requires a transient events file that contains the user-defined transient events that happen at specified times in the transient. The transient event editor allows the user to change time step at any time during the simulation. For the first part of the transient (to ~15s) a time step of 0.1 seconds is used from then to where the auxiliary cooling system starts a time step of 1 second is used and then 0.1s is again used for the auxiliary cooling. Table 3-39 includes the transient events as modelled in Flownet Nuclear.

Table 3-37. Primary Pressurized Water Cooler results comparison

<b>Parameter</b>	<b>HTTR Estimate</b>	<b>Flownet Nuclear – Pre-scribed Shaft speeds</b>	<b>Flownet Nuclear – 15 MW Shaft speeds</b>
Circulator speed [rpm]	9400	9400	8000
Helium inlet temp. [°C]	850	789.2	856.8
Helium outlet temp. [°C]	395	401	388.4
Helium pressure [kPa]	4000	3977	3984
Helium mass flow rate [kg/s]	8.3	10.45	8.54
Water inlet temp. [°C]	135	135	135
Water outlet temp. [°C]	175	177	177
Water pressure [kPa]	3500	3500	3500
Water mass flow rate [kg/s]	115	115	115

Table 3-38. Secondary Pressurized Water Cooler results comparison

<b>Parameter</b>	<b>HTTR Estimate</b>	<b>Flownet Nuclear – Pre-scribed Shaft speeds</b>	<b>Flownet Nuclear – 15 MW Shaft speeds</b>
Circulator speed [rpm]	8100	8100	8130
Helium inlet temp. [°C]	782	719.9	723.4
Helium outlet temp. [°C]	240	256	256
Helium pressure [kPa]	4100	4054.7	4054
Helium mass flow rate [kg/s]	3.6	3.87	3.88
Water inlet temp. [°C]	135	135	135
Water outlet temp. [°C]	175	171	171
Water pressure [kPa]	3400	3400	3400
Water mass flow rate [kg/s]	60	60	60

Table 3-39. Transient events as modeled in Flownet Nuclear

Simulated event	Time [s]	
	15 MW	30 MW
Coast down of the five main loop circulators commences. The speed of the circulators is hard-coded to be the same as the speed decrease prescribed by JAERI.	0	0
Water flow in the PPWC and SPWC is stepped down to 5 kg/s	0	0
The circulator models become unstable in Flownet due to operation far out of the circulator map's range. The circulators are replaced by mass flow sources that decrease at the same rate as the circulator mass flows did before 3.4 seconds	3.4	3.4
Reactor SCRAM is initiated. The decay heat curve is read as input into the reactor model.	8.2	8.2
The IHX circulator flow has reached 0 kg/s	12	10.8
The PPWC circulators have reached 0 kg/s	13.4	14.5
The SPWC mass flow has reached 0 kg/s	15	18
The auxiliary cooling system is activated	52	52

### ***15 MW Loss of power transient results***

Flownet Nuclear models circulators and other turbo-machinery by means of their characteristic curves or “maps”. Hence, in a situation such as the loss of power benchmark where the performance of the circulators determines the behavior of the rest of the system it is very important that the maps used are correct. In this case the speeds of the circulators are specified as input to the circulator models and an appropriate operating point is found on the maps. Figure 3.36 shows the speed of three of the circuit's circulators over the first part of the transient. Figure 3.37 shows a locus plot of the movement of the operating point of the PPWC circulator on its map. It is clear that during normal operation at 15MW power the operating point is already off the map. As speed is reduced the operating point moves toward the origin. It is for this reason that the circulator models become unstable after 3.4 seconds – the models cannot extrapolate far enough to simulate the performance of the machine.

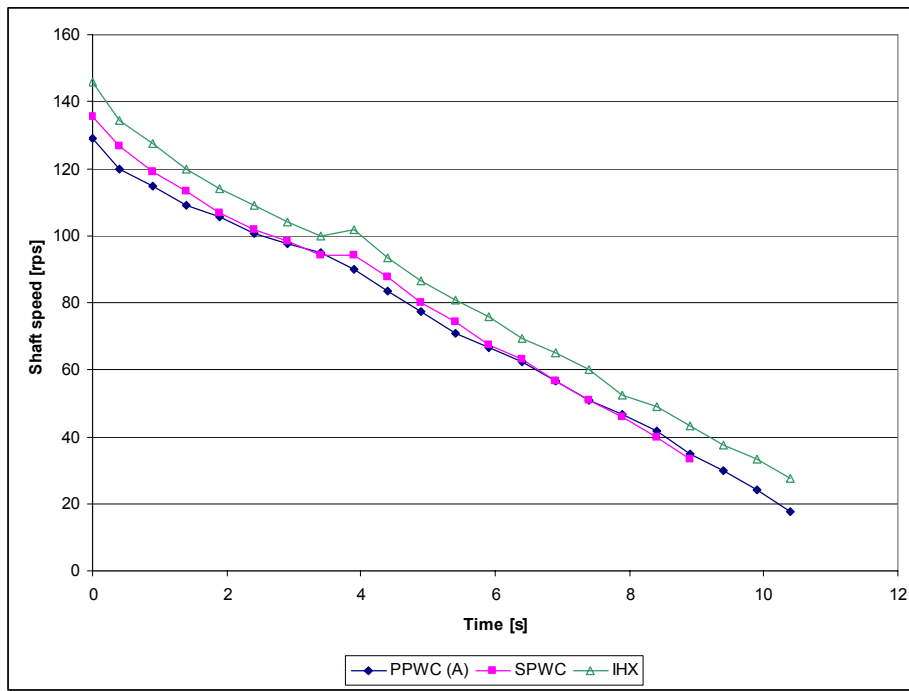


FIG. 3.36. Circulator shaft rotational speed – JAERI result used as input to Flownet Nuclear.

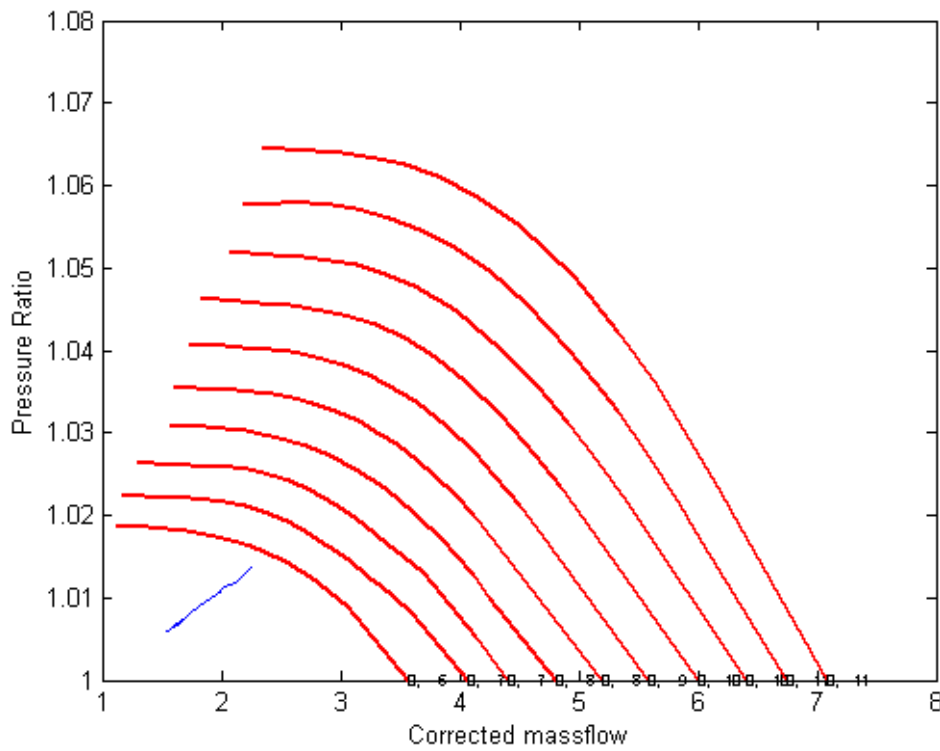


FIG. 3.37. Locus plot of circulator coast down on a Pressure ratio vs. corrected mass flow rate map.

The fact that the circulator is operating of the map has an effect on the mass flow that is produced. Figure 3.38 shows the simulation mass flows plotted against the test results. The figure shows that the simulation predicts a much faster decline in mass flow rate than what happens in reality. It is assumed that the form of the map in the inner quadrant is different to the actual map and as such actual coasting down cannot be accurately simulated.

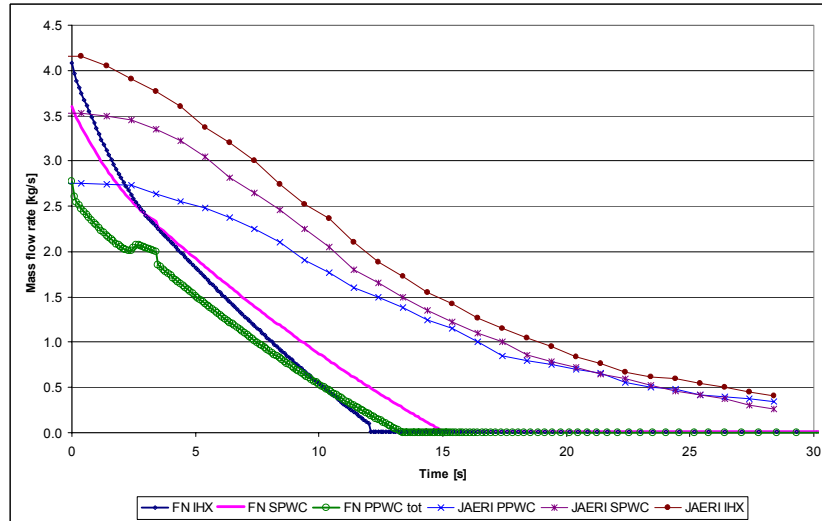


FIG. 3.38. Circulator mass flow rates of Flownet Nuclear compared to JAERI test results

The transient performance of the reactor is shown in Figure 3.39. Note the sudden drop in reactor power at 8 seconds due to the SCRAM. The remaining decay heat curve is as prescribed by JAERI. Figure 3.40 shows the transient fluctuation in system pressure. It is expected that once the system is not running the pressure will be close to the operating pressure due to the lack of a high system pressure ratio and large volumes of helium.

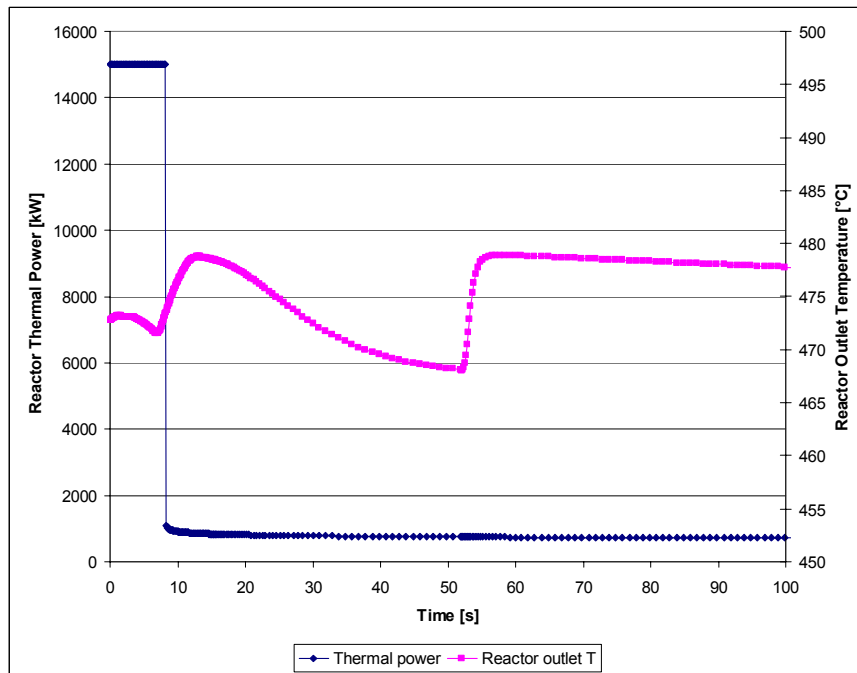


FIG. 3.39. HTTR Reactor thermal power and outlet temperature.

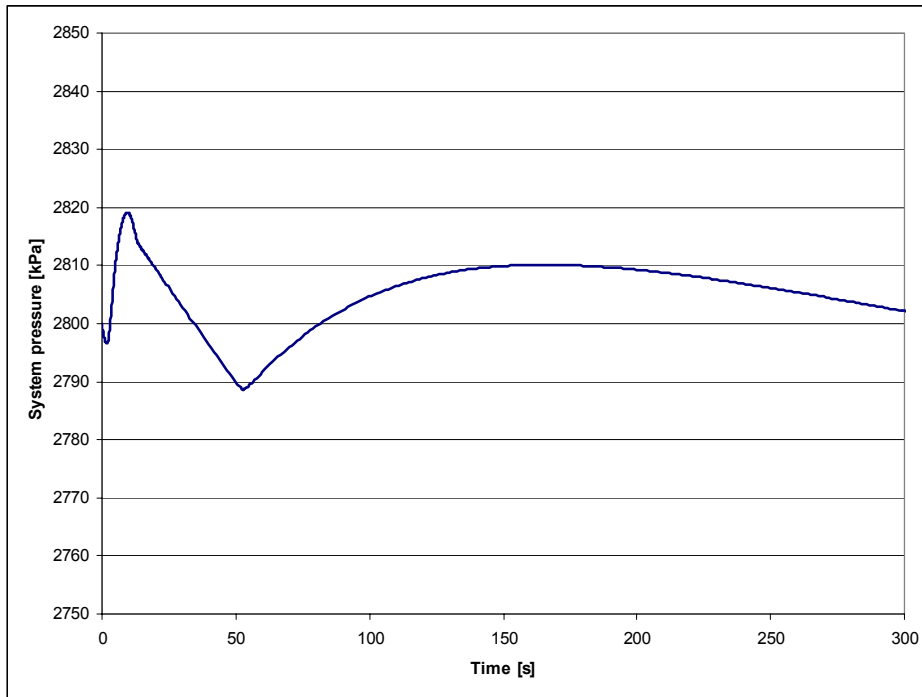


FIG. 3.40. HTTR system pressure (at PPWC circulator outlet).

In Figure 3.41, the mass flow rates are plotted for the whole transient from coast down to when the auxiliary cooling system is activated. Figure 3.42 provides the HTTR inlet and outlet temperatures while Figure 3.43 is a plot of the heat removed by the auxiliary heat exchanger throughout the loss of power transient from 15MW.

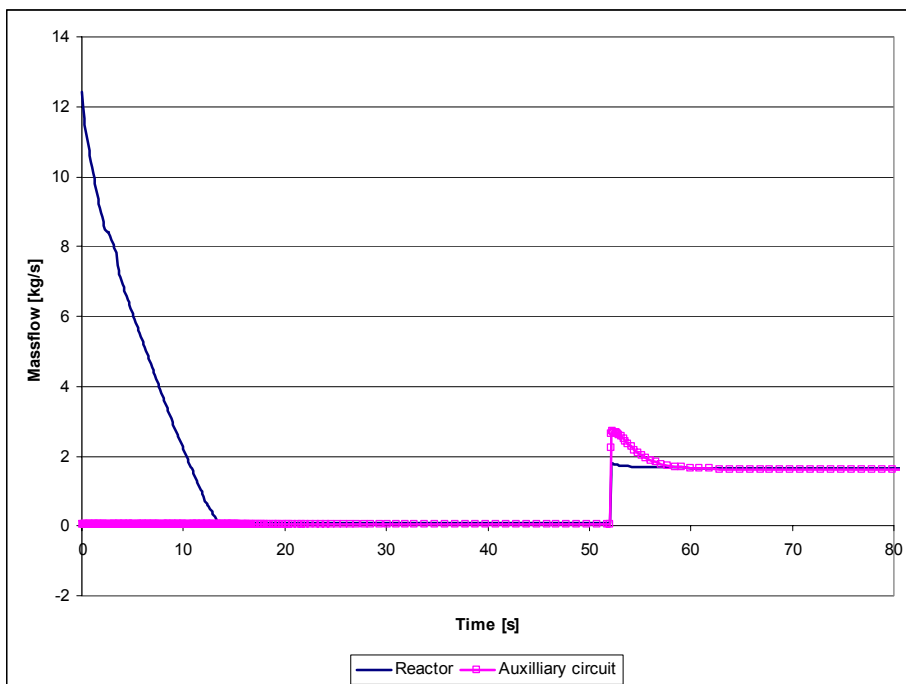


FIG. 3.41. Reactor and auxiliary mass flow rates.



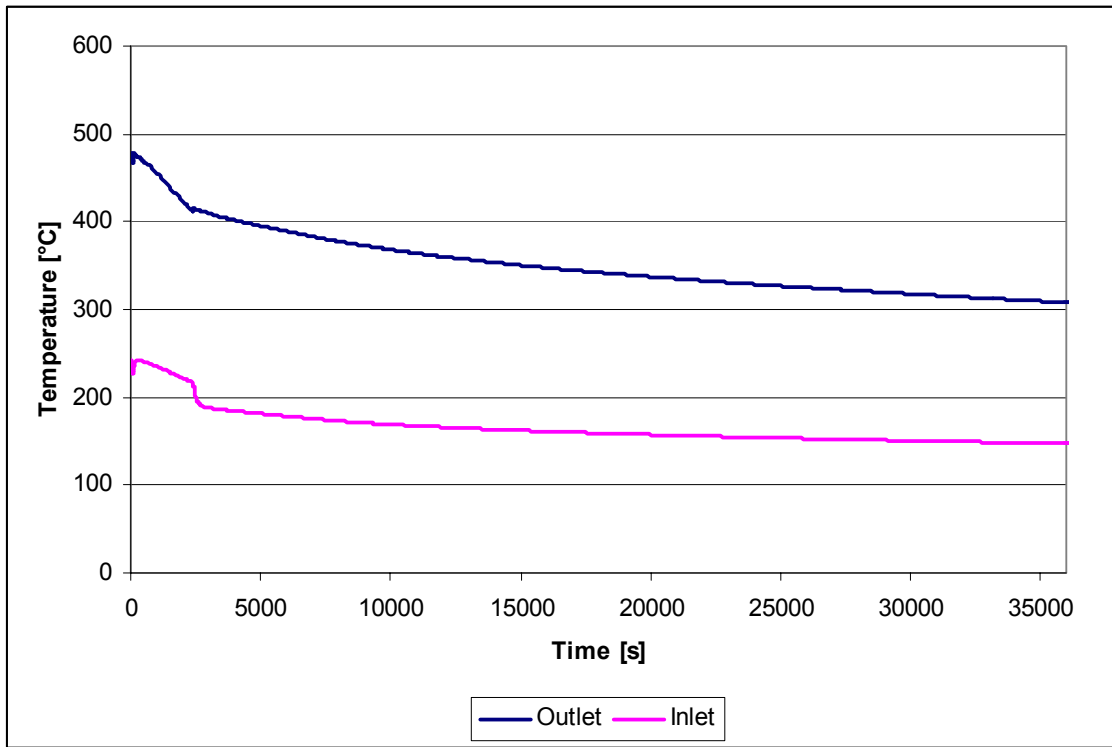


FIG. 3.42. Inlet and outlet temperatures of the HTTR core for the 15 MW loss of power transient.

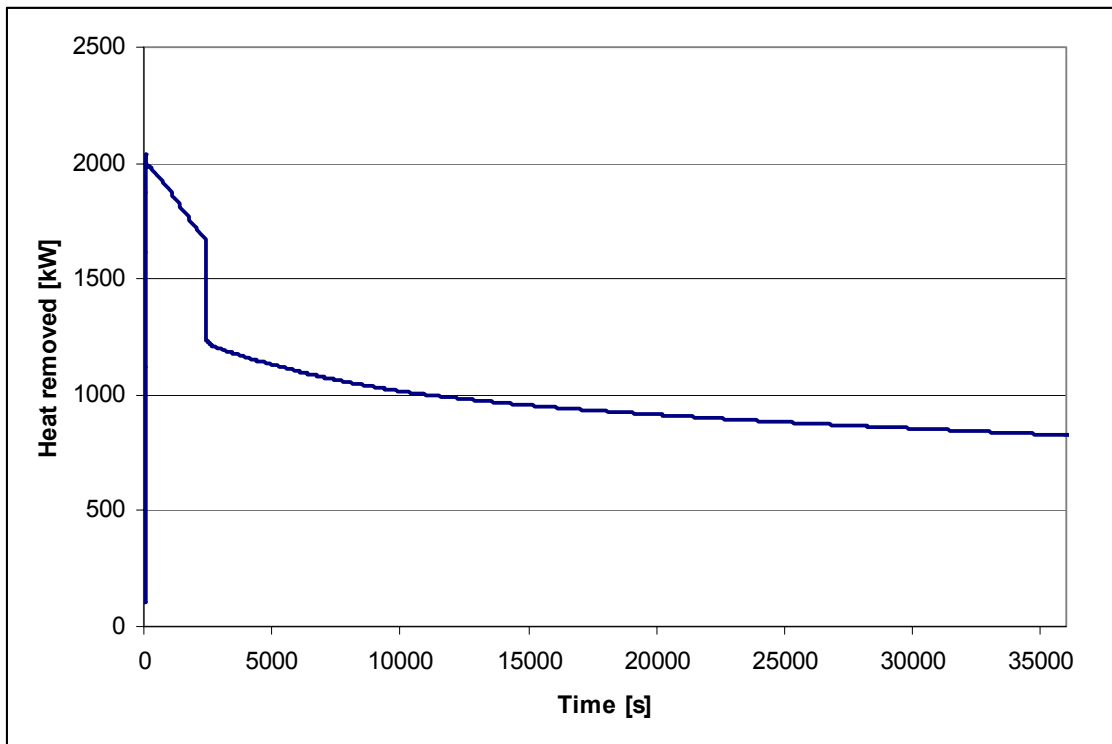


FIG. 3.43. Heat removed by Auxiliary heat exchanger for the 15 MW loss of power transient.

The assumptions described in this section are based on simulation judgment and as such the actual cause of discrepancies may be due to totally different reasons. The only way to really verify these assumptions is to get more experimental data from JAERI, specifically temperatures and pressures through the system as measured during the experiment.

### 30 MW Loss of power transient results

At the time of documenting these results no 30MW experimental data was available from JAERI. Therefore the 30MW simulation results are unverified but are still presented in this section. Figure 3.44 and 3.45 show the system mass flows during the coast-down and entire periods of the transient respectively. As with 15MW it is expected that the actual mass flow rate decrease will slower than in the simulation.

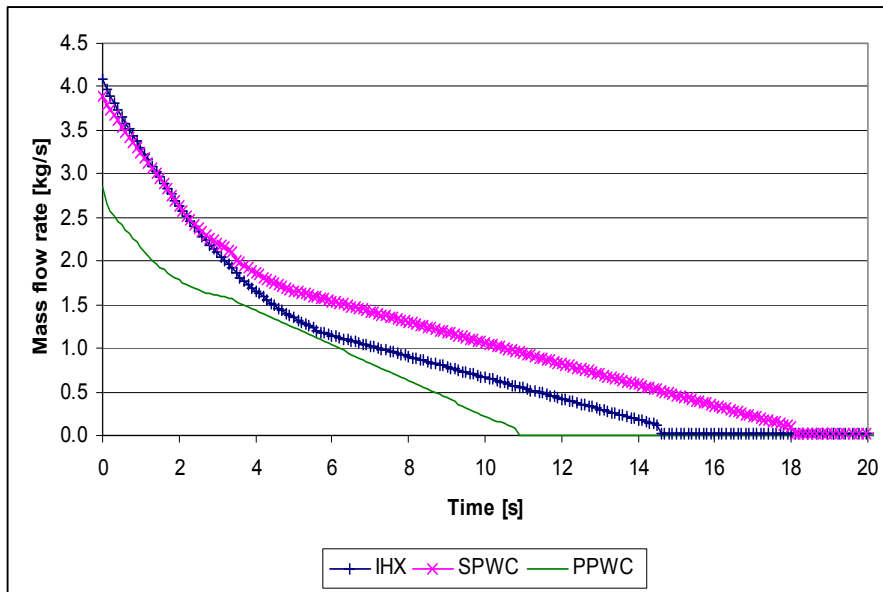


FIG. 3.44. Circulator mass flow rates in coast-down phase.

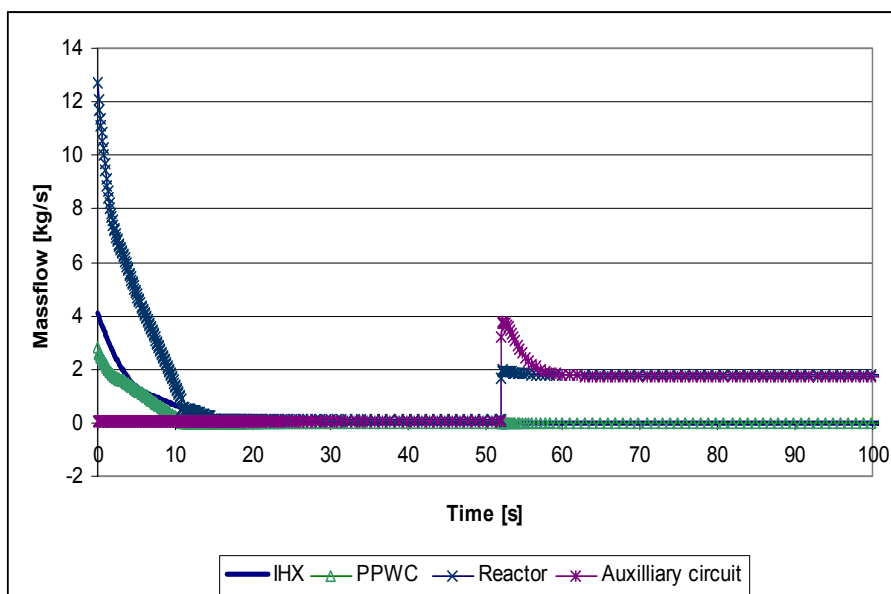


FIG. 3.45. Reactor and auxiliary mass flow rates.

The reactor performance is shown in Figure 3.46 while the change in system pressure is given in Figure 3.47.

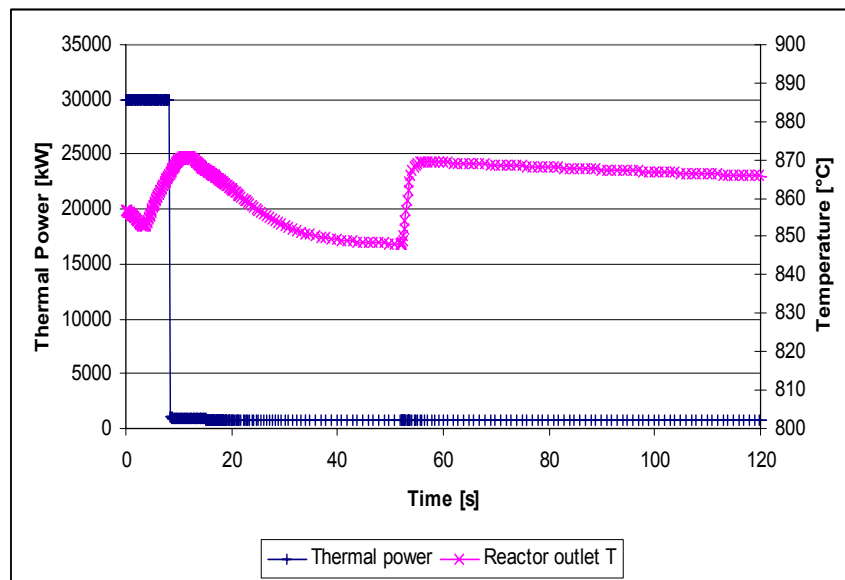


FIG. 3.46. HTTR Reactor thermal power and outlet temperature.

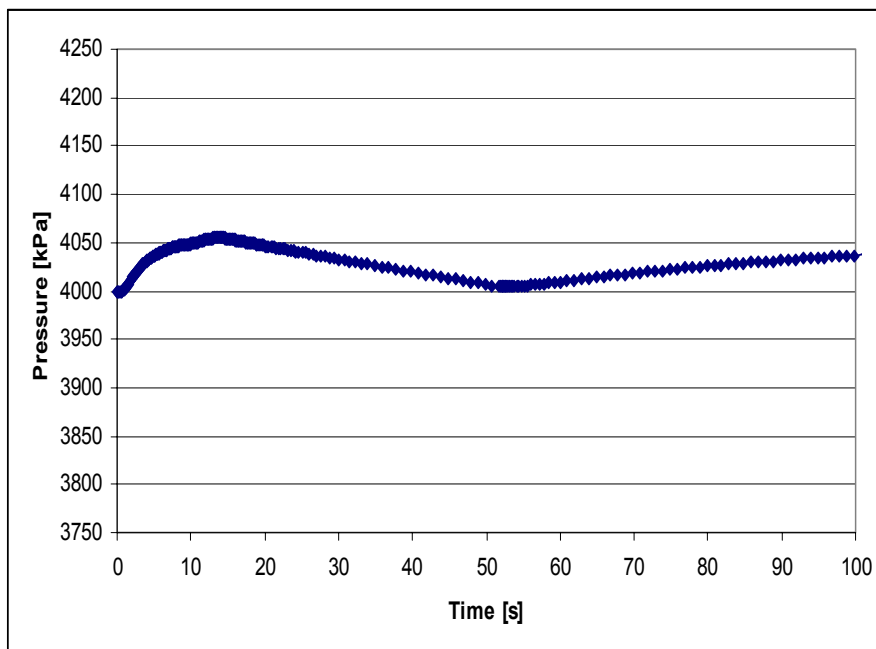


FIG. 3.47. HTTR system pressure (at PPWC circulator outlet).

Figures 3.48 and 3.49 provide HTTR inlet and outlet temperatures and the heat removed by the auxiliary heat exchanger throughout the loss of power transient from 30MW, respectively. Figure 3.50 shows a locus plot of the PPWC circulators' operating point movement on the map. Due to the same steady-state speed being used as for 15MW the starting point is off the map. Hence the same conclusions can be drawn as for 15MW.

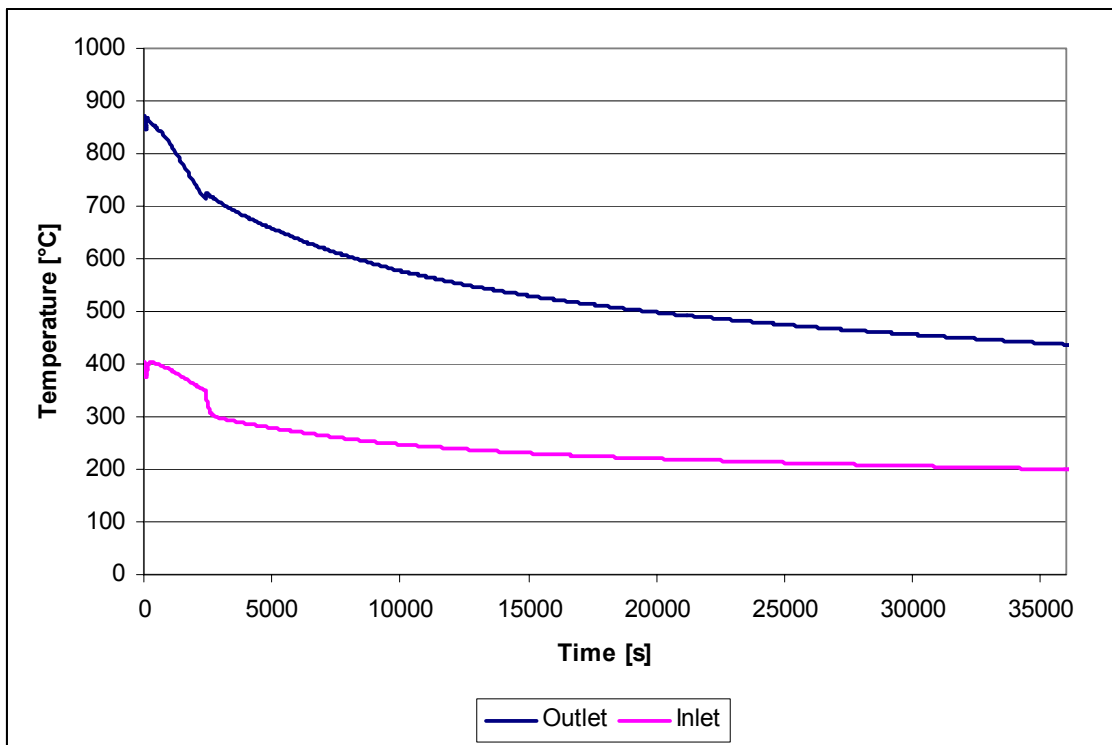


FIG. 3.48. Inlet and outlet temperatures of the HTTR core for the 30 MW loss of power transient.

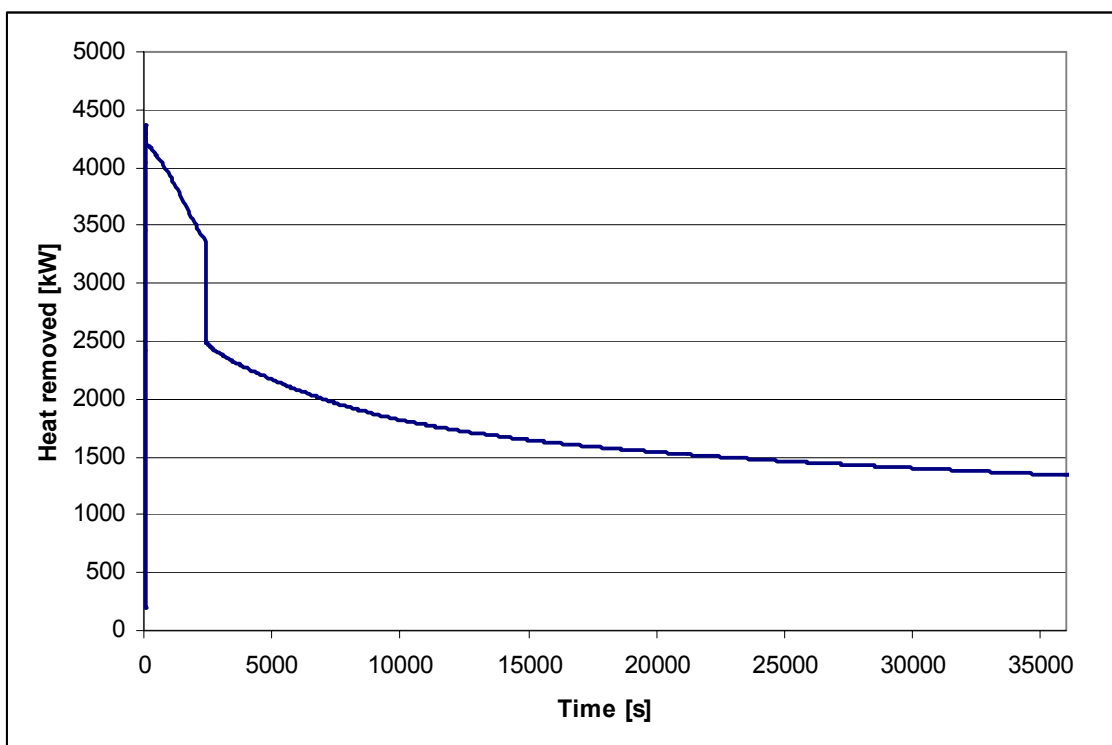


FIG.3.49. Heat removed by Auxiliary heat exchanger for the 30 MW loss of power transient.

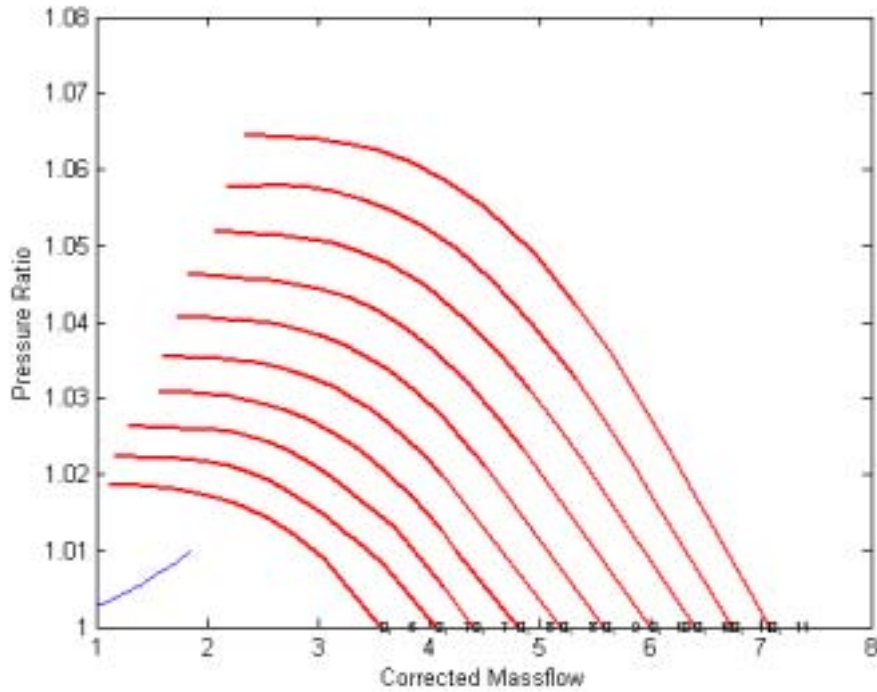


FIG. 3.50. Locus plot of circulator coast down on a Pressure ratio vs. corrected mass flow rate map.

As for 15MW, the assumptions made in the 30MW transient can only really be verified with more detailed transient data from JAERI.

### 3.2.3.6 Conclusions

The steady-state results obtained from Flownet Nuclear for the 15MW case showed good correlation to the experimental data when the circulator speeds were lowered to the actual values. It appears that original speed estimates provided by JAERI were based on over-conservative system resistance predictions. This resulted in the circulators being oversized for the required operating point. The working points are just off the circulator map ranges and as such the accuracy of the analyzed working point is dependent on the extrapolation routines in the simulation code. This accuracy becomes more evident in the loss of power event where the working point moves further away from the map. The effect can be seen in the mass flow plots where the experimental flow rates take longer to decrease than the simulated results.

However, other factors cannot be ruled out as temperature and pressure effects can also play a role in the discrepancies. The only way to verify that they are not the cause is to get more detailed pressure and temperature data of the HTTR system for the loss of off-site power experiments.

In general, the steady state simulation results and those obtained from the actual plant are in fair agreement. The transient results show greater deviations but the trends simulated are still the same as the actual performance.

### **3.2.4. United States of America [3-14]**

#### *3.2.4.1. Analysis methodology and code description*

The ORNL Graphite Reactor Severe Accident Code (GRSAC) was used for the IAEA CRP-5 HTTR-VC and HTTR-LP benchmark problems, utilizing both steady state and dynamic code features. These calculations relate to the HTTR initial rise to power sequence and safety demonstration tests. An existing HTTR model in GRSAC [3-15] was upgraded to provide more detail in certain critical areas.

The GRSAC (Graphite Reactor Severe Accident Code) software is a general-purpose program developed at Oak Ridge National Laboratory (ORNL). It is based on the ORNL MORECA code for simulating accident scenarios for selected gas-cooled reactor (GCR) design types [3-13]. The MORECA code and its predecessors were originally developed at ORNL under the sponsorship of the U.S. Nuclear Regulatory Commission (NRC) to perform confirmatory licensing-related studies of a variety of High-Temperature Gas-Cooled Reactor (HTGR) designs, including the Fort St. Vrain HTGR and subsequently the 350-MW(t) steam-cycle Modular HTGR (MHTGR). MORECA was later developed - under U.S. Department of Energy (DOE) sponsorship - to simulate the MHTGR design for the 600 MW(t) direct cycle gas turbine modular helium reactor (GT-MHR).

Since MORECA is a "hard-wired" code, configured only for a particular reactor design, the conversion of MORECA to GRSAC was motivated by the need to generate the connectivities necessary to assemble, verify, and run simulations for a wide variety of graphite-moderated GCR designs.

Since GRSAC was developed to study a wide variety of core transient and heatup accident scenarios, it includes a detailed 3-D thermal-hydraulic model for the core, plus models for the reactor vessel, shutdown cooling system (SCS), and shield or reactor cavity cooling systems. There is an option to include neutronics (point kinetics) with xenon and samarium poisoning to study Anticipated Transients Without Scram (ATWS) transients.

The 3-D, hexagonal geometry core model allows for detailed investigations of azimuthal temperature asymmetries in addition to axial and radial profiles. Variable core thermal properties are computed functions of temperature and are dependent on orientation and radiation damage. An annealing model for graphite accounts for the increase in thermal conductivity that occurs during heatup accidents.

The primary coolant flow models cover the full ranges expected in both normal operation and accidents, including pressurized and depressurized accidents (and in between), for forced and natural circulation, for up-flow and down-flow, and for turbulent, laminar, and transition flow regimes. The primary loop pressure calculation can consider variable inventory (due to depressurization actions) and loop temperature changes and may use a simplified model for balance-of-plant temperatures. The models for the reactor pressure vessel and the shield or cavity cooling system are different for each of the various basic reactor models. Models are also included for oxidation of graphite (including carbon deposits), cladding, and fuel. Time-at-temperature failure models for TRISO fuel, fission product release models for metal fuel, and Wigner stored energy release models are also available.

Other GRSAC features of interest are: fast-running (typically >2000 times faster than real time on a SUN SparcStation-20 workstation), interactive user interface with on-line and off-line plotting options, automated sensitivity study capabilities, and on-line documentation and help screens. The basic designs that can be simulated using GRSAC, which the user may modify via the interface to a large (but limited) extent, include the British and French Magnox types (including the Calder Hall, G-2/3, and Bugey-2), Windscale (U.K.), G1 (France), and the HTTR (Japan). Adaptations and analyses are planned for pebble-bed reactor designs, and the GT-MHR Plutonium burner (U.S.-Russia). More details on GRSAC are summarized below, and [3-15] is a complete users manual.

### ***GRSAC code features***

#### Reactor Design Setup

Specific design features for a chosen reactor type can be input by the user via design screen selections in the following categories: fuel element, nuclear parameters, core layout design, primary cooling system, vessel design, reactor cavity, fission product release, and oxidation parameters. Program setup screens allow the user to activate or deactivate oxidation, Wigner energy, or ATWS features, and to select the coolant gas, core flow direction and computation time parameters. In some cases, such as for the radial and axial power peaking factor inputs and flow coast-down curves, graphical displays and automated consistency check features are included. Pop-up HELP windows and a choice of metric or English unit entries are available for all user input screens in GRSAC. The user can also select a "run with validation" option, which is a smart front-end check of the entire set of inputs for data inconsistencies.

#### Initial Condition Runs

GRSAC accident sequence analyses require a large set of initial condition values which are created automatically via the Initial Condition (IC) mode. The user can change operational inputs such as power level, flow, pressure, etc., and observe the resulting detailed temperature and flow distributions attain steady state conditions. At any point in the run, one can store initial condition values in a RUN file.

#### Interactive and Programmed Inputs

The interactive input screen for accident simulations allows for user inputs (scram, depressurization, changes in emergency and/or cavity cooling, etc.) at any time during a run. Such inputs can also be pre-programmed, however, via a programmed input screen that is available to the user during the run setup procedure.

#### Accident Sequence Runs

Long-term Loss Of Forced Convection (LOFC) accidents begin with a programmed flow coastdown transient. LOFC transients in GCRs are generally characterized by slow heatups due to low power densities and large heat capacities associated with the core. They may be simulated both with and without total or partial depressurization of the primary coolant and with or without scram. Optionally, both the active or passive shutdown cooling systems can be made to be either unavailable or available only intermittently in degraded states. For helium or CO<sub>2</sub>-cooled cores, there is an option to allow air ingress following a

depressurization, and subsequently to initiate oxidation models for graphite (and clad and metal fuel, if applicable).

### Sensitivity Study Option

Many variations of transient and LOFC accident scenarios have been studied to observe the sensitivities of the predictions to parametric model and operational assumptions. These provide guidance in design studies for determining plant operating parameters (including design power level) and in identifying which physical properties and correlations are most crucial to the outcome of postulated accidents.

In the GRSAC automated sensitivity study feature, the rationale is to seek out a set of parameters within user-specified uncertainty bands that result in the worst (or best) case accident consequences using a gradient search algorithm. Sets of 13 model or design parameters (such as heat transfer correlations, etc.) and 12 operational/run parameters (such as time of scram) have been set up to be available for automatic variation (from run to run). The program allows the user to select up to 10 from this set for any given study. To study the effects of a single parameter variation in more detail, a single-parameter option can be used. That parameter is varied uniformly within the uncertainty band (reference run plus 4 others). A report generator creates a summary of the sensitivity run results.

### ***GRSAC model of the HTTR***

The ORNL GRSAC code has a number of general-purpose features that allow modification of design attributes and operational sequence options via a variety of user interface screens. However, to accommodate HTTR calculations for previous IAEA CRP-3 benchmark calculations and for activities during a JAERI Foreign Researcher Inviting Program [3-16], special models were added for HTTR-specific features not readily changed via the input screens. These included a special model for the vessel cooling system (VCS or Reactor Cavity Cooling System - RCCS), modeling of the core inlet flow path up the annulus between the core and reactor vessel, accommodation of the smaller core/larger side reflector design than is standard in GRSAC, and inclusion of an annealing model to account for thermal conductivity changes in the higher grade graphite (GRSAC reference case is for Magnox reactor type cores).

The RCCS (VCS) model used by ORNL previously for the HTTR-RCCS experiment IAEA benchmark calculations (with success) was adapted to the actual HTTR VCS. Preliminary calculations showed somewhat higher than expected heat removal rates. Sensitivity studies (also preliminary) showed the predicted rates to be very sensitive to the assumed emissivity values for the radiation shields, as would be expected. Also, it was not clear how to estimate convection heat transfer (if any) between the radiation shields and the cooling panel itself. The GRSAC model used for convection heat transfer (Keyhani) predicted an increase in effective  $h$  due to the addition of separation baffles in the vessel-to VCS annulus, which were installed to break up the convective flow path and reduce the heat transfer coefficient value.

A significant improvement in the GRSAC model for the HTTR VCS was made for the CRP-5 benchmarks on the basis of a study done by Akio Saikusa while at ORNL (1998-9) as a visiting JAERI Research Fellow. Using the PATRAN P/Thermal code system, he created a very detailed model of the upper vessel region, including the individual control rod drive standpipes, and the upper portion of the VCS. The objective was to derive an accurate



“effective view factor” for this complex geometry that could be used in the GRSAC VCS-vessel model. This effort was successful, and the results were incorporated.

#### 3.2.4.2. Results of HTTR-VC

##### ***Previous V&V calculations of HTTR vessel cooling (VC) performance***

The ORNL GRSAC code was used previously to predict steady state core and VC heat removal conditions for initial power levels of 10%, 30%, 50%, 75%, and 100%. The GRSAC calculations were benchmarked against the JAERI code ACCORD [3-16] with the results generally in very good agreement. For this exercise, the axial peaking factor curve was modified from the IAEA case to correspond to the no-burnup profile that would apply to the initial rise to power. The 100% reference case core flow distribution was also modified so that 9.8% of the total flow of 12.51 kg/s would bypass the active core.

The model of the fuel pin accounts for the outer sleeve, gap, and hollow-center fuel geometry. The gap model was revised to give a reference case 0.25 mm gap with helium, where the helium conductivity is included as a function of temperature, along with radiant heat transfer also across the gap.

The GRSAC special model of the HTTR core consists of 61 radial regions (representing the HTTR’s 954 channels). The balance of GRSAC’s 205 radial regions represent the removable and permanent side reflectors. There are 14 axial core nodes, where the middle 10 represent the active core. The 10 axial peaking factors (inlet to outlet) are: .5373 .6545 .8043 .9313 1.241 1.286 1.471 1.293 1.136 .6447

These values are very close to those derived from the CRP-5 VC and LP benchmark descriptions, and are used in those calculations as well. The core model was also modified so that the control rod channels are accounted for explicitly. Hence the axial flow modeling in the 61 representative (GRSAC) active core model regions are no longer assumed to have equal orificing characteristics. The 61 regions represent 37 assemblies (30 fuel elements and 7 control rod drive [CRD] elements), so some of the 61 regions partially represent the CRDs. The center radial node in GRSAC is “100% CRD channel,” but the next 3 rings of radial nodes contain a mixture of CRD and fuel channel elements. The fourth ring consists of only fuel element channels. To estimate the maximum temperatures in rings 2-4, the hot-channel option in GRSAC is used, which allows one node in each of these rings to have an increased peaking factor to represent a typical region in this ring where there is no CRD element.

##### Reference case V&V results - calculations

In Table 3-40 below, for each power level case, the predicted core mean coolant outlet temperatures (one for the active core only, and the other with side reflector coolant mixed in – w/SR) are given, along with the maximum gas outlet (T-o max) and fuel temperatures (T-f max). The maximum predicted fuel temperatures do not occur in the center radial region, as they would in a homogeneous core representation. The peak fuel temperature for the 100% power case appears in the outlet axial region for the active core (AR#12); for all other cases, it is in AR#8 (just below mid-core). The maximum vessel temperature (T-v max) and rate of heat removal from the vessel cooling system (P-vcs, in kW) are also shown (all temperatures in degree C).

Table 3-40: VC Reference Case Calculations as a Function of % Power

<b>%P</b>	<b>T-o active, w/SR</b>	<b>T-o max.</b>	<b>T-f max [J]</b>	<b>T-v max.</b>	<b>P-vecs</b>
10	184-180	186	207 [179]	120	100
30	328-315	334	387 [375]	159	180
50	488-468	498	570 [560]	210	280
75	687-658	701	785 [788]	273	460
100	884-848	900	993 [1017]	336	650

The corresponding JAERI values (using the ACCORD code) for maximum fuel temperatures for these cases are shown in the T-f max column in brackets [ ]. Considering the differences in codes and many modeling assumptions, the results are in very good agreement, the GRSAC temperature estimates being slightly lower for the full power case.

#### Sensitivity study

Several variations in parameters and assumptions were made to determine the sensitivity of the predicted results to uncertainties. Since the effects would be most pronounced at 100% power, they were run for that case. Two factors that could affect the flow redistribution in the active core are the friction factor and pressure drop of the “orificing.” A 50% increase in the assumed friction factor had no effect on flow redistribution or resulting predicted temperatures. A 100% increase in the assumed core pressure drop (from 0.12 kg/cm<sup>2</sup> to 0.24 kg/cm<sup>2</sup>) also had a negligible effect on gas and fuel temperatures.

An assumed 25% increase in the effective core heat transfer coefficient had some effect on maximum fuel temperatures (the lower boundary of the given h-correlation curve was used in the reference case). For this case, the maximum fuel temperature in axial region 12 was ~ 6°C lower.

The nominal value of gap resistance was increased by 50%, increasing the maximum fuel temperature in axial region 12 by ~ 25°C.

A significant effect in predicted active core temperatures was seen for changes in the assumed core bypass flow percentage (vs. the reference 9.8%). For a bypass of 5.3%, the mean outlet gas temperature was 864 with a maximum of 873, and the maximum fuel temperature (region 12) was ~ 27°C less. For a bypass of 12.3%, the mean active core outlet gas temperature was 897°C with a maximum of 914°C, and the maximum fuel was ~ 15°C more than for the reference case.

#### HTTR-VC V&V results – experimental

Table 3-41 provides a comparison of steady state VCS measurements during the 1997 heat-up tests (circulator power was the heat source) with the reference model predictions:

Table 3-41: Measured vs. Predicted VC Results in kW

<b>Run Time</b>	<b>Measured kW</b>	<b>Predicted kW (GRSAC)</b>
1997/9/10 9:00	99	93
1997/9/11 13:00	85	98
1997/9/15 12:00	130	98

In the second case, the vessel temperatures are higher than those in the first, so the model predicts a higher heat transfer rate. In the third case, the adjustment panel flow is “on” and additional heat is removed. The model prediction does not include adjustment panel cooling (39 kW measured).

There was a large uncertainty in the active core bypass flow in Fort St. Vrain, with estimates up to about 15%. One reason for the large uncertainty was the variability due to the positioning of the orifices controlling flow in each of the refueling regions (not a consideration for HTTR). However, some uncertainty should be accounted for in the HTTR.

The cooler bypass gas exiting the side reflector and CRD channels mixes with the hot gas from the active core fuel regions - and the differences in these are larger for the higher bypass flow assumption cases. It may be possible to infer bypass flow percentages from outlet temperature readings, and perhaps detect hot plenum mixing characteristics from fluctuations in mixed (average) gas outlet temperature readings.

### ***Steady-state benchmark calculations (HTTR-VC) for evaluation of vessel system performance***

The GRSAC code was run in the initial condition (steady-state) mode to determine VCS heat removal rates and resulting vessel temperatures. For the 100% (30 MW) power case, the predicted VCS heat removal rate was 0.670 MW. In this case the maximum vessel temperature was 333°C, and the average temperature of the vessel top head was estimated at 203°C.

GRSAC was also run at 50% power (15 MW), giving a VCS power of 0.285 MW, maximum vessel temperature of 209°C, and a peak top head temperature of 128°C. For this case, the reactor inlet temperature was assumed to be 241.4°C.

#### ***3.2.4.3. Results of HTTR-LP***

### ***Transient benchmark calculations for loss of off-site power (HTTR-LP) with scram and post-trip cooling via the auxiliary cooling system***

The GRSAC HTTR model was run in the accident mode for the LP benchmark, representing a loss of off-site power (LOSP) event. The LP benchmark problem involves a main circulator trip and flow coastdown, with a reactor trip (scram) and startup of the Auxiliary Cooling System (ACS) shortly thereafter. The performance of the ACS for the LP benchmark is approximated by running GRSAC’s simplified Shutdown Cooling System (SCS) model with the prescribed helium cooling flow-rates (1.2 kg/s for the first 40 minutes and 0.8 kg/s thereafter). An SCS cooling water flow of 18.3 kg/s is maintained throughout. Primary pressure calculations are very approximate, since the HTTR balance of plant (BOP) is not modeled explicitly. However, the results of the temperature predictions are not very sensitive to pressure, and the GRSAC estimates of pressure changes were fairly close to those shown in the JAERI results.

The results of the LOSP from 100% power are shown in Figures 3.51 and 3.52. They show a gradual decline in core temperatures, with the rate of decline slowing somewhat after the decrease in coolant flow after 40 minutes, as expected. In the GRSAC figures, the Auxiliary Cooler heat removal rate (requested in the benchmark) is approximated by the core heat removal rate (curve labeled “Primary”) in the figure.

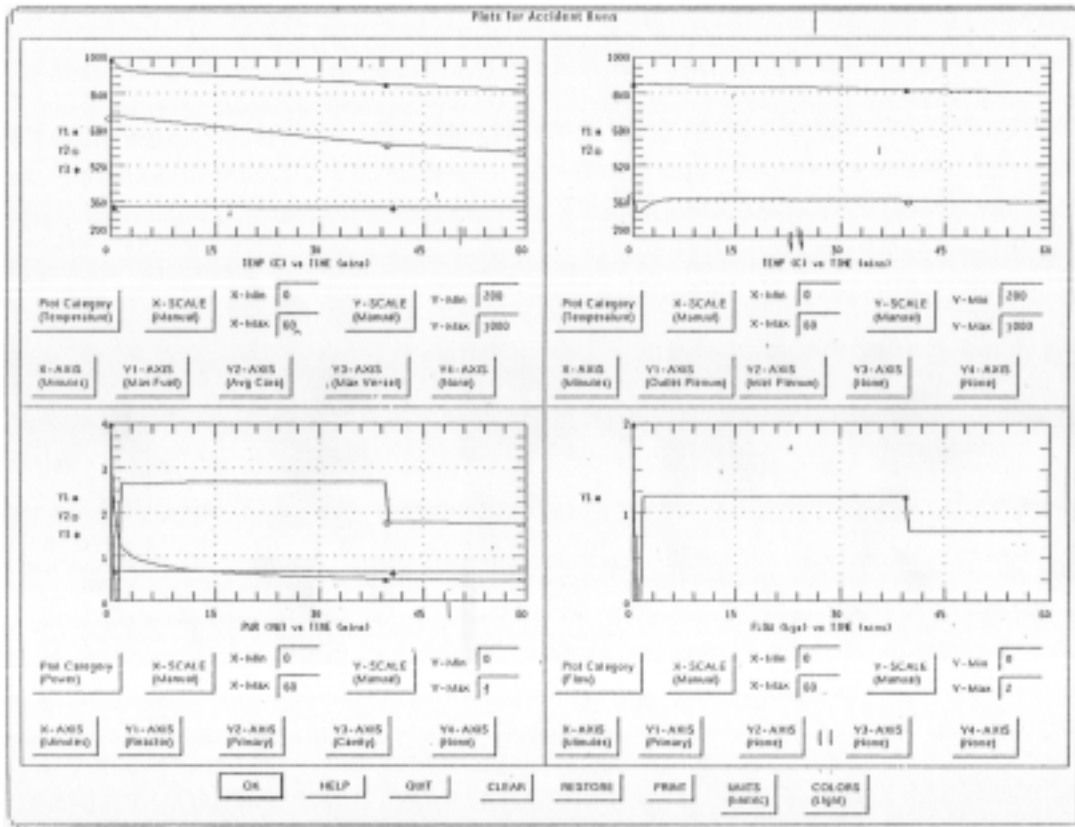


FIG. 3.51. HTTR-LC Benchmark: LOSP from 100% Power (1 hr).

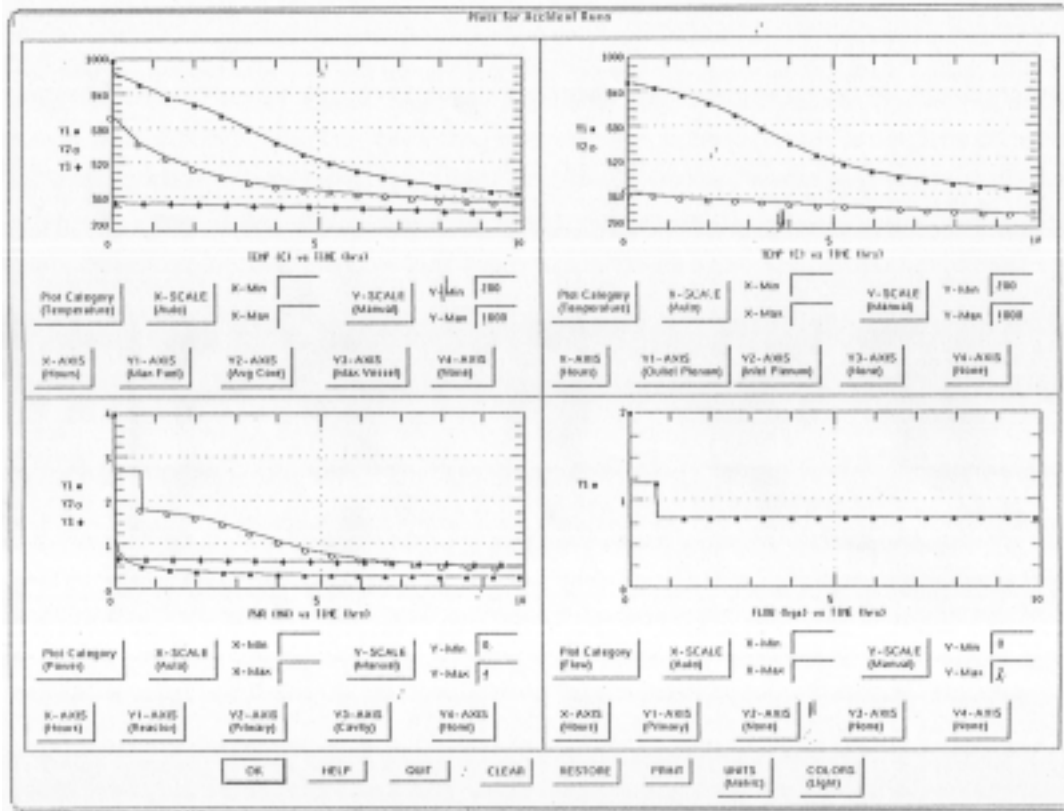


FIG. 3.52. HTTR-LC Benchmark: LOSP from 100% Power (10 hr).

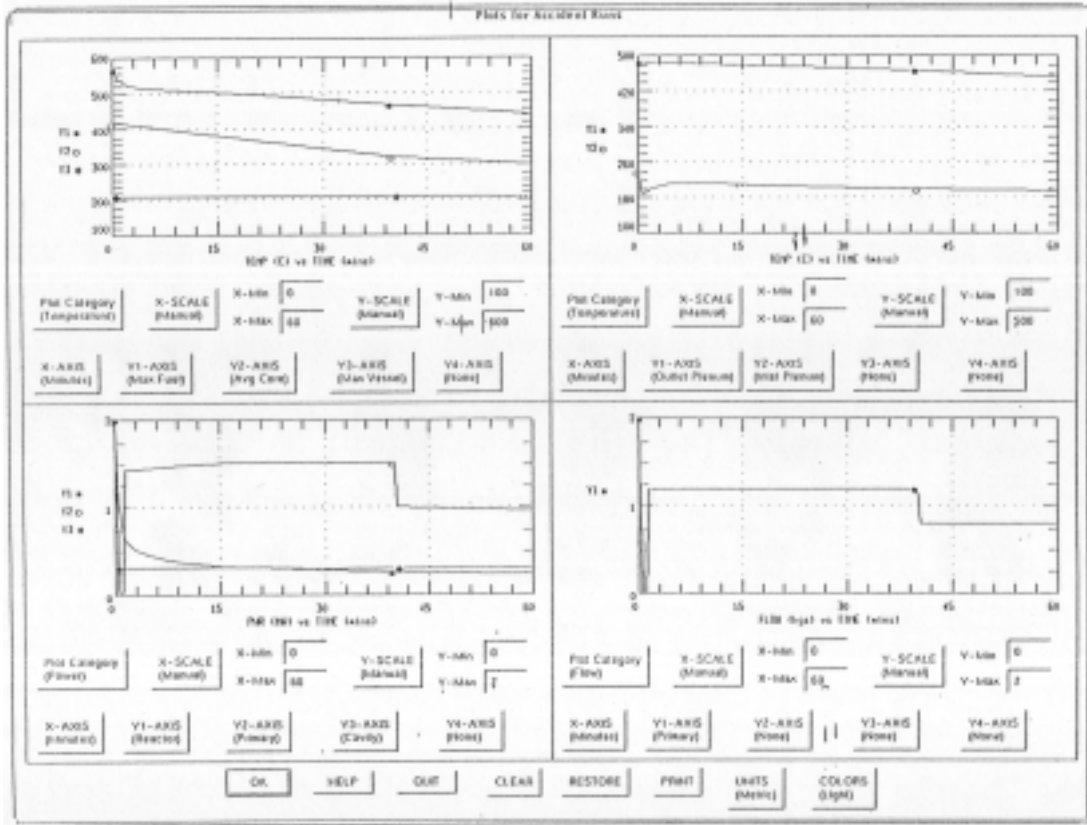


FIG. 3.53. HTTR-LC Benchmark: LOSP from 50% Power (1 hr).

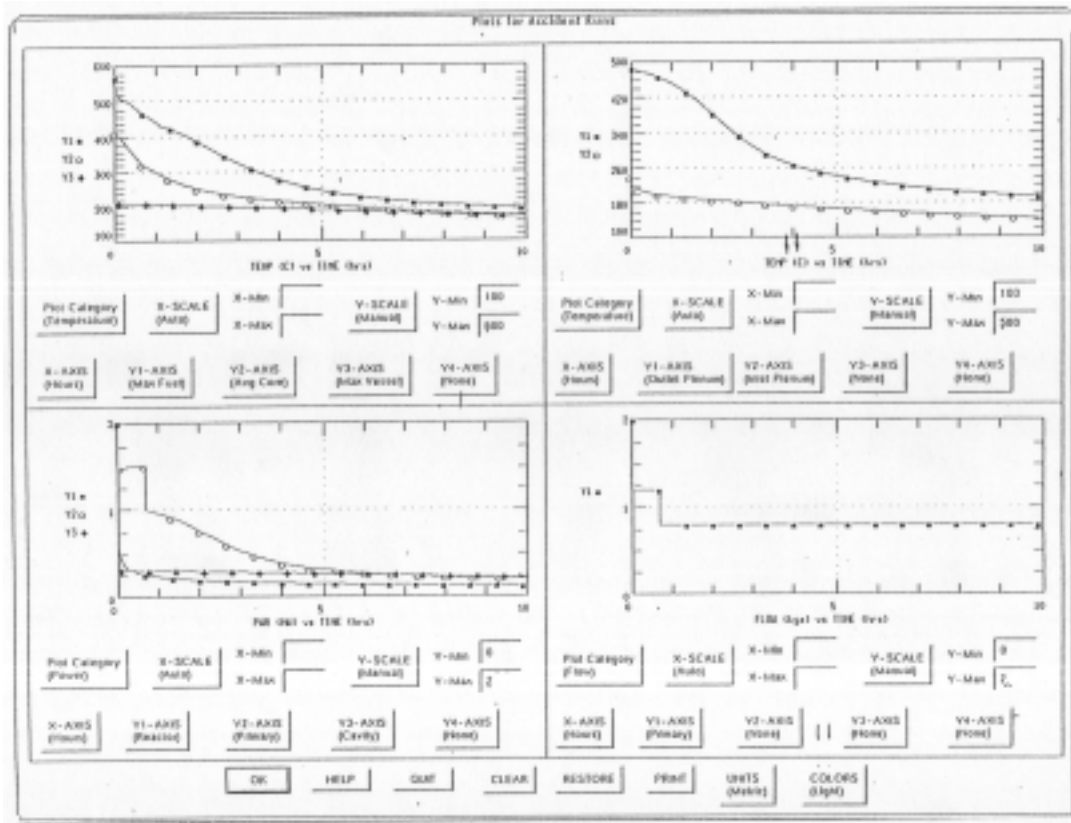


FIG. 3.54. HTTR-LC Benchmark: LOSP from 50% Power (10 hr).

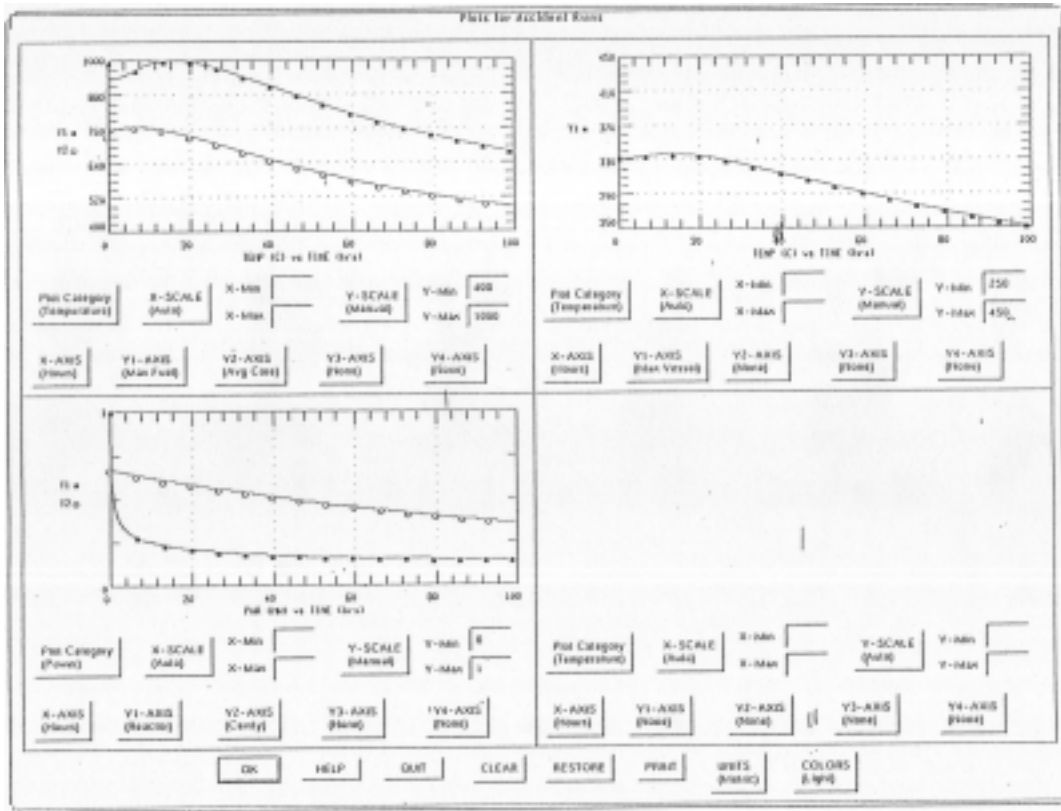


FIG. 3.55. HTR LOSP with no auxiliary cooling flow (pressurized).

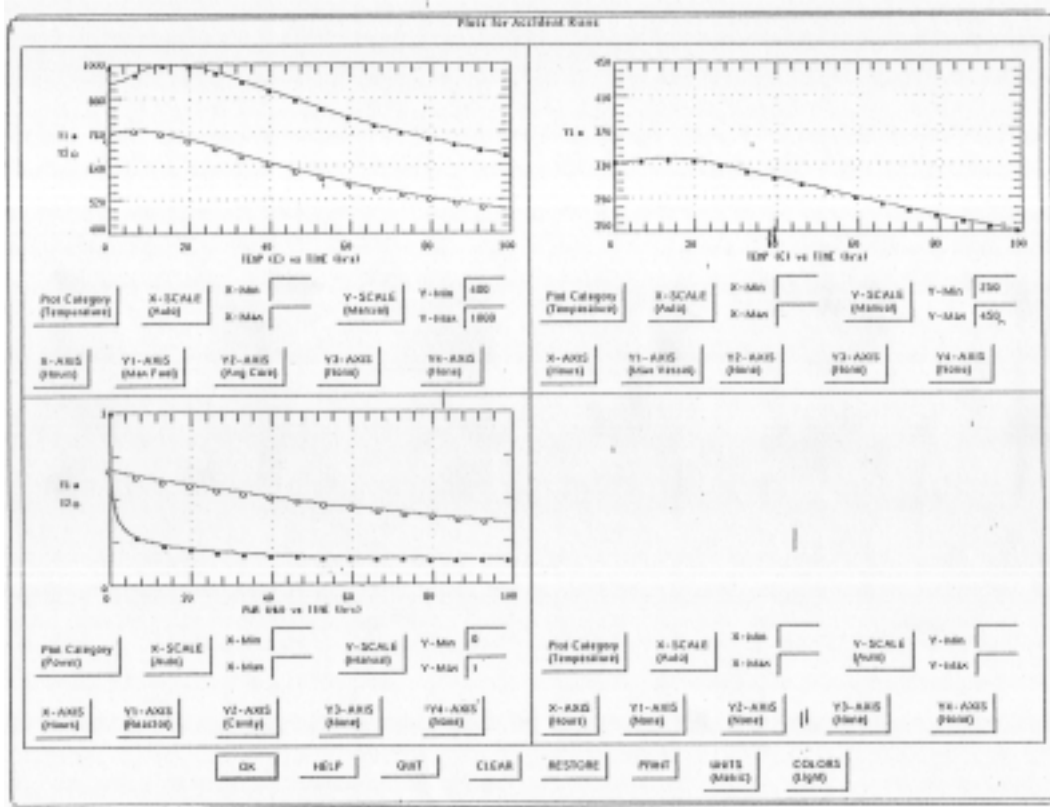


FIG. 3.56. HTR LOSP with no auxiliary cooling flow (depressurized).

The results of the LOSP from 50% power are shown in Figure 3.53 and 3.54. They also show a gradual decline in core temperatures, at a slower rate than the first case since the Auxiliary Cooler is removing less heat. Both sets of results appear to be in good agreement with the JAERI results published in the benchmark problem descriptions.

Figures 3.55 and 3.56 provide the results of HTTR loss of electric power even with no auxiliary cooling flow in the pressurized and depressurized condition, respectively.

It can be seen that no temperature excursions are expected, and the cool-downs proceed in an orderly fashion according to both GRSAC and JAERI (ACCORD) predictions.

### 3.2.5. France

A model of the High Temperature Engineering Test Reactor was developed for the benchmark concerning the evaluation of the Performance of Vessel Cooling System during normal operation. The CAST3M code [3-17] has been used to model the HTTR. CAST3M is a multi-purpose finite element code developed at CEA, which allows 3D, 2D and R-Z axisymmetric calculations. Domains of applications are structural mechanics, fluids mechanics heat transfer and hydro-magnetics. As far as heat transfer is concern, the code is able to take into account radiation phenomenon.

#### 3.2.5.1. Relevant Hypothesis and Limits of the Model

##### *The geometry*

Considering on the one hand the design geometry and its symmetry and on the other hand, the boundary conditions, a R-Z model has been chosen to describe the problem as shown on Figure 3.57. The heat transfers in the core and the coolant channels are supposed to be approximately 2D axisymmetric.

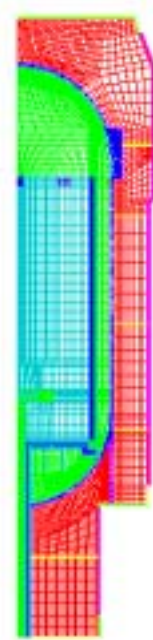


FIG. 3.57. 2D-reactor mesh.

#### Main correlations used for the reactor modelling

Helium hydraulics and heat transfer in the reactor core as well as in the reflectors, the bottom replaceable reflector block and the hot plenum, are modelled using an equivalent **porous media formulation** comprising a solid phase and a gaseous phase (Figure 3.58). Indeed, the number of helium channels per blocks as well as the number of fuel holes disables to use a complete description of the geometry that would be required for solving the convection equation in the coolant holes and the conduction and radiation equations in the graphite and in the gas.



In the porous medium, the relation between the pressure drop and the mass flow is computed from the Blasius equation used for turbulent fully developed flow in a smooth pipe. The heat exchange coefficient between the helium in convection inside the fuel element is computed from the correlation given in [3-18]. It is nevertheless weighted by the channels related specific surface. Moreover, the Dittus-Boelter equation is used for the heat transfer in the reflector and in the plenum.

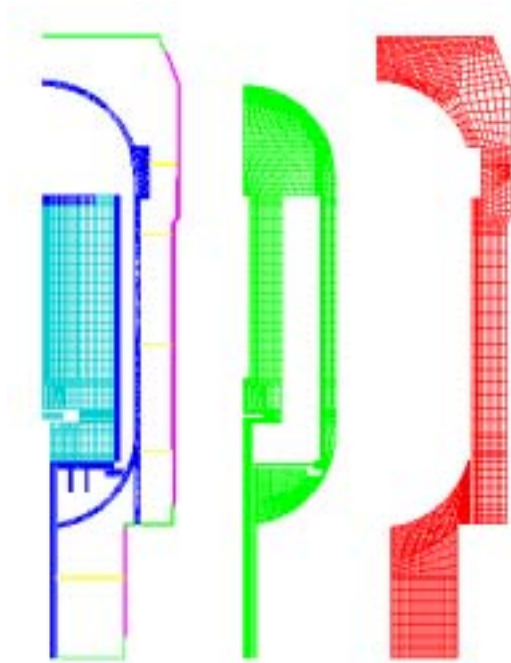


FIG. 3.58. Solid, helium and air related meshes.

As far as the bottom replaceable reflector is concern, the characteristics (hydraulic diameter, porosity, specific surface) correspond to the average characteristics of the hot plenum. It is assumed that this assumption won't influence strongly the benchmark results (external temperature of the vessel, panel temperatures and power evacuated by the Vessel Cooling System).

Concerning the solid phase of the porous medium in the core, the equivalent horizontal thermal conductivities have been determined, as function of the temperature. In order to evaluate these equivalent conductivities, a simplified model of the fuel block has been used taken into account the radiation heat transfer across the channels.

Finally, neither the standpipe nor the thermal shield of the upper cavity, have been considered in the model. The helium is modelling as an incompressible fluid. Although the mass density variation is important, the Boussinesq's approximation is used. Besides, the thermal conductivity as well as the viscosity is function of the temperature. In all the pressure vessel cavities the radiation heat transfer is taken into account, the helium being transparent for this kind of transfer.

***The reactor cavity***

Four ring plates are located in the reactor cavity (50 mm from the pressure vessel and 100 mm from the external panels). There is also a ring plate in the lower part of the reactor (under the pressure vessel). As depicted in Figure 3.59, additional plates have been added at the extremity of the ring plates [3-21]. Therefore, it is assumed that these ring plates share the lower part of the reactor cavity in two independent hydraulic zones and the side reactor cavity

in five others hydraulic zones. In these zones, the air is considered as an incompressible fluid and transparent for the radiative heat transfer. The Boussinesq's approximation is also used. The thermal conductivity and the viscosity have been assumed constant and considered at 60°C.

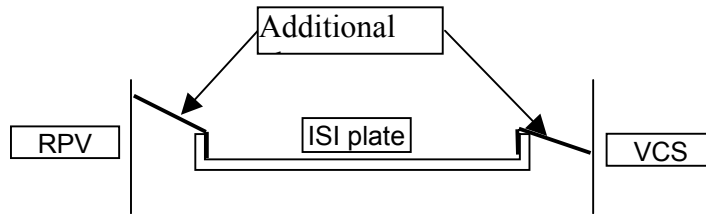


FIG. 3.59. Ring plate between the reactor and the panel.

### The side and bottom panels

For the reactor cavity cooling system (VCS), the water temperature distribution is supposed constant along the tubes. Calculations show that the axial flow is fully turbulent in the tubes of the side panel (Re number about  $10^4$ ). Then, the Mc Adams correlation allows evaluating the Nusselt number from which a local exchange coefficient has been deduced. Considering the R-Z model an average exchange coefficient has been used for the whole panel.

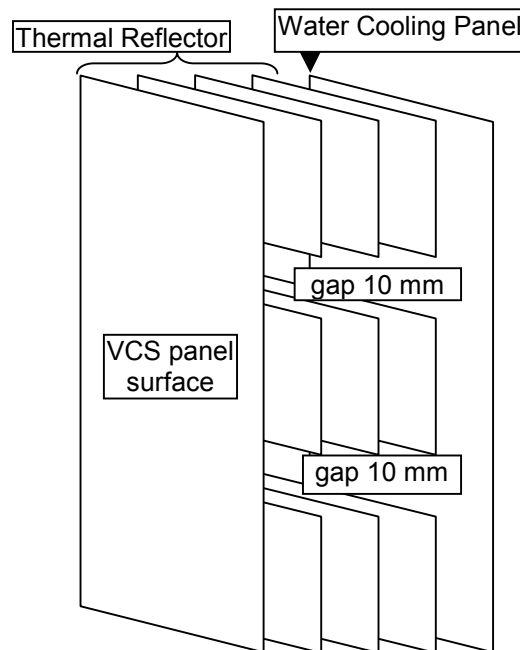


FIG. 3.60. Thermal reflector plates along the VCS.

In front of the tubes, there are some vertical thermal reflector plates (2 or 5). These plates have gaps to manage the axial thermal expansion of the whole reflector. Nevertheless, they enhance air circulation between the panels. As it is difficult to take into account the presence of these gaps in the whole reactor model, some preliminary calculations have been carried out on the thermal reflector zone.

A simplified model has been developed where all the reflector plates have been described. In order to simulate the radiative heat transfer between the pressure vessel and the

thermal reflector, a part of the pressure vessel at constant temperature (300°C) radiating to the reflector is assumed as a boundary condition. In this simplified model, the emissivities corresponding to those of the global model. On the side panel, the heat exchange is represented by similar parameters than those used in the global reactor model. Three calculations have been performed: the first taking into account radiation between the reflector plates but without air circulation, the two last considering air circulation with and without the gaps. The results are presented in Table 3-42.

Table 3-42. Refined Calculations on the Thermal Reflector

<i>Configurations</i>	Radiative Power
Without air circulation	116 kW
With air circulation and without gaps	124 kW
With air circulation and with gaps	131 kW

The air circulation observed inside the thermal reflector corresponds to an upward flow in the front cavities (reactor side) and a downward flow in the cavity along the VCS panel (cooling tubes side). The calculations show a more equilibrated flow in the central cavities. By taking into account the gaps, the air velocities are enhanced and this leads to lower temperature on the reactor side part of the thermal reflector. The extracted power is therefore higher in this case. Finally, according to these preliminary results, the air circulation has been considered in the final model without describing the gaps.

### 3.2.5.2. Results for the global model

#### **Algorithm and finite element**

The finite elements are linear quadrangle for the heat transfer, quadrangles isoQ2 and isoP1 for respectively velocity and discontinuous pressure [3-19, 3-20]. A finite element isoQ2 comprises 4 linear elements. The global model has 11580 elements for the solid parts, 2808 elements in the helium and 1513 in the air (Figure 3.58). Hydraulics and energy transport equation are treated by an implicit scheme and simultaneously solved in order to get a stable solution for the fluid-structure heat transfer. Two calculations have been performed corresponding respectively to core power 9 MW and 30 MW (Table 3-43).

Table 3-43. Reactor Characteristics Used in the Calculations

	9 MW	30 MW
Reactor Power	9 MW	30 MW
Helium Inlet Temperature	175 °C	395 °C
Flow Rate	12.4 kg/s	12.4 kg/s
Cooling Water Temperature of VCS	28.5 °C	25 °C
Helium Outlet Temperature	312 °C	850 °C

#### **Benchmark Results**

The surface temperature of the thermal reflector (reactor cavity side) is also given for both envisaged power level. The outlet temperature of the helium ranges from 321 °C at 9 MW to 863 °C at nominal power. Moreover, the temperature distributions in the helium, the air and the solid are provided in Figure 3.61 for the whole reactor. Finally, the Table 3-44 gathered the main results of the total power extract by the Vessel Cooling System in steady state.

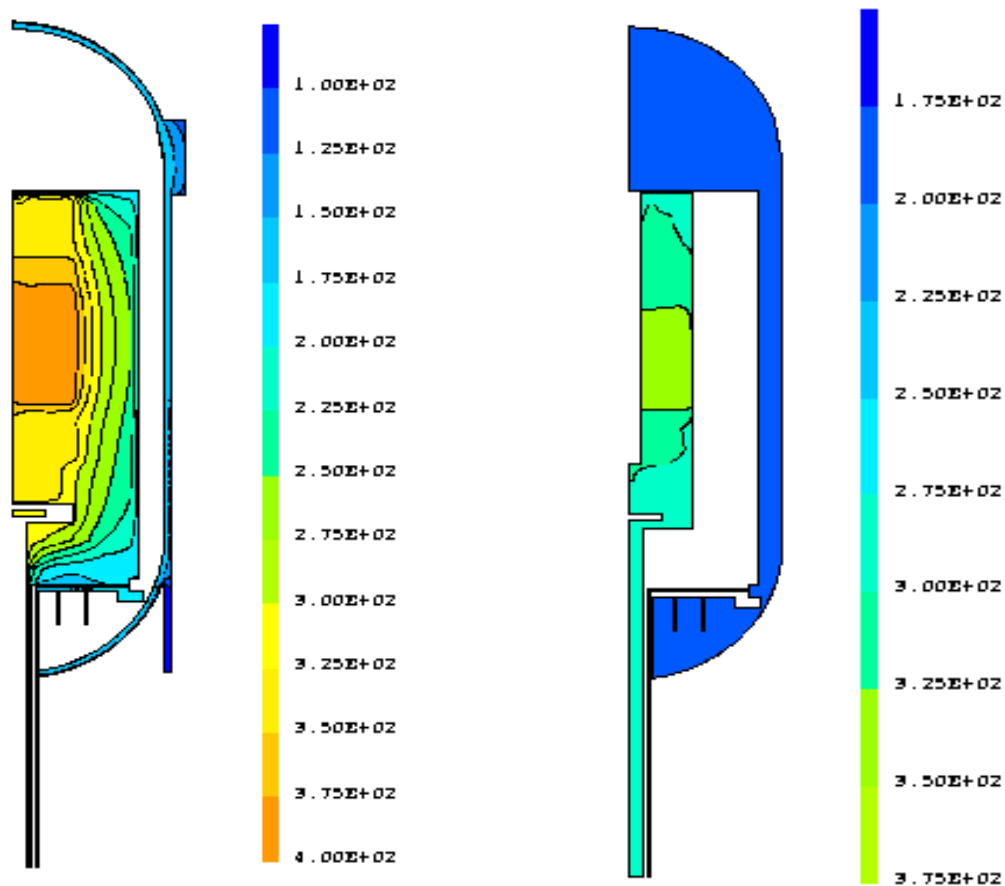


FIG. 3.61. Solid and gas temperature distributions for 9 MW.

Table 3-44. VCS Performance in Steady State

Power level	9 MW	30 MW
Upper panel	20 kW	60 kW
Side panel	150 kW	480 kW
Lower panel	8 kW	15 kW
<b>Total extracted power</b>	<b>178 kW</b>	<b>555 kW</b>
<b>PRV Temp. (19-27 m)</b>	<b>168 °C</b>	<b>340-353 °C</b>

### 3.2.5.3. Concluding remarks

The results obtained in the upper part of the cavity could be better. Indeed the:

- Thermal reflector located to the top head dome (reactor pressure vessel) has not been taken into account and lead to an overestimation of vessel temperature.
- Standpipes were also not considered. They certainly slow down the air circulation and modify the radiative exchanges between the divers walls.
- Heat exchange in the upper part of the VCS tubes is assumed to be uniform

The additional plates to the ring plates constitute a *thermal continuity* between the reactor pressure vessel and the thermal reflector of the VCS. The impact of this assumption on the temperatures remains to be tackled.

Finally, the concrete, beyond the VCS tubes, has not been considered. Therefore, a part of radiation heat coming from the vessel (directly or through the support panel of the VCS tubes) must induce a temperature gradient in the concrete. These thermal leakages are not taken into account in the present model.

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